

NASA Contractor Report 178292

Development of Response Models for the Earth Radiation Budget Experiment (ERBE) Sensors:

Part I – Dynamic Models and Computer Simulations for the ERBE Nonscanner, Scanner and Solar Monitor Sensors

(NASA-CR-178292) DEVELOPMENT OF RESPONSE MODELS FOR THE EARTH RADIATION BUDGET EXPERIMENT (ERBE) SENSORS. PART 1: DYNAMIC MODELS AND COMPUTER SIMULATIONS FOR THE ERBE NONSCANNER, (Information and Control Systems) G3/35
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FOREWORD

This report entitled "Development of Response Models for the Earth Radiation Budget Experiment (ERBE) Sensors" consists of the following four parts.

This is Part I, NASA CR-178292, entitled "Dynamic Models and Computer Simulations for the ERBE Nonscanner, Scanner and Solar Monitor Sensors".

Part II, NASA CR-178293, is entitled "Analysis of the ERBE Integrating Sphere Ground Calibration".

Part III, NASA CR-178294, is entitled "ERBE Scanner Measurement Accuracy Analysis Due to Reduced Housekeeping Data".

Part IV, NASA CR-178295, is entitled "Preliminary Nonscanner Models and Count Conversion Algorithms".

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LIST OF SYMBOLS

<u>Symbol</u>	<u>Definition</u>
A_c	Area of conductive transfer in an element
A_r	Area of radiative transfer in an element
A_{k-i}	Contacting area between elements k and i
a	Radius of a contacting point, e.g. Fig. 7
A_i	Inside surface area for i^{th} surface
\bar{A}	Constant defined in Eq. 14
A_r	Area of radiative interaction
A_c	Area of nodes covered by a heater
a	Thermal wave velocity, e.g. Eq. 58
b	Bias term, e.g. Eq. 25
B	Vector of coefficients for irradiance from the source
B_Q	Coefficient which describes heating contributions from the control systems
b_E	Electronic bias
B_k	Constants related with the irradiance at a FOV limiter
C_k	Specific heat capacity of element k
\bar{C}	Constant defined in Eq. 15
C	Number of counts
ΔC_B	Baseband signal
D	Enclosure coupling factor through the interactive boundaries

LIST OF SYMBOLS (CONTINUED)

<u>Symbol</u>	<u>Definition</u>
e_k	Energy in element k
E_n^m	Rate of incoming radiant energy to surface n in enclosure m
\tilde{E}_n^m	Rate of incoming radiant energy to non-opaque boundary surface n in enclosure m from a neighboring surface
e	Noise term, Eq. 25
\hat{E}	Total incident radiation estimate
ΔE_{os}	Baseband signal
$E_\lambda(t)$	Spectral irradiance at the instrument due to radiation from the footprint at time t
$e_i(t)$	Combined error of the measurement and count conversion accuracy
$E_{Bi}(T_B)$	Output of count conversion when viewing the blackbody at temperature T_B
$E_{B\lambda}(T_B)$	Spectral irradiance from the blackbody at T_B
F_{k-i}	Configuration factor between elements k and i
\bar{F}_{k-i}	Exchange factor between elements k and i
F	System boundary conditions imposed by surroundings
$F_E(s)$	Control electronics transfer function.
h_{k-i}	Thermal conductance of the contact between elements k and i
H_z	Hertz

LIST OF SYMBOLS (CONTINUED)

<u>Symbol</u>	<u>Definition</u>
I	Identity Matrix
i	Calibration sequence
J	Number of measurements in the scanner mode.
k_k	Thermal conductivity of element k
K_i	Electronic gain for component i
k_{fo}	Post amplification gain
l	Longwave
L_{k-i}	Shortest distance between the center of mass of element k and the contacting point of element k to its neighboring element i
L	Length of a solar monitor shutter
l	Exposure length to solar radiation
M_i^*	Opaque radiosity - the rate of outgoing radiant energy from surface i from diffuse and specular reflectance
M_i	Total radiosity - opaque radiosity plus energy flux transmitted through surface i
ms	Milliseconds
m	Meter
n	Number of contacts per unit area
n_i	Noise from source i

LIST OF SYMBOLS (CONTINUED)

<u>Symbol</u>	<u>Definition</u>
\dot{Q}_k	Dynamic heat transfer due to behavior of element k as a heat source or sink
\dot{q}_c''	Heat flux by conduction through the contacting area between two elements
R_{k-i}	Thermal resistance through the contacting area between elements k and i
R_i	Inside radius of solar monitor barrel
R_o	Outside radius of solar monitor barrel
S_n^m	Heat flux arriving at the system boundary at surface n in enclosure m
s	Shortwave
T	Total radiation channel
T_k	Temperature (vector) of element k
\dot{T}_k	Governing equation for a radiometer with a dynamic response to an incoming radiance at element k
T(i)	Temperature change for thermal node i or voltage change for electronic node i
T_∞	Ambient temperature
T_b	Current shutter temperature in solar monitor
Δt	Total scan period
t_k	Time at end of space-look
t	Sampling instant

LIST OF SYMBOLS (CONTINUED)

<u>Symbol</u>	<u>Definition</u>
T_B	Temperature of a blackbody
$u(x)$	Unit step function
V_k	Volume of element k
$V(t), v(t)$	Instrument output (in volts) at time t
$\bar{v}(t_k)$	Space clamp value at time t_k , i.e. at the end of space look
$v(t_{ki})$	Instrument output (in volts) when viewing space at time t_k
$V_B(t)$	Detector bridge bias voltage at time t or most recent value (v)
$V_d(t_k)$	Drift balance DAC voltage measurement at time t or most recent value (v)
$v_{Bi}(T_B)$	Steady-state output of the i^{th} channel when viewing blackbody temperature T_B
W	Watts
x	Distance from end of barrel in a solar monitor
x	Vector consisting of electrical signal processing variables, e.g. Eq. 73
x	Vector consisting of the offset temperature and radiative constant, e.g. Eq. 109
$\Delta \bar{x}_i$	Lower frequency signal portion of Δx for component i, e.g. Eq. 79
$\Delta \tilde{x}_i$	Noise and irradiance portion of Δx for component i, e.g. Eq. 79

LIST OF SYMBOLS (CONTINUED)

<u>Greek Symbol</u>	<u>Definition</u>
γ_k	Density of element k
δ_1	Surface roughness of surface 1
δ	Thickness of solar monitor shutter
δ_k	Estimate of unaccounted drift during k th scan period (v)
ϵ_k	Emissivity of element k
ϵ_{k-i}	Exchange factors between specular surfaces k and i
η_k	Attenuation factor for element k
λ_k	Wavelength k
ρ_k	Reflectivity of surface k
ρ	Density of solar monitor shutter
σ	Stefan-Boltzmann constant
σ_k^2	Noise variance estimate during space look
τ_k	Transmissivity of surface k
τ_i	Time constant for electronic element i
τ	Average time lag (in seconds)
$\tau_i(k)$	Transmittance of i th channel
ϕ_k	Radiative energy supplied to surface k

Superscript

d	Diffuse
s	Specular
m	Number of an enclosure in a system

LIST OF SYMBOLS (CONCLUDED)

<u>Superscript</u>	<u>Definition</u>
M	Medium field-of-view
W	Wide field-of-view
P	Photo-diode
<u>Subscript</u>	
s	Shortwave band
l	Longwave band
s	Surface, Eq. 6
k-i	between k and i elements
n	Number of an enclosure's surface in a system
H	Heater, Eq. 35
A	Precision aperture
F	Field-of-view limiter
B	Bridge
A	Amplifier
E	Electronic
v	Volatage
B	Blackbody
o	Initial
s	Solar, Eq. 58
H	Heat sink, Eq. 99

LIST OF ACRONYMS

<u>Acronym</u>	<u>Definition</u>
ACR	Active Cavity Radiometer
Al Heat Sink	Aluminum Heat Sink
A/D	Analog-to-Digital (conversion)
B.C.	Boundary Conditions
CCA	Count Conversion Algorithm
Cu Heat Sink	Copper Heat Sink
DAC	Digital-to-Analog Conversion
ERBE	Earth Radiation Budget Experiment
EXP	Exponential
FOV	Field-of-view
FOVL	Field-of-view limiter
G.E.	Governing Equations
HK	Housekeeping
HKD	Housekeeping data
IBB	Internal blackbody
I/O	Input/Output (data processing)
I.C.	Initial Conditions
LW	Longwave
MFOV	Medium Field-of-View
MFVOT	Medium Field-of-View Total
MFOVSW	Medium Field-of-View Shortwave
MAM	Mirror Attenuator Mosaic
MRBB	Master Reference Blackbody

LIST OF ACRONYMS (CONCLUDED)

<u>Acronym</u>	<u>Definition</u>
ms	Millisecond
NS	Nonscanner
NFOV	Narrow Field-of-View
RHS	Right hand side
RMS	Root mean square
SW	Shortwave
SWICS	Shortwave Internal Calibration Source
SiPD	Silicon Photo-Diode
T	Total radiation channel
WFOV	Wide Field-of-View
WFOVT	Wide Field-of-View Total
WFOVSW	Wide Field-of-View Shortwave
WRT	With respect to

1. INTRODUCTION

The radiometric measurement system for the Earth Radiation Budget Experiment (ERBE) is designed to measure the earth's radiation fluxes which result from the earth and its atmosphere's emitted energy as well as from reflected energy, i.e., the earth's albedo. A general overview of the concept behind the three-satellite ERBE radiometric system and its instruments has been described in many internal NASA documents, such as the Science Team Minutes and Contractor Status Reports, as well as in journal publications, (e.g., Ref. 1).

The radiometer package on each ERBE satellite consists of eight instruments distinguished by their mode of operation, field-of-view and spectral response. Four of the eight instruments are nonscanner (NS) type radiometers, two with a Medium Field-of-View (MFOV) and two a Wide Field-of-View (WFOV). Two of the nonscanner radiometers, one MFOV and one WFOV, have a suprasil filter which eliminates the instrument's response to longwave radiative effects so that they only remain sensitive to shortwave radiative effects. Shortwave (SW) radiative effects are defined here as having wavelengths between 0.2 and 5 μm and longwave radiative effects as having wavelengths between 5 and 50 μm . Since radiometers without filters measure total radiation, they are designated "Total" (T). The four nonscanner radiometers are therefore the Medium Field-of-View Total (MFOVT), Medium Field-of-View Shortwave (MFOVSW), Wide Field-of-View Total (WFOVT) and Wide Field-of-View Shortwave (WFOVSW). Three of the remaining instruments or "channels" are Narrow Field-of-View (NFOV) scanning radiometers, two of which have filters. One scanner radiometer has a filter which eliminates longwave radiative effects and another has a filter which eliminates shortwave radiative effects. Hence the three scanner radiometers are the Narrow Field-of-View Shortwave (NFOVSW), the Narrow Field-of-View Longwave (NFOVLW) and the

Narrow Field-of-View Total (NFOVT). The eighth instrument in the ERBE radiometric package is the solar monitor radiometer.

After the ERBE radiometers perform their earth, space and/or sun view and calibration measurements, the measurement results undergo an analog-to-digital (A/D) conversion and the data now in counts is transmitted to the earth through the spacecraft telemetry system. Counts for the nonscanners are measured every 0.8 seconds. Heat sink temperatures are read every 8 seconds and other housekeeping (HK) temperatures every 16 seconds. Therefore, in 16 seconds the data record for one nonscanner consists of 20 radiometric count readings, 2 heat sink temperature readings and 1 reading for each of the other HK temperatures.

The scanners perform a measurement cycle once every 4 seconds. During a normal, earth-viewing scan each scanner performs 8 space-look measurements, 62 earth-view measurements and 4 calibration measurements. The Longwave and Total NFOV scanners utilize the Internal Blackbody (IBB) for calibration measurements, whereas the Shortwave NFOV scanner utilizes the Shortwave Internal Calibration Source (SWICS). During a Mirror Attenuator Mosaic (MAM) scan cycle, the NFOVT and NFOVSW scanners measure solar flux performing 8 space-look measurements, 46 MAM measurements and 4 calibration measurements. Housekeeping (HK) data is sampled once per scan.

The solar monitor uses the same type of Active Cavity Radiometer (ACR) that is used by the nonscanners. The three major differences between these ACR's are as follows:

- a) the solar monitor has a shutter to cut off the solar flux every 32 seconds,
- b) the configuration of the solar monitor housing is totally different and

c) the solar monitor has no heat sink heater.

Solar Monitor count measurements are made every 0.033 seconds and HK measurements are made every 4 seconds.

The radiometric instruments described above will have inherently different behavior and responses to arbitrary input signals since no instrument can be exactly duplicated or subjected to exactly the same calibration or operating environment. Even when viewing the problems of sensor performance from this macroscopic point of view, there are many things that have to be dealt with and accounted for. For example, environmental contamination and aging effects influence the material degradation of the sensors. Sensor behavior may also be questioned when the thermal load and leakage exceed the design level. One example of abnormal thermal load is in the sun-blip case, where the sensor is subjected to a sunrise and sunset during its orbit.

In attempting to account for such problems found in the sensor's radiometric performance and capability, it is therefore necessary to develop a radiometer sensor model. A sensor model should describe both a transient and steady-state sensor response for a given radiative input. It must provide sufficiently simple analytical expressions for parameters of particular interest and describe the model's behavior, e.g., the model's sensitivity to changes in the modeling assumptions. The model must also be able to incorporate changes resulting from instrument design requirements during its construction as well as instrument aging, i.e., sensor material degradation and unexpected thermal load and leakage during its long-term orbital operation. The products obtained from the sensor model and simulation must be in a form that is useful to the users. The sensors' response must therefore be validated by updating model parameters and converted into data with useful engineering units.

This report entitled "Development of Response Models for the Earth Radiation Budget Experiment (ERBE) Sensors" consists of the following four parts.

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2. SENSOR MODELING

The following ERBE radiometer sensor models were developed by Information & Control Systems, Incorporated (ICS) to consider thermal and radiative exchange effects. These models consider the surface specularity and spectral dependence of a filter as well as the thermal conductive and radiative interactions among the nodes in an enclosure of a radiometer. An enclosure is a domain in which an instrument's geometry is divided into appropriate elements. These enclosures in an instrument's model are set up using real or fictitious boundaries to effectively alleviate the computational load for defining configuration factors. The instrument's model also includes the electronic thermal control systems which are coupled with the energy equations. A general form of the energy equations can be used for all the sensors, despite differences in the mode of operation, field-of-view and spectral response, as long as the radiative enclosures and thermal nodes in the system's sensor configuration comply with the first law of thermodynamics.

A sensor model can be tested and evaluated by comparing its behavior with the actual results obtained from the sensor's ground and inflight calibrations. The sensor model includes additional uncertainty since it incorporates outputs from the calibration source models. Once the sensor and calibration source models produce sufficiently accurate results, they can be used to define the parameters for the count conversion procedure.

The characteristics of a sensor's simulation model are such that it has a dynamic response to the time dependence of an arbitrary input signal. The model also has a spectral response which is due to the spectral dependence of a sensor's filter and its exposed surfaces. Thermal effects for the model can be determined from the conductive and radiative properties of the sensor. The

model's measurement error can therefore be determined after its time response, spectral response and thermal effects have been obtained. The model can simulate the sensor's response to the temporal incident radiation received through the field-of-view limiter in both the transient and steady-states.

The purpose of modeling and simulating is therefore 1) to understand sensor performance, 2) to improve measurement accuracy by updating the model parameters and 3) to provide appropriate constants necessary for the count conversion algorithms.

2.1 General Description of the ERBE Sensor Models

2.1.1 Models

A sensor's geometry can be appropriately divided by considering its thermal and radiative interactions as a consequence of its design. Figures 1 through 6 provide diagrams of how sensor elements have been geometrically partitioned for the nonscanner, scanner and solar monitor. Each element has a time response to the energy flowing to and from its neighboring elements.

One might assume that maximizing the number of sensor nodes would maximize the accuracy of the model's prediction of energy from a target source. However, increasing the number of nodes would introduce additional sources of error since the large set of equations required would incorporate additional parameter approximations. Such equation approximations would need to be made to calculate constants such as thermal resistance, exchange factors and surface spectral and specular characteristics. There is an accuracy tradeoff between increasing the number of nodes and approximating the sensor's size, material properties, geometry, enclosure surface characteristics, temporal incident

1 ACTIVE CAVITY
 2 Cu HEAT SINK
 3 REFERENCE CAVITY
 4 Al HEAT SINK
 5 BASE PLATE
 6 FOV LIMITER

- H HEATER
P TEMPERATURE PROBE

7

NONSCANNER MEDIUM FIELD-OF-VIEW SHORTWAVE (MFOVSW)

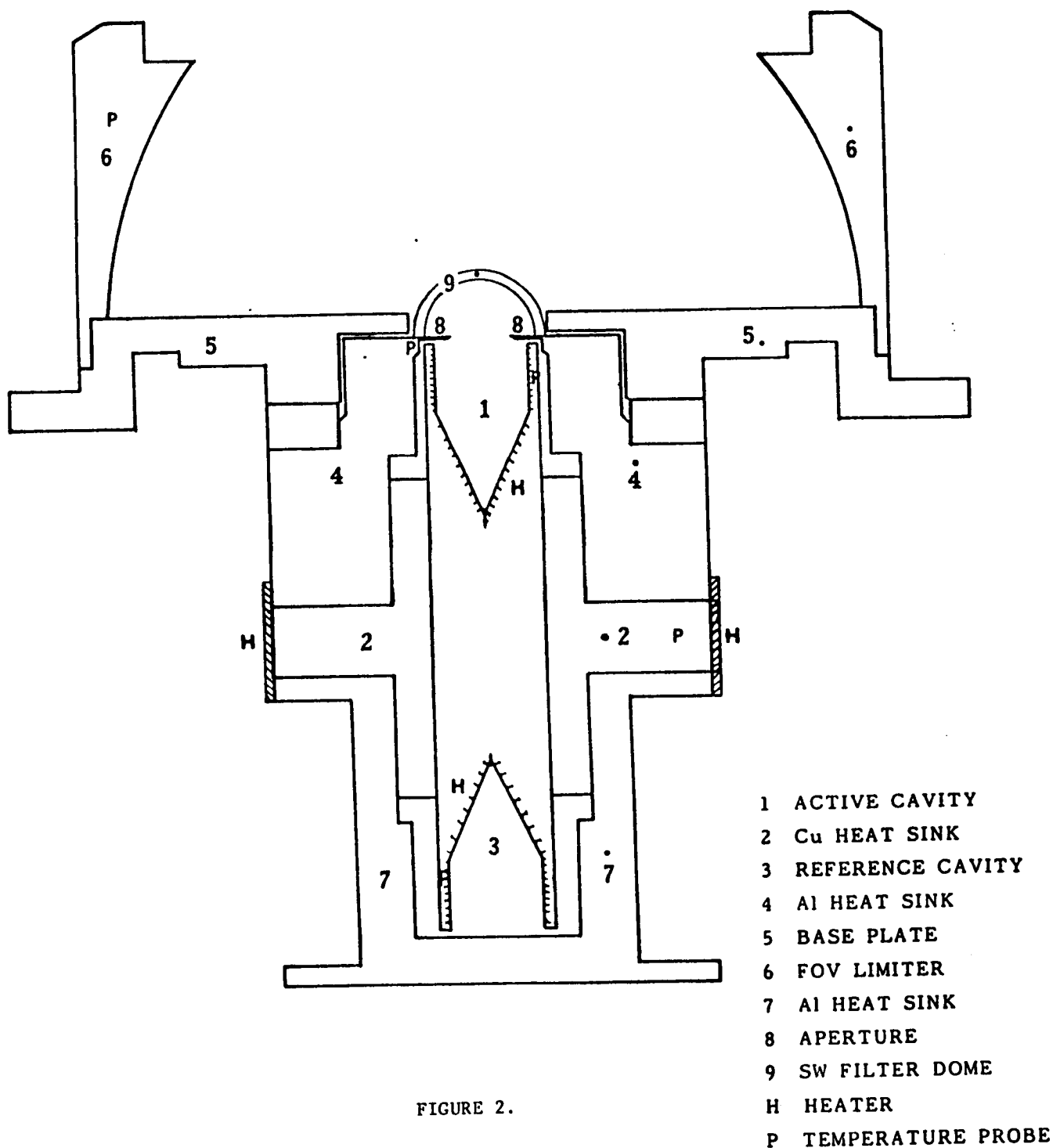


FIGURE 2.

NONSCANNER WIDE FIELD-OF-VIEW TOTAL (WFOVT)

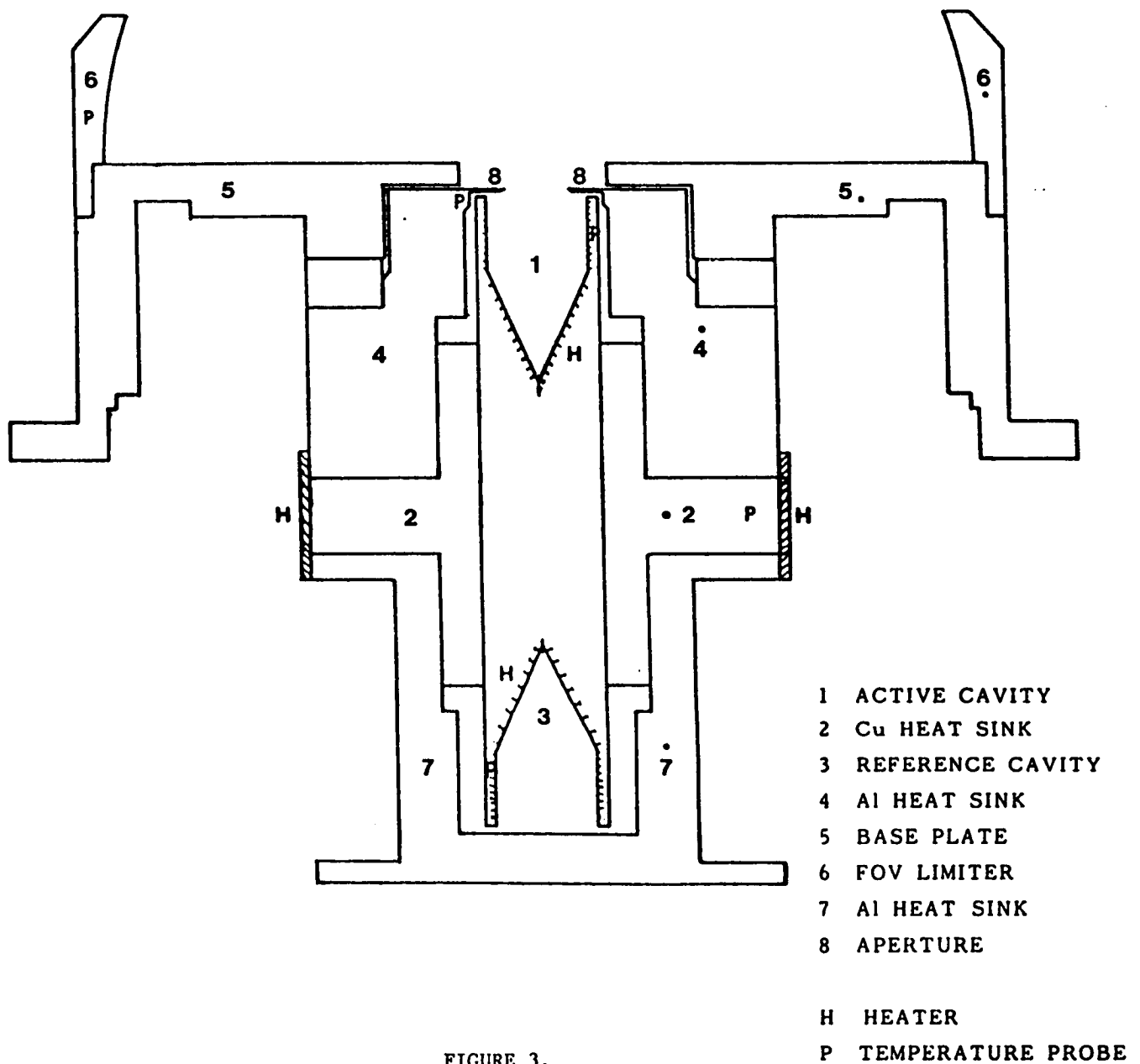


FIGURE 3.

NONSCANNER WIDE FIELD-OF-VIEW SHORTWAVE (WFOVSW)

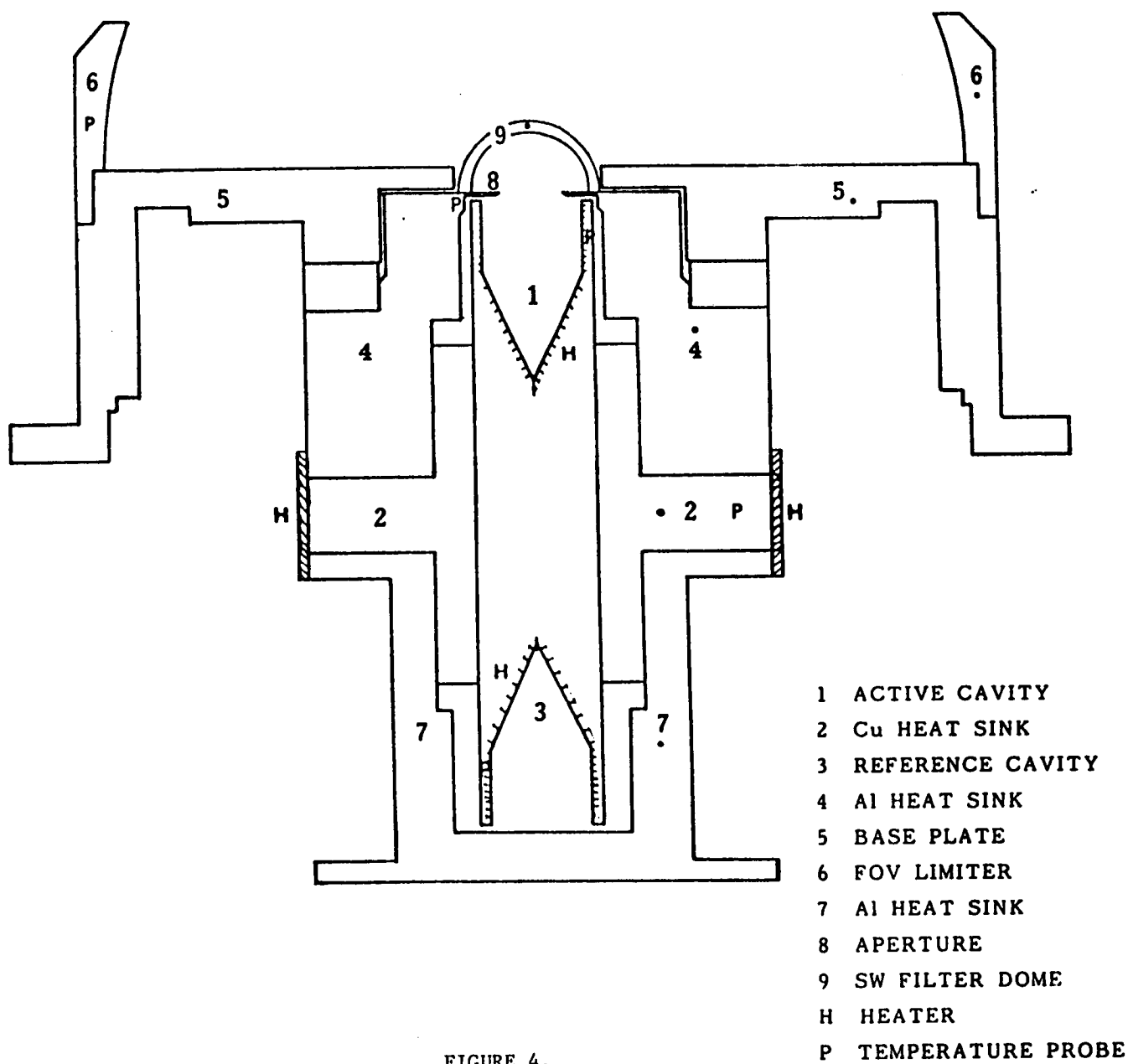
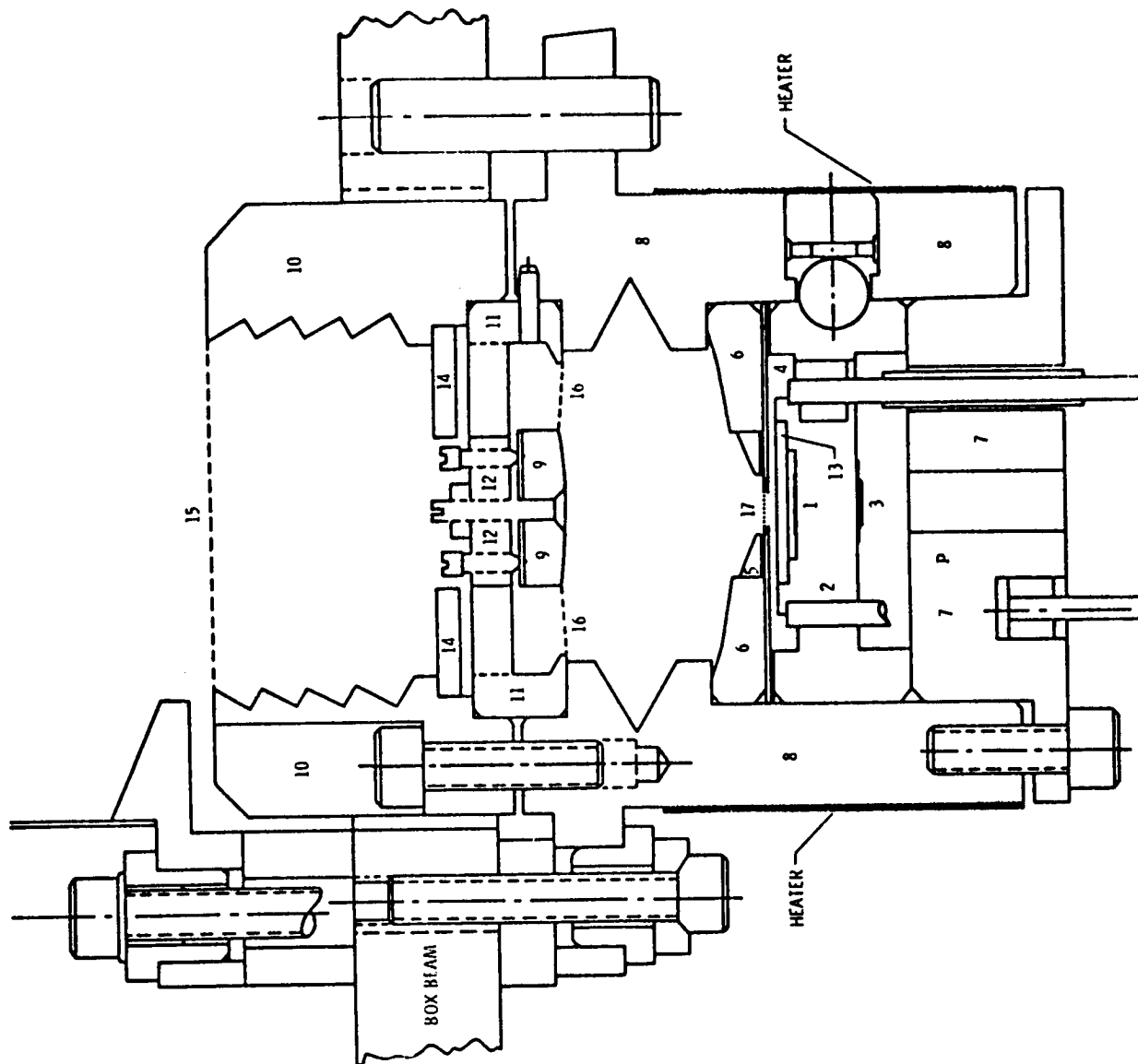


FIGURE 4.

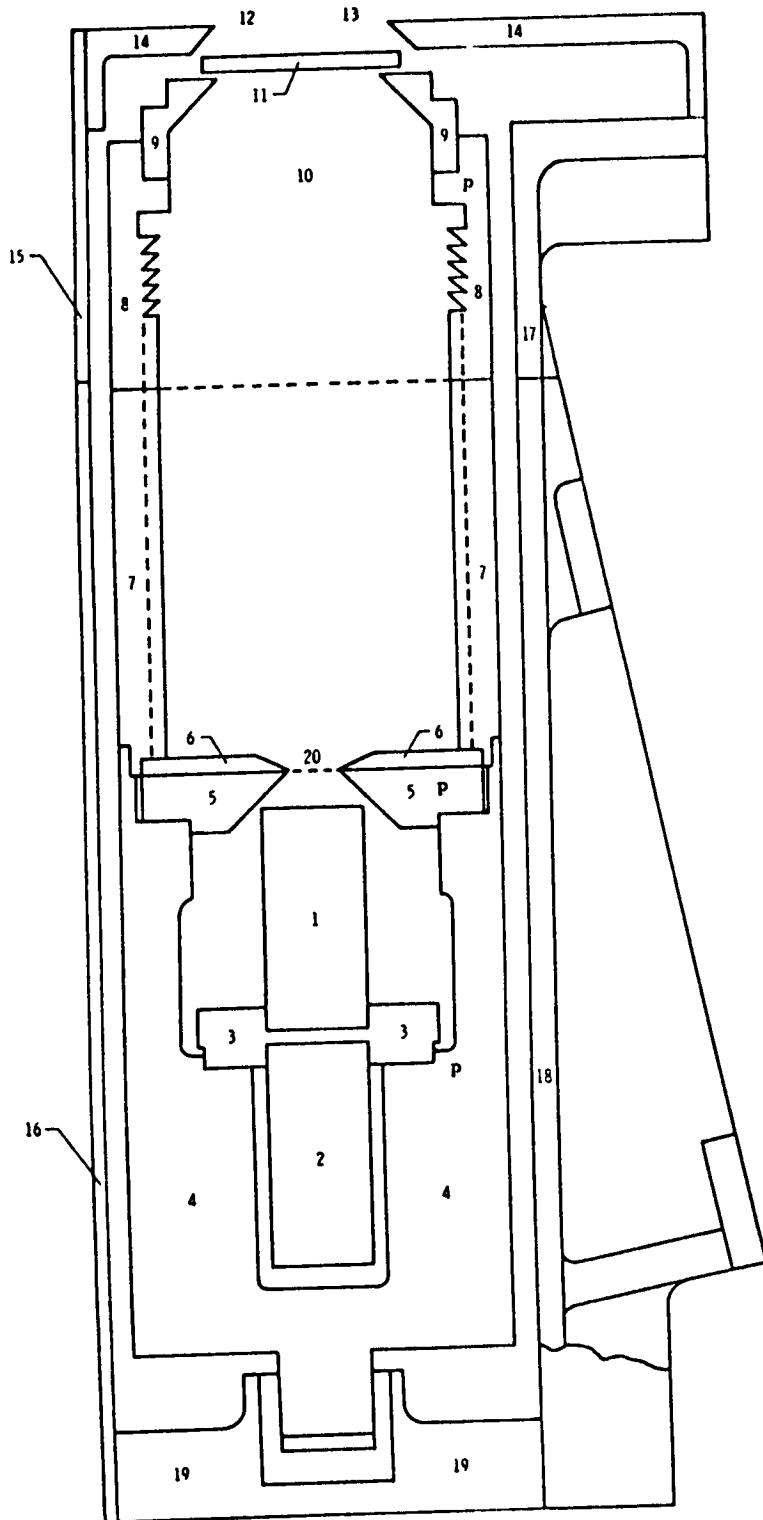
SCANNER NARROW FIELD-OF-VIEW TOTAL (NFOVT)



1. ACTIVE FLAKE
 2. HEAT SINK & DETECTOR ASSY.
 3. COMPENSATING FLAKE
 4. PRECISION APERTURE
 5. INSERT, PRIMARY
 6. PRIMARY MIRROR
 7. CAP ASSEMBLY
 8. HOUSING & HEATER
 9. SECONDARY MIRROR
 10. FOV LIMITER
 11. MOUNT, SECONDARY(A)
 12. MOUNT, SECONDARY(B)
 13. LW OR SW FILTER
 14. SW FILTER
 15. FICTITIOUS SURFACE AT FOV
 16. FICTITIOUS SURFACE AT SECONDARY
 17. FICTITIOUS SURFACE AT APERTURE
- P- TEMPERATURE PROBE

FIGURE 5.

SOLAR MONITOR



1. ACTIVE CAVITY
2. REFERENCE CAVITY
3. BUTTON, DUPLEX CONE CAVITIES
4. HEAT SINK
5. APERTURE, PRIMARY
6. SHIELD, APERTURE
7. BODY (REAR BARREL)
8. BODY (FRONT BARREL)
9. APERTURE, SECONDARY (FOR FOV LIMITER)
10. MOTOR HOUSING
11. CHOPPER BLADE
12. CHOPPER PENDULUM
13. MOTOR, CHOPPER
14. FOV LIMITER
15. FRONT TOP HOUSING COVER
16. REAR TOP HOUSING COVER
17. FRONT HOUSING
18. REAR HOUSING
19. HOLDER, HEAT SINK
20. FICTITIOUS SURFACE
- P. TEMPERATURE PROBES

FIGURE 6.

energy flux intensity and electronic control system. If the temperature gradients within a node and between nodes are small enough, a model approximation with only a few nodes would not jeopardize the accuracy required to perform the parameter estimates and error analysis. Factors which would decrease these temperature gradients within a sensor include uniform thermal properties, simple and symmetrical geometry, uniformity in directional and spectral properties of an enclosure surface and small temporal change in incident energy flux.

A nonscanner sensor has 9 nodes if it has a suprasil filter dome (Figures 2 and 4) and 8 nodes if it does not (Figures 1 and 3). The sensor's geometry is divided into nodes based on material homogeneity, overall sensor size, thermal conductivity of materials, thermal resistance between materials and surface radiative properties. The 9 nodes of the MFOVSW model (Figure 2) are therefore, as numbered: 1) the active cavity with a heating and sensor coil 2) the copper heat sink with a heater, 3) the reference cavity with a sensor coil, 4) the aluminum heat sink, which is partly covered by a heater, 5) the base plate or substrate, 6) the field-of-view limiter, 7) the aluminum heat sink which houses the reference cavity, which is partly covered by a heater, 8) the precision aperture and 9) the filter dome for shortwave channels.

Similarly, the scanner model (Figure 5) is divided into 12 nodes for the total channel, 13 nodes for the longwave channel and 14 nodes for the shortwave channel. The solar monitor is divided into 19 nodes (Figure 6).

In addition to their nodes, these sensor models incorporate fictitious surfaces to define and simplify enclosure geometries. Since the specular property of each surface in an enclosure is taken into consideration, the simple geometry facilitates the exchange factor calculation, alleviating the computational work load for determining the exchange factors. A net radiation

analysis is performed for each enclosure which has surfaces with specular and spectral characteristics. This analysis appropriately deals with any partial specular and diffuse characteristics of an enclosure surface (Ref. 2). The model can also easily include transient response and thermal conduction. This net radiation method utilizes the waveband approximation to an enclosure's surface which has a specific spectral dependence.

2.1.2 Thermal Conductive and Radiative Interactions

A volumetrically defined node, or control volume, complies with the first law of thermodynamics as in the following equation

$$\frac{de}{dt} = A_c \dot{q}_c'' + A_r \phi \quad (1)$$

The first and second terms of the right hand side of Equation 1 represent the conductive transfer through contacting surfaces and radiative transfer from surrounding elements of an enclosure, respectively. The rate of energy change in a single node, therefore, depends on the difference between the ingoing and outgoing energy fluxes through a control surface or "boundary". This rate of energy gain or loss is composed of the rate of energy stored in a node as well as the rate of energy source or sink as indicated in the following equation

$$\frac{de}{dt} = V\gamma C \frac{dT}{dt} - \dot{Q} \quad (2)$$

where V is the volume of a node, γ the density of a node, C the specific heat, T the temperature and \dot{Q} the rate of energy, or heat, at a source or sink.

Combining Equations 1 and 2 yields

$$V\gamma C \frac{dT}{dt} = A_c \dot{q}_c'' + A_r \phi + \dot{Q} \quad (3)$$

Equation 3 represents an energy balance in a single node. In the geometrically partitioned model, the conduction term can be written in the following form

$$\left(A_c \dot{q}_c'' \right)_{i-j} = \frac{T_j - T_i}{R_{i-j}} \quad (4)$$

where A_{i-j} is the contacting area of i element to j element, \dot{q}_c'' the heat flux by conduction through the contacting area between i and j elements. R_{i-j} is the thermal resistance between i and j elements. The thermal resistance between nodes is defined by the following relationship

$$R_{i-j} = \frac{L_{i-j}}{k_i A_{i-j}} + \frac{1}{h_{i-j} A_{i-j}} + \frac{L_{j-i}}{k_j A_{j-i}} \quad (5)$$

where L_{i-j} is the distance from the thermal center of a node to the center of the contacting surface, k the thermal conductivity and h the thermal conductance of the contact between i and j elements. The thermal center is normally coincident with the geometrical center if a node material has homogeneous and isotropic thermal properties. The contact resistance ($1/hA$) is mainly dependent on the surface roughness and the contacting pressure. The well-machined surface has a regular and uniform roughness. In such a case, the contact resistance can be determined by the following empirical relations (Ref. 3) as functions of the surface roughness and the thermal conductivities of two contacting materials.

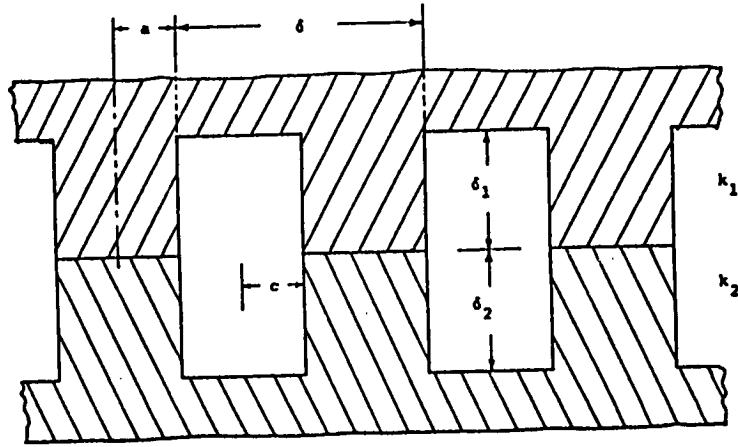


Figure 7. A Simple Thermal Contact Resistance Model

Let surfaces contact each other with surface roughness δ_1 and δ_2 and thermal conductivities k_1 and k_2 , respectively. Assume the contacting area per periodic contact is πa^2 . Then the number of contacts per unit area $n = \frac{1}{\pi a^2}$ and the total number of contacts for two surfaces with total contacting area A is $\frac{A}{\pi a^2}$. Conductance can therefore be calculated:

$$h = \frac{k_s}{\delta} \left(\frac{\epsilon^2}{1 - \epsilon^2} \right) \left(\frac{\eta}{\epsilon + \eta} \right) \quad (6)$$

where

$$\frac{1}{k_s} = \frac{1}{2} \left(\frac{1}{k_1} + \frac{1}{k_2} \right)$$

$$\delta = \delta_1 + \delta_2 \quad (\approx 2\delta_1 \text{ or } 2\delta_2)$$

$$\eta = 2.13\delta \sqrt{n}, \text{ where } \eta \text{ is a parameter for roughness}$$

$$\epsilon = \frac{c}{a + c}, \text{ where } c \text{ is the half-width of the surface cavity}$$

$$n = \frac{1}{\pi a^2} = \text{number of contacts per unit area}$$

$$k_1, k_2 = \text{thermal conductivity of nodes 1 and 2}$$

$$\delta_1, \delta_2 = \text{root mean square surface roughness of nodes 1 and 2}$$

$$A = \text{total node interface area}$$

$$a = \text{radius of a contacting point}$$

For the radiative interchanges in an enclosure including both specular and diffuse surfaces, a radiative property such as a reflectance, ρ , can be divided into a diffuse and a specular component as well as shortwave and long-wave bands.

That is, for shortwave bands:

$$\rho_s = \rho_s^d + \rho_s^s$$

and for longwave bands:

$$\rho_l = \rho_l^d + \rho_l^s$$

where the superscripts, d and s, signify the diffuse and specular, and the subscripts the short and longwave bands. For a semi-transparent or an absorbing medium, the radiative properties have the following relationship among the reflectance ρ , emittance or absorptance ϵ , and transmittance, τ .

$$\epsilon_s + \rho_s + \tau_s = 1 \quad \text{for shortwave}$$

$$\epsilon_l + \rho_l + \tau_l = 1 \quad \text{for longwave}$$

The energy fluxes incident upon and leaving a typical semi-transparent surface of the enclosure are described in Figure 8. Consider the i^{th} inside surface area, A_i , of the enclosure which neighbors with other enclosures. The quantities E_i and M_i^* are the rates of incoming and outgoing radiant energy while considering diffuse and specular reflectance of the surface per unit inside area, respectively. The quantity M_i^* does not include the rate of transmitted incident radiant energy from the other neighboring enclosures. The quantity \tilde{E}_i is the rate of incoming radiant energy to the non-opaque boundary surface from the neighboring enclosure. The quantity ϕ_i is the energy flux supplied by some external means to the surface to make up for the net radiative loss and thereby maintain the specified surface temperature. A heat balance at the surface provides the relation

$$\phi_i = E_i - M_i \tag{7}$$

A second equation results from the fact that the energy flux leaving the

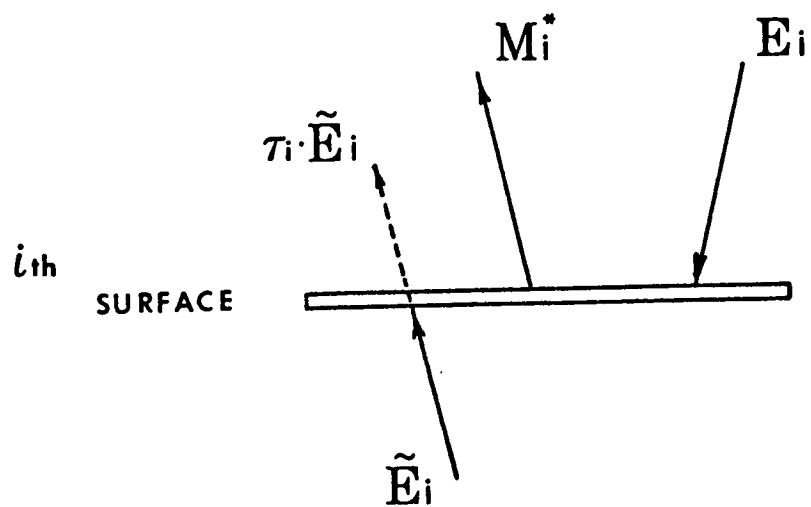


FIGURE 8. Radiation Flux Interchange

surface having diffuse and specular characteristics is composed of directly emitted plus reflected energy. This gives

$$M_i^* = \epsilon_i \sigma T_i^4 + \rho_i^d E_i + \rho_i^s E_i \quad (8)$$

Equation 8 is called an opaque radiosity. The total radiosity, considering the rate of transmitted radiant energy which is originally coming from the neighboring enclosure, is composed of the opaque radiosity M_i^* plus the energy flux transmitted through the surface with transmissivity, τ_i . This gives

$$M_i = M_i^* + \tau_i \tilde{E}_i \quad (9)$$

The diffuse total radiosity considering only a diffuse factor in Eq. 9 and excluding the specular reflection term is described as the following

$$M_i^d = \epsilon_i \sigma T_i^4 + \rho_i^d E_i + \tau_i \tilde{E}_i \quad (10)$$

The contribution to E_i from the other diffusely reflecting surfaces in an enclosure can be represented in accordance with the relationship defined by the configuration factors in an enclosure with diffuse surfaces, except that now the configuration factors are replaced by the exchange factors, or

$$E_i = \sum_j \bar{F}_{ij} M_j^d \quad (11)$$

where \bar{F}_{ij} is the exchange factor defined between i element and j element through j 's images on the element i . Equations 7 through 11 can be represented by the short and longwaves, respectively.

To obtain the heat balance (Eq. 7) as functions of surface temperatures in an enclosure and the irradiance, or incident radiation, from the target source at the field-of-view limiter of the radiometer models, first Eq. 9 is substituted into Eq. 7, yielding

$$\phi_i = (1 - \rho_i^d - \rho_i^s) E_i - \epsilon_i \Omega_i - \tau_i \tilde{E}_i$$

in which $\rho_i^d + \rho_i^s = \rho_i$ where ρ_i is the total reflectance of the surface and $\Omega = \sigma T^4$.

Thus

$$\phi_i = (1 - \rho_i) E_i - \epsilon_i \Omega_i - \tau_i \tilde{E}_i \quad (12)$$

Then, combining Eqs. 10 and 11 and rearranging, yields the following equation

$$E_i = \sum_j (\delta_{ij} - \bar{F}_{ij} \rho_i^d)^{-1} \bar{F}_{ij} (\epsilon_j \Omega_j + \tau_j \tilde{E}_j). \quad (13)$$

For brevity, we introduce new variables from Eq. 13, that is

$$A_{ij} = (\delta_{ij} - \bar{F}_{ij} \rho_j^d)^{-1} \bar{F}_{ij} \epsilon_j. \quad (14)$$

$$C_{ij} = (\delta_{ij} - \bar{F}_{ij} \rho_j^d)^{-1} \bar{F}_{ij} \tau_j. \quad (15)$$

Thus, Eq. 12 with the above constants can be written

$$E_i = \sum_j A_{ij} \Omega_j + C_{ij} \tilde{E}_j \quad (16)$$

Equation 16 has an important role to couple an enclosure with the neighboring enclosures through a semi-transparent or transparent boundary surface.

The above relation (Eq. 16) can be used for the cases when an enclosure is partly or totally surrounded by the neighboring enclosures and also when part or all of the surfaces of the enclosure are non-opaque, i.e., semi-transparent or transparent. To include all the configurations of the enclosures and to couple them together through interactive boundary surfaces, the above relation (Eq. 16) can be generalized in the matrix form, that is,

$$E = \bar{A} \Omega + \bar{C} \tilde{E} \quad (17)$$

where \bar{A} and \bar{C} denote Eqs. 14 and 15, respectively. The bars above the notation A and C are to distinguish Eq. 14 and 15 from the notation for the area and the conduction constant of governing equation to be discussed in the Section 2.1.4.

2.1.3 Coupling Enclosures

In the above Eq. 17, \tilde{E} can be directly related to E by defining the geometrical coupling factor. The relationship between E and \tilde{E} is apparently dependent on the geometrical formation of enclosures through the interactive (non-opaque) surfaces as in the following example. Consider the system shown in Fig. 9 which consists of nine enclosures and interacts at the boundaries with the incident radiation from the environment. The incoming radiation E can be written

$$E = \begin{bmatrix} E^1 \\ E^2 \\ E^3 \\ \vdots \\ E^m \end{bmatrix} \quad (18)$$

where the superscript m denotes the number of enclosures in the prescribed system. Each enclosure of the system has a certain number of interacting surfaces. In such a case, the incident radiation upon the surfaces of the i^{th} enclosure can be described by

$$E^i = \begin{bmatrix} E_1^i \\ E_2^i \\ E_3^i \\ \vdots \\ E_n^i \end{bmatrix} \quad \text{and } 1 \leq i \leq m \quad (19)$$

where the subscript n denotes the number of a surface in an enclosure of the system. As shown in Fig. 9, E_4^1 and E_2^2 define the incident radiation on an interactive boundary surface between the first and second enclosures. The subscripts 4 and 2 here, respectively denote the surface numbers as seen or counted from each enclosure. These surfaces designated by 4 and 2 in the enclosure 1 and 2, respectively, are supposed to be the interactive boundary surface co-owned by enclosures 1 and 2. On the contrary, incident radiation \tilde{E} coming from the opposite, or neighboring, enclosure upon the boundary surface is defined by \tilde{E}_2^2 and \tilde{E}_4^1 , respectively. Accordingly, observing enclosures 1 and 2 with respect to the incident radiation on the interactive boundary surface, we find the following coincidences:

$$E_4^1 = \tilde{E}_2^2 \quad \text{and} \quad E_2^2 = \tilde{E}_4^1 \quad . \quad (20)$$

Equation 20 tells us that the incident radiation E_4^1 , upon the interactive boundary surface of the first enclosure is virtually regarded as the opposite of incident radiation \tilde{E}_2^2 , on the back surface of the same boundary, whether the second enclosure is considered to be formulated or vice versa. This coincidence on each interactive boundary surface is uniquely determined by the formation of the enclosure in a given system.

To generalize the above relationship, Eq. 20, for the enclosure which has interactive boundaries and also system boundaries, let us consider the first enclosure shown in Figure 9. The first enclosure has not only two interactive boundaries at 3rd and 4th surfaces, but also two system boundaries at 1st and 2nd surfaces (Fig. 9). Extending the above relationship (Eq. 20) to include the system boundaries and describing it in a matrix form, we have

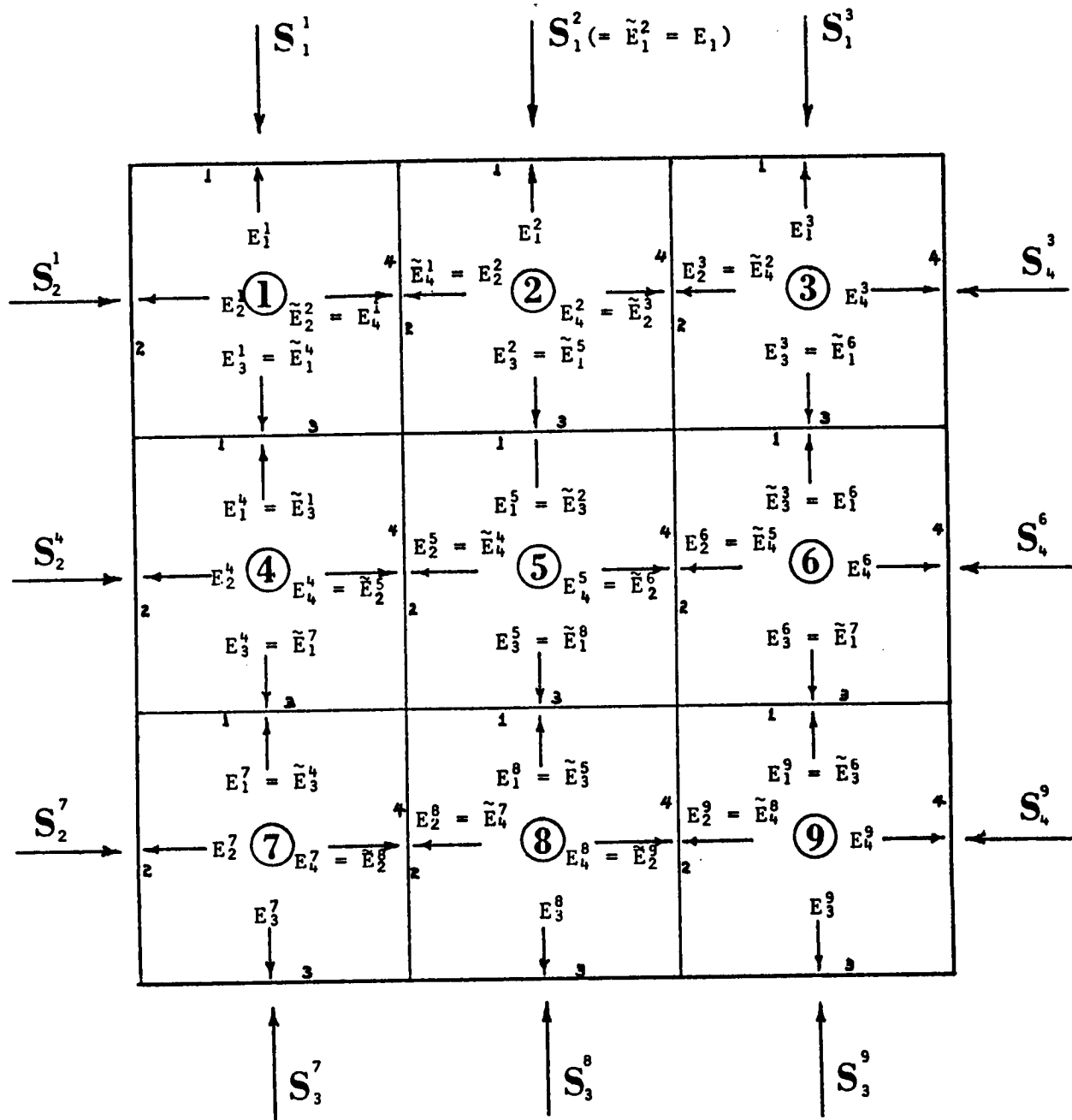


FIGURE 9. Interactive Enclosure Model

$$E^1 = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix} E^2 + \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} E^4 + \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} S \quad (21)$$

Likewise, the other enclosures have the same expression as depicted for the first enclosure. Thus, introducing a new variable D for the enclosure coupling factor, the relationship (Eq. 21) can be written

$$\tilde{E} = DE + FS \quad (22)$$

where D is a matrix representing the enclosure coupling factors through the interactive boundaries, F is a matrix representing the system boundary conditions imposed by surroundings, and S is a column vector of the heat fluxes arriving at the system boundaries. Now substituting Eq. 22 into Eq. 17 and collecting the same variables, we have, with I being an Identity matrix

$$E = (I - \bar{C}D)^{-1} \bar{A}\Omega + (I - \bar{C}D)^{-1} \bar{C}FS \quad (23)$$

Equation 23 illustrates that the incident radiation E upon a surface of the enclosure accounts for all the effects of radiation from the surfaces with different temperatures not only in its own enclosure, but also in neighboring enclosures, and the heat flux at the system boundaries.

The energy balance at a node or a surface in the enclosure due to the incident radiation is then described by inserting Eqs. 22 and 23 into Eq. 12. Then, by eliminating E and \tilde{E} from Eq. 12, the radiative energy balance is obtained as only functions of the temperatures of enclosure surfaces and the irradiance from the target source or any other kinds of heat flux which interacts with the system boundaries. That is,

$$\phi = \begin{bmatrix} (I - \rho - \tau D)(I - \bar{C}D)^{-1} \bar{A} - \epsilon & \Omega \\ (I - \rho - \tau D)(I - \bar{C}D)^{-1} \bar{C} - \tau & FS \end{bmatrix} \quad (24)$$

When the radiative energy balance described by Eq. 24 is applied to the enclosures obtained by properly dividing the geometry of the system, the number of the radiative energy balance equations, ϕ , obtained from Eq. 24 is more than the number of nodal points because consideration is given twice to the interactive boundary between two enclosures. Accordingly, in the multiple-enclosure system the radiative energy balance is calculated for each surface of enclosures. Then, for the overlapped boundary surface, the radiative energy balances from both enclosures are added together. The radiative energy balance (Eq. 24) is then plugged into the energy equation (Eq. 3) of the system.

The enclosure coupling technique developed here has many potential applications and advantages over the conventional methods currently used.

Among the advantages, this technique can be used for any kind of geometry by only setting up enclosures via dividing the geometry and defining the D and F matrices in Eq. 22. The enclosure space with irregular geometry can be properly divided into many smaller regular sub-spaces to adopt the enclosure coupling technique. For an enclosure with complicated geometry, the division of the space geometry into well-defined sub-spaces facilitates the computation of the configuration factor or the exchange factor, especially for an enclosure with specular surfaces which is totally dependent on the geometry.

This technique can also be extended to treat radiation interactions including the scattering and absorption with optically thin or thick participating media in an enclosure. Even though there are temperature and density gradients in participating media, by dividing the enclosure space into sub-spaces which are small enough to assume the sub-spaces to be isothermal and isotropic, the

enclosure theory can be directly applied for the system without loss of generality.

In short, the advantages and potential applications of the enclosure coupling technique are listed below.

ADVANTAGES:

1. It minimizes the effort required to define the exchange factors among the surfaces in an enclosure with complex geometry, and so may be able to decrease a degree of possible approximation in the computation of exchange factors.
2. It may take the wavelength characteristics of materials or participating media into account in a simple manner.
3. It can be easily coupled with the thermal model of the system so that the material and thermal properties are considered in the coupled dynamic system without approximations and assumptions such as those in the boundary definitions.
4. It enables the system to deal with any imposed initial conditions and prescribed boundary conditions.
5. It may enhance the accuracy in the computation of a geometrically partitioned conductive and radiative model.
6. As a consequence, this technique can avoid the length of time for computation and a cumbersome manipulation which is often revealed in the other methods such as ray tracing or Monte Carlo methods.

APPLICATIONS:

1. It can be used for any enclosure geometry.
2. It can be used for either the enclosure with a vacuum space or the enclosure with participating media (solid, liquid or gas).

3. It can be applied to any optical system's analyses and design regardless of the directional, spectral, and/or temperature dependences of its enclosure spaces.
4. Applications of the enclosure coupling technique to optical systems may vary from a typical radiometer to any waveguide system.

2.1.4 Governing Equation

The governing equation which describes the performance of a radiometer that has a dynamic response to an incoming radiance can be obtained by substituting Eqs. 4 and 24 into Eq. 3 and collecting terms. Then we can separate the incoming radiation flux from the radiation balance term. The equation can be written in a general form to describe the energy balance in a node as follows

$$\dot{T} = AT + DT^4 + BE + B_Q \dot{Q} + b + e \quad (25)$$

where

$$A = \frac{1}{MR}$$

$$D = \frac{\sigma A_r}{M} \left[(I - \rho - D)(I - \bar{C}D)^{-1} \bar{A} - \epsilon \right]$$

$$B = \frac{A_r}{M} \left[(I - \rho - D)(I - \bar{C}D)^{-1} \bar{C} - \tau F \right]$$

$$B_Q = \frac{A_c}{M}$$

b = a bias term

e = a noise term

R = the thermal resistance between nodes (see Eq. 5)

$$M = V\gamma C$$

A_r = the area for the radiative interaction

A_c = the node area covered by the heat sink heater

Equation 25 represents a general form that consists of a set of equations for a multinode system. Considering only the thermal model for the nonscanner total radiometer which is divided into 8 nodes, we have a set of 8 equations which represent the energy balance in each node separately. The inclusion of the electronic thermal control system equations will increase the total system by the number of the transfer functions. The constants of Eq. 25 for the nonscanner, the scanner, and the solar monitor are described in Appendices C, D, and E.

2.1.5 Electronic Control System Model for the Nonscanner

The block diagram shown in Fig. 10 illustrates the nonscanner ACR control system model. The control electronics transfer function is of the form (cf. ERBE Instruments, Monthly Status Report - Vol. 1, Nov. 1980, unpublished)

$$F_E(s) = \frac{K_E(\tau_1 s + 1)(\tau_2 s + 1)}{s(\tau_3 s + 1)(\tau_4 s + 1)} \quad (26)$$

To obtain the differential equations describing the electronics transfer function, we define the state variables T_{10} , T_{11} , T_{12} , as follows

$$\dot{T}_{10} = V_B K_a K_E \left[\frac{R_r(T_3)}{R_a(T_1) + R_r(T_3)} - B_E \right] + K_E n_E \quad (27)$$

Note $B_E = \frac{R_4}{R_3 + R_4}$ is an input constant, where R_3 and R_4 are the bridge resistances.

$$R_i(T_j) = A_i + B_i T_j + C_i T_j^2, \quad i = a, r; \quad j = 1, 3 \quad (28)$$

$$\dot{T}_{11} = \frac{1}{\tau_3} (-T_{11} + T_{10} + \tau_1 \dot{T}_{10}) \quad (29)$$

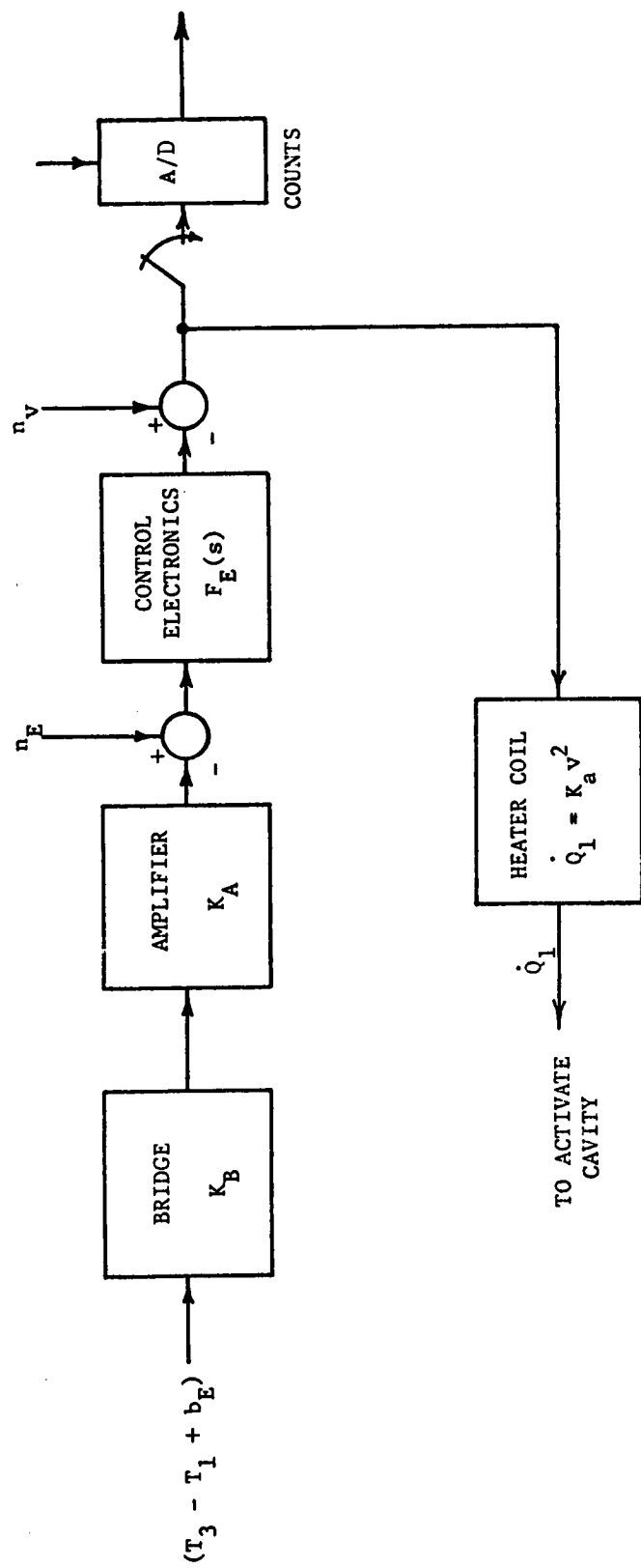


FIGURE 10. Active Cavity Radiometer (ACR) Control System Model

Substituting Eq. 27 into Eq. 29 gives

$$\dot{T}_{11} = \frac{1}{\tau_3} (-T_{11} + T_{10}) + \frac{\tau_1}{\tau_3} V_B K_a K_E \left[\frac{R_r(T_3)}{R_a(T_1) + R_r(T_3)} \right] + \frac{1}{\tau_3} K_E n_E \quad (30)$$

$$\dot{T}_{12} = \frac{1}{\tau_3} (-T_{12} + T_{11} + \tau_2 \dot{T}_{11}) \quad (31)$$

For full expression, substitute \dot{T}_{11} in Eq. 30 into Eq. 31.

Note that T_{12} is now the output of the electronics, so that the voltage output is

$$v = T_{12} + n_v \quad (32)$$

To convert to counts, C, use the following relation:

$$C = K_c v = K_c T_{12} + K_c n_v, \quad K_c = 819.1 \text{ counts/volt.}$$

The heat flow provided to the active cavity to maintain its temperature at a constant level is given by

$$\dot{Q}_1 = K_a v^2 = K_a T_{12}^2 + 2K_a T_{12} n_v + K_a n_v^2 \quad (33)$$

This expression for \dot{Q}_1 brings to light the major effects that the noise terms introduce into the operation of the radiometer. Note that due to the use of square feedback, the linear noise terms acquire a gain of $2K_a T_{12}$, while a quadratic noise term (having non-zero mean) is introduced. Coils which are wrapped around the active and reference cones generate heat and are expressed by the following terms.

$$\dot{Q}_{J_1} = V_B^2 \frac{R_a(T_1)}{[R_a(T_1) + R_r(T_3)]^2}$$

$$\dot{Q}_{J_3} = V_B^2 \frac{R_r(T_3)}{[R_a(T_1) + R_r(T_3)]^2}$$

To obtain the closed-loop system, the expression Eq. 33 is simply substituted into the equation governing the active cavity temperature.

$$\begin{aligned} \dot{T}_1 = & A_{12}(T_2 - T_1) + \sum_k D_{1k} T_k^4 + B_1 E + \frac{k_a}{M_1} T_{11}^2 + \frac{2K_a T_{11} n_v}{M_1} \\ & + \frac{K_a n_v^2}{M_1} + V_B^2 \frac{R_a(T_1)}{[R_a(T_1) + R_r(T_3)]^2} \end{aligned} \quad (34)$$

The block diagram shown in Fig. 11 illustrates the non-scanner ACR heat sink heater control system model.

The control electronics transfer function can be approximated in the form

$$F_{HE}(s) = K_{HE} \frac{(\tau_H s + 1)}{s} \quad (35)$$

The differential equations describing the heat sink heater controller can be expressed as

$$\dot{T}_{13} = K_{H13} K_{HE} (T_H - T_2) + K_{HE} n_H \quad (36)$$

$$v_H = T_{13} + \tau_H K_{HE} K_{H13} (T_H - T_2) + \tau_H K_{HE} n_H \quad (37)$$

$$\dot{Q}_2 = K_H v_H^2 \quad (38)$$

In Eq. 25 the heat source term, \dot{Q} , is substituted by Eq. 38 for the aluminum and copper heat sinks.

2.1.6 Simulation of the Nonscanner Model

The simulation of the ERBE nonscanner sensors is divided into two major parts; the first determines the coefficients of the system, the second solves the systems over time and plots the results. Figure 12 is a flow diagram

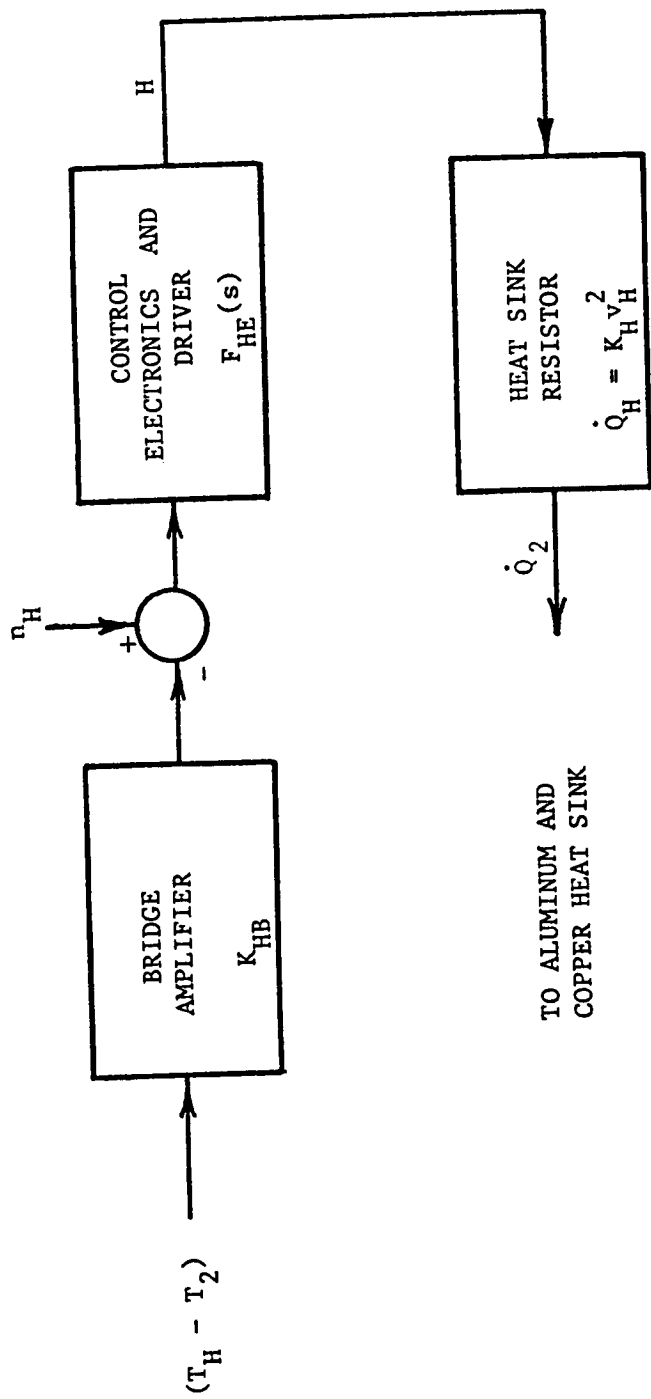


FIGURE 11. Nonscanner Heat Sink Heater Control System Model

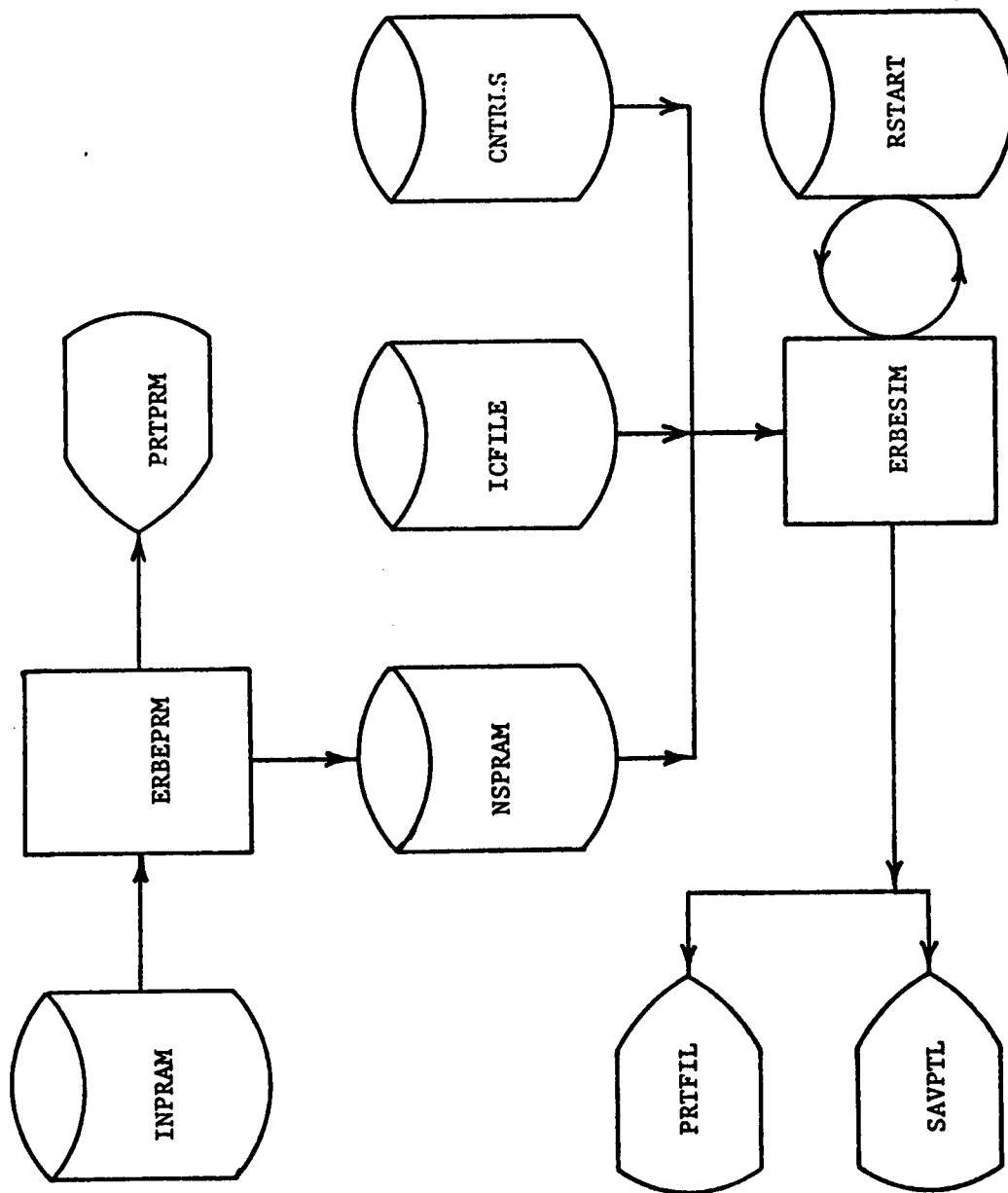


FIGURE 12. Flow Diagram Indicating File and Program Usage in Simulation

indicating file and program usage. The equation describing the change of temperature in a given node at time i is

$$\dot{T}(i) = AT(i) + DT^4(i) + BE + B_Q \dot{Q} + B_n n \quad (39)$$

Where A is the conductance coefficient matrix, D is the radiative coefficient matrix, B is the irradiance source vector and B_Q is a vector of coefficients which describe heating contributions from the control system. $T(i)$ represents either the temperature change for a thermal node or a voltage change for an electronic node. The general method of solution is a fourth order Runge-Kutta algorithm, all of which is discussed in great detail in the following sections.

2.1.6.A Parameter and Coefficient Determination

The coefficient matrices and vectors are determined in the first program in the sequence ERBEPRM. The data inputted to this is determined from material specifications and engineering drawings supplied by the manufacturer. The code is partitioned into 5 major sections, not including the program input/output. The first of these is the subroutine "COMPUTA". This routine takes the heat capacities and node to node thermal resistances and returns the conductance coefficient matrix. The program then enters the routine ADDELC which is designed to modify the conductance matrix in such a way as to allow the ACR heater controls to use the same equations in the simulation as the node temperatures. The output from this routine is an enlarged conductance matrix and several electronic vectors. Once these two routines have been computed COMPUTH calculates the radiative exchange coefficient between surfaces for each cavity. Due to the linking of cavities a and b , as defined in Appendix C, dome filter or fictitious surfaces, the radiative exchange coefficients for these cavities

are stored for later use. The results of COMPUTH are then added together to give the total radiative exchange coefficient matrix in COMPUTD, COMPUTB and CMPUTBP. The results are then issued to two files, one formatted for inspection and one binary for use by the simulation program.

2.1.6.B Instrument Simulation

The program consists of four logical sections, an Input, the Integration, the intermediate Output and the Plotting section. The actual solution to the system of equations that describes each instrument is accomplished by the program ERBESIM. The program steps through a determined time interval using an evenly spaced discrete time step and solves each system using a Fourth Order Runge-Kutta scheme to evaluate the system at the next step. Each system is sampled at another rate controlled by the user until 20 samplings are taken or the end of simulated time is reached. The results are then stored on magnetic medium for use by the plotting routines. The final product is the sensor's response described by plots and tables of the modeled response of the instrument, i.e., time-dependent temperatures of the nodes, instrument response (in counts), electronic voltages (in volts) and input source irradiance (in W/m^2).

The first section of the program reads the coefficient matrices and vectors, initial conditions from binary files, and the simulation control variables from a namelist file. The Namelist file contains variables for controlling simulation time step size, sampling rates and the combination of instruments and radiation profiles, etc. which are to be simulated. The simulation is

capable of allowing the user to choose either a sawtooth, Fourier series, step function, constant or ramp radiance profile.

The Runge-Kutta method used was developed by Fehlberg and solves the system

$$Y_{i,m+1} = Y_{i,m} + h \sum_{j=0}^5 A_j K_{ji}, \quad i = 1, 2, 3, \dots \text{number of nodes}$$

where

$$K_{ji} = f(t_m + C_j h, y_m + h \sum_{\ell=0}^i b_{j\ell} K_{\ell j})$$

where f is the function describing $T(i)$ from Eq. 39 as functions of time and temperature, a_j , $b_{j\ell}$, and C_j are constants determined by Fehlberg for solving first order non-stiff systems of ordinary differential equations.

2.1.7 Count Conversion for Nonscanner

2.1.7.A Nonscanner Total Channel

In the active cavity radiometers, the electronics control the amount of heat added to the active cavity to maintain the temperature of the cavity about half a degree above the reference cavity. The control of the heat input to the active cavity is inversely proportional to the incident radiation entering through the field of view and precision aperture from a given source. In other words, the control system provides more heat input for the weak incident radiation and less heat input for the strong incident radiation. Thus, the voltage driving the active cavity heater provides the sensor output proportional to the incident radiation. The voltage reading, after A/D conversion, provides the sensor output in counts. The sensor output in counts is then converted into engineering units. The algorithms for the count conversion are of the gain/offset type, and are derived based on the sampled instrument data, e.g., sensor response (in counts), HK temperatures, etc. and the instruments radiative and conductive interactions.

The following is the count conversion equation for NS total channels.

$$E(t - \tau) = A_v v^2(t) + A_H [T_H(t) - T_{H_o}] + A_A [T_A(t) - T_{A_o}] + A_F [T_F(t) - T_{F_o}] + B \quad (40)$$

The constants in Eq. 40 are defined as follows

$$A_v = - \frac{B_a K_a}{B_1} \quad (41)$$

$$A_H = - \frac{4}{B_1} \sum_{k \neq A, F} D_{1k} T_{k_o}^3 \quad (42)$$

$$A_A = - 4 \frac{D_{18}}{B_1} T_{8_o}^3 \quad (43)$$

$$A_F = - 4 \frac{D_{16}}{B_1} T_{6_o}^3 \quad (44)$$

$$B = \frac{A_{12}}{B_1} b_E - \sum_{k=1}^n \frac{D_{1k}}{B_1} T_{k_o}^4 - \frac{v_B^2}{4R_B M_1 B_1} \quad (45)$$

where

V = instrument output (in volts)

T_H = measured heat sink temperature ($^{\circ}K$) at time t

T_A = measured precision aperture temperature ($^{\circ}K$) at time t

T_F = measured FOV limiter temperature ($^{\circ}K$) at time t

The remaining quantities, i.e., B_a , K_a , B_1 , T , D_{1k} , A_{12} and b_E are constants which are specified for each instrument. These constants for the count conversion equations are determined for the simulation study of the instrument model using ground and inflight calibration data. These parameters or constants

may be altered at various points in time, usually after a flight calibration, in order to reflect any changes in the instrument characteristics.

2.1.7.B Nonscanner Filtered Channel for Shortwave

For the filtered channels, the count conversion has a set of two equations for short and longwaves as follows:

$$\alpha_{11} \hat{E}_s + \alpha_{12} \Delta T_9 = \beta_1 \quad (46)$$

$$\alpha_{21} \hat{E}_s + \alpha_{22} \Delta T_9 = \beta_2 \quad (47)$$

The coefficients of the terms in the above equations are given by the following equations,

$$\alpha_{11} = B_{1s} - B_{1l} \quad (48)$$

$$\alpha_{12} = 4D_{1,9} T_{9_o}^3 \quad (49)$$

$$\alpha_{21} = B_{9l} - B_{9s} \quad (50)$$

$$\alpha_{22} = A_{9,4} + A_{9,8} - 4D_{9,9} T_{9_o}^3 \quad (51)$$

$$\beta_1 = (A_{12} b_E - \sum_{k=1}^{n_s} D_{11} T_{k_o}^4 - \frac{V_B^2}{4R_B M_1}) - 4(\sum_{k \neq 8,6}^{n_s} D_{1k} T_{k_o}^3 - \frac{V_B^2 \beta_r}{16R_B^2 M_1})$$

$$* (T_H - T_{H_o}) - 4D_{18} T_{8_o}^3 (T_A - T_{A_o}) - 4D_{16} T_{6_o}^3 (T_F - T_{F_o})$$

$$- B_a K_a v_s^2 - B_{ie} \hat{E} \quad (52)$$

$$\beta_1 = \sum_{k=1}^{n_s} D_{9k} T_{k_o}^3 + A_{98}(T_A - T_{9_o}) + A_{d8}(T_H - T_{9_o}) + 4(\sum_{k \neq 4,8,9}^{n_s} D_{9k} T_{k_o}^3)$$

$$* (T_H - T_{H_o}) + 4D_{98} T_{8_o}^3 (T_A - T_{A_o}) - 4D_{16} T_{6_o}^3 (T_F - T_{F_o}) + B_{1l} E \quad (53)$$

The filtered channel count conversion is obtained from the solution of two simultaneous equations (Eqs. 46 and 47). Thus the solution to the equations is given by

$$\hat{E}_s = \frac{1}{\alpha_{11} \alpha_{22} - \alpha_{12} \alpha_{21}} (\alpha_{22} \beta_1 - \alpha_{12} \beta_2) \quad (54)$$

$$T_9 = \frac{1}{\alpha_{11} \alpha_{22} - \alpha_{12} \alpha_{21}} (-\alpha_{21} \beta_1 + \alpha_{11} \beta_2) \quad (55)$$

The longwave incident radiation estimate is then defined by subtracting the shortwave incident radiation from the total incident radiation estimate. That is,

$$\hat{E}_\ell = \hat{E} - \hat{E}_s \quad (56)$$

The constants in Eqs. 46 - 56 are

v_s = shortwave channel output (in volts)

\hat{E}_s = shortwave incident radiation estimate (W/m^2)

\hat{E}_ℓ = longwave incident radiation estimate (W/m^2)

\hat{E} = total incident radiation estimate (W/m^2)

The temperatures of the heat sink, aperture and FOV limiter, T_H , T_A , and T_F , are measured as before in total NS channel.

2.1.8 Solar Monitor Model Analysis

The solar monitor is different from the nonscanner in size, shape and design feature as shown in Figure 6. It has a barrel structure ~ 9.5 cm long between the primary and secondary apertures. The tail of the core body of the instrument is attached to the housing and includes the primary and secondary apertures, barrel, heat sink and active and reference cavities. This instrument has a shutter located in front of the secondary aperture. The shutter operates on a 64 second "on and off" cycle in which it is open for 32 seconds and then closed for 32 seconds.

Since this instrument deals with a strong intensity of the incident radiant energy and does not have the heat sink temperature controller, the heat sink has a larger heat capacity for damping out the effect of large input temperature gradients.

However, it is desirable to check whether there is any transient effect due to the periodical on and off mode of the shutter, the strong intensity of solar radiation, and the excessively long barrel geometry. In other words, it is desirable to check whether the steady-state approximation is appropriate to develop the count conversion algorithm for the solar monitor. Thus, we introduce two simplified geometries here to check how fast the temperature reaches a stable condition, as every 32 seconds the solar radiation is viewed.

First, let us assume that the shutter has a rectangular shape of thin aluminum plate and the shutter is partly exposed to the incident solar radiation for a 32 second interval of the full 64 second cycle and has a radiation interaction with the isothermal neighboring components at any time. Figure 13 shows

the shutter model.

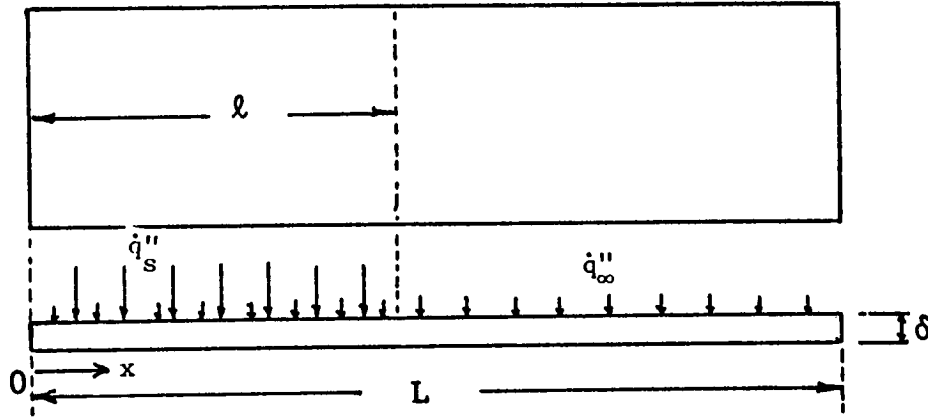


FIGURE 13. The Shutter Model

It is assumed that the left end of the shutter is insulated and the right end is attached to a temperature reservoir which has a large heat capacity. Then the energy balance equation, the initial and boundary conditions governing the above model can be written as

$$\text{G.E.} \quad \rho C \frac{\partial T}{\partial \tau} = \frac{k}{\rho C} \frac{\partial^2 T}{\partial x^2} + \frac{\dot{q}''}{\delta} \quad (57)$$

$$\text{I.C.} \quad T(\tau = 0) = T_o$$

$$\begin{aligned} \text{B.C.} \quad T_x(x = 0, \tau) &= 0 \\ T_x(x = L, \tau) &= T_b \end{aligned}$$

The above equation with Neumann and Dirichlet boundary conditions can be easily solved by introducing a new variable and separating the variables. Thus the solution to the equation is described by

$$\begin{aligned} T(x, \tau) = T_b + 2(T_o - T_b) \sum_{n=0}^{\infty} \frac{(-1)^n}{(\lambda n L)} e^{-a \lambda n^2 \tau} \cos(\lambda n x) \\ + \frac{\dot{q}'' L^2}{2k\delta} \left\{ \left[1 - \left(\frac{x}{L} \right)^2 \right] - 4 \sum_{n=0}^{\infty} \frac{(-1)^n}{(\lambda n L)^3} e^{-a \lambda n^2 \tau} \cos(\lambda n x) \right\} \end{aligned} \quad (58)$$

where T_0 is an initial temperature set by 273.16°K , L the total length of the shutter (6.5 cm), δ the thickness of the shutter (0.2 cm), k the thermal conductivity ($1.557 \text{ w/cm}^\circ\text{K}$), ℓ the exposure length to solar radiation (2.5 cm), ρ the density of the shutter (2.713 g/cm^3), c the specific heat of the shutter ($0.887 \text{ J/g}^\circ\text{K}$), a the thermal wave velocity ($0.647 \text{ cm}^2/\text{sec}$)

$$\lambda_n = \frac{(2n+1)\pi}{2L} \quad n = 0, 1, 2, \dots$$

\dot{q}_s'' is the solar flux (0.142 w/cm^2) $\times 0.2$

$$\dot{q}_\infty'' = F \epsilon \sigma [T_\infty^4 - T^4]$$

$$\dot{q}'' = \dot{q}_s'' u(\ell-x) u(32-\tau) + \dot{q}_s'' u(\ell-x) u(\tau-64) u(96-\tau) \dots + 2 \dot{q}''$$

F the configuration factor between the shutter and the neighboring components (0.85)

ϵ the emissivity of the shutter surface (0.8)

σ = the Stephan-Boltzmann constant

T_∞ = the ambient temperature (use $T_\infty = T_b$)

T_b = the reservoir temperature set by 273°K ,

T = current shutter temperature, but use $T(\tau-1)$

$u(x)$ = the unit step function.

The graphical description of the solution is shown in Figures 14 and 15. Figure 14 shows that the shutter temperature not only responds to the on and off mode of the solar flux, but the peak temperatures as well as the minimum temperatures of each cycle (64 sec) also stay at the same ranges. That is, the time averaged temperature of the shutter does not change significantly. And Figure 14 also shows that the temperatures before reaching minimum and maximum points become stable. Figure 15 shows the temperatures along the length of the shutter after 100 seconds passed.

Secondly, the long barrel which is placed in the housing and is between the primary and secondary apertures is considered by the following assumptions:

- 1) no direct exposure to the solar flux

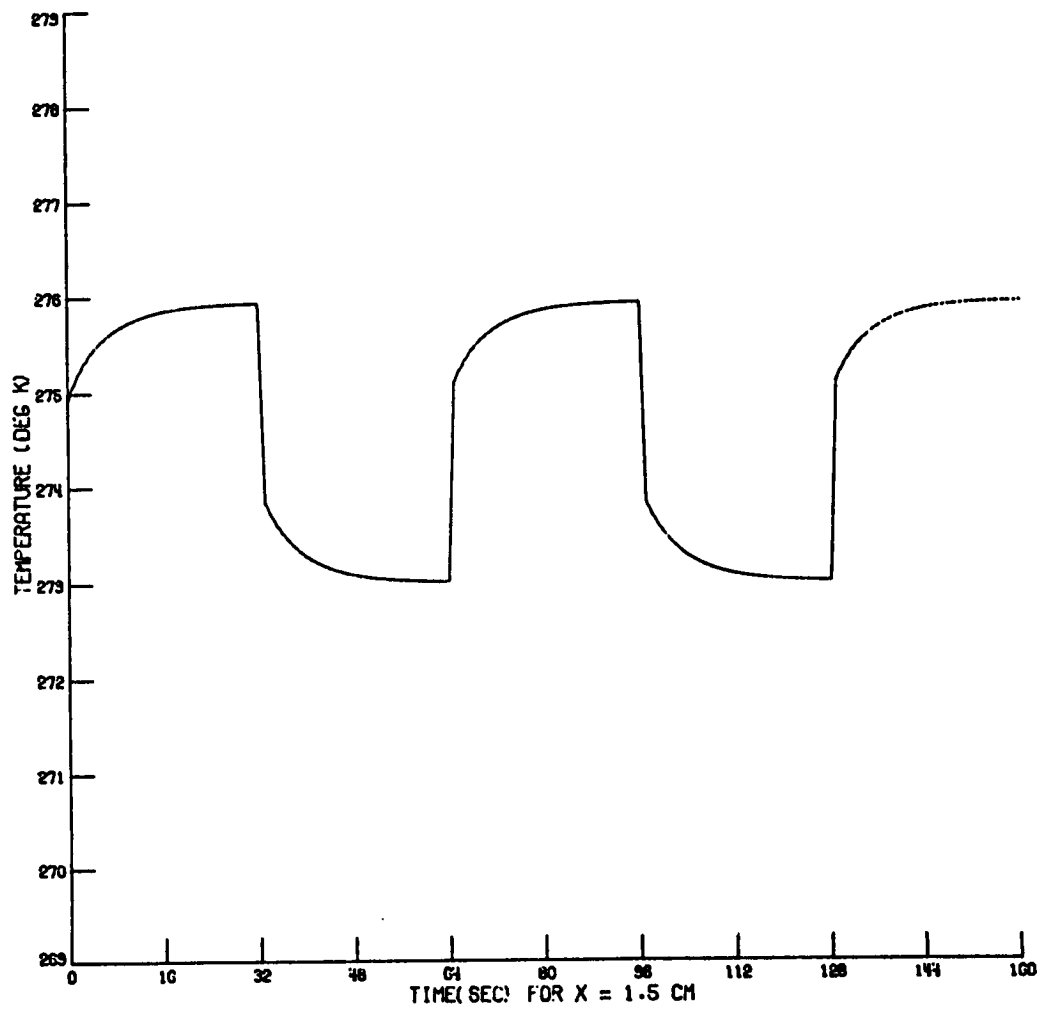


FIGURE 14. SHUTTER TEMPERATURE AT $X = 1.5$ CM WITH 20% ABSORPTION OF SOLAR FLUX $\dot{Q}_s = 0.142$ W/CM².

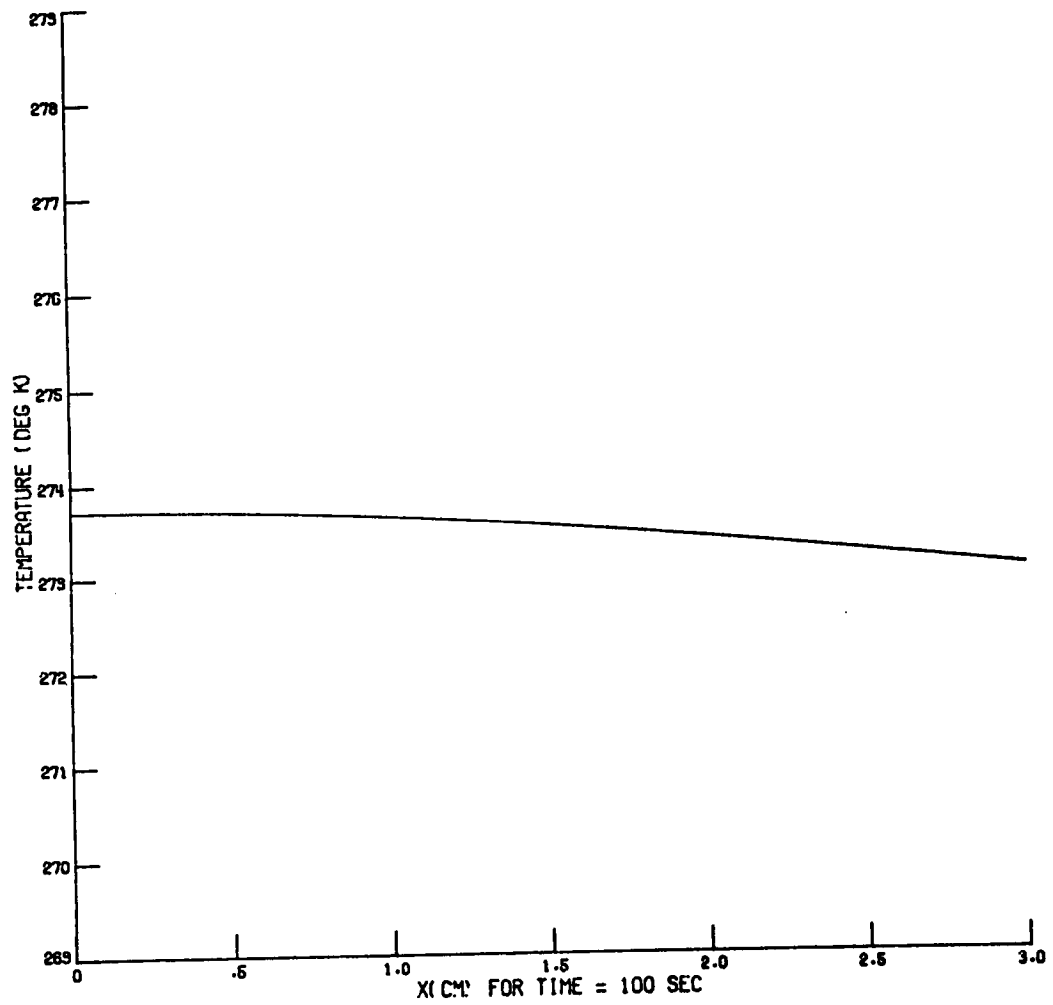


FIGURE 15. SHUTTER TEMPERATURE AT 100 SECONDS WITH 20% ABSORPTION OF SOLAR FLUX $\dot{Q}_s = 0.142 \text{ W/CM}^2$.

- 2) radially uniform temperature since the wall thickness is thin
- 3) no circumferential temperature gradient
- 4) radiation exchanges only at inside and outside walls

The configuration of the model is shown in Figure 16.

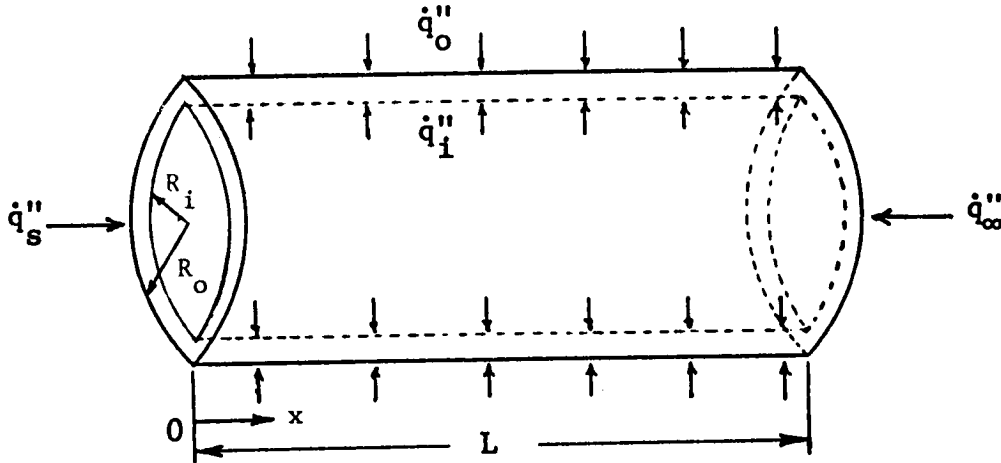


FIGURE 16. A SIMPLIFIED CONFIGURATION OF SOLAR MONITOR BARREL

The equations and boundary conditions governing the thermal conditions imposed upon the above geometry can be written as

$$\text{G.E.} \quad \rho c \frac{\partial T}{\partial \tau} = k \frac{\partial^2 T}{\partial x^2} + \frac{\dot{q}''}{R_o - R_i} \quad (59)$$

$$\text{I.C.} \quad T(\tau=0) = T_b$$

$$\text{B.C.} \quad T_x(x=0) = \dot{q}''_{\infty} \quad (\text{set to be constant})$$

$$T(x=L) = T_b .$$

Then the solution to the equation is obtained as

$$T = T_b + \frac{\dot{q}'' L^2}{2k(R_o - R_i)} \left\{ \left[1 - \left(\frac{x}{L} \right)^2 \right] - \frac{2k(R_o - R_i)}{\dot{q}'' L} \left(1 - \frac{x}{L} \right) \right. \\ \left. - 2 \sum_{n=1}^{\infty} \left[\frac{(-1)^n}{(L\lambda n)^3} - \frac{2k(R_o - R_i)}{\dot{q}'' L} \frac{1}{(L\lambda n)^2} e^{-a\lambda n^2 t} \cos(\lambda n x) \right] \right\} \quad (60)$$

where

$$\dot{q}'' = \dot{q}_i'' + \dot{q}_o''$$

$$\dot{q}_o'' = F_{1-1} \epsilon_1 \alpha [(T^4(x, t-5) - T^4(x, t-1))]]$$

$$\dot{q}_i'' = |\dot{q}_s'' - \dot{q}_\infty''| \epsilon_2 F_{1-2}$$

$$\dot{q}_\infty'' = F_{1-3} \epsilon_3 \sigma [T_{av}^4(\text{in } x) - T^4(x=0, t-1)] [u(64-\tau) u(\tau-32) + u(128-\tau) u(\tau-96) + \dots]$$

$$T_b = 273.16 \text{ } ^\circ\text{K}$$

$$L = 10 \text{ cm}$$

$$k = 1.557 \text{ w/cm } ^\circ\text{K}$$

$$a = 0.647 \text{ cm}^2/\text{sec}$$

$$R_o - R_i = 0.5 \text{ cm}$$

$$\lambda_n = \frac{(2n+1)}{2L} \pi$$

$$\dot{q}_s'' = 0.146 \left[u(32-\tau) + u(\tau-64) u(96-\tau) + \dots \right]$$

$$F_{1-1} = 0.75$$

$$\epsilon_1 = 0.8$$

$$F_{1-2} = 0.4$$

$$\epsilon_2 = 0.08$$

$$F_{1-3} = 0.4$$

$$\epsilon_3 = 0.5$$

The above values are used in both the shutter and barrel cases. These estimates were selected to best represent the solar monitor based on the material and thermal properties of the shutter and barrel. The results of the calculations described in Figures 17 and 18 demonstrate that the temperatures do not change very much with respect to time or in the axial direction of the barrel since they exhibit only about a half degree of temperature change in both cases.

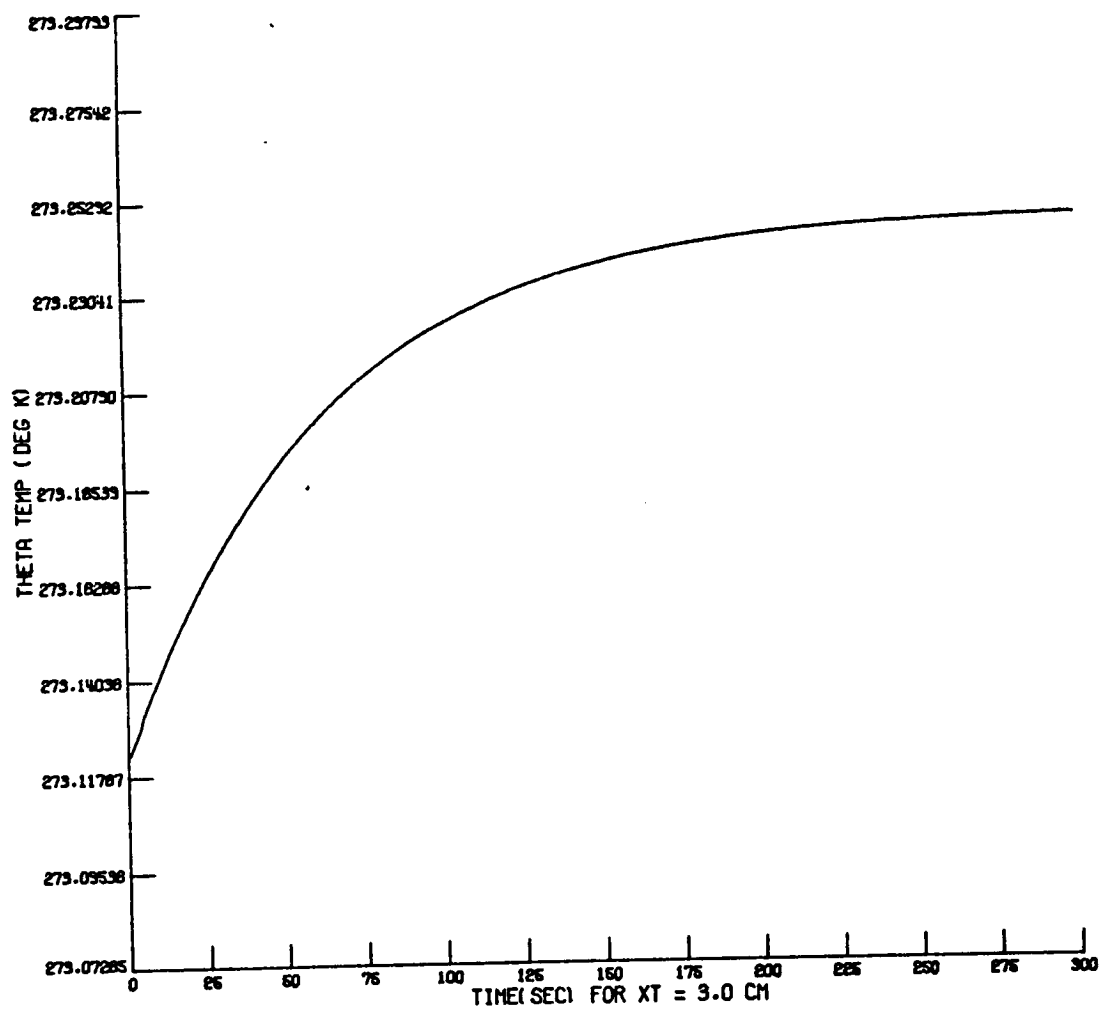


FIGURE 17. BARREL TEMPERATURE WRT TIME AT X=3 CM FROM THE SECONDARY APERTURE.

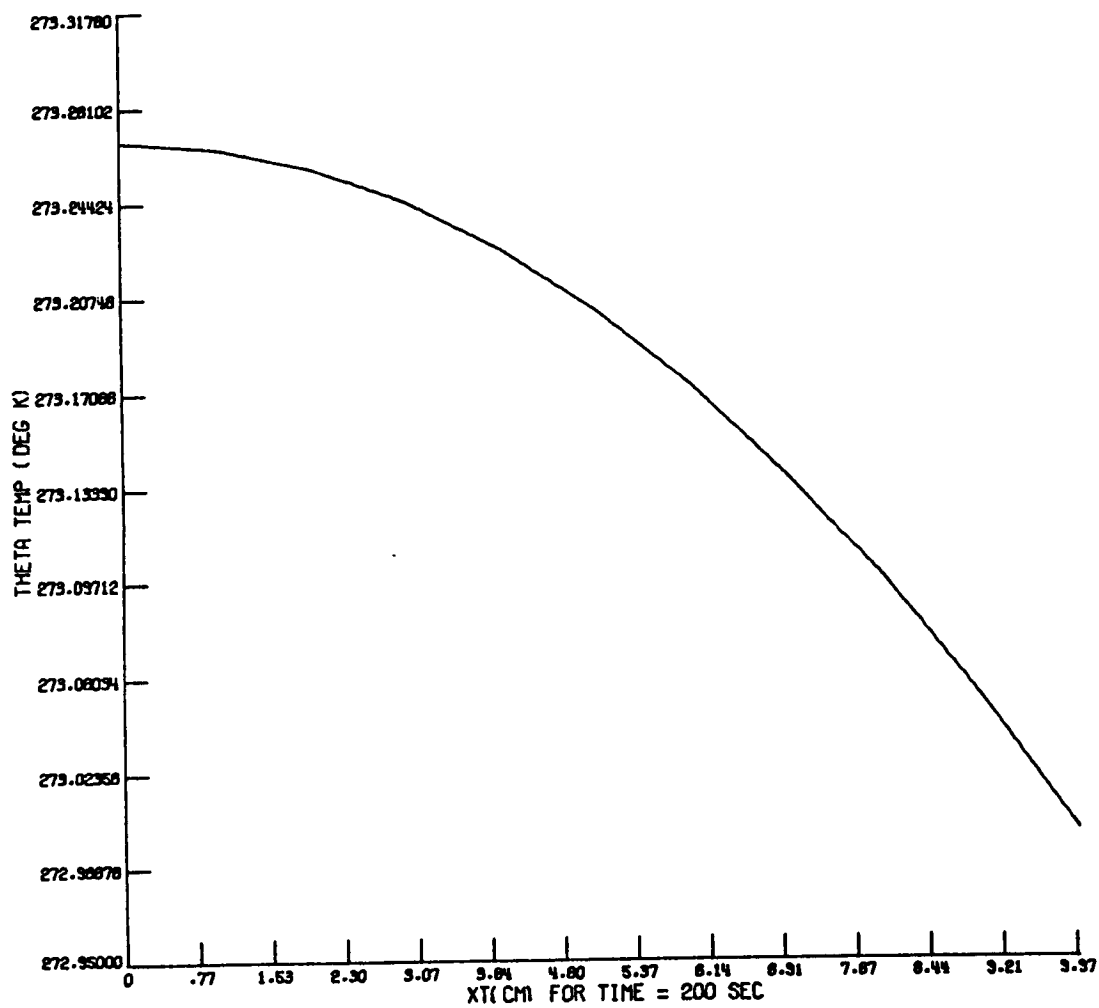


FIGURE 18. BARREL TEMPERATURE ALONG THE AXIAL DIRECTION AT THE TIME OF 200 SECONDS.

Therefore, these results show that the steady-state approximation for the count conversion algorithm does not contain any crucial errors. Therefore, the open and closed modes of the shutter were considered in the algorithm development. Namely, the output responds to the direct radiation which falls into the active cavity according to the shutter's on and off modes. Even though the instrument has a small time constant and hence it is stabilized quickly, within the 32 seconds of a half cycle, the output data which is used for the conversion algorithm has to be picked up at the near-end of a half cycle just before the shutter closes the aperture.

2.1.9 Solor Monitor Count Conversion Algorithm

The approximations made here including the steady-state approximation are the following:

1. The nodal points are selected to be adjacent to where the temperature probes are so that the temperatures measured by the probes may be regarded as the temperature of the nodes.
2. $T_1 = T_2 + b_E$ where b_E is the electronical bias.
3. The temperature of the reference cavity is the same as that of the heat sink, that is, $T_2 = T_3$. Accordingly, $T_{10} = T_{30} + b_E$ where the subscript o signifies the initial temperature.
4. $T_1 - T_{10}$ can be replaced by $T_3 - T_{30}$.

With the above approximations, the equation for count conversion was derived as

$$\hat{E}(t-\tau) = A_v V^2(t) + A_3 [T_3(t) - T_{30}] + A_5 [T_5(t) - T_{50}] + A_7 [T_7(t) - T_{70}] + B \quad (61)$$

where T_3 , T_5 , and T_7 are temperatures in $^{\circ}\text{K}$ which are measured in the instrument. These temperatures are measured and telemetered every 16 seconds. If these temperatures are measured twice during the open period of the shutter, then the second measured temperatures can be used for the count conversion algorithm. Otherwise, as the temperatures are measured arbitrarily without regard to the shutter on-and-off mode, the temperature measured at or near the time before the shutter closure must be used. As well as the temperatures measured, the count has to be picked up at every 40th data point or near the time before the shutter closure. $V(t)$ in the equation is the instrument output in volts and is telemetered at every 0.8 second intervals. The constants in the equations are obtained from the thermal control electronics and conductive and radiative heat transfer, all of which interact with the active cavity and the nodes

which have temperature probes. In the equation,

$$A_v = - \frac{B_a K_a}{B_1} \quad (62)$$

$$A_3 = - \frac{4}{B_1} \left[D_{1-1} T_{10}^3 + D_{1-2} T_{20}^3 + D_{1-3} T_{30}^3 + D_{1-4} T_{40}^3 + D_{1-6} T_{60}^3 \right. \\ \left. + D_{1-18} T_{180}^3 + D_{1-19} T_{190}^3 \right] \quad (63)$$

$$A_5 = - \frac{4}{B_1} \left[D_{1-5} T_{50}^3 + D_{1-6} T_{60}^3 \right] \quad (64)$$

$$A_7 = - \frac{4}{B_1} \left[D_{1-7} T_{70}^3 + D_{1-8} T_{80}^3 + D_{1-9} T_{90}^3 + D_{1-11} T_{110}^3 + D_{1-14} T_{140}^3 \right. \\ \left. + D_{1-15} T_{150}^3 + D_{1-17} T_{170}^3 \right] \quad (65)$$

$$B = \frac{A_{13}}{B_1} b_E - \sum_k \frac{D_{1k}}{B_1} T_{k0}^4 - \frac{B_J K_J V^2}{B_1} - \frac{\epsilon}{B_1}. \quad (66)$$

the quantities such as B_a , B_J , K_a , K_J , V , B_1 , T , D_{1k} , T_{k0} , A_{13} and b_E are the constants to be specified for the instrument.

2.1.10 Scanner Control Electronics

The control electronics for the scanner have a different feature from that of the nonscanner. The difference between them is attributed to the type of sensor in the scanner. The scanner uses thermister bolometers for both active and reference flakes. When the temperature of the thermister varies due to the energy exchange, the current through the thermister varies with the change of the temperature-dependent thermister resistance. In the scanner instrument, the active flake that is attached to the heat sink indirectly responds to the incident radiation from the source through its optical arrangement. The reference flake is also attached to the heat sink and is placed where the space is totally sealed by the surrounding materials such as the heat sink and housing. Thus it is assumed that the reference flake's temperature is equal to that of the heat sink and its housing. Accordingly, the observed difference in resistances between the active and reference flakes, is proportional to the incoming radiation.

Figure 19 is the block diagram for the scanner control electronics and their transfer functions.

The thermister resistance as a function of temperature is given by

$$R_i(T) = R_i(T_o) \exp \beta_i \left(\frac{1}{T_i} - \frac{1}{T_o} \right), \quad i = 1, 3 \quad (67)$$

Using the polynomial expansion of the exponential in Eq. 67, the difference in resistances between the active and reference flakes is approximated as

$$R_1(T_1) - R_3(T_3) = \frac{R_{1o} \beta_1}{T_o^2} T_1 + \frac{R_{3o} \beta_3}{T_o^2} T_3 + \frac{R_{1o} \beta_1 - R_{3o} \beta_3}{T_o} \quad (68)$$

The constant in the bridge is defined by

$$K_B \approx \frac{2v_B}{R_o} \frac{k(1+k)}{(1+2k)^2} \quad (69)$$

The transfer function of the bridge is transformed into a differential equation as shown in the following

$$\tau_1 \dot{x}_1 + x_1 = K_B(R_1 - R_3) \quad (70)$$

where τ_1 is a time constant which, for the case of the scanner is less than 10 ms.

For the preamplifier, we end up with the equation

$$\tau_2 \dot{x}_2 + x_2 = x_1 + \tau_3 \dot{x}_1 - K_c C_B + (E_{os} + n) + \tau_3 (\dot{E}_{os} + \dot{n}) \quad (71)$$

Thus, the following equations describe the energy balance equation (Eq. 72) and the differential forms of the transfer functions, (Eq. 73) which include generalized forms of the amplifier and filter and a model of each scanner channel

$$\dot{T} = AT + DT^4 + B_Q \dot{Q}_8 + BE \quad (72)$$

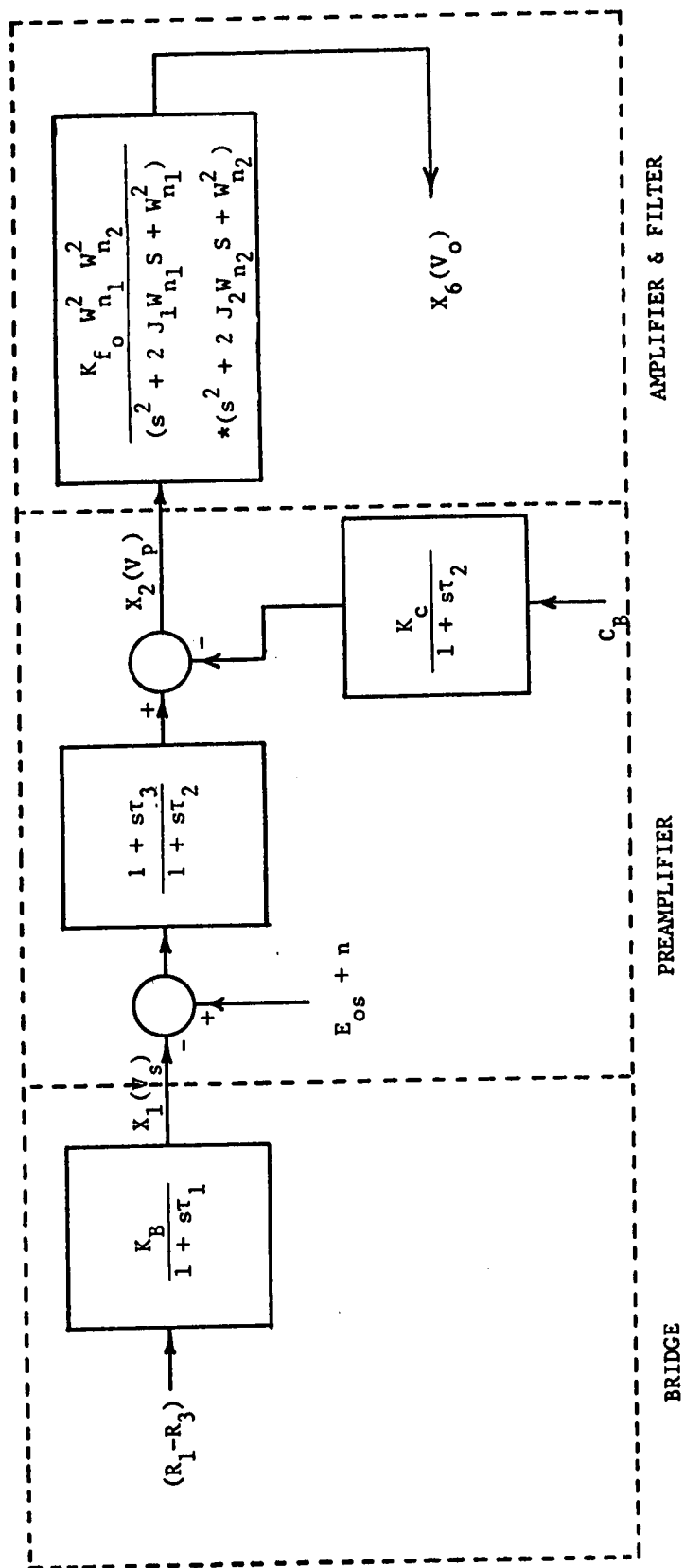


FIGURE 19 Scanner Electronic Control Model

$$\dot{x} = A_x x + B_T T + B_{os} (E_{os} + n) + B_c C_B + b \quad (73)$$

where T is a 14 component vector consisting of node temperatures, x is a 6 component vector consisting of electrical signal processing variables, \dot{Q}_8 is the heater heat flow rate, E_{os} , C_B , and b are electrical and flake related variables, n is the total noise as viewed at the preamplifier input, and E is the incoming radiant incidence vector. The measured quantities which are telemetered to ground are the raw counts, x_6 , and the heat sink temperature T_7 , measured by a thermister.

$$y = \begin{bmatrix} x_6 \\ T_7 \end{bmatrix} + v \quad (74)$$

where v is the measurement noise including quantization error due to A/D conversion.

2.1.11 Scanner Steady-State Analysis

Two types of transient behavior will be noticeable in the instruments. One type occurs when the heater \dot{Q}_8 is activated, i.e., when the instrument is turned on. The transient behavior in this case is rather slow, as the heat sink time constant is large and the conductivity of the filters (for SW and LW instruments) is low. Thus, the time period necessary for the various node temperatures to reach their relatively steady operating temperatures, particularly the filters, will be long. The precise period can be determined by simulation or measurement.

The second and more important transient behavior is due to quick variations in the input (or observed) radiation incidence, E , after the operating condition has been reached. The time constants, in this case, are determined largely by the electronic signal processing parameters and the

flake characteristics. After the operating condition has been reached, the temperature variations of most nodes should be negligible. The active flake temperature then varies with the incoming radiation. The flake resistance and other electrical variables also vary with the incoming radiation and can produce transient behavior which can be observed if the target level is changed suddenly, as a step function. On the other hand, if the incoming radiation varies slowly relative to the system time constants, then the system will operate in quasi-steady state condition. A preliminary analysis indicates that, for the parameters selected by the designer, the applicable time constants are in the vicinity of 10 ms. Thus, if the input radiation does not contain frequencies above 10 Hz, the transient will be negligible, while frequencies above 17 Hz, will be attenuated. Whereas the earth scan may have low content in the high frequencies (which are of less interest to ERBE), the transition from space to Earth may produce a necessarily short period of transient behavior. It may be of interest to analyze the magnitude and duration of such transients, which probably last less than 50 ms.

Since the quasi-steady state condition is of interest to this analysis, let us first consider one scan period. The scanner looks at space for a period "sufficient" for the transient effect to die down and obtains an average of the last few values (to reduce noise and errors) as the "space clamp". Then the scanner sees the Earth and atmosphere, then looks at two black bodies at ambient, but known, temperatures as well as the SWICS in the off condition. The $k\Delta t$ correspond to the k th space clamp, so that $T(k\Delta t)$ and $x(k\Delta t)$ are the temperature and electronic vectors at the space clamp measurement. Now let

$$\Delta x(t) = x(t) - x(k\Delta t), \Delta T(t) = T(t) - T(k\Delta t), \quad k\Delta t \leq t \leq (k+1)\Delta t. \quad (75)$$

From Eq. 72 and Eq. 73

$$\dot{\Delta T} \approx A \Delta T + 4D T_k^3 \Delta T + B_Q \Delta \dot{Q}_8 + BE \quad (76)$$

$$\Delta \dot{x} \approx A_x \Delta x + B_T \Delta T + \Delta d + B_{os} \Delta n, \quad (77)$$

where it has been assumed that the thermal and radiative properties are included in A, D, B_Q and B and the time constants for the electronics in A_x , B_{os} and B_c have remained unchanged over the scan period of 4 seconds. Also we can define Δd , the drift component of the voltage as,

$$\Delta d = \Delta B_T T(k\Delta t) + B_{os} \Delta E_{os} + \Delta b + B_c \Delta C_B \quad (78)$$

Considering the amplifier and Bessel filter to be in a quasi-steady state condition, it can be seen that

$$\Delta x_6(t) \approx K_{f_o} \overline{\Delta x}_2(t) + K_{f_o} \tilde{\Delta x}_6(t) \quad (79)$$

where $\overline{\Delta x}_2(t)$ is the part of $\Delta x_2(t)$ with lower frequency content (below 10 Hz) consisting of an averaged part of $\Delta x_2(t)$, which may be considered the signal part, and $\tilde{\Delta x}_6(t)$ is the part of $\Delta x_6(t)$ consisting largely of noise and irradiance variations over short distances at the top of the atmosphere with zero mean. As the filter attenuates high frequencies, $\tilde{\Delta x}_6(t)$ will have a low amplitude.

The preamp equations are expressed as

$$\tau_2 \Delta \dot{x}_2 + \Delta x_2 = \Delta x_1 + \tau_3 \Delta \dot{x}_1 + \Delta E_{os} + \Delta n + \tau_3 (\Delta \dot{E}_{os} + \Delta \dot{n}), - K_c \Delta C_B \quad (80)$$

Now assume that

$$\tau_2 \overline{\Delta \dot{x}_2} - \tau_3 \Delta \dot{E}_{os} \approx 0 \quad (81)$$

then

$$\overline{\Delta x}_2 = \overline{\Delta x}_1 + \tau_3 \overline{\Delta \dot{x}_1} + \Delta E_{os} - K_c \Delta C_B + \overline{\Delta n} \quad (82)$$

where $\overline{\Delta x_1}$ is the low frequency portion of Δx_1 , and where it has been assumed that ΔE_{os} and ΔC_B are baseband signals. Note that

$$\Delta x_2 = \overline{\Delta x_2} + \tilde{\Delta x_2}, \quad (83)$$

where $\tilde{\Delta x_2}$ contains the higher frequency components of Δx_2 and causes the term $\tilde{\Delta x_6}$. The bridge equations are now expressed as

$$\begin{aligned} \tau_1 \dot{\Delta x_1} + \Delta x_1 = & \tau_1 B_{T_{11}} \Delta T_1 + \tau_1 B_{T_{13}} \Delta T_3 + \Delta b_1 + \tau_1 \Delta B_{T_{11}} T_1 \\ & + \tau_1 \Delta B_{T_{13}} T_3 + \tau_1 \Delta B_{T_{11}} \Delta T_1 + \tau_1 \Delta B_{T_{13}} \Delta T_3 \end{aligned} \quad (84)$$

It is assumed that the only variable on the RHS containing higher frequency components is T_1 , whereas the remaining variables are slow moving drift components.

$$\overline{\Delta x_1} = \tau_1 (B_{T_{11}} \overline{\Delta T_1} + B_{T_{13}} \Delta T_3 - \dot{\Delta x_1}) \quad (85)$$

Combining Eqs. 85, 82, and 79,

$$\begin{aligned} \overline{\Delta x_2} = & \tau_1 (B_{T_{11}} \overline{\Delta T_1} + B_{T_{13}} \Delta T_3) + (\tau_3 - \tau_1) \dot{\overline{\Delta x_1}} - \tau_2 \dot{\overline{x_2}} \\ & + E_{os} - K_c \Delta C_B + \overline{\Delta n} \end{aligned} \quad (86)$$

$$\begin{aligned} \Delta x_6 = & K_{f_o} \tau_1 (B_{T_{11}} \overline{\Delta T_1} + B_{T_{13}} \Delta T_3) \\ & + K_{f_o} \left[(\tau_3 - \tau_1) \dot{\overline{\Delta x_1}} - \tau_2 \dot{\overline{x_2}} + \Delta E_{os} - K_c \Delta C_B + \overline{\Delta n} + \tilde{\Delta x_6} \right] \end{aligned} \quad (87)$$

Now consider the thermal equation

$$\dot{\Delta T_1} = A_{12} (\Delta T_2 - \Delta T_1) + 4 \sum_{k=1}^N D_{ik} T_k^3 \Delta T_k + B_1 E \quad (88)$$

$$B_1 \bar{E} = A_{12} (\Delta \bar{T_1} - \Delta T_2) - 4 \sum_{k=1}^N D_{1k} T_k^3 \Delta T_k + \dot{\bar{T_1}} \quad (89)$$

$$\Delta \bar{T}_1 = \frac{1}{K_{f_o} \tau_1 B_{T_{11}}} \Delta x_6 - \frac{B_{T_{13}}}{B_{T_{11}}} \Delta T_3 - \frac{1}{\tau_1 B_{T_{11}}} \delta_1 \quad (90)$$

$$\delta_1 = (\tau_3 - \tau_1) \dot{\bar{x}}_1 - \tau_2 \dot{\bar{x}}_2 + \Delta E_{os} - K_c \Delta C_B + \Delta \bar{n} + \Delta \tilde{x}_6 \quad (91)$$

Substituting Eq. 88, 90, and 91 into Eq. 89, we have

$$\begin{aligned} B_1 \bar{E} = & \frac{(A_{12} - 4 D_{11} T_1^3)}{K_{f_o} \tau_1 B_{T_{11}}} \Delta x_6 - A_{12} \dot{\Delta T}_2 - \frac{B_{T_{13}}}{B_{T_{11}}} (A_{12} - 4 D_{11} T_1^3) \Delta T_3 \\ & - \sum_{k=2}^N 4 D_{ik} T_k^3 \Delta T_k + \dot{\Delta \bar{T}}_1 - \frac{(A_{12} - 4 D_{11} T_1^3)}{\tau_1 B_{T_{11}}} \delta_1 \end{aligned} \quad (92)$$

2.1.12 Scanner Count Conversion

The scanner count conversion algorithm below is based on steady state approximation. The structure (i.e., the form of the equations) of the algorithm is the same for each of the three scanner channels. The parameters have different values for each channel.

The average of scan points during space clamp is given below

$$\bar{v}(t_k) = \frac{1}{n} \sum_{i=1}^n v(t_{ki}), \quad n = 8 \quad (93)$$

The mean variance of the counts compared with the space clamp during a scan cycle is obtained by the equation

$$\sigma_k^2 = \frac{1}{n} \sum_{i=1}^n \left[v(t_k) - \bar{v}(t_k) \right]^2 \quad (94)$$

The scanner count conversion equation is then defined by the following equations

$$E(t - \tau) = A_v \left[v(t) - v(t_k) \right] + A_H \left[T_H(t) - T_H(t_k) \right] + A_\delta \delta_k \frac{t - t_k}{\Delta t}, \quad (95)$$

$$t_{k-1} \leq t \leq t_k$$

The coefficients in Eq. 95 are the following

$$A_v = \frac{A_{12} - 4 D_{11} T_1^3}{K K_{f_o} \tau_1 B_{T_{11}}} \frac{V_{B_o}}{V_B(t)} \quad (96)$$

$$A_H = - \frac{1}{K} 4 \left[D_{11} T_1^3 + D_{12} T_2^3 + D_{13} T_3^3 + D_{14} T_4^3 + D_{15} T_5^3 + D_{16} T_6^3 + D_{18} T_8^3 + D_{113} T_{13}^3 \right] - A_{12} \left(1 + \frac{B_{T_{13}}}{B_{T_{11}}} \right) \frac{1}{K} \quad (97)$$

$$A = c \frac{A_{12} - 4 D_{11} T_1^3}{K \tau_1 B_{T_{11}}} \frac{V_{B_o}}{V_B(t)} = c K_{f_o} A_v \quad (98)$$

$$\begin{aligned} \delta_k = & \frac{1}{K_{f_o}} \left[\bar{v}(t_k) - \bar{v}(t_{k-1}) \right] - \tau_1 (B_{T_{11}} + B_{T_{13}}) \left[T_H(t_k) - T_H(t_{k-1}) \right] \\ & - \frac{\tau_1}{V_{B_o}} (B_{T_{11}} T_1 + B_{T_{13}} T_3) \left[V_B(t_k) - V_B(t_{k-1}) \right] \\ & + K_d \left[V_d(t_k) - V_d(t_{k-1}) \right] \end{aligned} \quad (99)$$

$$t_k = t_{k-1} + \Delta t \quad (100)$$

In the last term of Eq. 95, $(t-t_k)/\Delta t$ can be replaced by the number of measurements in the scan mode as follows:

$$\frac{t - t_k}{\Delta t} \approx 0.0135 (J-8) \quad (101)$$

where J is the number of measurements in the scan mode. The constants in the above equation are the following:

$\bar{v}(t_k)$ = space clamp value at time t_k

$v(t_{ki})$ = instrument output (in volts) when viewing space at t_k

σ_k^2 = noise variance estimate during space look

$v(t)$ = instrument output sample at time t

$T_H(t)$ = heat sink temperature measurement at t or most recent value ($^{\circ}\text{K}$)

Δt = total scan period (4 sec.)

t_k = time at end of space look (sec.)

t = sampling instant (sec.)

$V_B(t)$ = detector bridge bias voltage measurement at time t or most recent value (v)

$V_d(t_k)$ = drift balance DAC voltage measurement at time t or most recent value (v)

δ_k = estimate of unaccounted drift during k^{th} scan period (V)

τ = average time lag (sec.)

k_{f_o} = post amplification gain

Note that to compute δ_k , it is necessary to have the following space clamp value; i.e., $\bar{v}(t_k)$. If the term with δ_k in Eq. 95 is neglected, then the count conversion may be done without using $\bar{v}(t_k)$. The variance σ_k^2 is computed for diagnostic purposes and should be output as described below. The converted value $E_i(t - \tau)$, $i = s, l, T$ (for shortwave, longwave, and total, respectively) can be interpreted as follows

$$E_i(t) = \int \tau_i(\lambda) E_{\lambda}(t) d\lambda + e_i(t) \quad (102)$$

where $\tau_i(\lambda)$ is the transmittance of the i^{th} channel, $E_{\lambda}(t)$ is the spectral irradiance at the instrument due to radiation from the footprint at time t ,

and $e_i(t)$ is the error in the measurement combined with the count conversion process, and can be estimated through analysis of calibration data and model results.

2.1.13 Accuracy and Diagnostic Checks

During each scan, the instruments view two black bodies at ambient, but known temperatures. These data can be used to determine if the basic operation of the instruments have changed as well as provide an approximate accuracy check of the combined error $e_i(t)$ in Eq. 101.

1. Store a table of elements

$(v_{Bs}(T_B), v_{Bl}(T_B), v_{BT}(T_B), E_{Bs}(T_B), E_{Bl}(T_B), E_{BT}(T_B))$ where $T_B = T_{Bj}$, $1 \leq j \leq N$. The values $v_{Bi}(T_B)$ represent the steady state output of the i^{th} channel when viewing the black body at temperature T_B , and may be obtained in calibration tests. The values $E_{Bi}(T_B)$ represent the output of count conversion when viewing the black body at temperature T_B , given by

$$E_{Bi}(T_B) = \int \tau_i(\lambda) E_{B\lambda}(T_B) d\lambda \quad (103)$$

where $E_{B\lambda}(T_B)$ is the spectral irradiance from the black body at temperature T_B .

2. Compute E_{Bi} using the measured black body temperature $T_B(t)$ as

$$E_{Bi}(T_B(t)) \approx E_{Bi} + \frac{T_B(t) - T_{B1}}{T_{B2} - T_{B1}} [E_{Bi}(T_{B2}) - E_{Bi}(T_{B1})] \quad (104)$$

where $T_{B1} \leq T_B(t) \leq T_{B2}$

3. Compute $\hat{E}_i(t)$ using the count conversion algorithm with the corresponding value $v_i(t)$.

4. Plot $v_i(t)$ and $v_{Bi}(T_B(t))$ on same axes.

5. Plot $\hat{E}_i(t)$ and $E_{Bi}(T_B(t))$ on same axes.

6. Compute and plot σ_k^2 , and

$$\bar{e}_{v_i, k+1} = a_{v_i} k + (1 - a)(v_1(t) - v_{B1}(T_B(t))) \quad (105)$$

$$\bar{e}_{i, k+1} = a \bar{e}_{ik} + (1 - a)(\hat{E}_i(t) - E_{B1}(T_B(t))) \quad (106)$$

$$\sigma_{x_i, k+1}^2 = a \sigma_{v_i}^2 + (1 - a) \left[(v_1(t) - v_{B1}(T_B(t))) - \bar{e}_{v_i, k+1} \right]^2 \quad (107)$$

$$\sigma_{i, k+1}^2 = a \sigma_{ik}^2 + (1 - a) \left[(\hat{E}_i(t) - E_{B1}(T_B(t))) - \bar{e}_{i, k+1} \right]^2 \quad (108)$$

3. INSTRUMENT PERFORMANCE AND COUNT CONVERSION TEST

The instrument models developed for the non-scanner, the scanner, and the solar monitor are based upon the theoretical approach with a certain degree of approximation. For instance, the energy equations for the instruments were formulated with the assumptions that the node (or control volume) has uniform temperature and homogeneous thermal and radiative properties. However, this lumped geometry of the node, in reality, cannot hold the uniform temperature through the node geometry unless the temperatures of the neighbor nodes are the same, or the node boundaries are perfectly insulated, or the thermal conductivity of the node material is infinity. Otherwise, if there is any heat flux allowed into or from the node, a temperature gradient from the center of node to the boundary will exist.

The radiative properties that are used in the ERBE radiometer instrument models may be the values picked out of the handbook or measured and averaged values over a certain band of wavelength. No matter where these values were obtained, they would introduce some errors when they are used for the computation of the radiation exchange between the surfaces. The spectral distribution of the emissive energy flux from a source of radiation such as MRBB may lie on the limited range of wavelength in which the averaged radiative properties used may be far above or below the real values within the spectral energy distribution. Therefore, the averaged values may not represent reality in instrument performance. Also, during long term operation in space, these properties will be subject to changes due to various reasons such as thermal and environmental contamination, and aging effect.

The electronic systems are used in the radiometer to measure the temperature difference between the active and reference cavities or flakes, and to

digitize the signal, then to transmit it through a telemetry system. The parts such as resistors of this electronic system are sensitive to a wide range of temperature changes. For instance, if the resistance of the bridge slightly changes from the reference point, then the error in irradiance measurement due to this temperature change may be in error by a few watts per square meter. Although the instrument models are well-conceptualized and designed, the real behavior of the radiometers are, as expected, different from what is was designed to be.

Other factors to be considered, which influence the sensor performance and measurement accuracy, are the inflight changes from the states that the sensor normally operates in and the calibration source variability in its operation mode or the changes in source characteristics. An environmental change was expected in the long-term operation of the instrument due to the orbital changes which caused cyclic variations in the heat load of the instrument. Another change, was due to the contamination or degradation of then sensor surfaces due to radiative interactions with other surfaces.

The internal calibration sources such as MRBB and SWICS are also subject to change. The monitoring of the change in intensity of these sources by possible variation in power supply and shift of the spectral distribution in the source output is possible by using the SiPD, although it is also subject to change, and intercomparing calibration sources.

To account for the possible errors due to the reasons discussed above a procedure was developed to detect those errors. The procedure uses the algorithm shown in Figs. 20 and 21 for that purpose. This algorithm is capable of evaluating the sensor model, adjusting the simulation model parameters for minimum errors, and establishing confidence intervals for the

ERBE INSTRUMENT VALIDATION PLAN

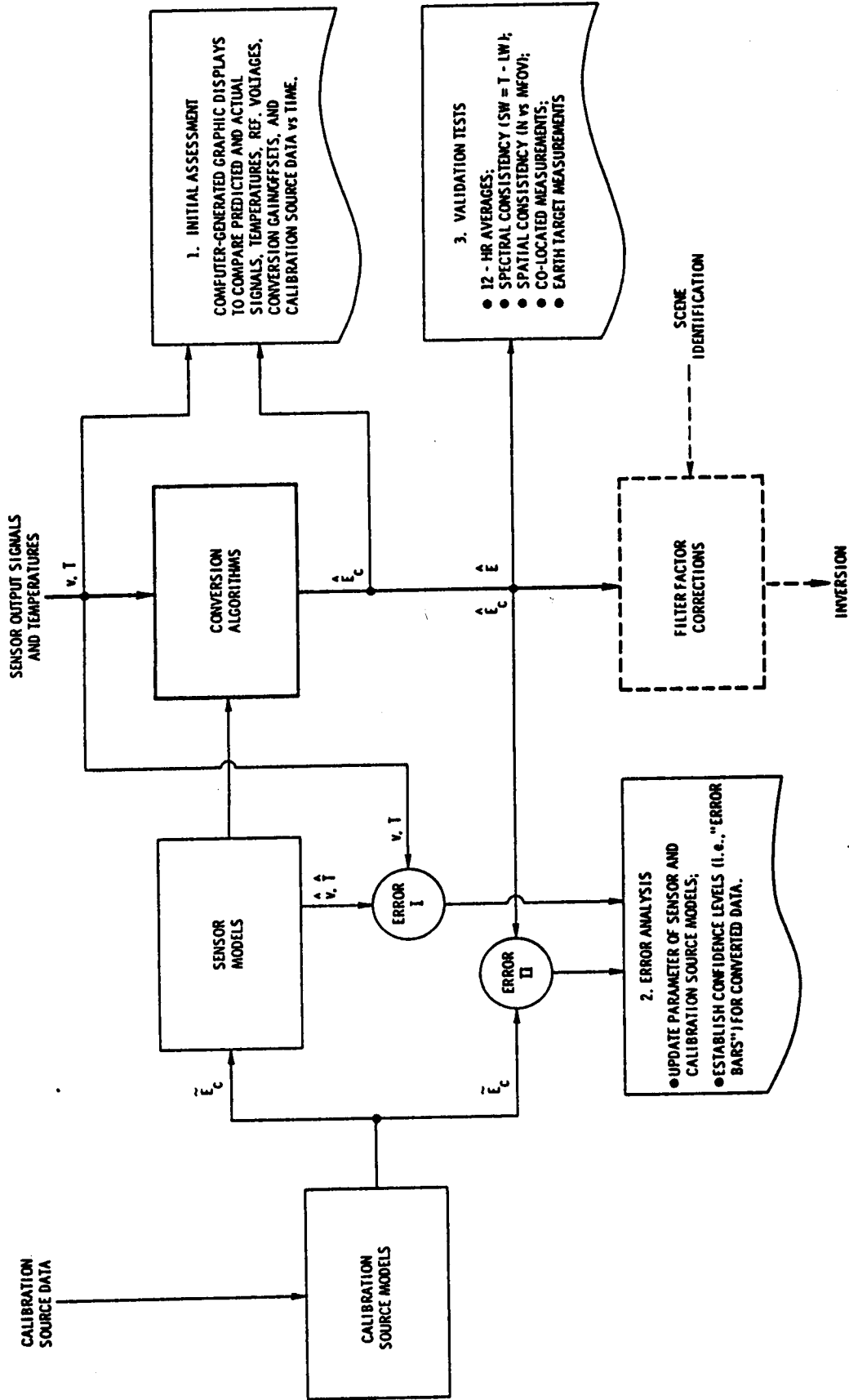


FIGURE 21.

converted data reflecting sensor characteristics and calibration sources.

The sensor simulation model responses, using known calibration sources as inputs to the model, are checked against the instrument calibration output. Any differences in the slope, offset and curvature between the simulation and calibration results were interpreted as uncertainties of the model simulating the instrument. These differences, however, generally reflect the problems discussed earlier. In the algorithm, these differences are narrowed by checking and adjusting the sensor model, source, and environmental parameters to minimize the errors. However, the estimated parameters must stay within the tolerance that can be allowed by the approximations made in the modeling or by the nature of reasons described earlier.

The algorithm consists of three parts (see Fig. 20):

The first part is a routine to adjust the sensor model parameters by comparing sensor simulation outputs such as counts and HK temperatures to the inflight and ground calibrations obtained for the known sources.

The second part is a procedure to check the errors generated by simulating a sensor by comparing them to the calibration data for the known sources. It checks the error generated from the sensor simulation target source, or environment at different time intervals for the same known irradiance source as in the inflight calibration.

The third part consists of routines to search for any changes in the target, environment and sensor performance due to the degradation of the instrument surface, vibration, contaminations, and thermal and aging effects on the radiation interactive surfaces.

To do the above job, the algorithm has been designed to have the following capabilities:

1. to adjust the sensor model parameters in comparison with the calibration data from known sources to minimize the error between the sensor model and the calibration data,
2. to periodically check any changes in sensor, target and environment parameters between inflight calibrations by comparing the calibration data for the same sources at different time intervals (e.g., 1 or 2 week periods),
3. to re-define the sensor, target and environment parameters, if they have been changed,
4. to test and check any differences between calibration data and responses from the simulated sensor model for the same known sources,
5. to repeat the routines 1 through 4 until the errors produced by step 4 are minimized,
6. to generate the count conversion parameters.

3.1 Description of Routines

3.1.1 Routine I

The calibration sources identified during either ground or inflight calibrations are numbered by $j = 1, N$ at time t_i and t_{i+1} where $i = 1, n$. The sensor simulation model generates temperatures and sensor responses corresponding to the calibration sources at a time t_i . These temperatures and sensor responses, which are obtained from the model simulation, are compared with measured data to compute errors of the simulation model. Once these errors fall within the pre-determined error bounds (after repetitive refinement of simulation parameters), then these errors will be used as criteria to determine the measurement errors

against known targets on the ground. If not, the parameters for both the target source and environment effects will be checked; then in turn the sensor parameters will be checked and adjusted, in case when the computed parameters are not within the acceptable range based on the materials properties, e.g. its optical properties. Such a routine will include a decision block to check targets and environmental effects to avoid unnecessary adjustments of sensor parameters when the errors may be caused by and come from the target and environment parameters.

3.1.2 Routine II

If the errors are within bounds, the sensor responses V_j , $j = 1, N$ at different times t_i and t_{i+1} will be sorted and evaluated in Routine I to check the errors from the sensor simulation model for the targets at different times t_i and t_{i+1} .

3.1.3 Routine III

After finishing Routine II, using all the target sources at different times, t_i and t_{i+1} , Routine III checks the sensor responses of the same target at a different time and computes the errors. If the error of the same target at a different time falls within the error bounds, the Routine will check the next target. If not, the history of target source variation will be studied, and at the same time the sensor and environmental conditions will be rechecked to determine the source of errors. This evaluation in Routine I is performed whenever the error signals the flag. The process continues until the error source is recognized and its correction of error is made by adjusting the related parameters in the simulation model. When this Routine finishes satisfactorily evaluating all of the target sources, the program will print all the decisions, errors, sensor

and calibration source HK data, sensor parameters and counts and will supply the final results to the count conversion algorithm.

3.1.4 Routine IV

If the error does not fall within the prescribed error bounds, the target and environment parameters are checked in order to find whether changes in these parameters created pseudo-errors in the sensor responses. If there has been a change in either the target or environmental parameters, they are re-defined and re-input to Routine I. In spite of repeated checks, if the error cannot be minimized, then the thorough review of instruments will be required to determine whether the error bounds are to be reassessed. In this routine, when the target and environment are checked and relatively free from the erroneous performance then we would postulate that the instrument's measurement capability might be deteriorated due partially to the thermal and environmental contamination or by other means such as aging of the parts and so on.

4. DISCUSSION OF PARAMETER ESTIMATION

The simulation code was developed to study the sensor instrument responses and behavior during the ground and inflight calibration procedures as described earlier in Section 2.1.6. This simulation code uses the irradiance at the FOV limiter of the instrument as the input data. All the constants representing the instrument geometry, material, thermal and radiative properties and the electronic thermal control systems are parameterized to be used as coefficients in the system of equations. Then the simulation code generates the counts and the nodal temperatures of the model corresponding to the input irradiance. When the simulation code uses the input irradiance which is the same as the calibration source, the results of counts and housekeeping temperatures from the simulation are compared to the calibration results. If the differences from these comparisons for various levels of irradiance are wide, then the simulation model parameters are adjusted to minimize those differences. This procedure for estimating the model parameters is repeated until the differences can no longer be reduced. These estimations also continue for various calibration sources until systematic response patterns for each calibration source are established. All procedures mentioned above are performed for each individual radiometer sensor. Even though two instruments of the same type are used in the calibration procedure, for example, the results may differ due to the inherent differences in the many factors involved, e.g. slight differences in the sensor assembly process of uniform surface roughness, coatings, glue or contact areas.

The following is an example that shows how the parameters for the nonscanner total channel are selected. Assume that the instrument output derived from calibration input irradiances are linear; then the factors which determine the slope and offset of the curve for simulated results would be judged to be the

offset temperature between the active and reference cavities, and the radiative constant for the term of irradiance in Eq. 34, i.e. B_1 . To determine these two factors using the measurement data, the linear form of Eq. 34 can be written

$$\alpha x = \beta \quad (109)$$

where x is a column vector consisting of the offset temperature and radiative constant, and α and β are defined later. The solution to Eq. 108 is

$$x = (\alpha' \alpha)^{-1} \alpha' \beta \quad (110)$$

where α' is the transpose of α .

The constant α is a matrix defined by

$$\alpha_{ij} = \begin{bmatrix} A_{12} + 4 D_{11} (T_2^3)_1, & E_1 \\ A_{12} + 4 D_{11} (T_2^3)_2, & E_2 \\ \vdots & \vdots \\ A_{12} + 4 D_{11} (T_2^3)_i, & E_i \end{bmatrix}$$

where i is the calibration sequence.

The constant β is a column vector

$$\beta_i = \begin{bmatrix} \beta_1 \\ \beta_2 \\ \vdots \\ \beta_i \end{bmatrix}$$

β as an explicit form is

$$\beta_1 = -B_a K_a V_i^2 - \phi_i - B_{J_1} \dot{Q}_{J_1} \quad (111)$$

$$\phi_1 = \sum_k D_{1k} T_{ki}^4, \quad \text{only when } k = 1, \text{ replace } T_{1i} \text{ by } T_{2i} \quad (112)$$

The nominal values are used for unmeasured T_{ki} and \dot{Q}_{J_1} except $T_3 \approx T_2 + \delta T$. The temperature of the reference cavity is a little higher than the heat sink temperature due to the Joule heating by electric current flowing probe wire of reference cavity. The temperature difference, δT , due to the Joule heating cannot be directly measured but can be observed from the simulation results.

The plots of the calibration and simulation results show that they are not actually linear but could be considered to be linear for estimation (see for instance, the steady-state response comparison of simulation of calibration data for the NS WFOVSW sensor on page 100 in Appendix A). Therefore, the above approach can still adjust the overall slope and offset of the curve by regarding them as a linear. The linearity and adjustment can be made by adjusting other parameters. But since almost all parameters are interrelated, the selection of a parameter that mostly influences the linearity of the curve in a plot of simulation results must be cautiously taken by testing the linearity with various parameters.

Since the simulation code was developed, the calibration data for the nonscanner radiometers using MRBB as a calibration source was the only source available. Hence, the nonscanner instrument models have been simulated using MRBB data which was used in the calibration. Figure 22 shows the calibration data for the MRBB temperatures in the 1st column and the count outputs for the four nonscanner channels in the 6th through 9th columns. The 2nd column shows the blackbody radiation corresponding to the temperatures in the 1st

1	2	3	4	5	6	7	8	9	DIFFERENCE BETWEEN CAL AND SIMULATION			
MRBB TEMP °K	σ_e^4 (w/m ²)	MODEL % SHORT- WAVE	MODEL % LONG- WAVE	IRRADIANCE SW/LW (w/cm ²)	MFOV (COUNTS) CALIB/	WFOV (COUNTS) C/M	MFOVSW (COUNTS) C/M	WFOVSW (COUNTS) C/M	10	11	WT	WS
188.23	71.55	.02%	99.980%	.0000015	8047	6912	6382	7327	+2	0	0	0
-84.93				.0071535	8049	6912	6382	7327	0	0	0	0
243.4	198.95	.243%	99.757%	.00000483	7936	6668	6367	7312	+6	-3		
-29.76				.0198467	7924	6665	6373	7315	+6	+3		
275.66	327.55	.692%	99.308%	.0002267	7828	6407	6355	7300	+4	0		
2.5				.0325283	7832	6407	6363	7301	+8	+1		
299.36	455.25	1.266%	98.734%	.0005763	7714	6144	6344	7282	+7	-5		
26.2				.0449487	7221	6139	6352	7285	+8	+3		
318.56	583.77	1.916%	98.084%	.0011185	7606	5858	6332	7265	+1	-1		
45.4				.0572585	7608	5857	6339	7266	+7	+1		
334.77	711.72	2.603%	97.397%	.0018526	7483	5563	6318	7244	+11	0		
61.61				.0693194	7494	5563	6324	7245	+6	+1		
348.86	839.62	3.307%	96.693%	.0027766	7378	5252	6306	7223	0	0		
75.70				.0811854	7378	5252	6307	7221	1	-2		

EXPLANATIONS: MRBB TEMPERATURE WAS OBTAINED FROM TRW IN °C, THE °K IS THE RESULT OF ADDING 273.16 HEAT SINK CONTROLLER SET AT 39.995°C OR 313.155°K IN THE MODEL.

FIGURE 22.

column. The 3rd and 4th columns show the percentages for the shortwave and longwave portions of the blackbody radiation described in the 5th column, respectively. These shortwave and longwave portions were calculated by the integration of Planck's equation for blackbody radiation with respect to the wavelength from 0 to 5 μm for shortwave and from 5 μm to 1000 μm for longwave. These portions were used for the MFOV and WFOV shortwave instrument simulations. The shortwave range selection of 0 to 5 μm was simply made by the shortwave filter (suprasil) characteristics. In the 6th through 9th columns, the numbers in the bottom part of the block signify the output (in counts) corresponding to MRBB radiation from the simulation. The 10th and 11th columns show the differences in count between calibration and simulation tests.

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2. Sparrow, E. M., and R. D. Cess, Radiation Heat Transfer, Hemisphere Publishing Corporation, 1978.
3. Rohsenow, W. M., Developments in Heat Transfer, The MIT Press, 1964.

APPENDIX A

**PRELIMINARY RESULTS ON ERBE NON-SCANNER INSTRUMENT
MODELING, COUNT CONVERSION AND SENSITIVITY ANALYSIS**

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GOALS

- DEVELOP A DYNAMIC SENSOR MODEL FOR EACH INSTRUMENT
- DEVELOP AN ADAPTIVE COUNT CONVERSION ALGORITHM THAT CAN ACCOMMODATE CHANGING CONDITIONS
- DETERMINE THE ACCURACY OF THE ERBE INSTRUMENTS
- DETERMINE THE ACCURACY OF THE ERBE MEASUREMENTS

APPROACH TO ASSESSMENT OF INSTRUMENT ACCURACY

- DEVELOP A MATHEMATICAL MODEL AND COMPUTER SIMULATION FOR EACH INSTRUMENT
- COMPARE GROUND CALIBRATION DATA TO CORRESPONDING SIMULATION DATA
- MINIMIZE DIFFERENCE BY ADJUSTING MODEL PARAMETERS
- PERFORM A SENSITIVITY ANALYSIS
- ESTIMATE CONFIDENCE INTERVALS FROM GROUND CALIBRATION DATA
- REPEAT SAME PROCEDURE FOR FLIGHT CALIBRATION DATA (ERROR ANALYSIS)
- ADJUST ESTIMATE OF CONFIDENCE INTERVALS AT EACH FLIGHT CALIBRATION

APPROACH TO ADAPTIVE COUNT CONVERSION

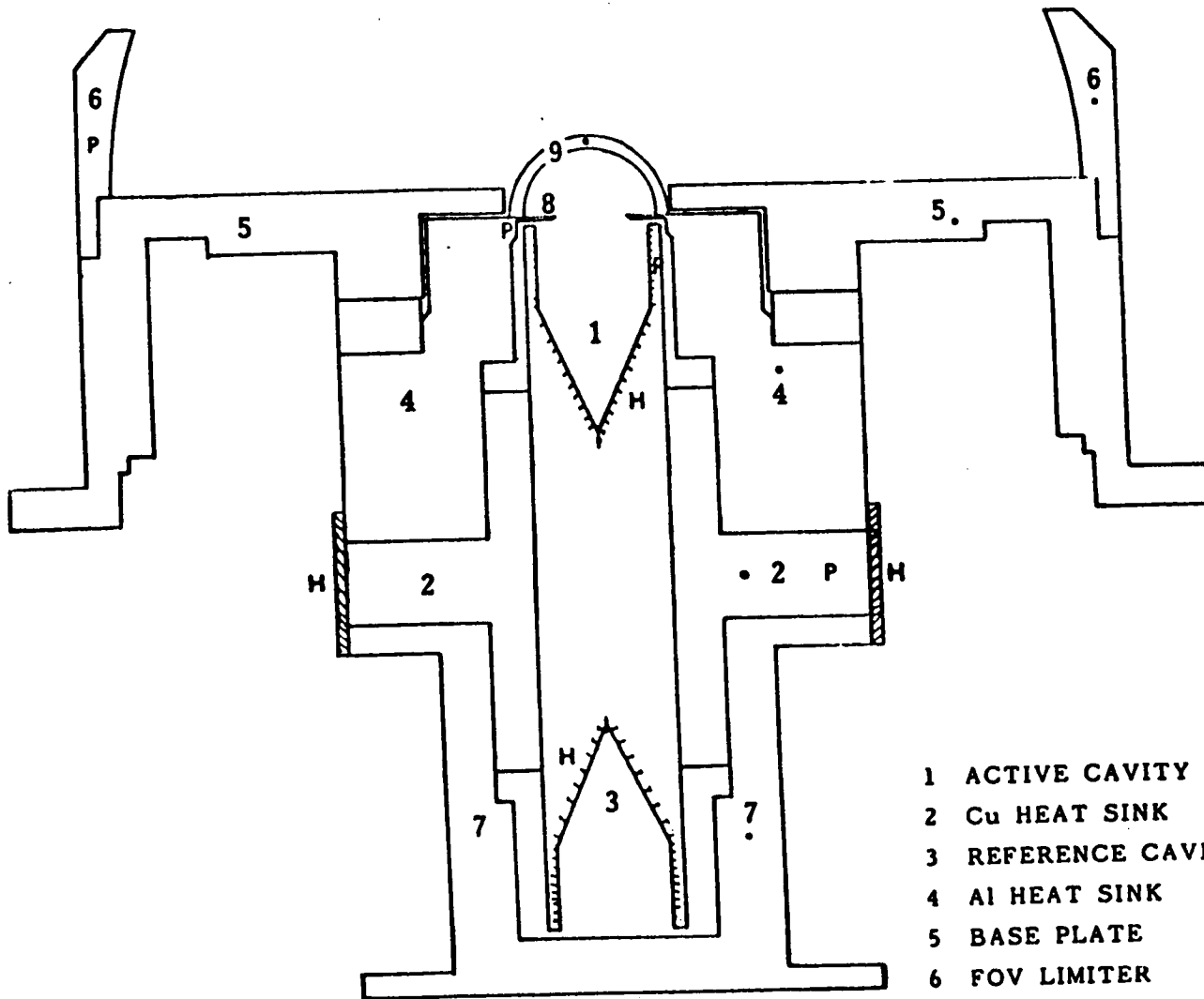
- DEVELOP A MATHEMATICAL MODEL AND COMPUTER SIMULATION FOR EACH INSTRUMENT
- COMPARE GROUND CALIBRATION DATA TO CORRESPONDING SIMULATION DATA
- MINIMIZE DIFFERENCE BY FINE TUNING MODEL PARAMETERS
- DEVELOP A COUNT CONVERSION ALGORITHM (CCA) BASED ON THE MODEL
- SIMULATE THE INSTRUMENT AND CCA
- ESTIMATE ERROR DUE TO CCA
- REPEAT PROCEDURE FOR EACH FLIGHT CALIBRATION
- FINE TUNE MODEL AND CCA PARAMETERS
- OBTAIN NEW ERROR ESTIMATES

The diagram is a cross-sectional view of a laser assembly. It features a central vertical cavity (1) with a V-shaped top and a triangular bottom. This cavity is flanked by two heat sinks: a copper heat sink (2) on the left and an aluminum heat sink (4) on the right. The entire assembly is mounted on a base plate (5). At the top, there are two curved components labeled 6, which are FOV limiters. The diagram also shows various internal structures and interfaces, labeled with numbers 1 through 8, and letters P and H. A legend on the right side of the diagram provides the following information:

- 1 ACTIVE CAVITY
- 2 Cu HEAT SINK
- 3 REFERENCE CAV
- 4 Al HEAT SINK
- 5 BASE PLATE
- 6 FOV LIMITER

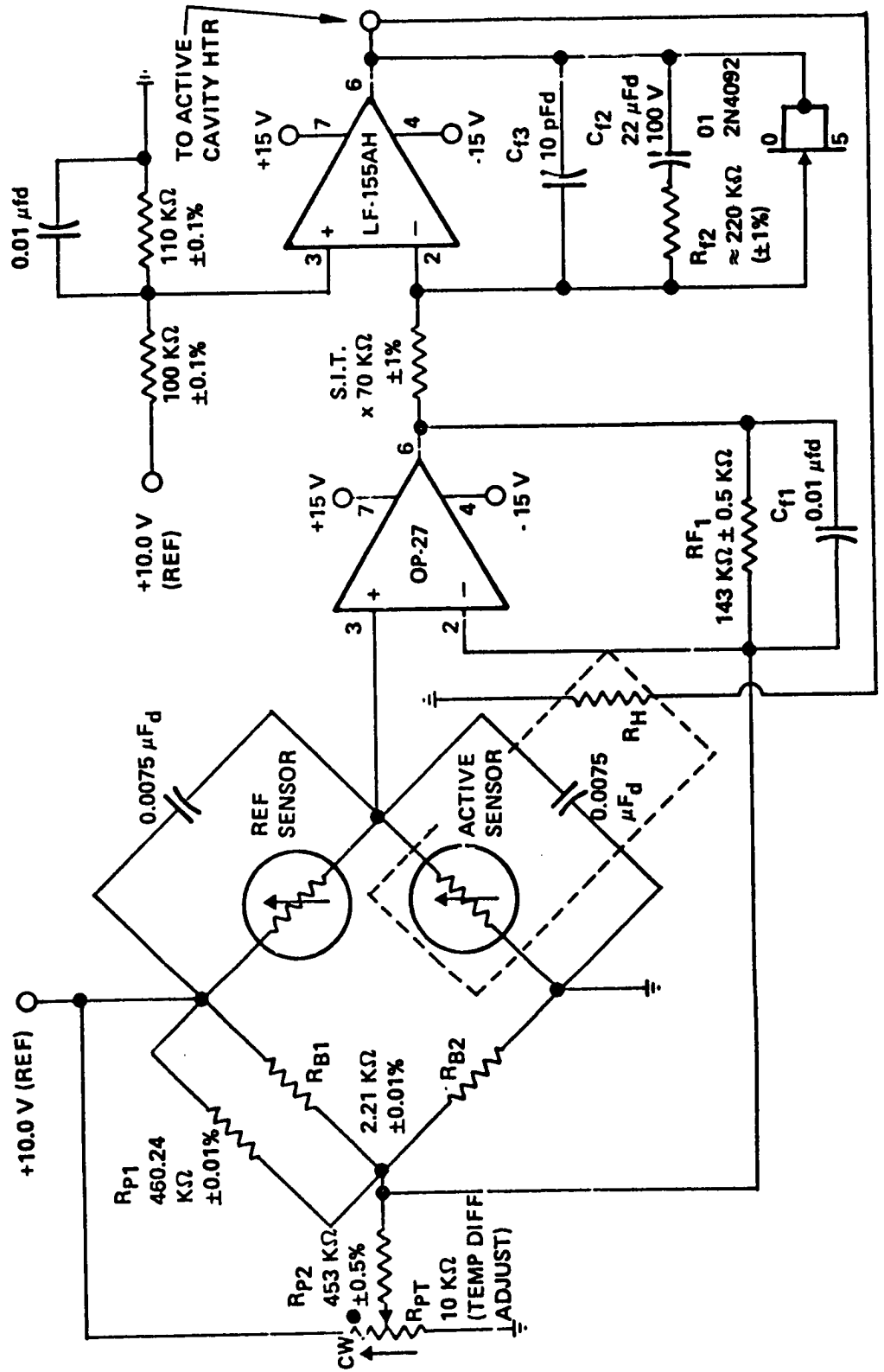
- H HEATER
P TEMPERATURE PROBE

NON-SCANNER WFOV SW



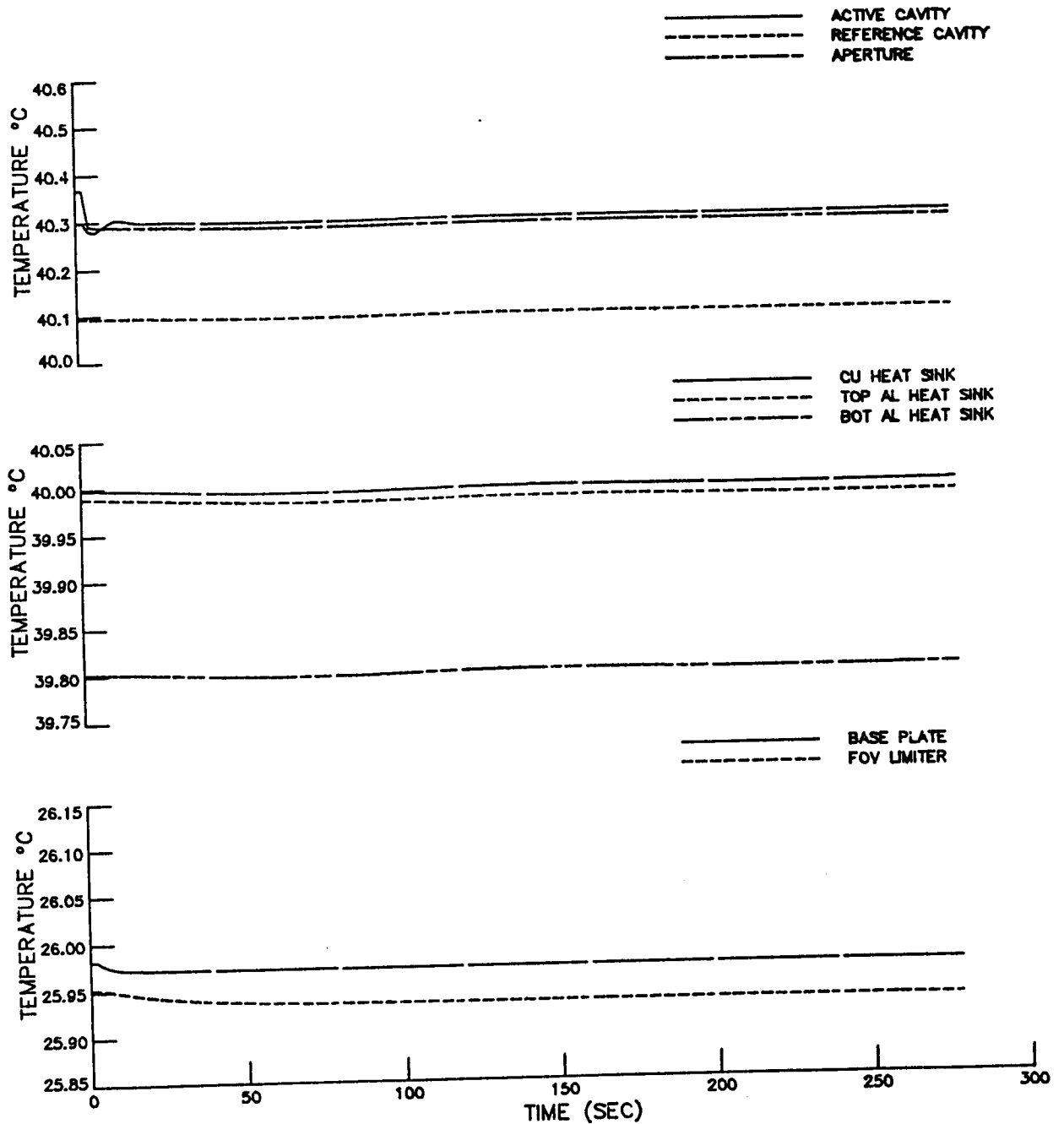
- 1 ACTIVE CAVITY
- 2 Cu HEAT SINK
- 3 REFERENCE CAVITY
- 4 Al HEAT SINK
- 5 BASE PLATE
- 6 FOV LIMITER
- 7 Al HEAT SINK
- 8 APERTURE
- 9 SW FILTER DOME
- H HEATER
- P TEMPERATURE PROBE

NON-SCANNER ELECTRONICS (ACR)



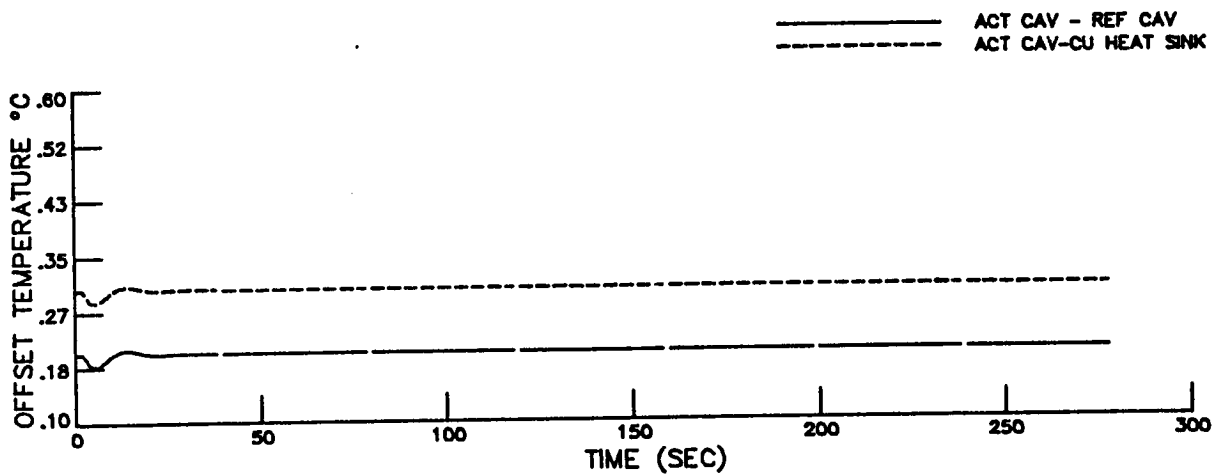
NON-SCANNER DYNAMIC RESPONSE

WFOV TOTAL SIMULATION



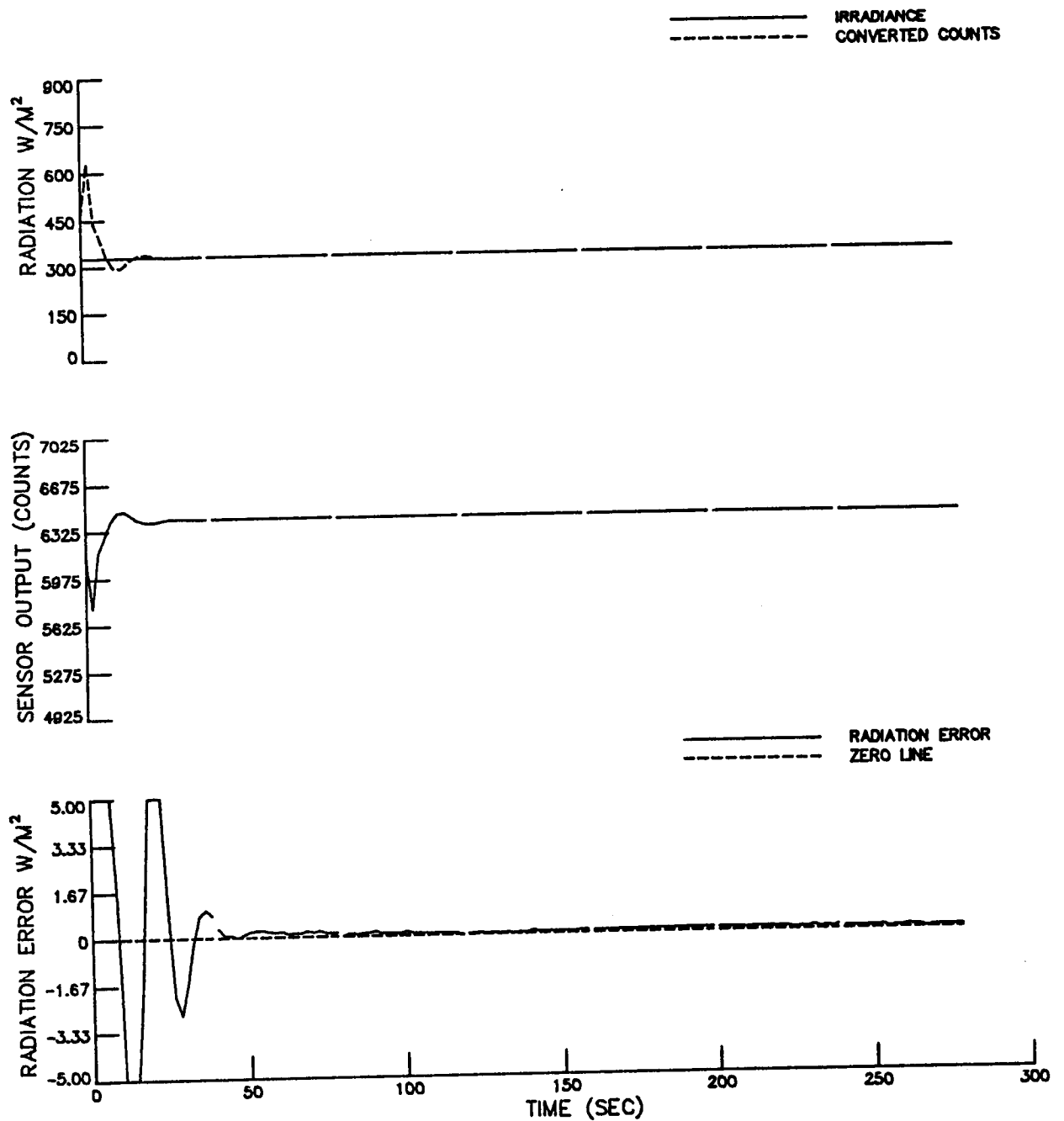
NON-SCANNER DYNAMIC RESPONSE

WFOV TOTAL SIMULATION



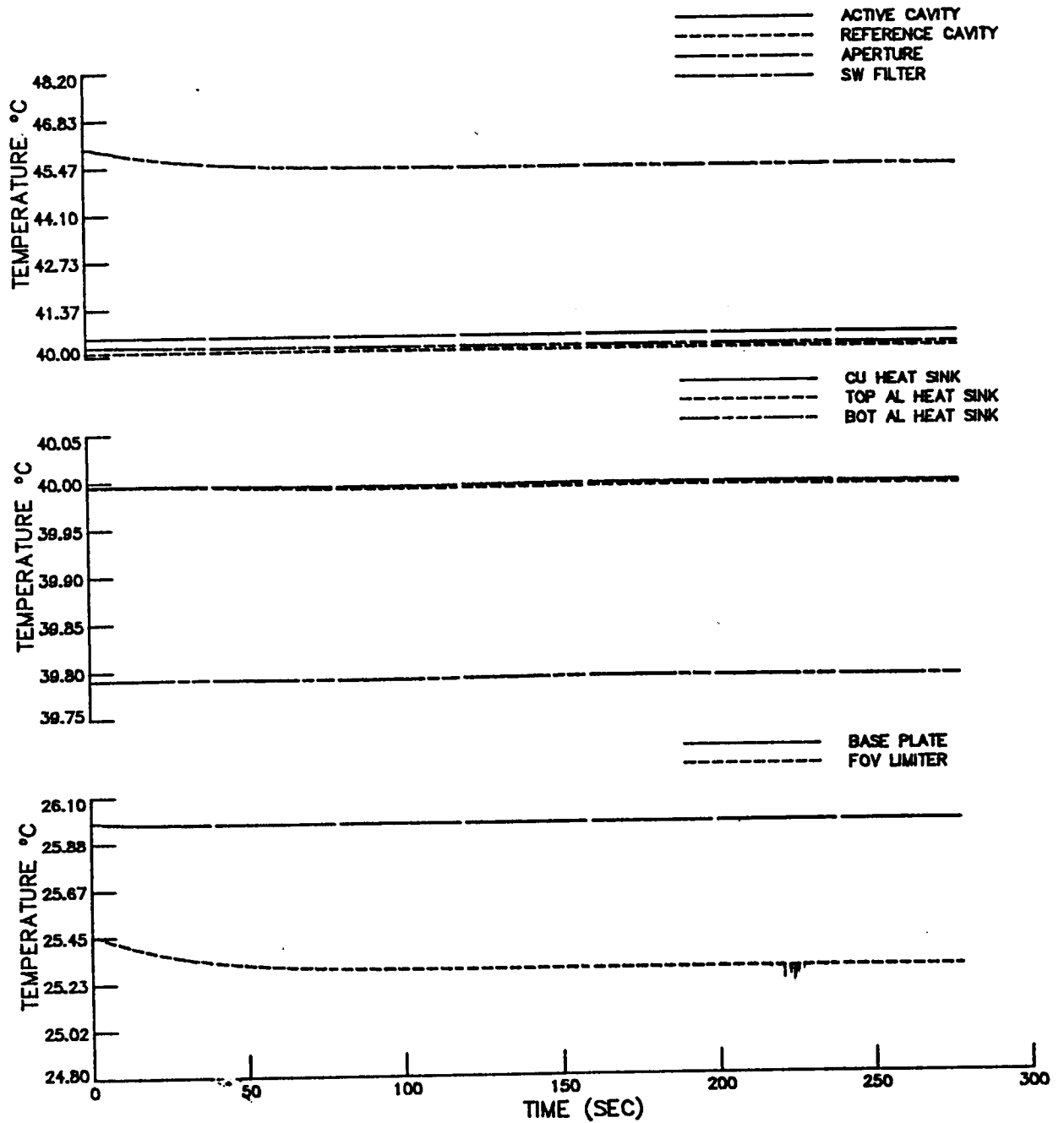
NON-SCANNER DYNAMIC RESPONSE

WFOV TOTAL SIMULATION



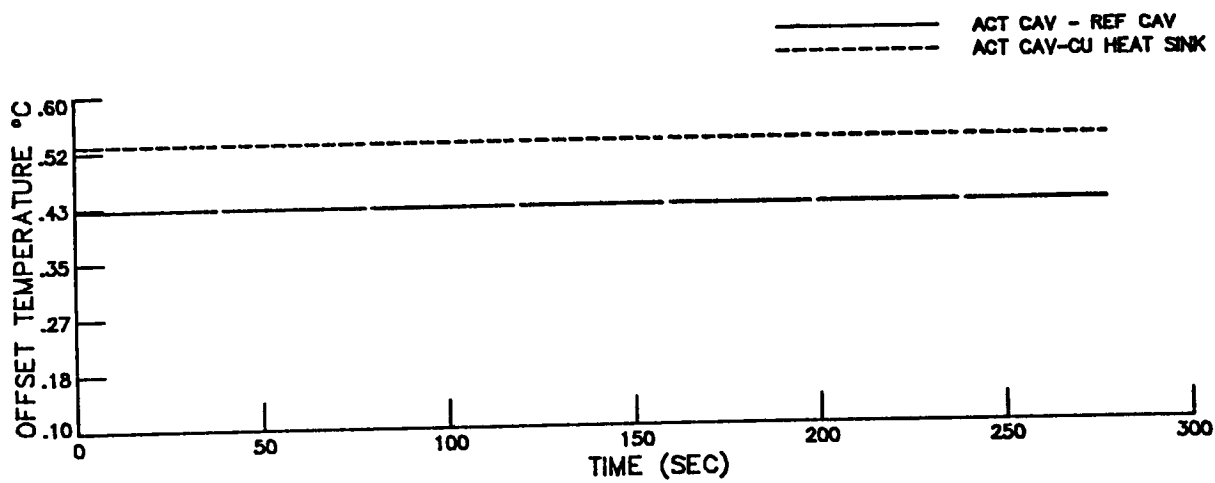
NON-SCANNER DYNAMIC RESPONSE

WFOV FILTERED SIMULATION



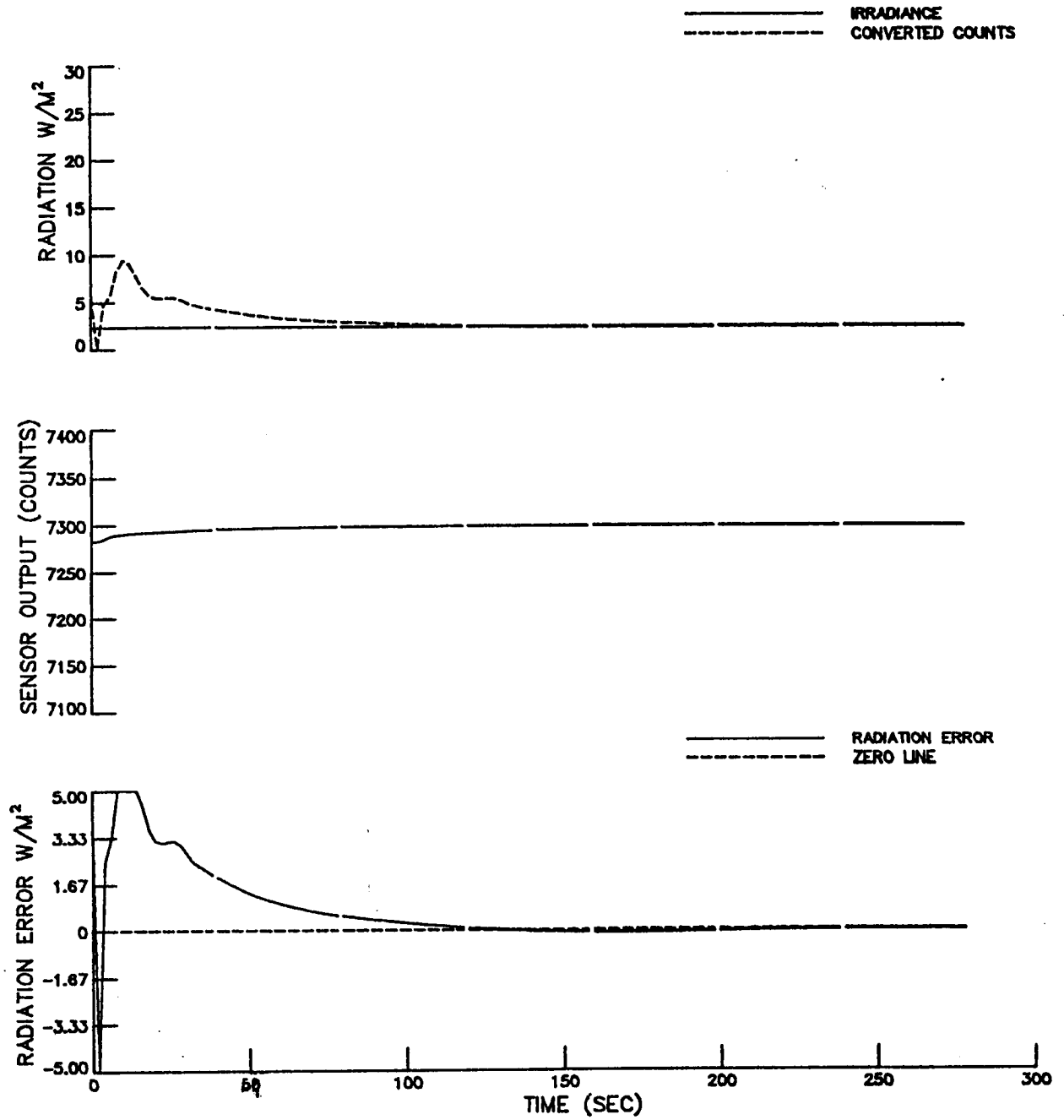
NON-SCANNER DYNAMIC RESPONSE

WFOV FILTERED SIMULATION



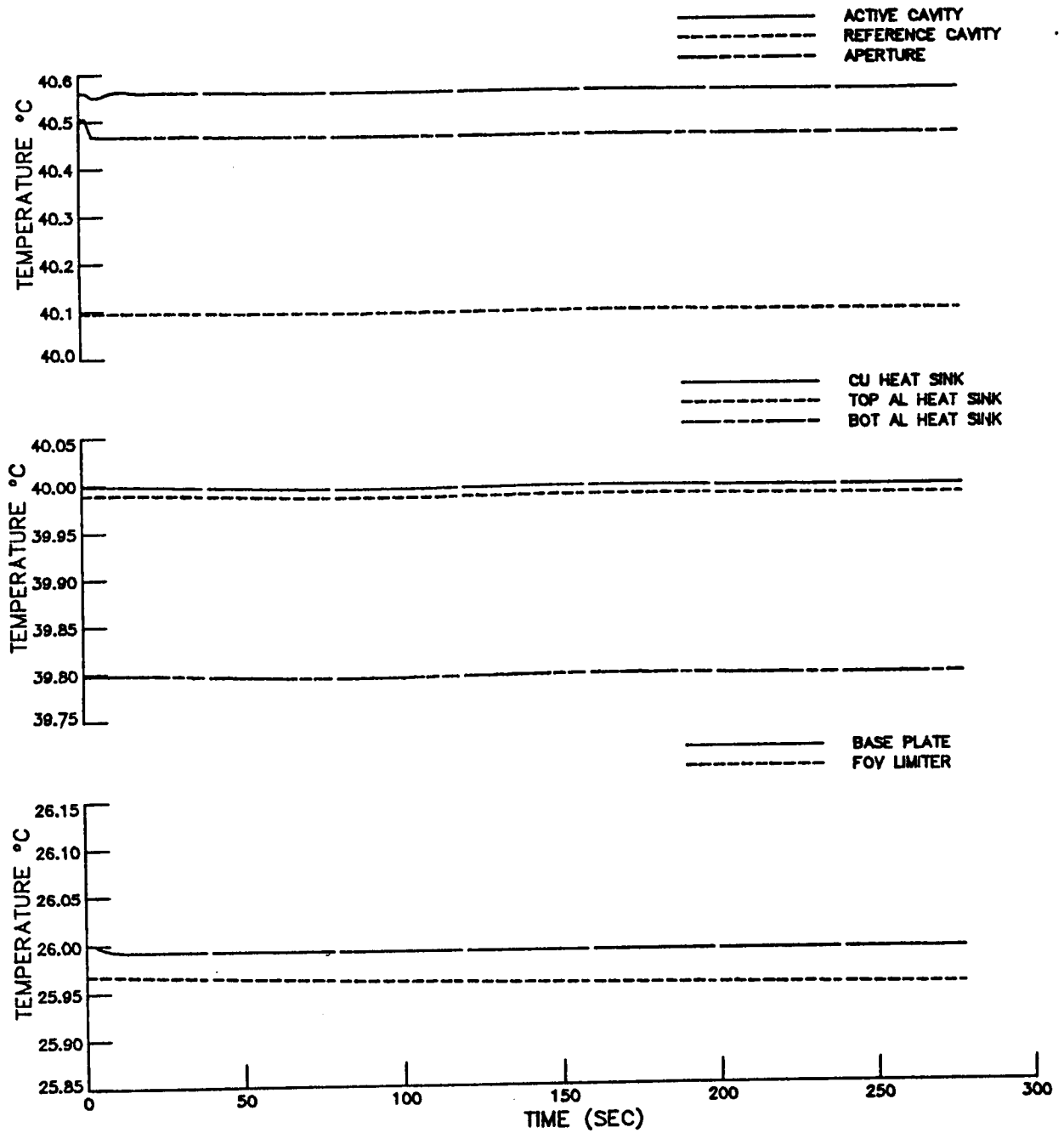
NON-SCANNER DYNAMIC RESPONSE

WFOV FILTERED SIMULATION



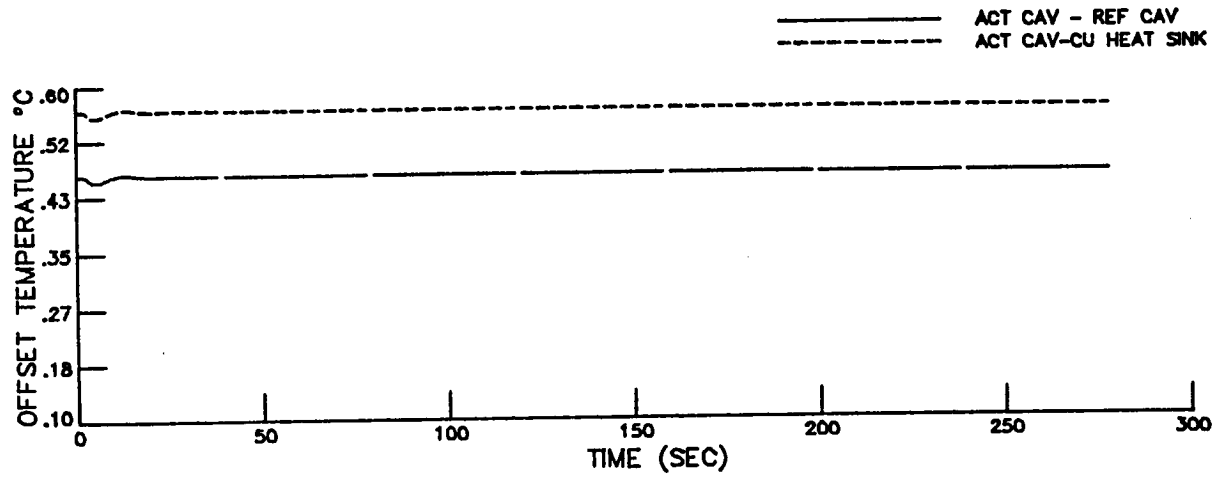
NON-SCANNER DYNAMIC RESPONSE

MFOV TOTAL SIMULATION



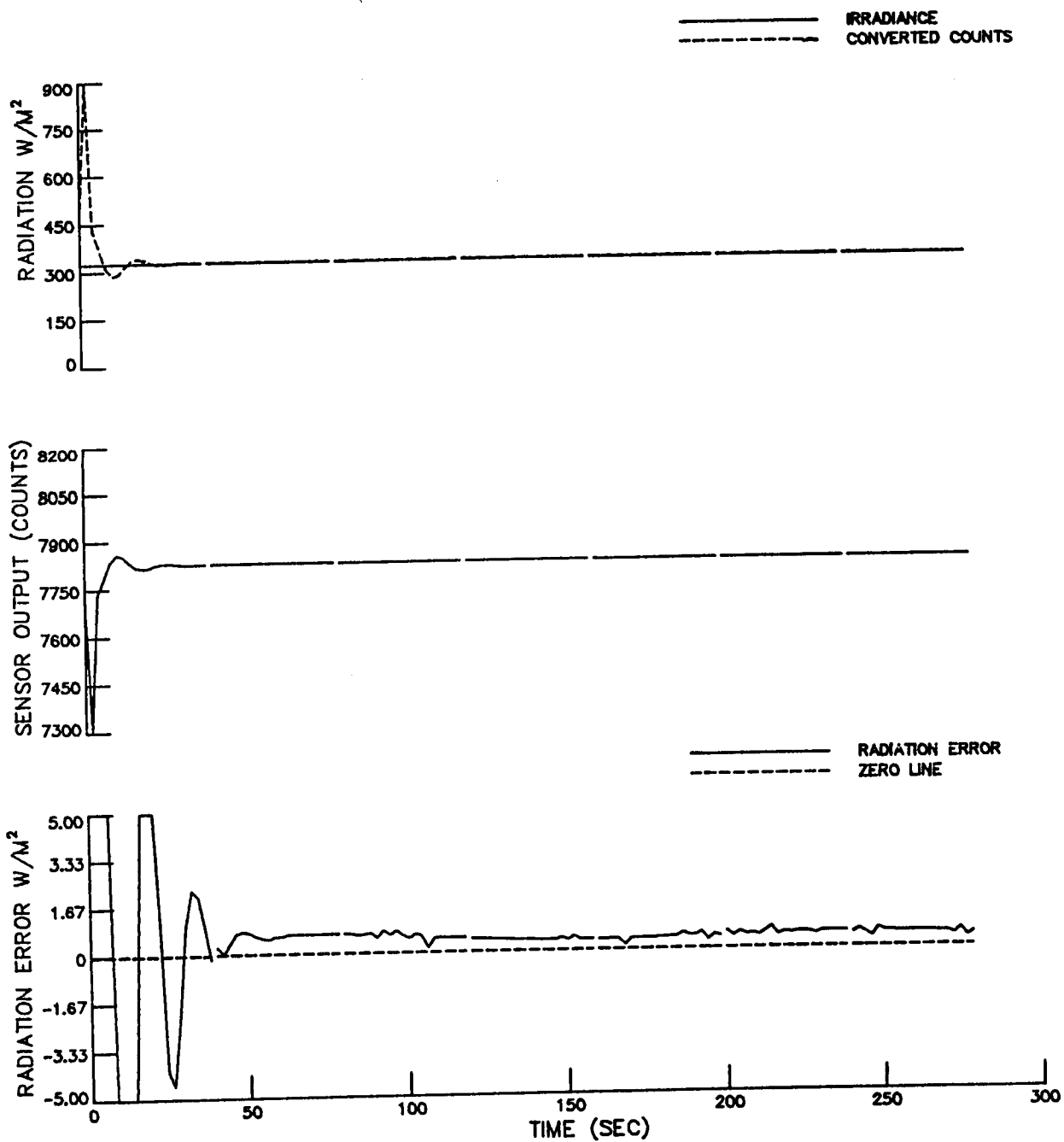
NON-SCANNER DYNAMIC RESPONSE

MFOV TOTAL SIMULATION



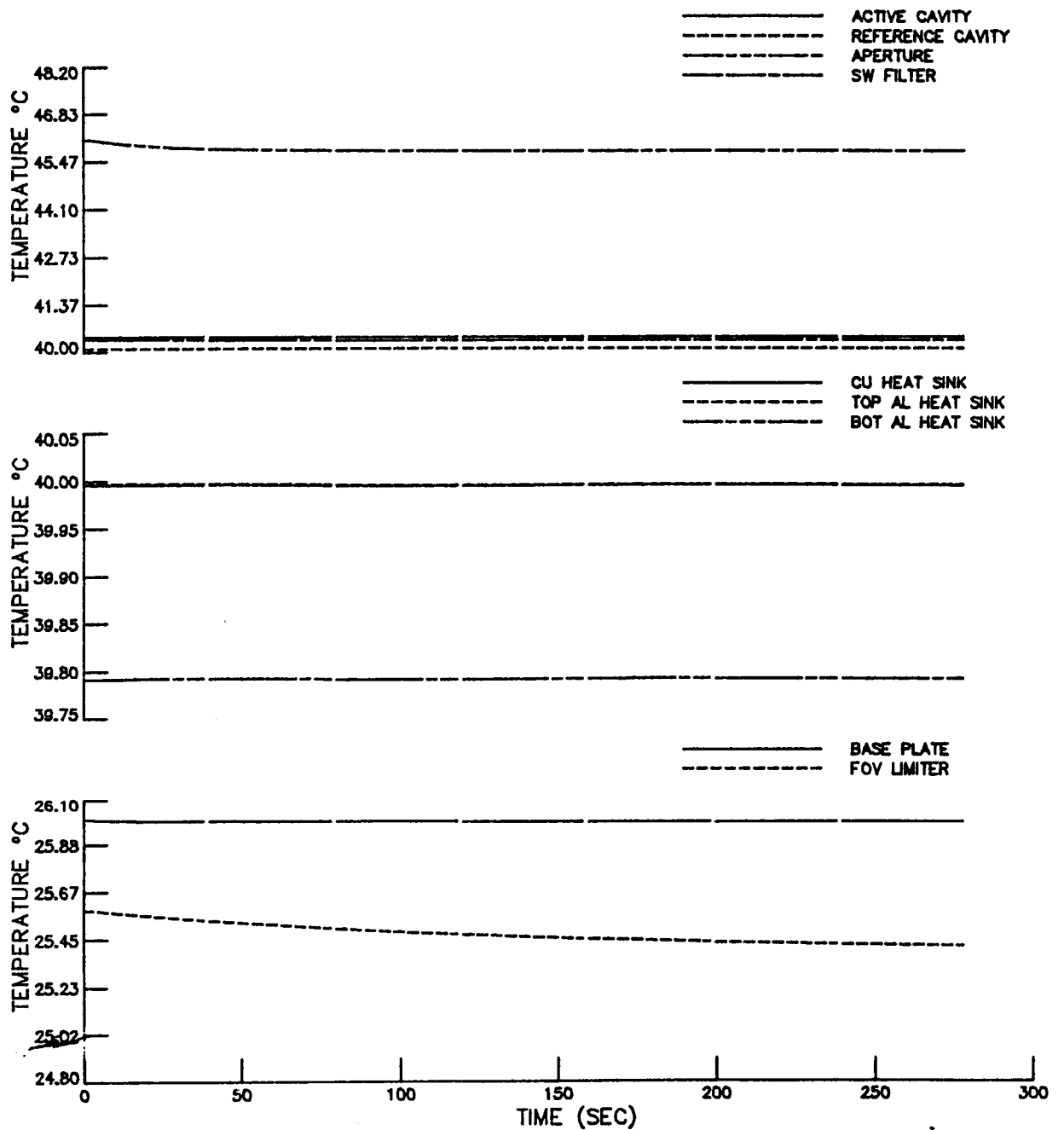
NON-SCANNER DYNAMIC RESPONSE

MFOV TOTAL SIMULATION



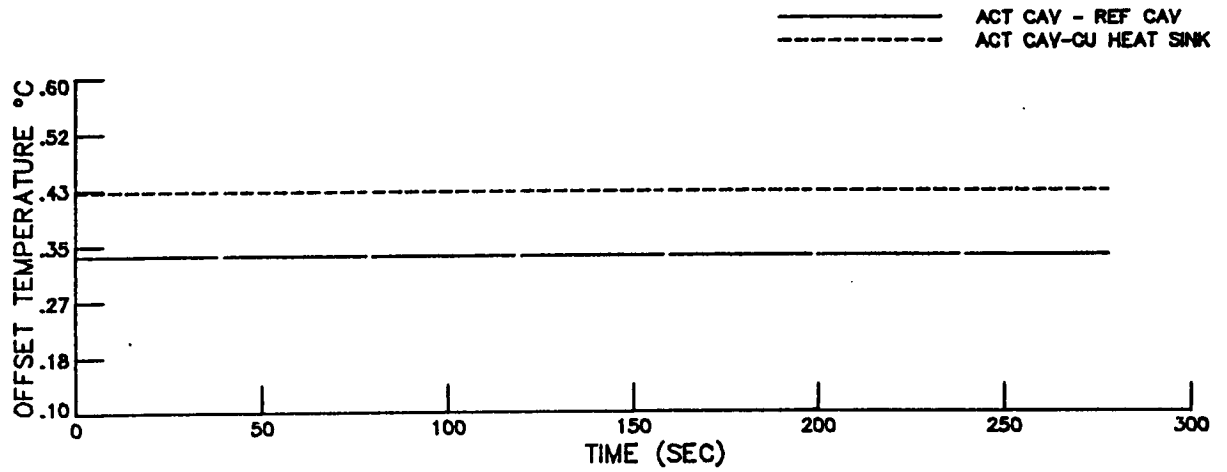
NON-SCANNER DYNAMIC RESPONSE

MFOV FILTERED SIMULATION



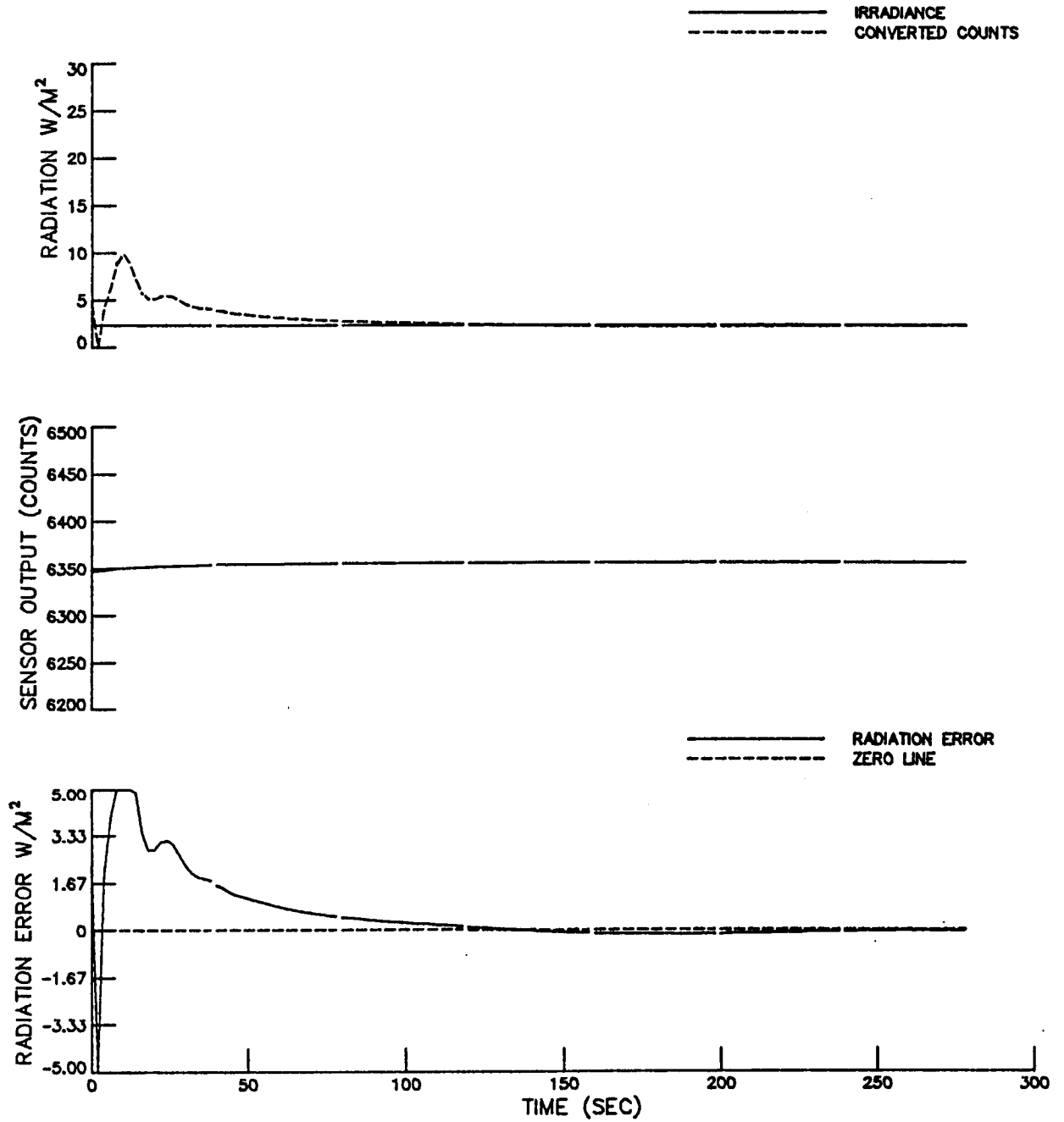
NON-SCANNER DYNAMIC RESPONSE

MFOV FILTERED SIMULATION



NON-SCANNER DYNAMIC RESPONSE

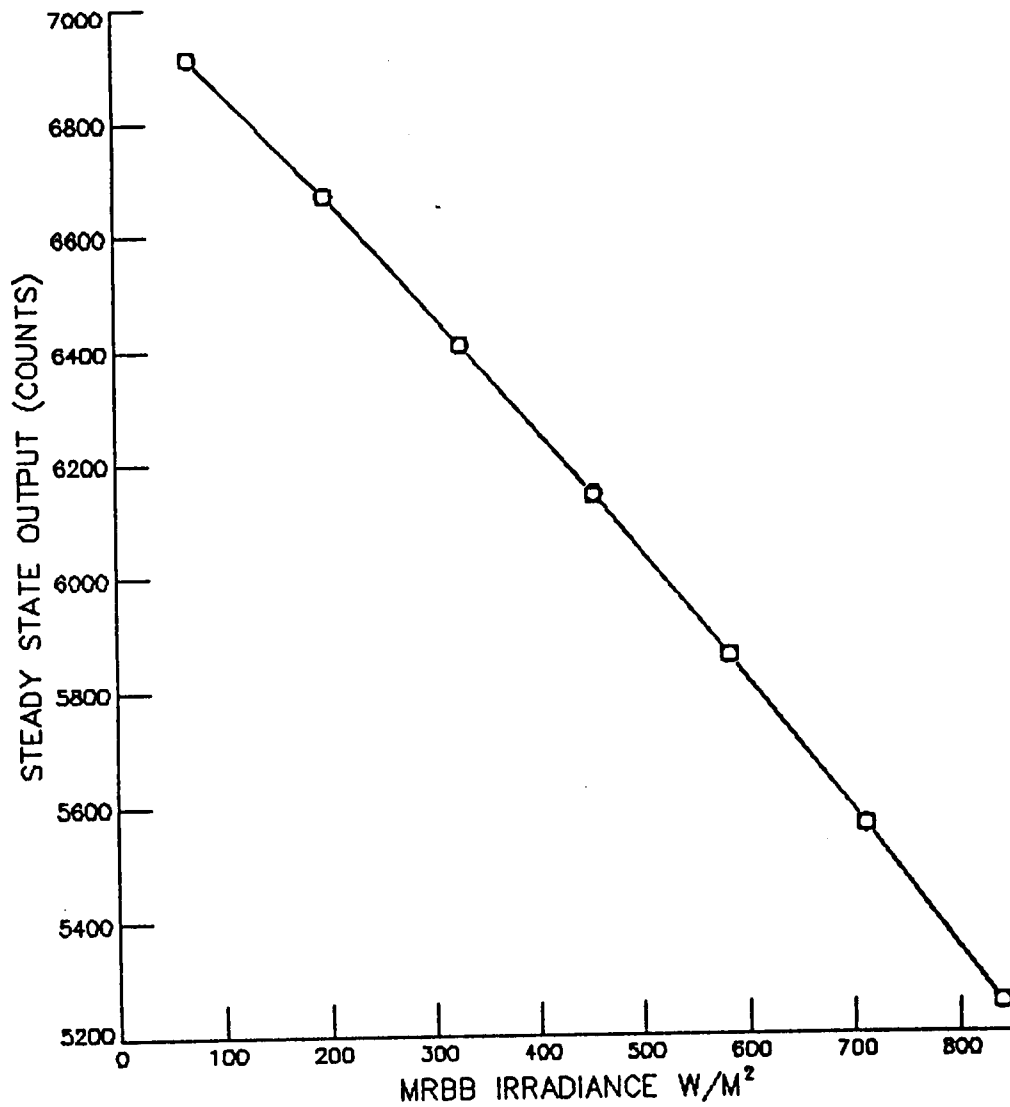
MFOV FILTERED SIMULATION



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

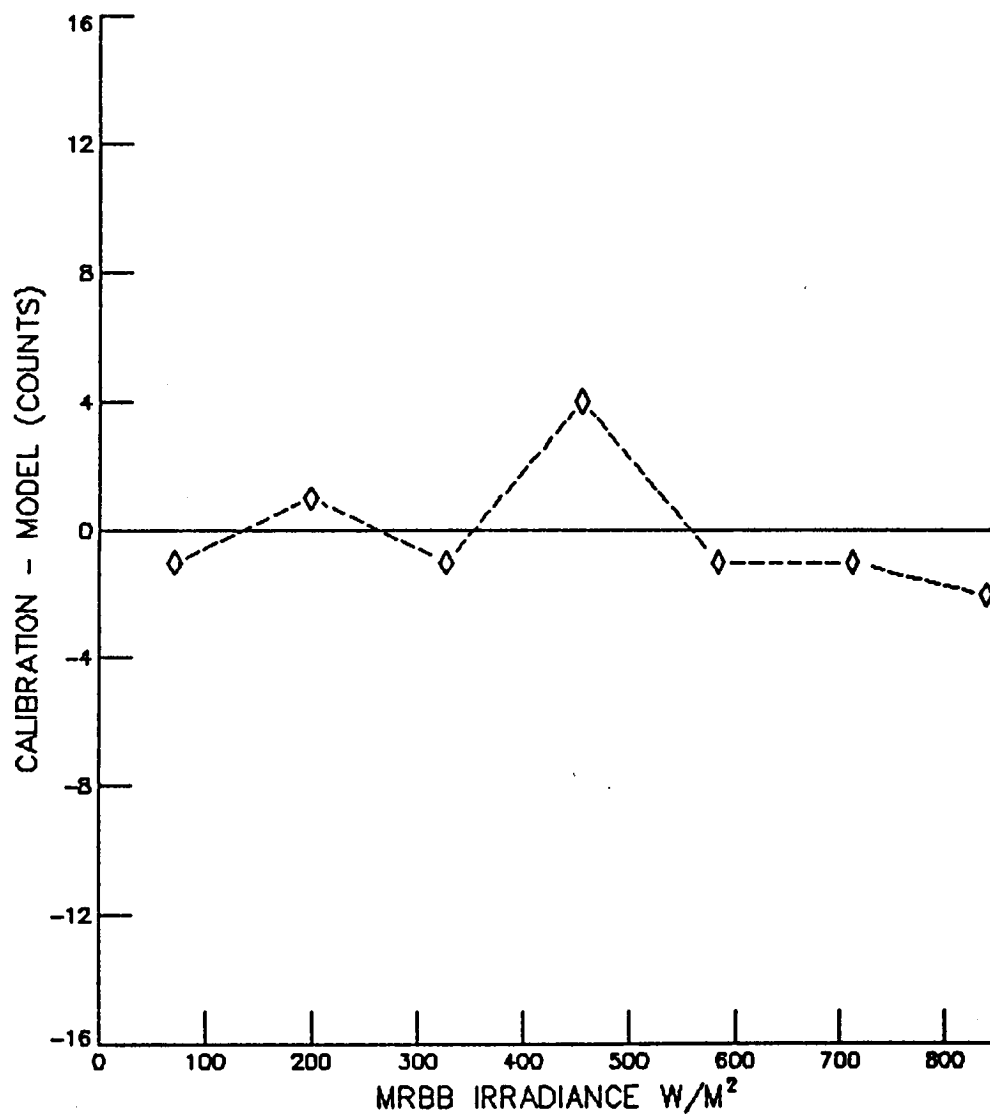
WFOV TOTAL

○ CALIBRATION
□ MODEL



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

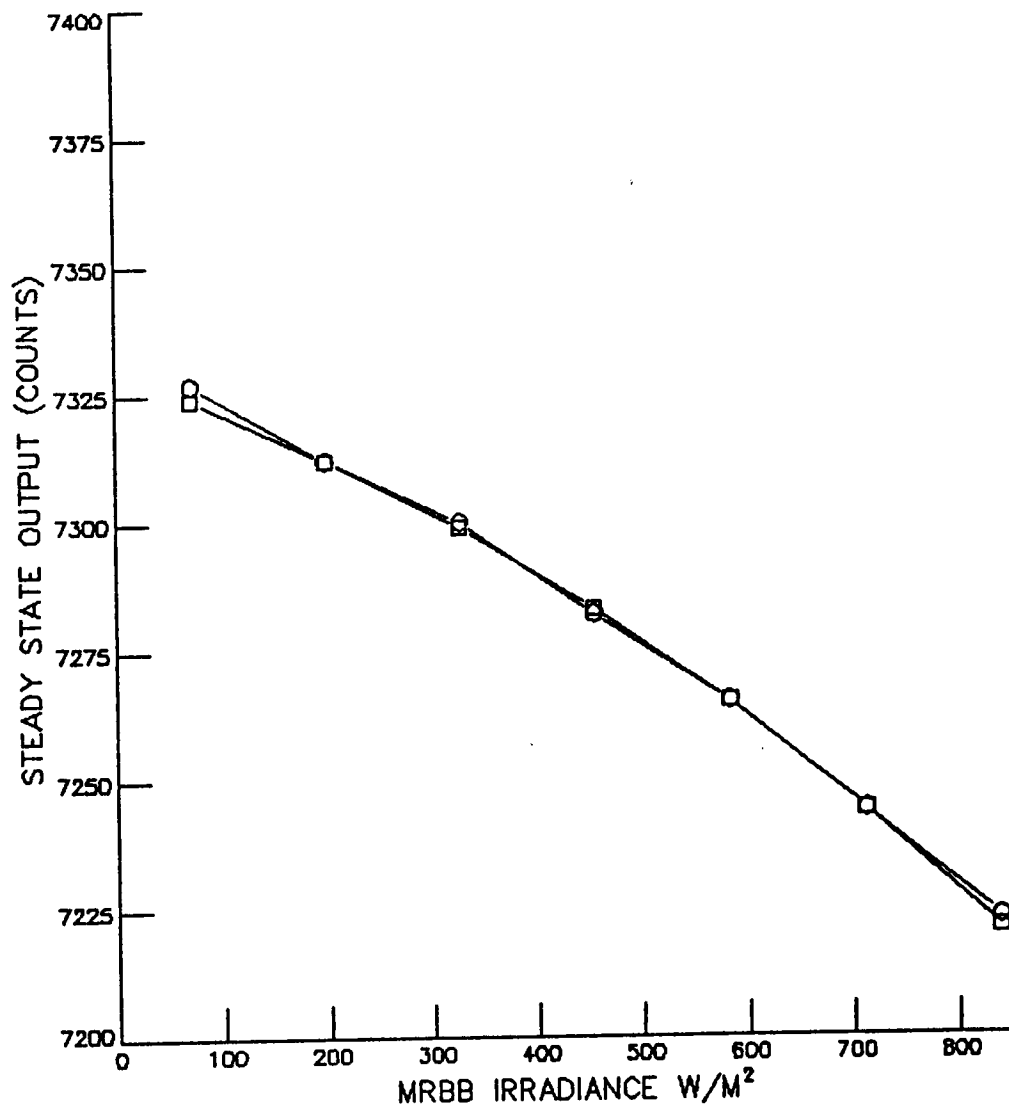
WFOV TOTAL



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION OF CALIBRATION DATA

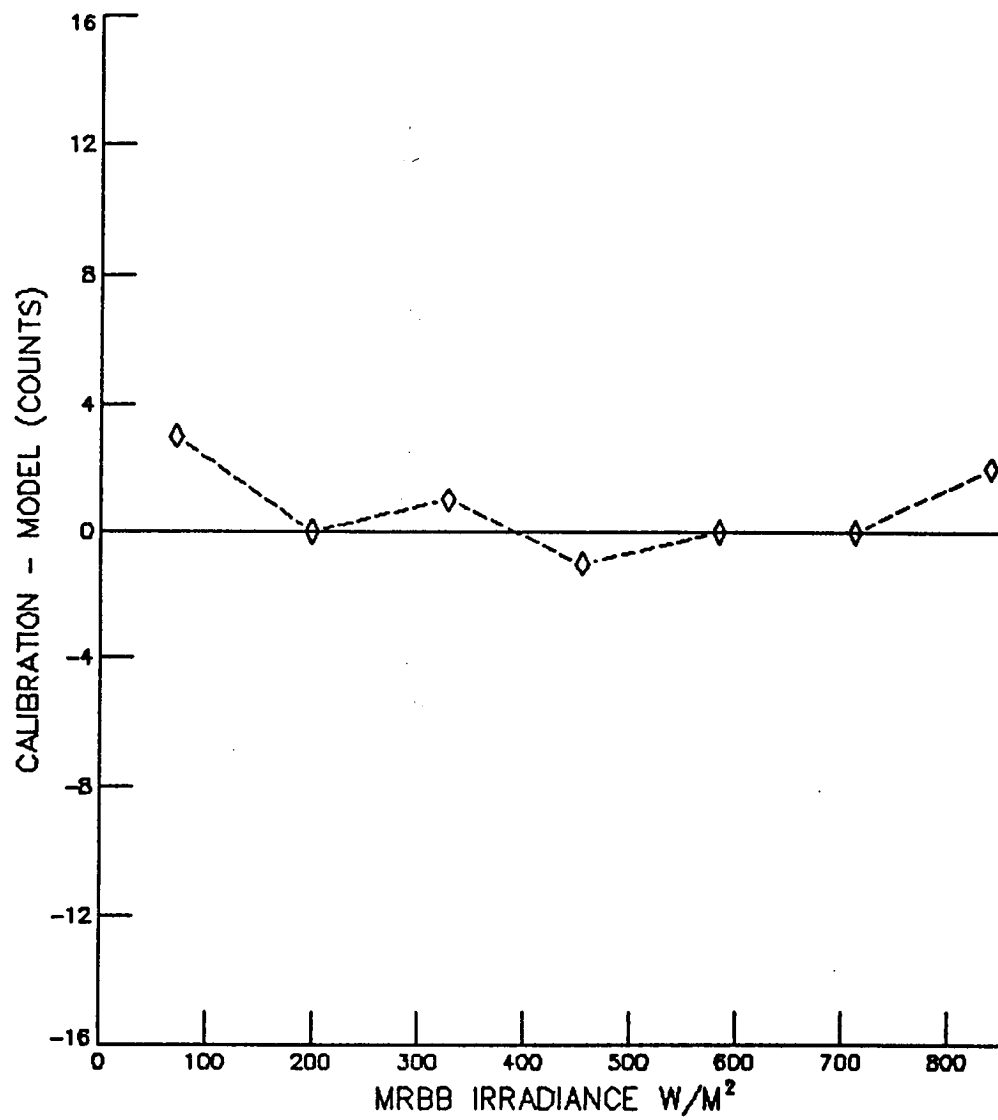
WFOV FILTERED

○ CALIBRATION
□ MODEL



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

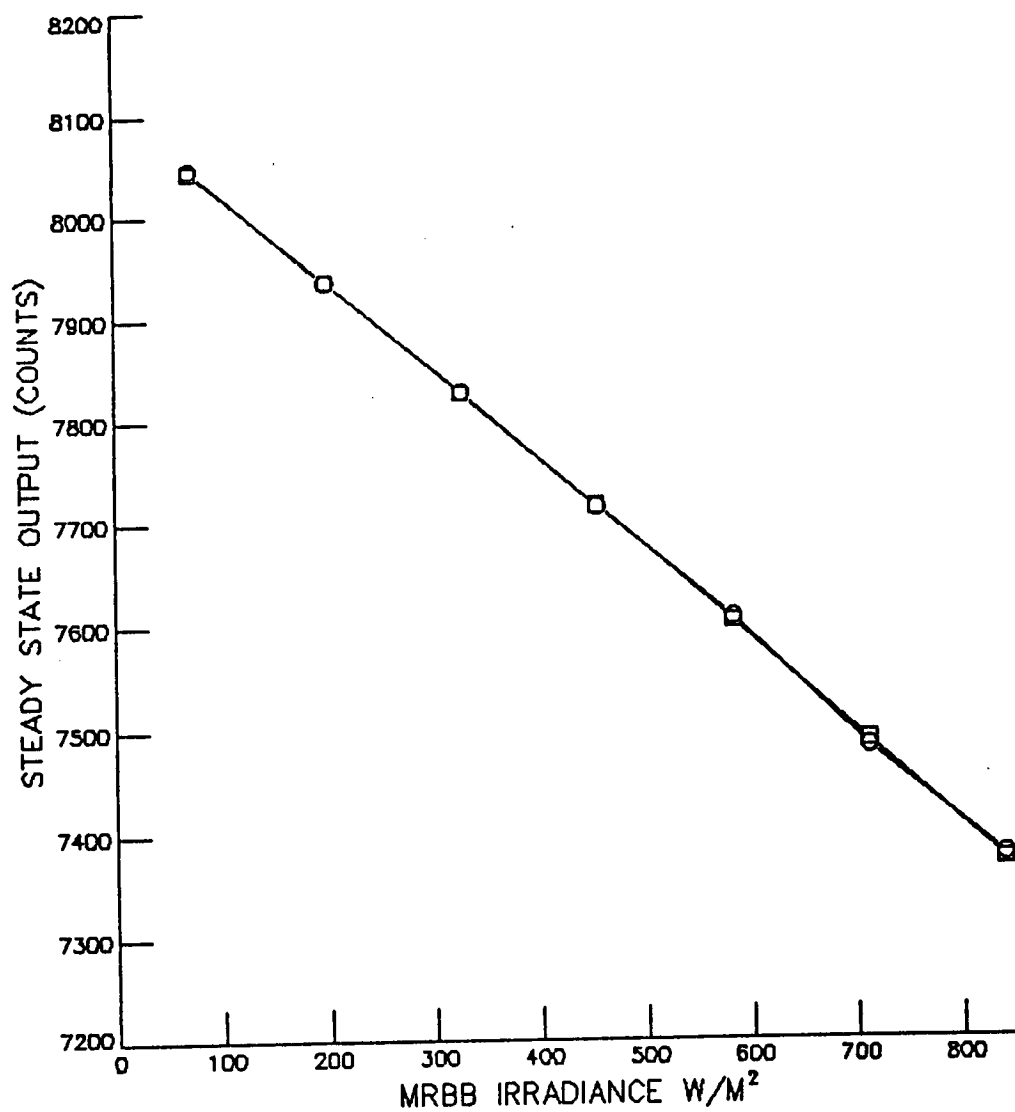
WFOV FILTERED



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

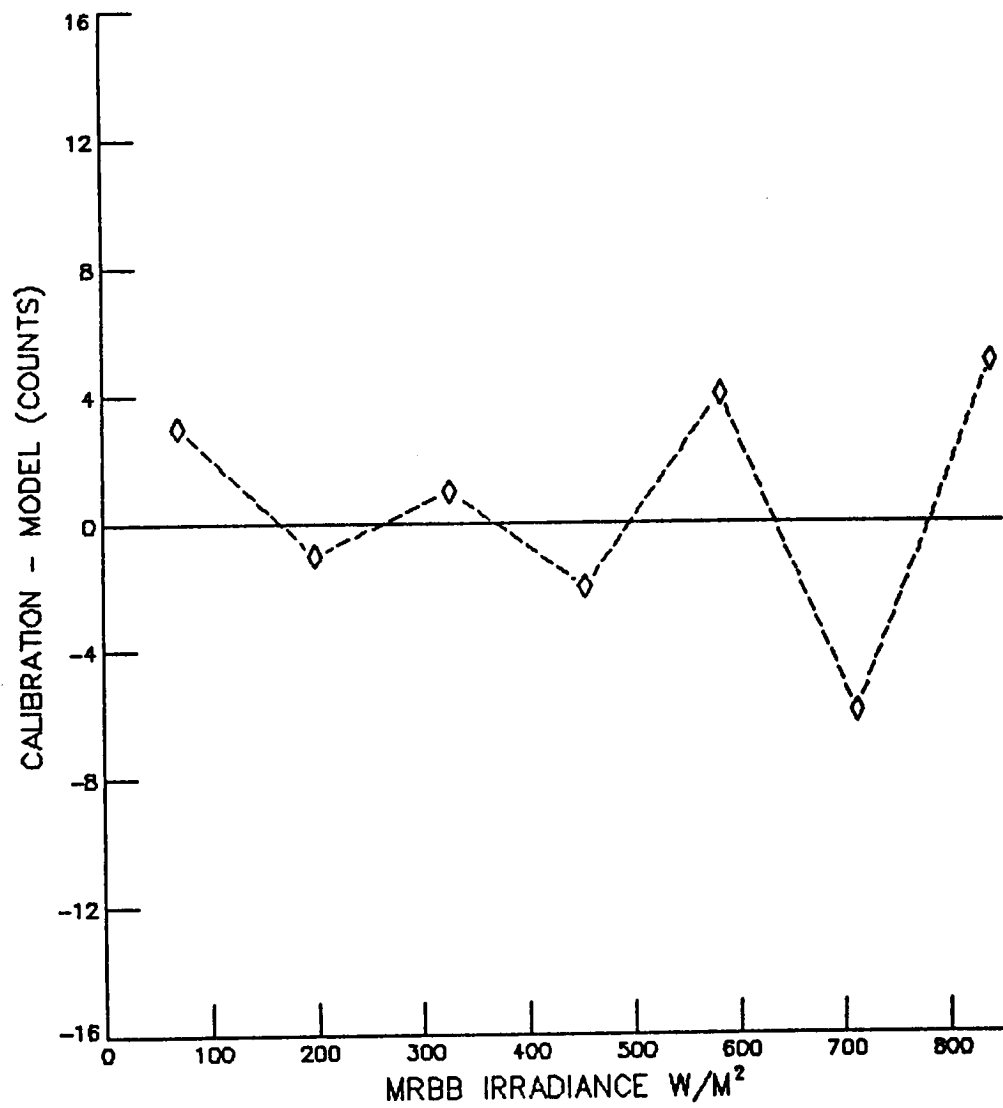
MFOV TOTAL

○ CALIBRATION
□ MODEL



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

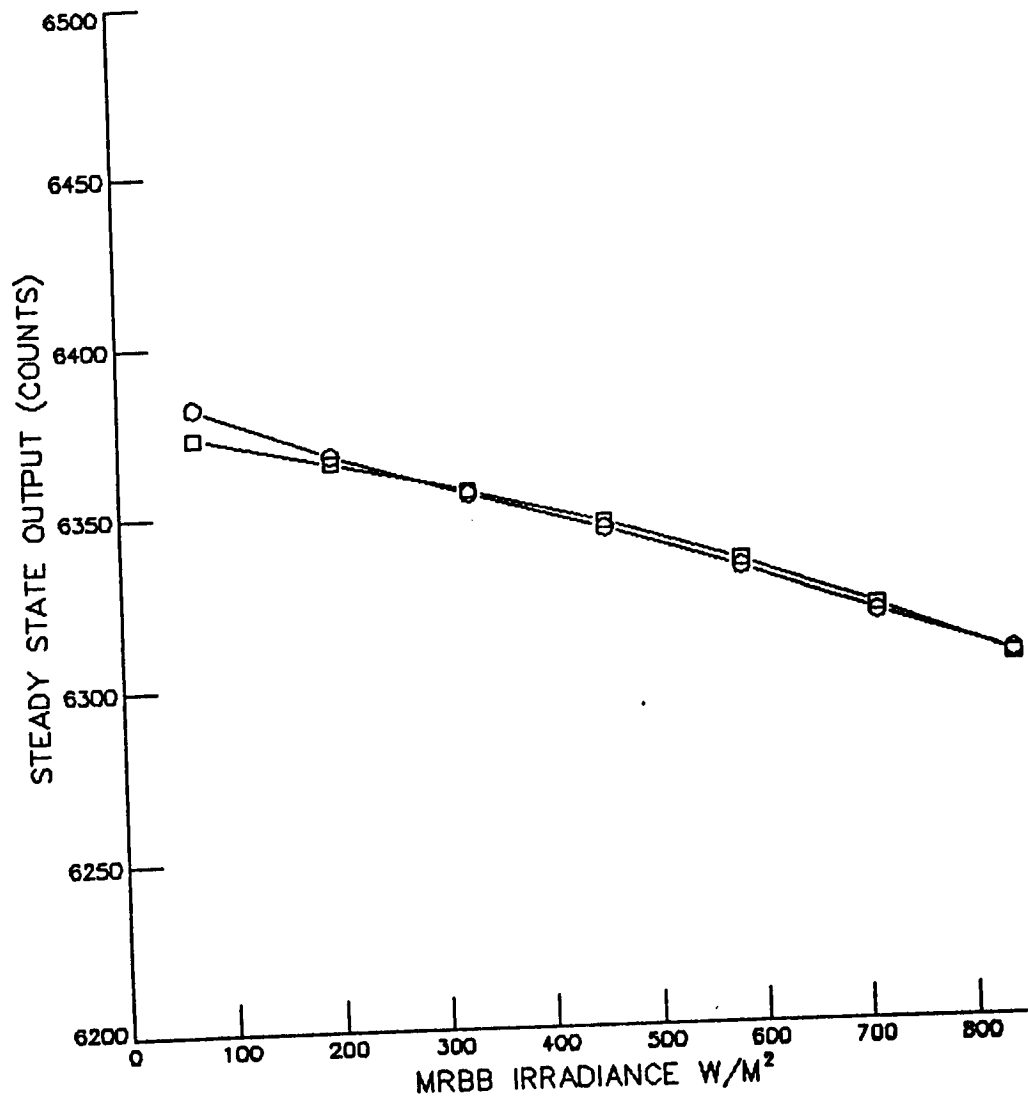
MFOV TOTAL



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

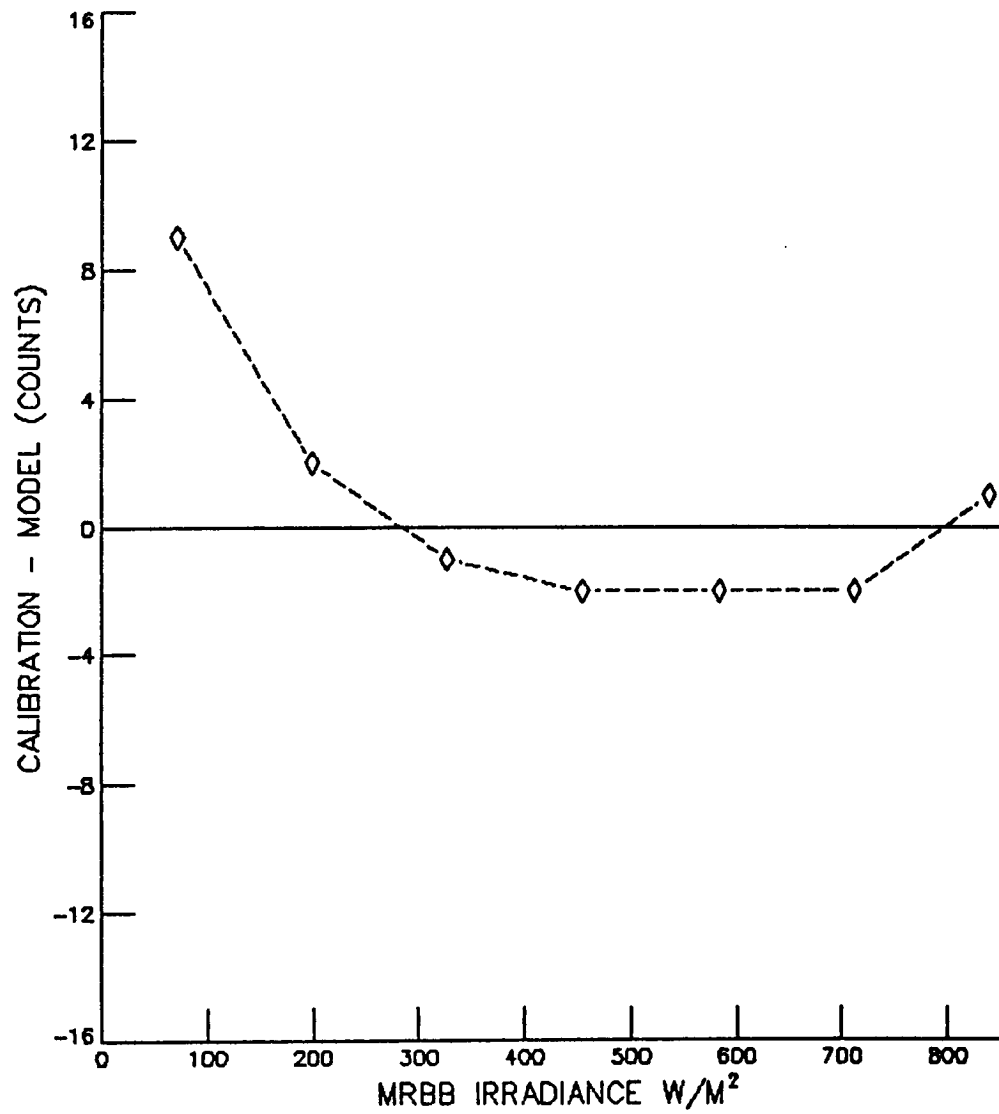
MFOV FILTERED

○ CALIBRATION
□ MODEL



NON-SCANNER STEADY STATE RESPONSE
COMPARISON OF SIMULATION TO CALIBRATION DATA

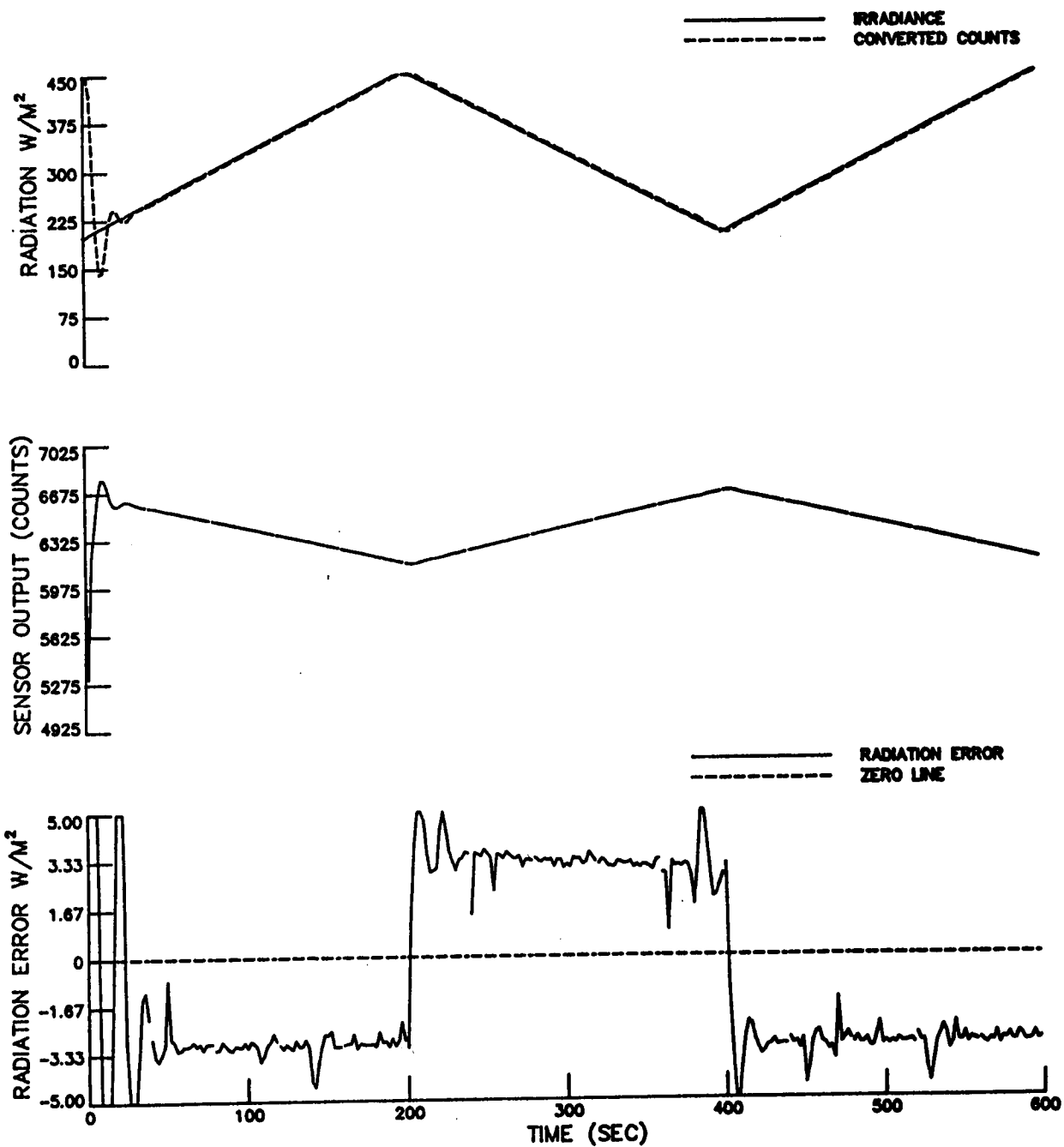
MFOV FILTERED



NON-SCANNER DYNAMIC RESPONSE

WFOV TOTAL SIMULATION

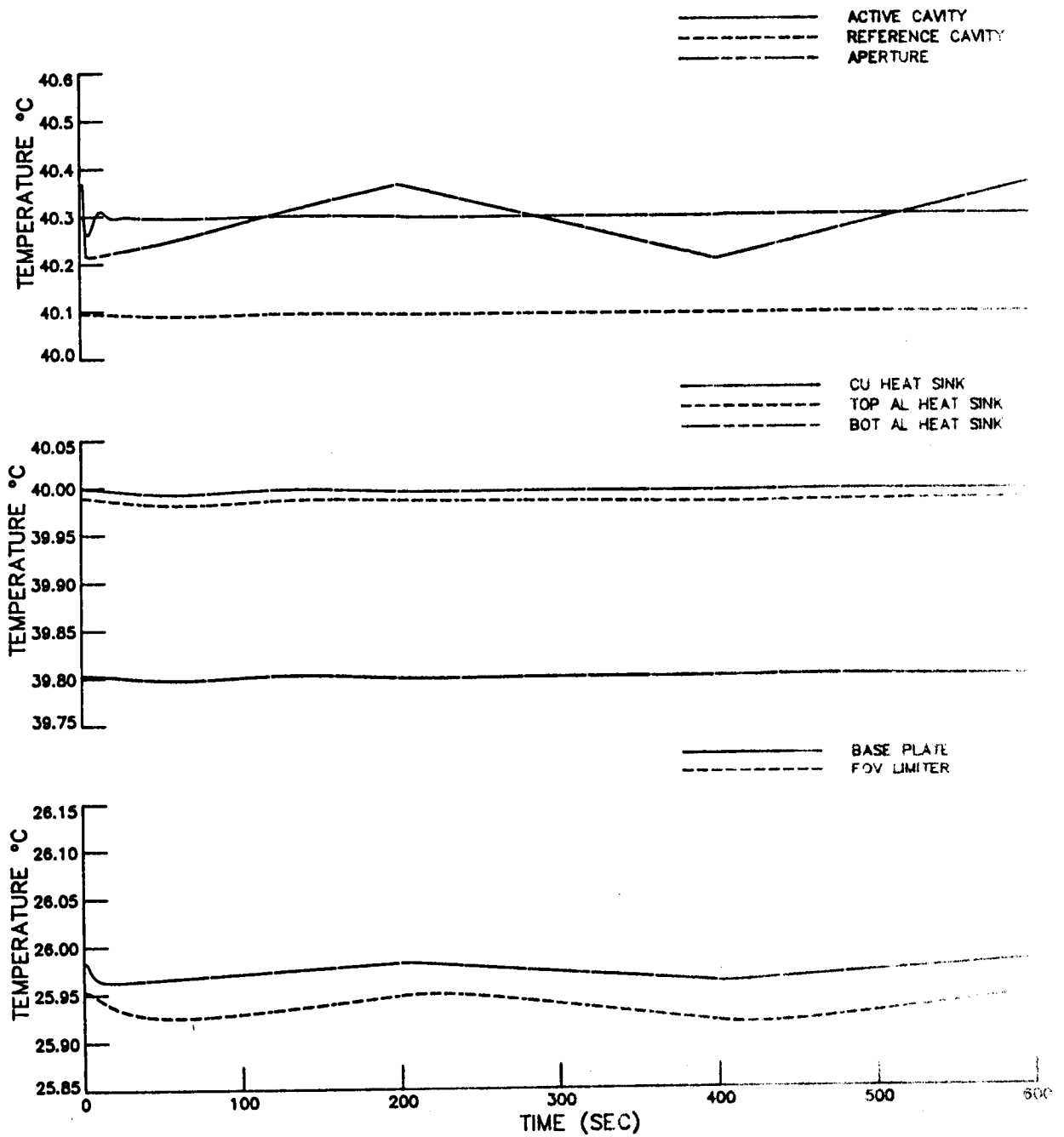
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

WFOV TOTAL SIMULATION

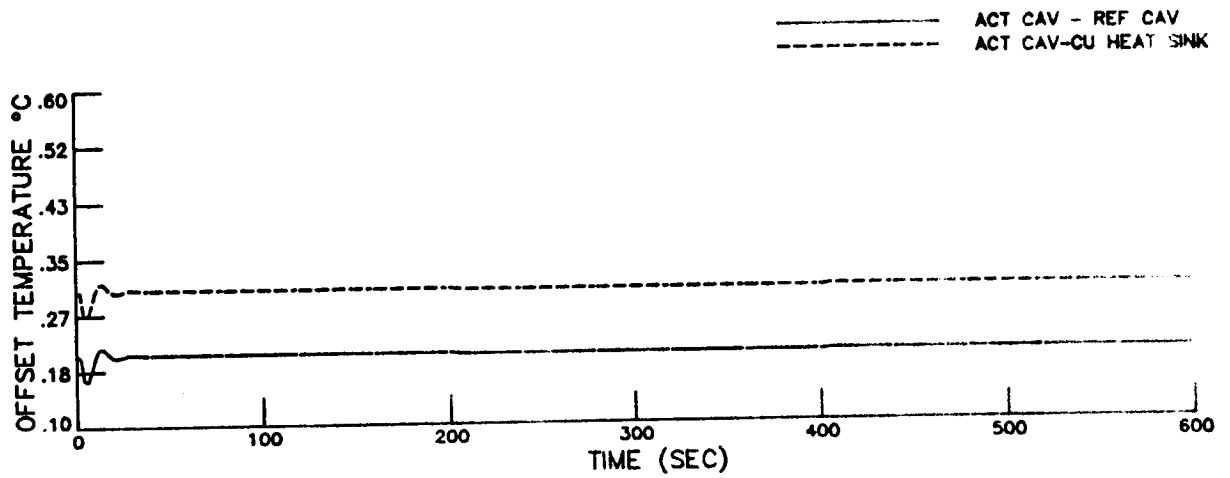
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

WFOV TOTAL SIMULATION

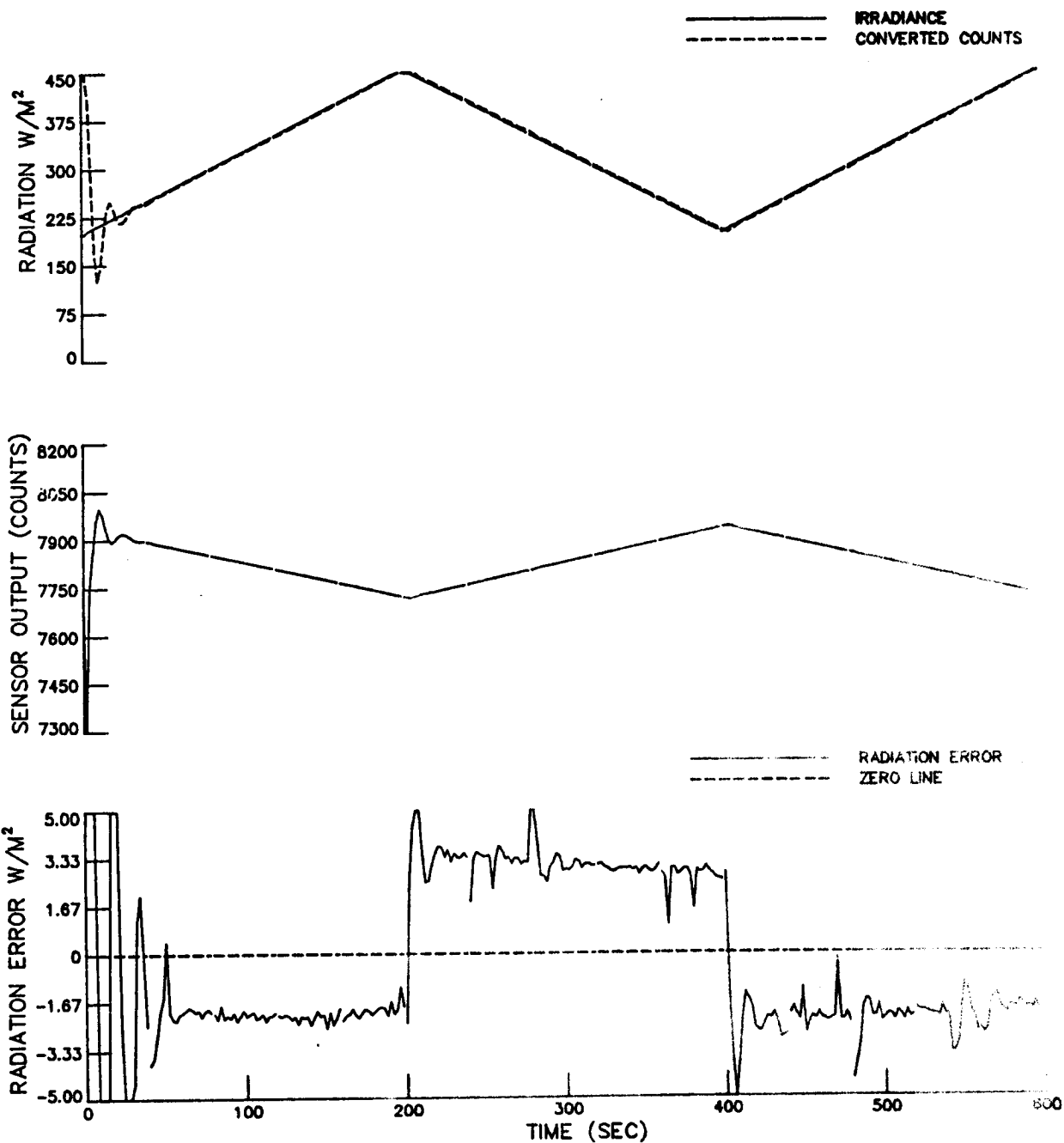
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

MFOV TOTAL SIMULATION

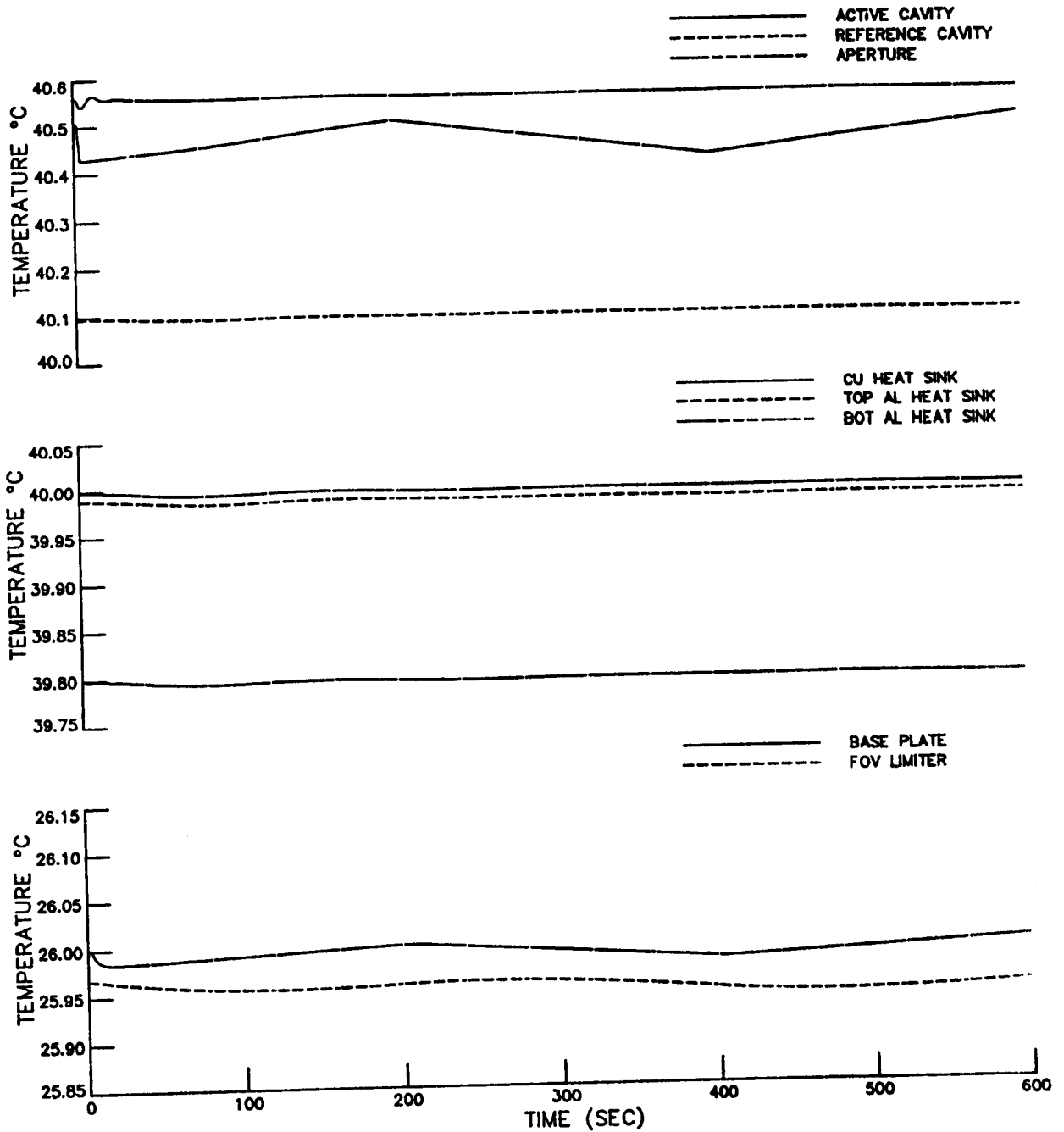
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

MFOV TOTAL SIMULATION

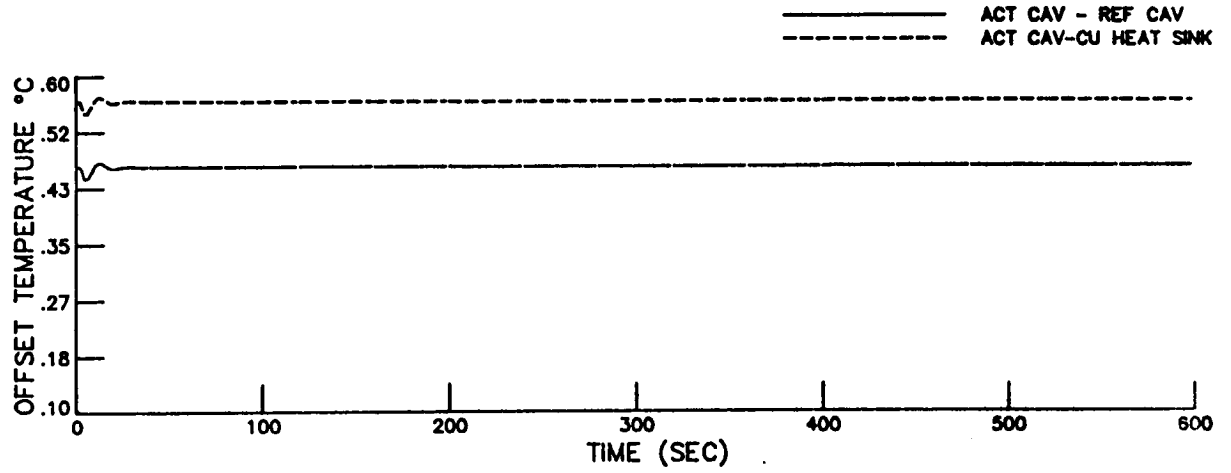
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

MFOV TOTAL SIMULATION

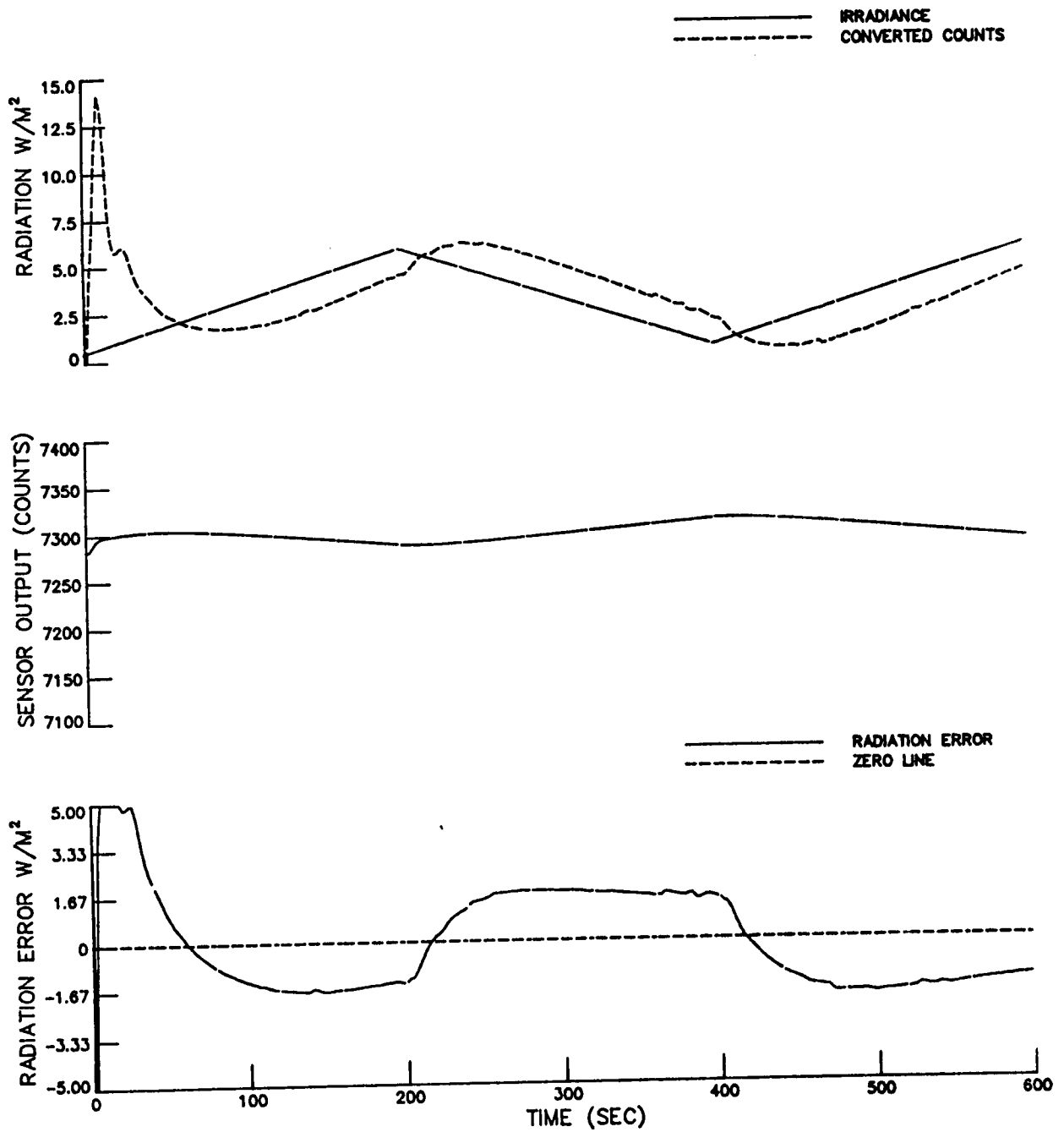
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

WFOV FILTERED SIMULATION

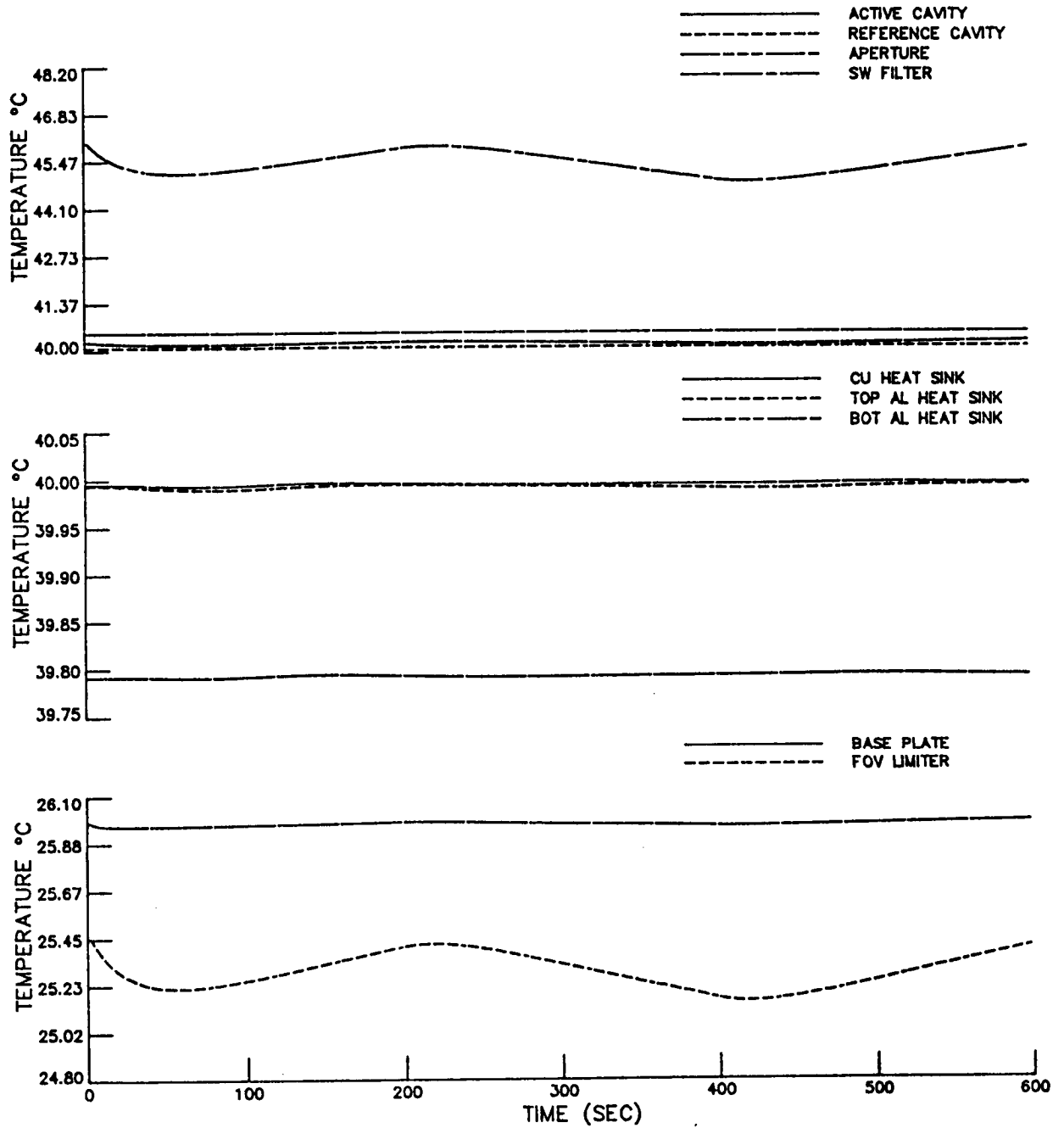
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

WFOV FILTERED SIMULATION

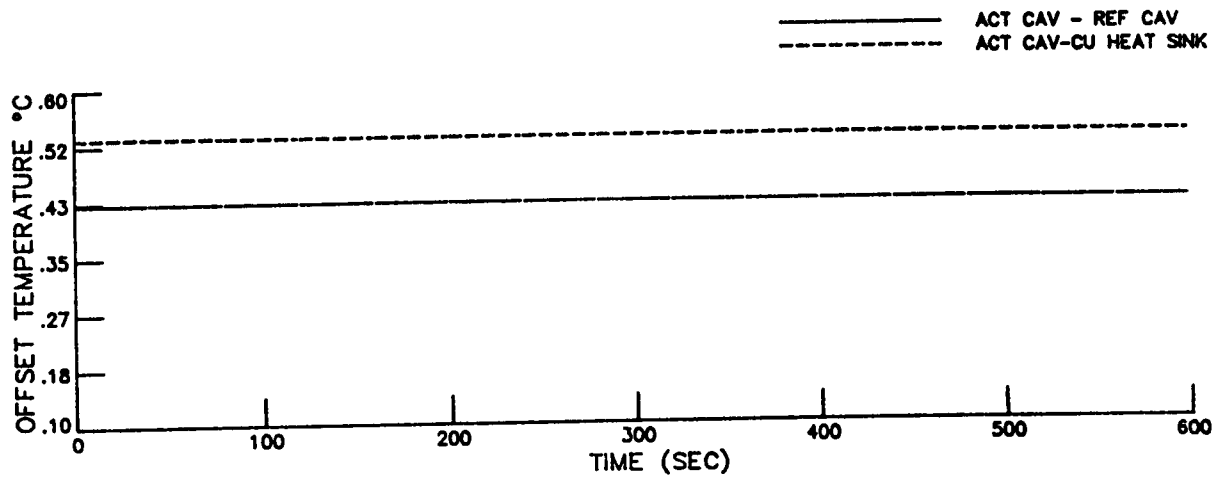
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

WFOV FILTERED SIMULATION

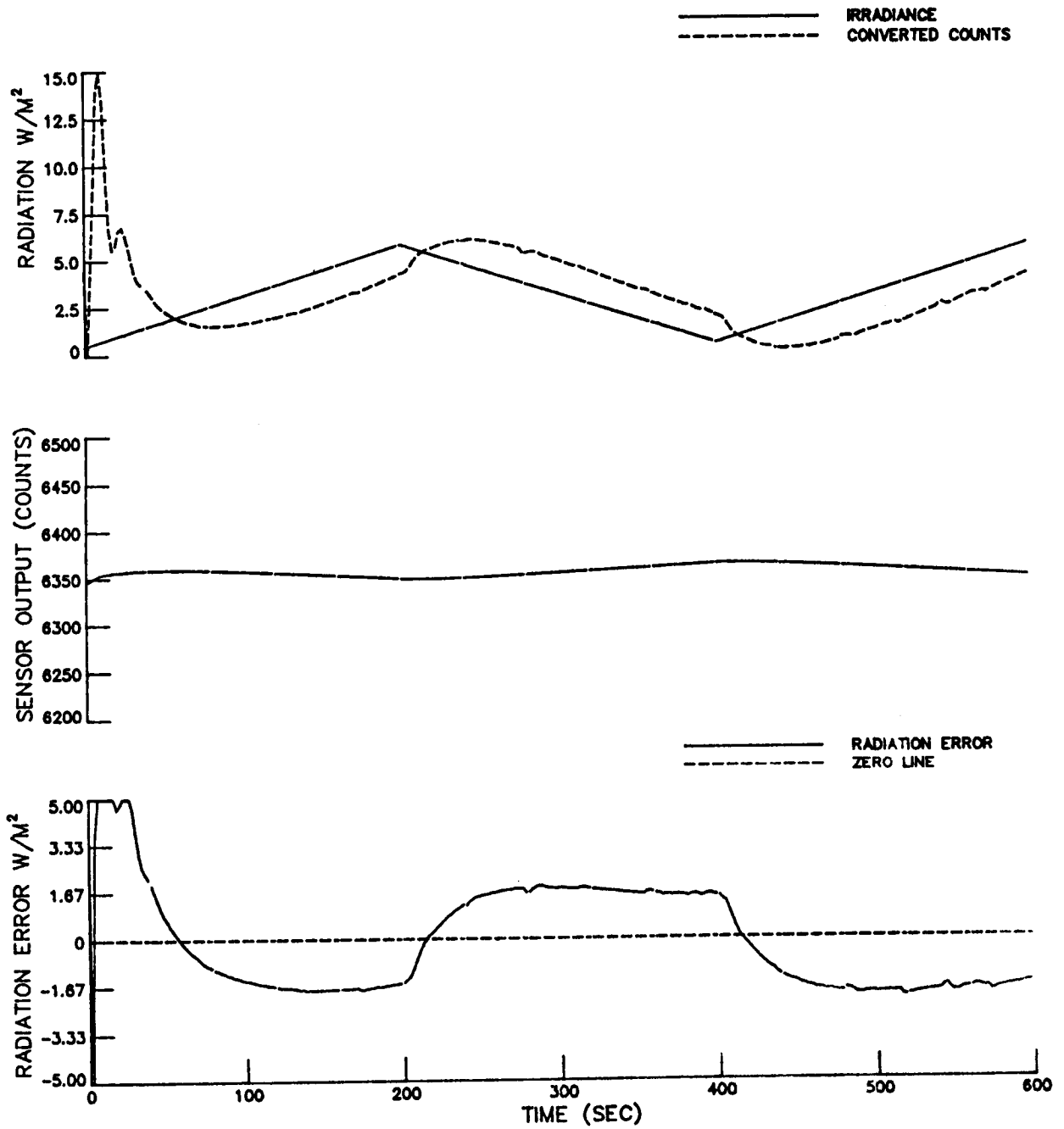
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

MFOV FILTERED SIMULATION

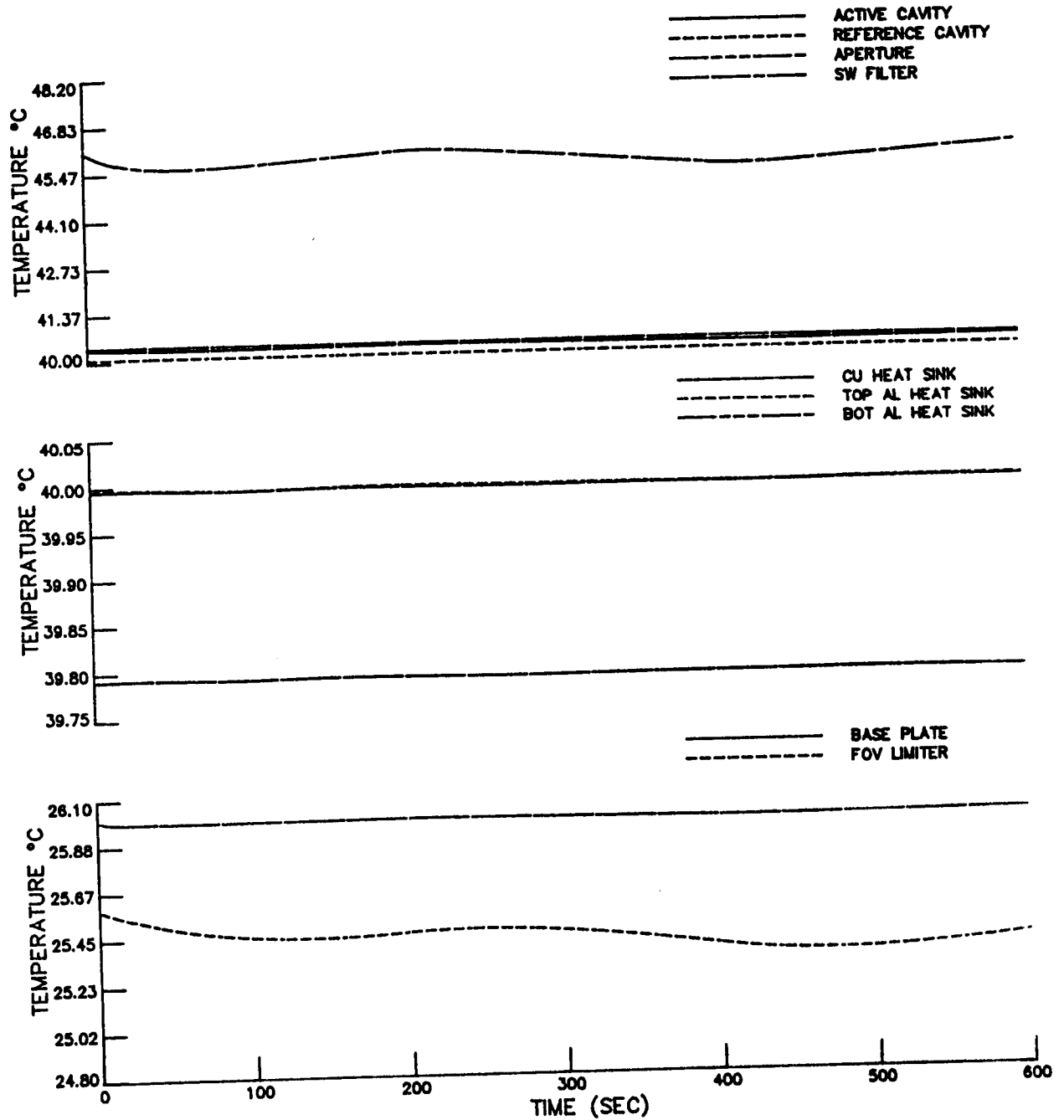
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

MFOV FILTERED SIMULATION

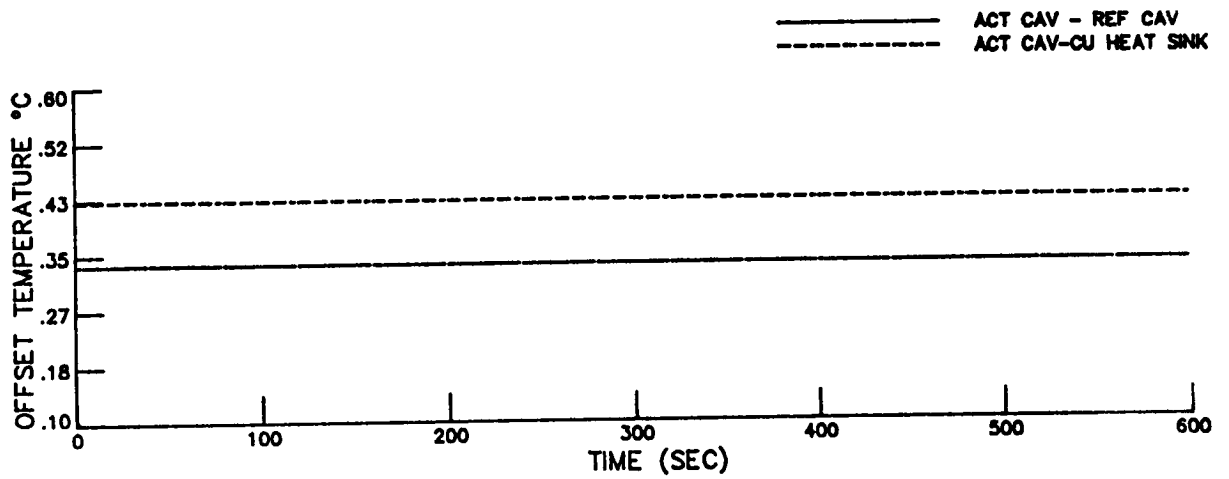
TIME DELAY (LAG) ESTIMATION



NON-SCANNER DYNAMIC RESPONSE

MFOV FILTERED SIMULATION

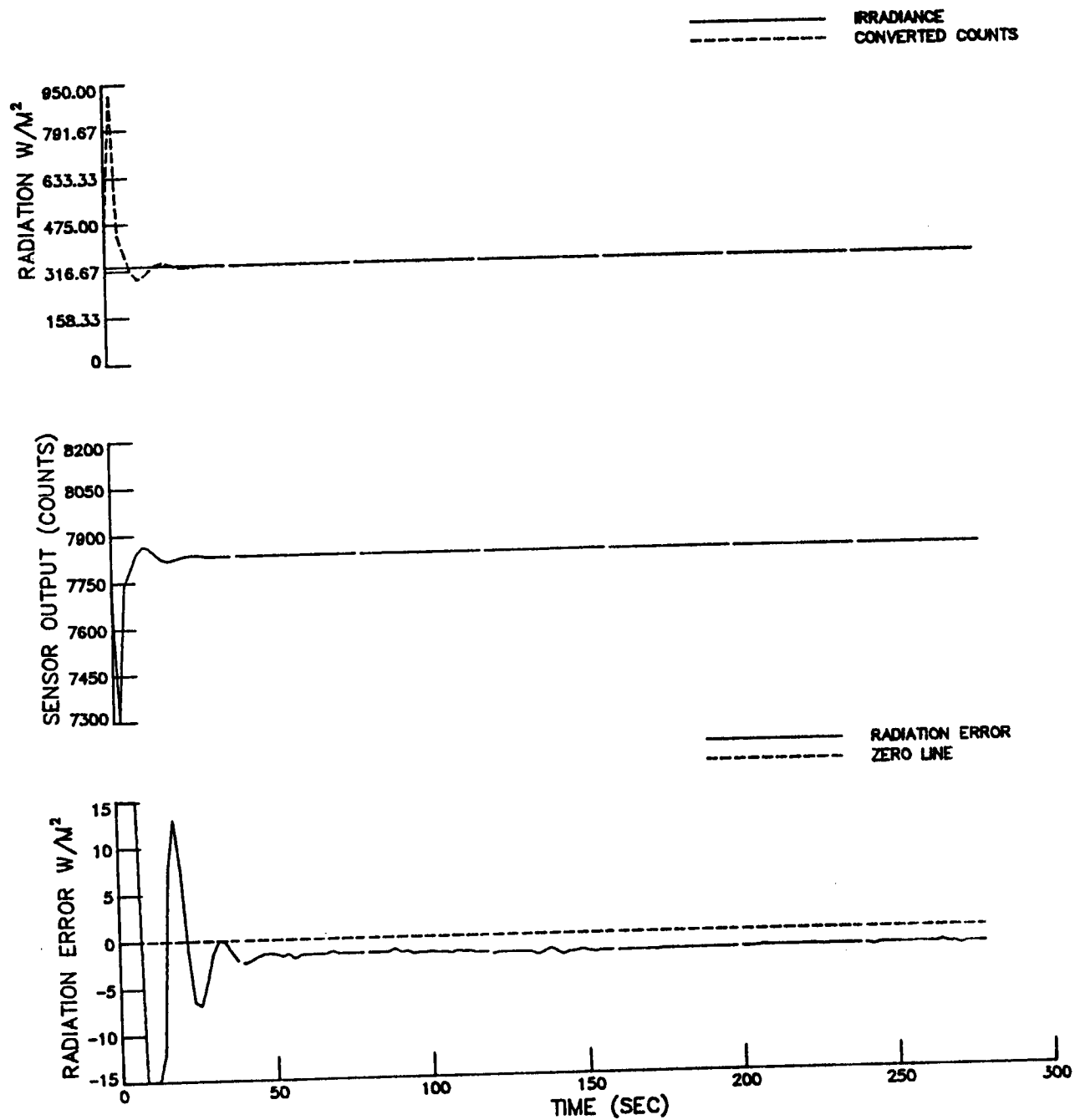
TIME DELAY (LAG) ESTIMATION



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

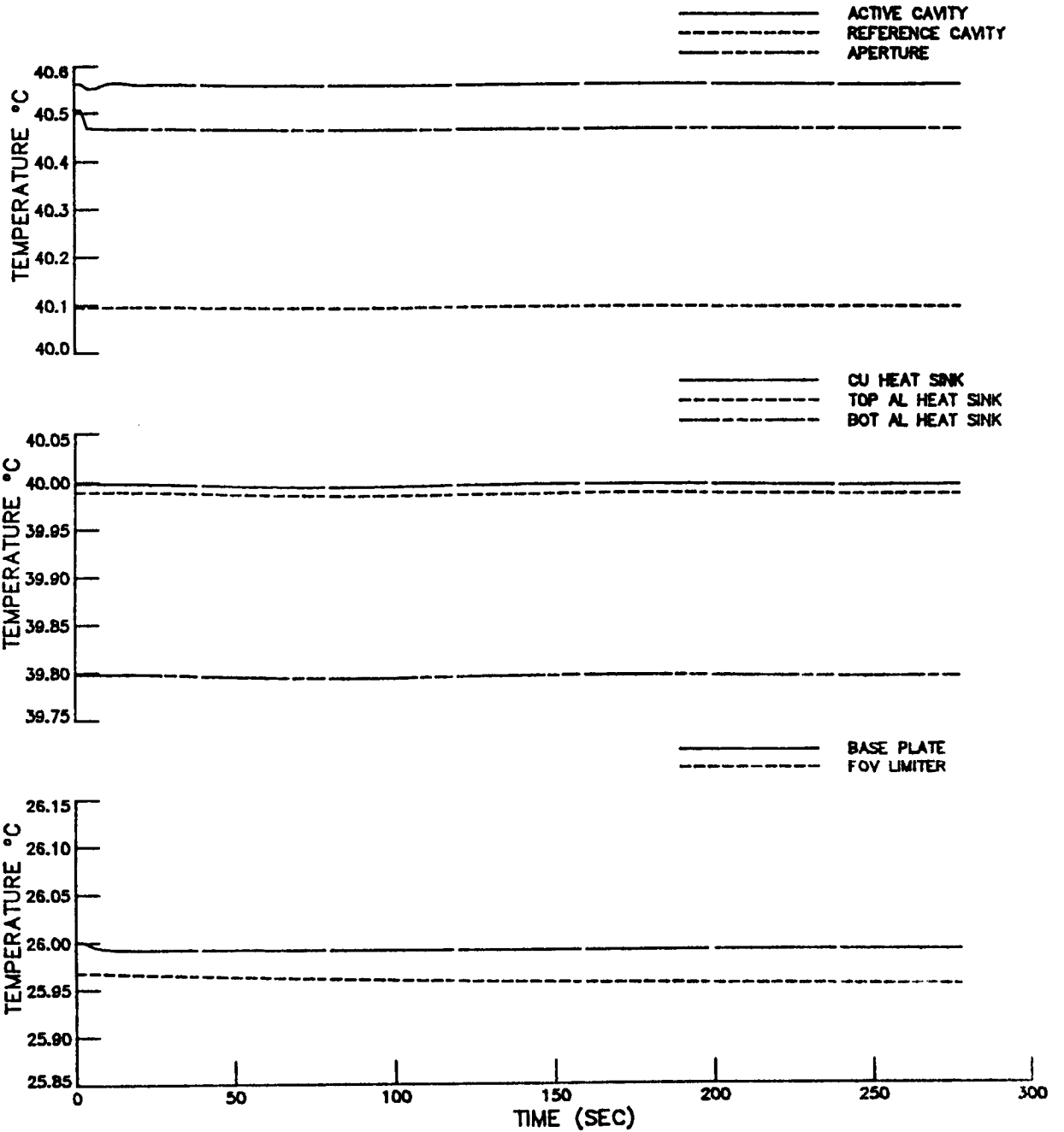
ACTIVE CAVITY EMISSIVITY CHANGED BY .02 (NOMINAL .92)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

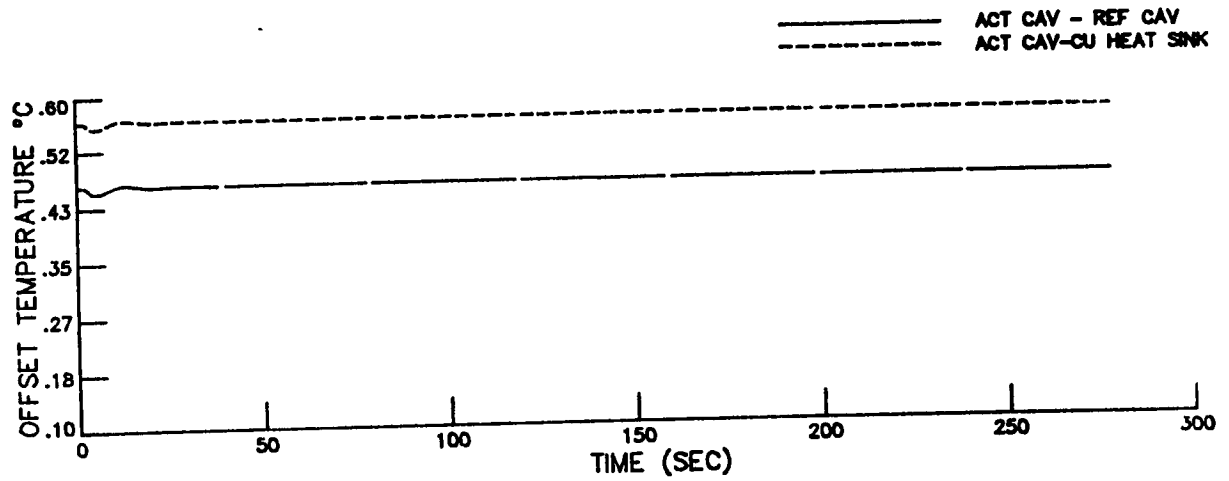
ACTIVE CAVITY EMISSIVITY CHANGED BY .02 (NOMINAL .92)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

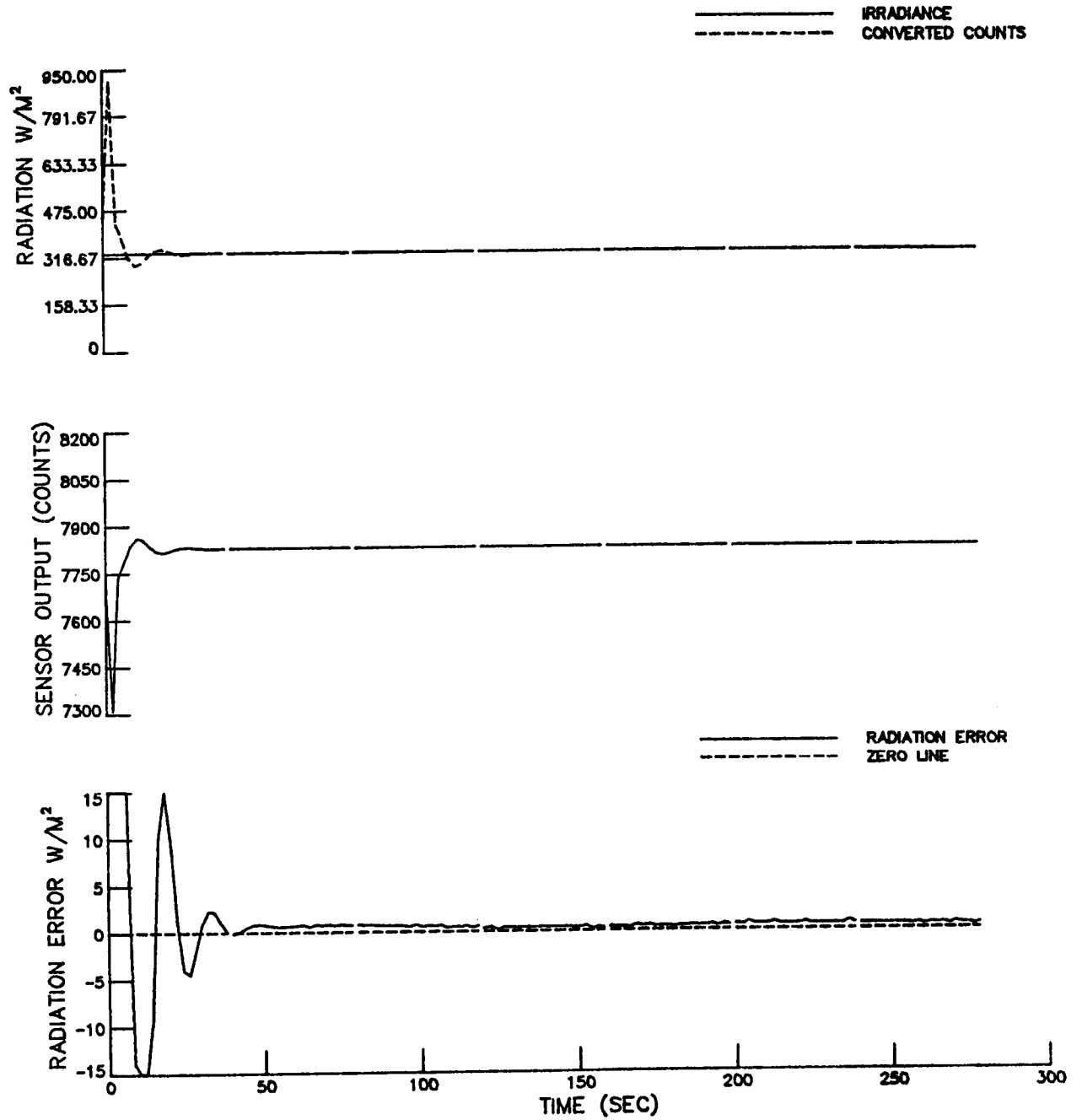
ACTIVE CAVITY EMISSIVITY CHANGED BY .02 (NOMINAL .92)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

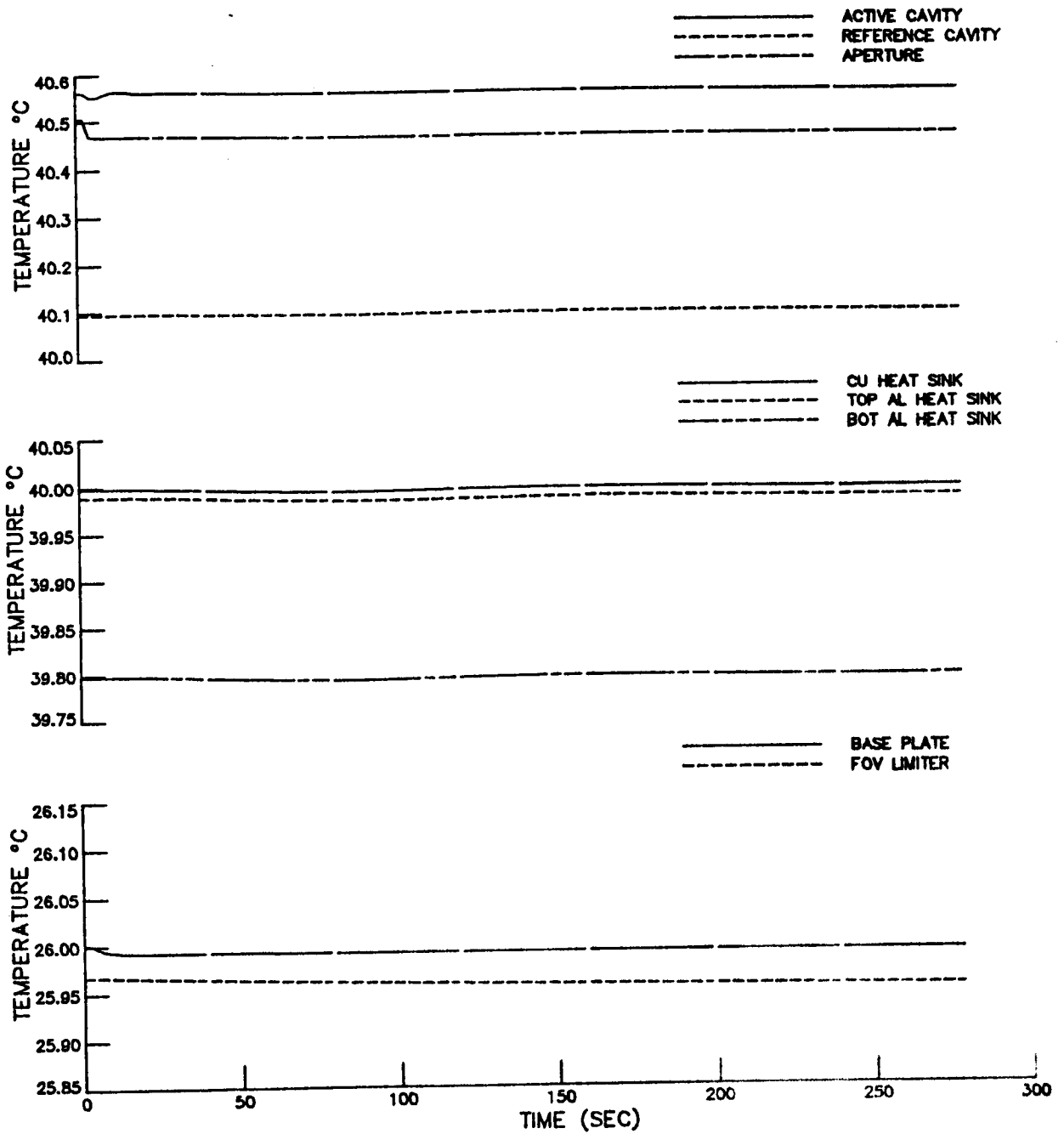
REFERENCE CAVITY EMISSIVITY CHANGED BY .02 (NOMINAL .95)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

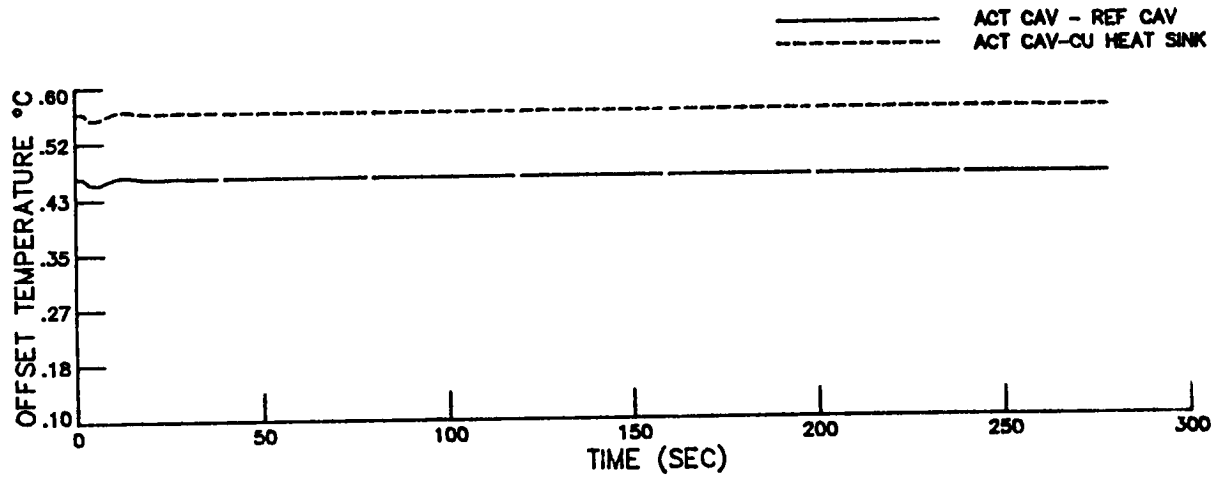
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SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

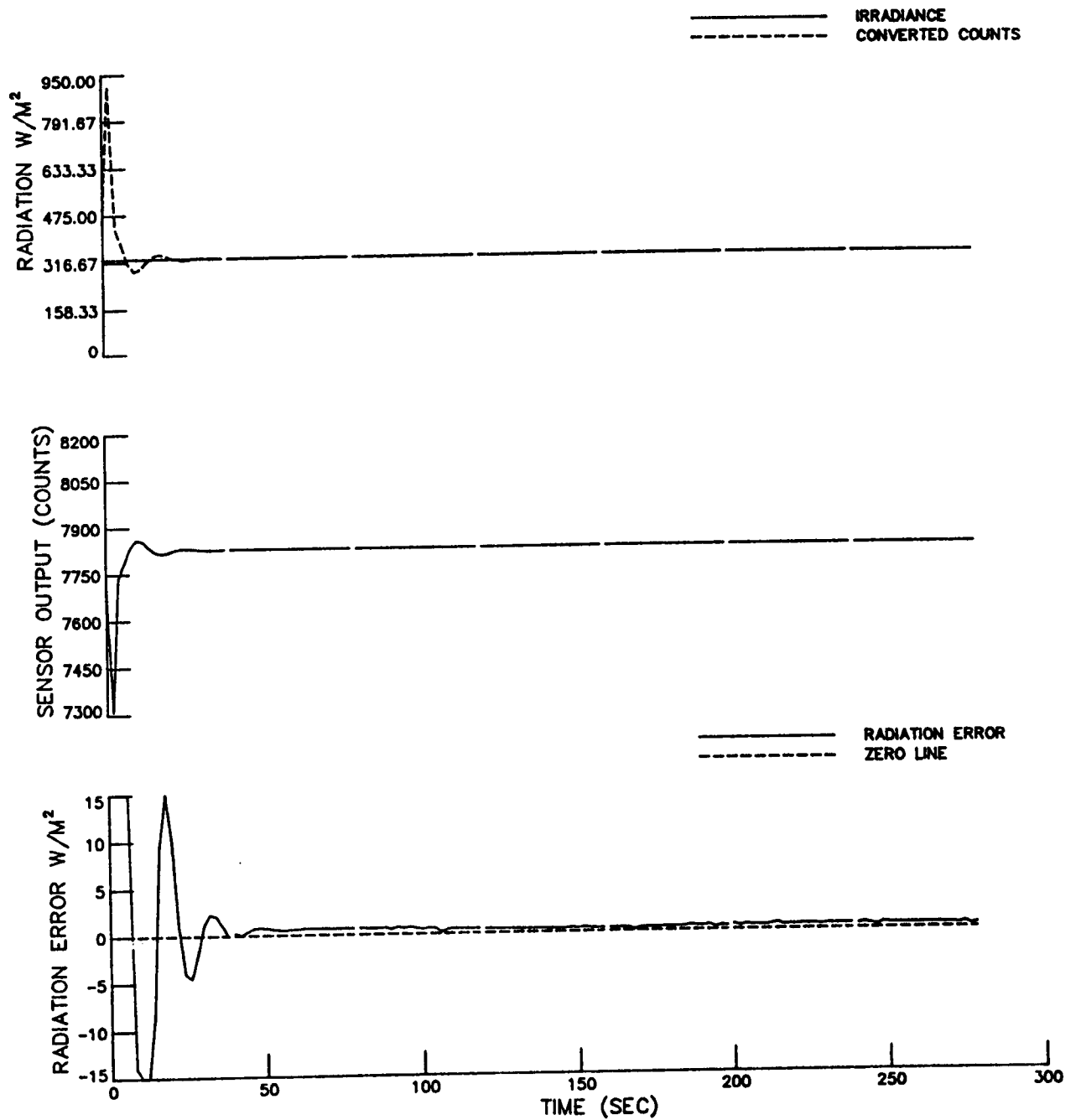
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SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

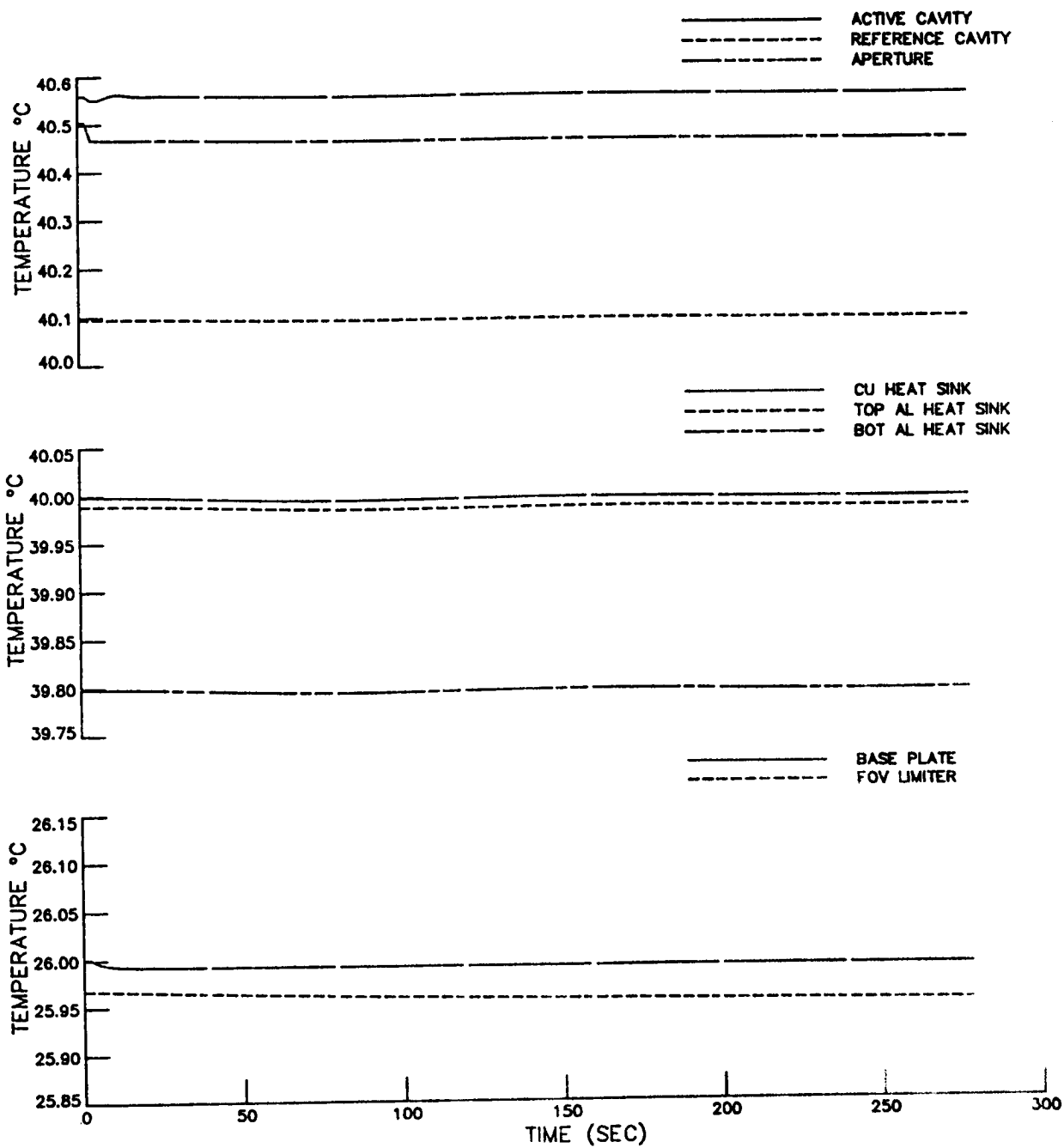
DOME FILTER LONG WAVE TRANSMITTANCE (LEAK) CHANGED BY .005 (NOMINAL .01)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

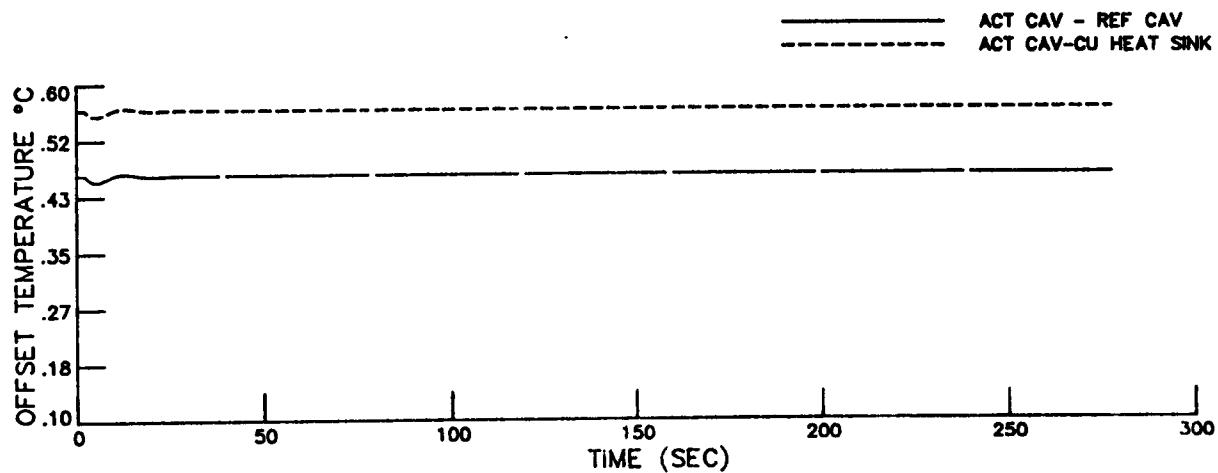
DOME FILTER LONG WAVE TRANSMITTANCE (LEAK) CHANGED BY .005 (NOMINAL .01)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

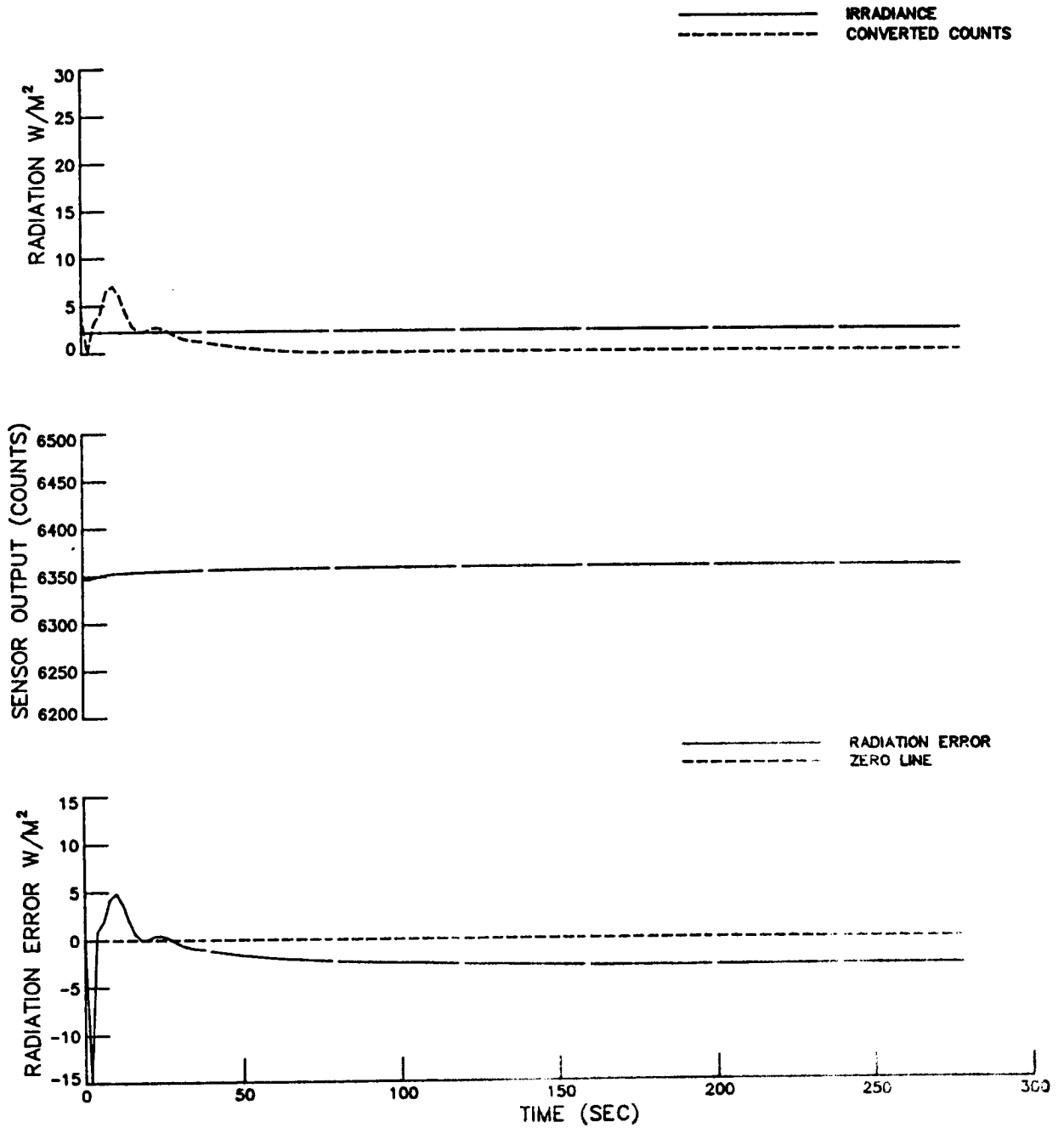
DOME FILTER LONG WAVE TRANSMITTANCE (LEAK) CHANGED BY .005 (NOMINAL .01)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

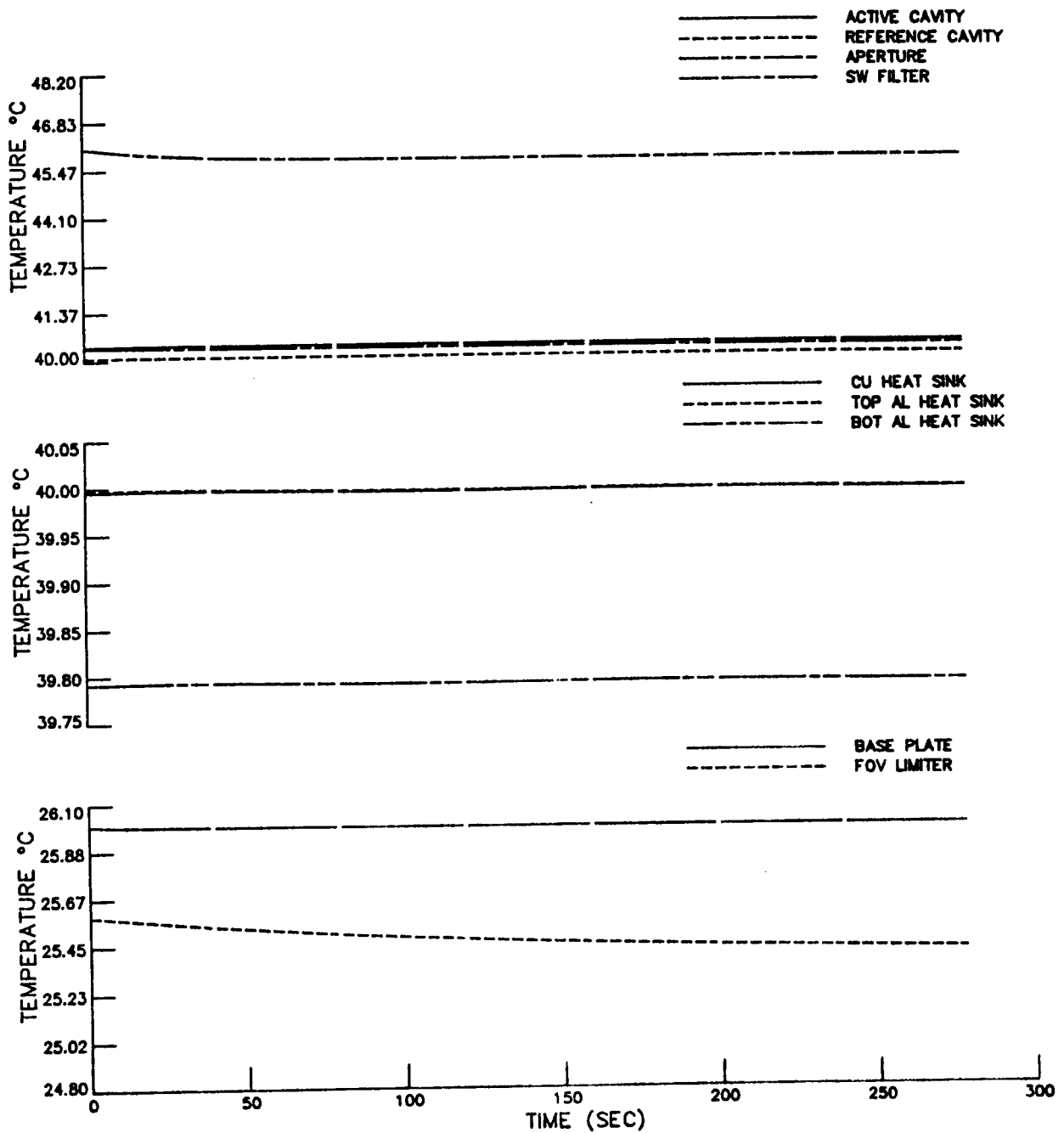
DOME FILTER LONG WAVE TRANSMITTANCE (LEAK) CHANGED BY .005 (NOMINAL .01)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

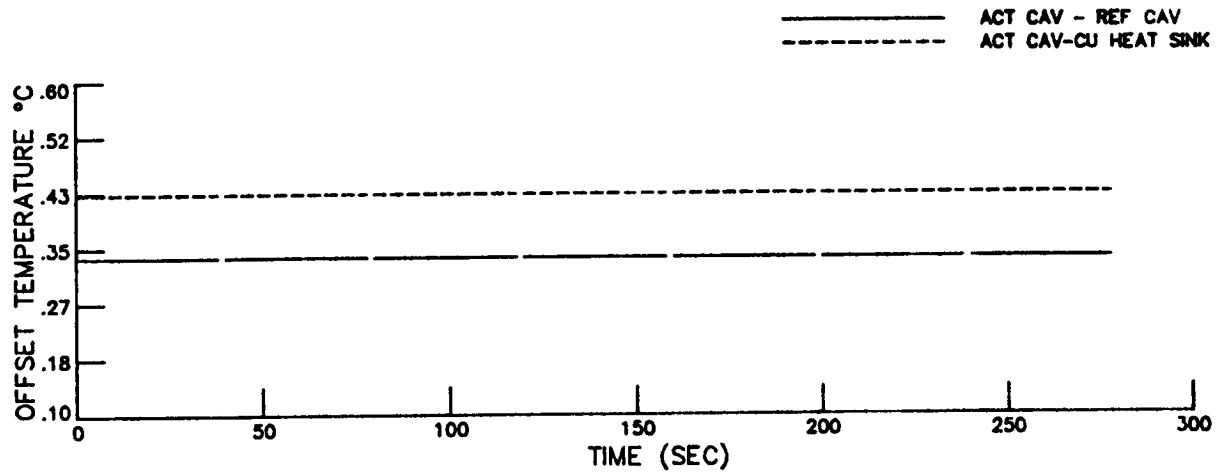
DOME FILTER LONG WAVE TRANSMITTANCE (LEAK) CHANGED BY .005 (NOMINAL .01)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

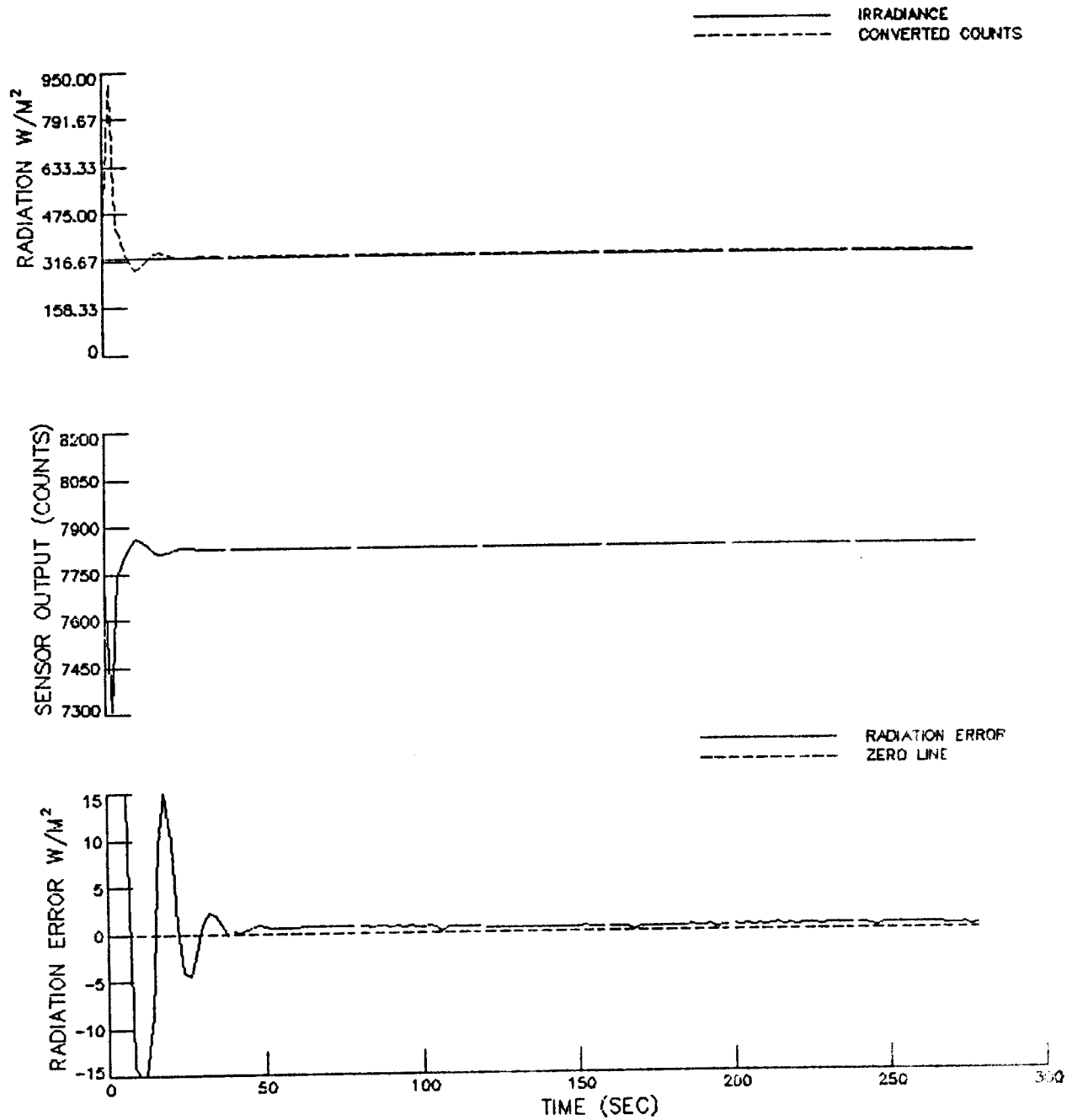
DOME FILTER LONG WAVE TRANSMITTANCE (LEAK) CHANGED BY .005 (NOMINAL .01)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

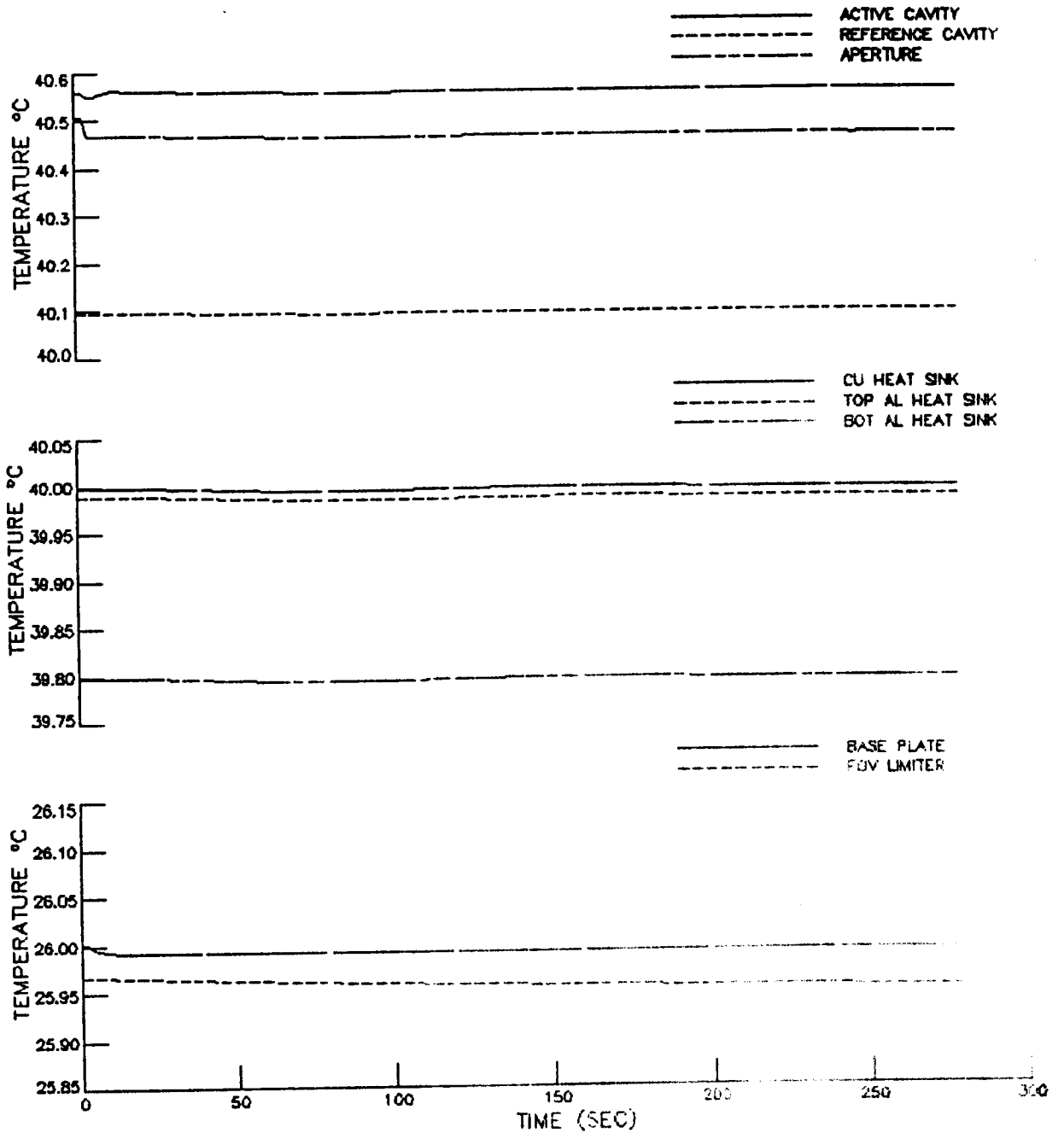
DOME FILTER SHORT WAVE TRANSMITTANCE CHANGED BY .005 (NOMINAL .946)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

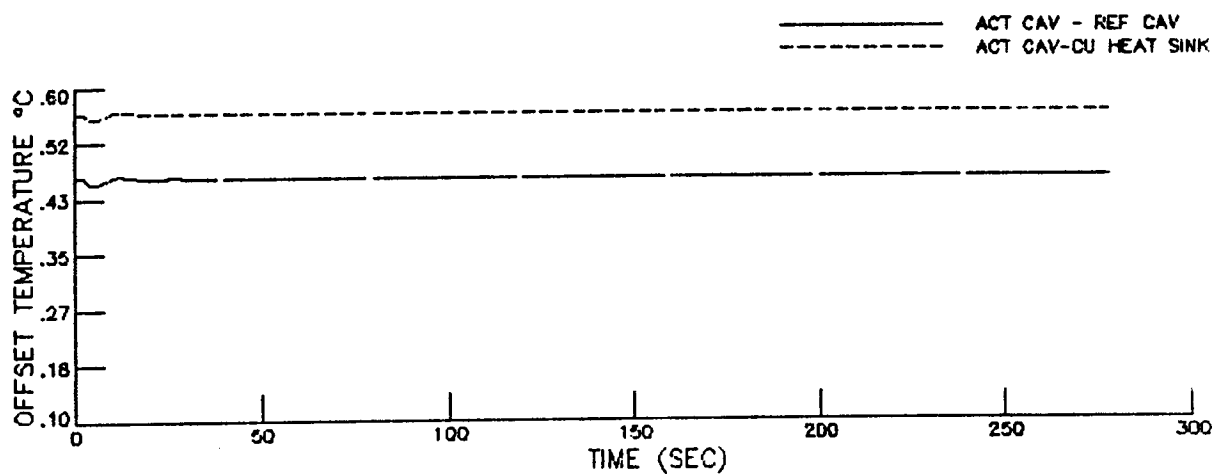
DOME FILTER SHORT WAVE TRANSMITTANCE CHANGED BY .005 (NOMINAL .946)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

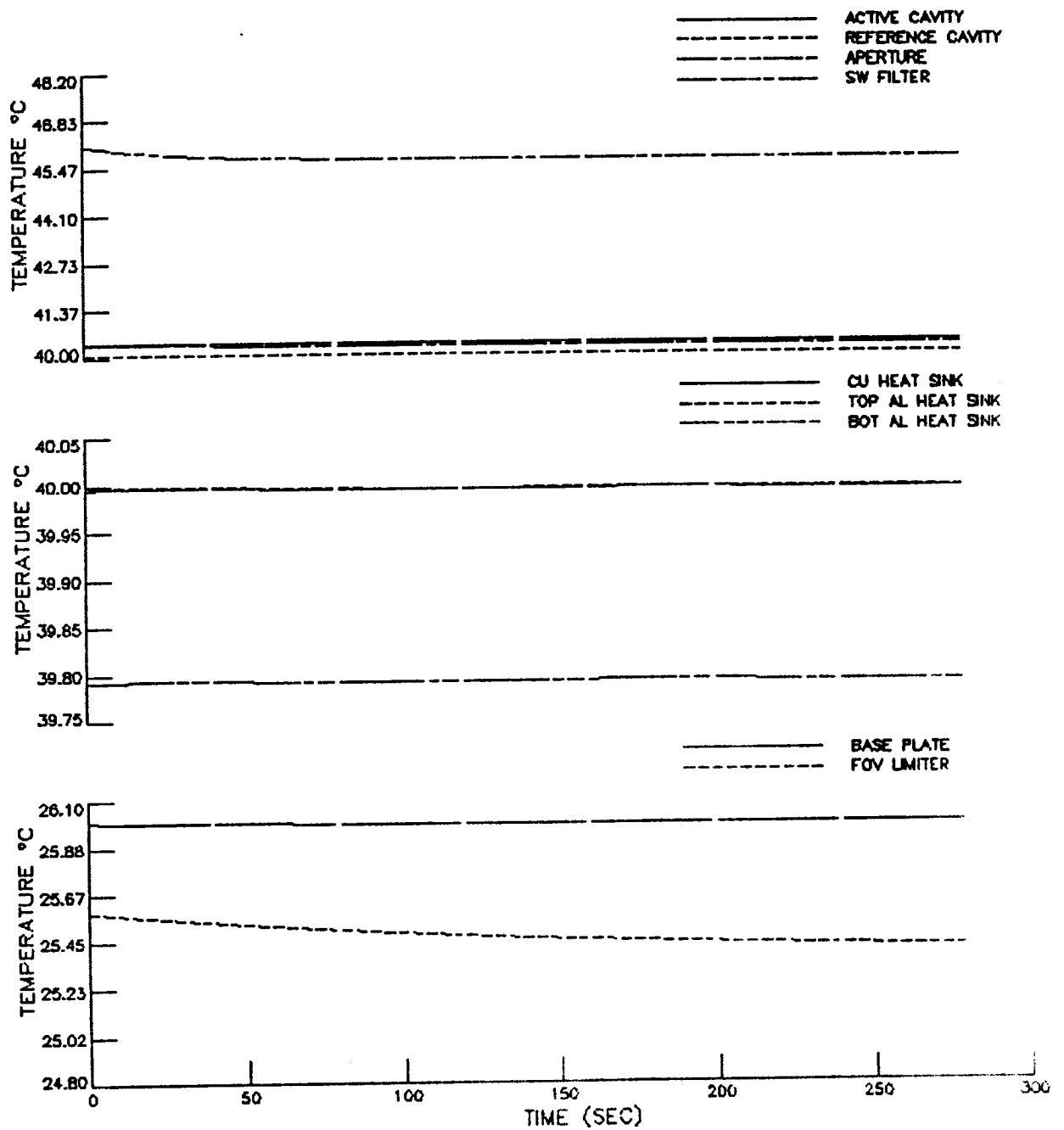
DOME FILTER SHORT WAVE TRANSMITTANCE CHANGED BY .005 (NOMINAL .946)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

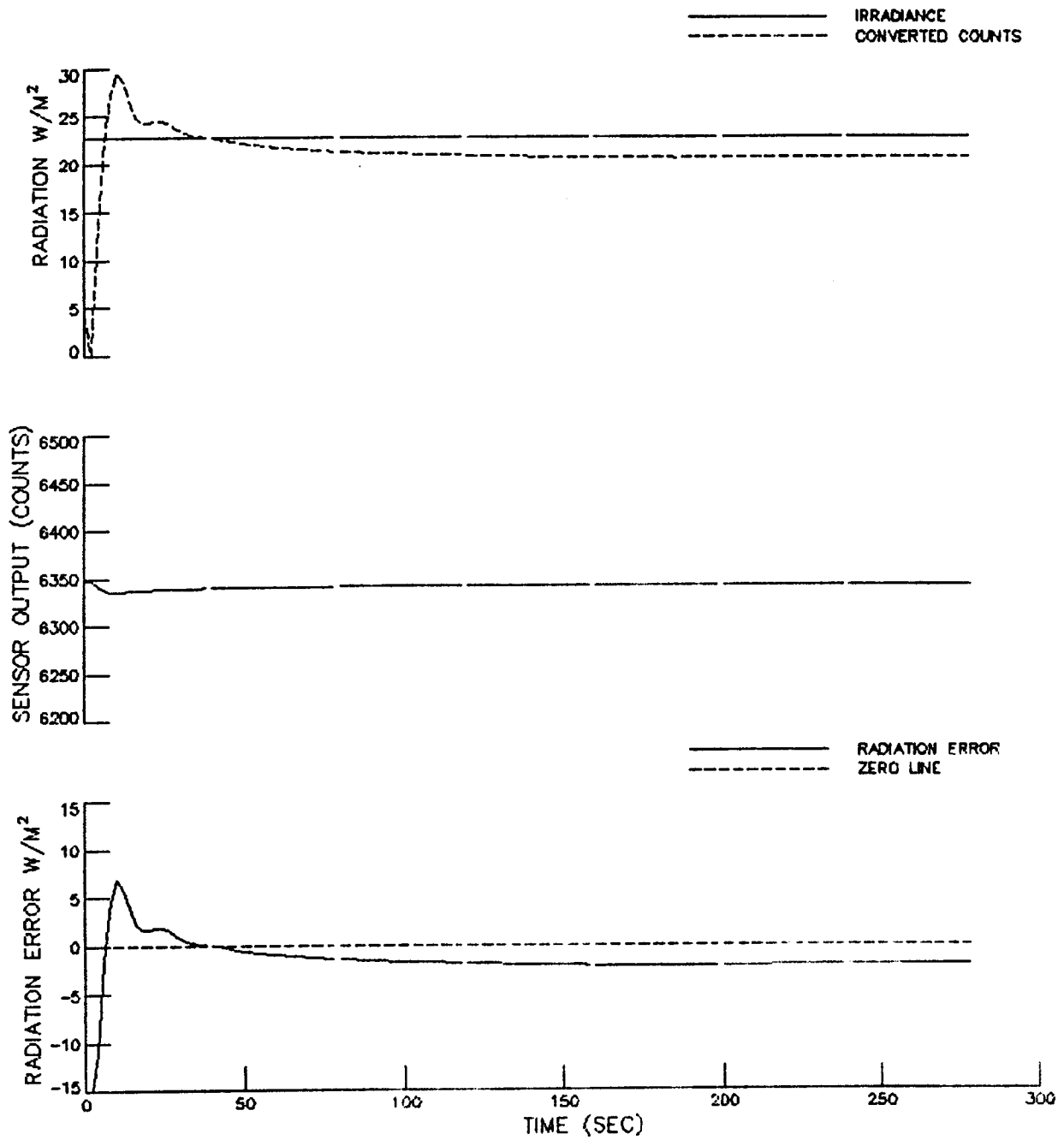
DOME FILTER SHORT WAVE TRANSMITTANCE CHANGED BY .005 (NOMINAL .946)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

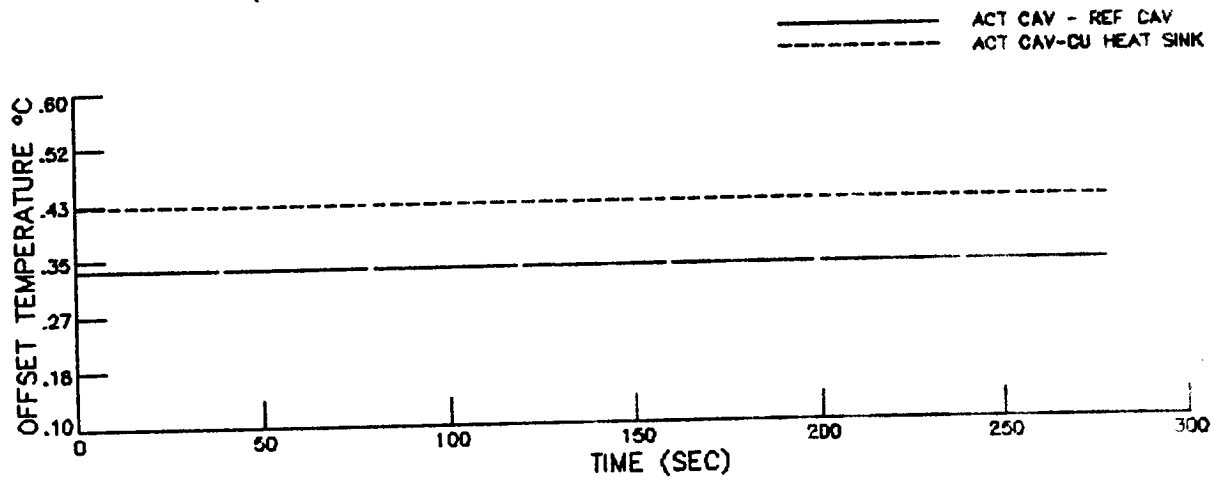
DOME FILTER SHORT WAVE TRANSMITTANCE CHANGED BY .005 (NOMINAL .946)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

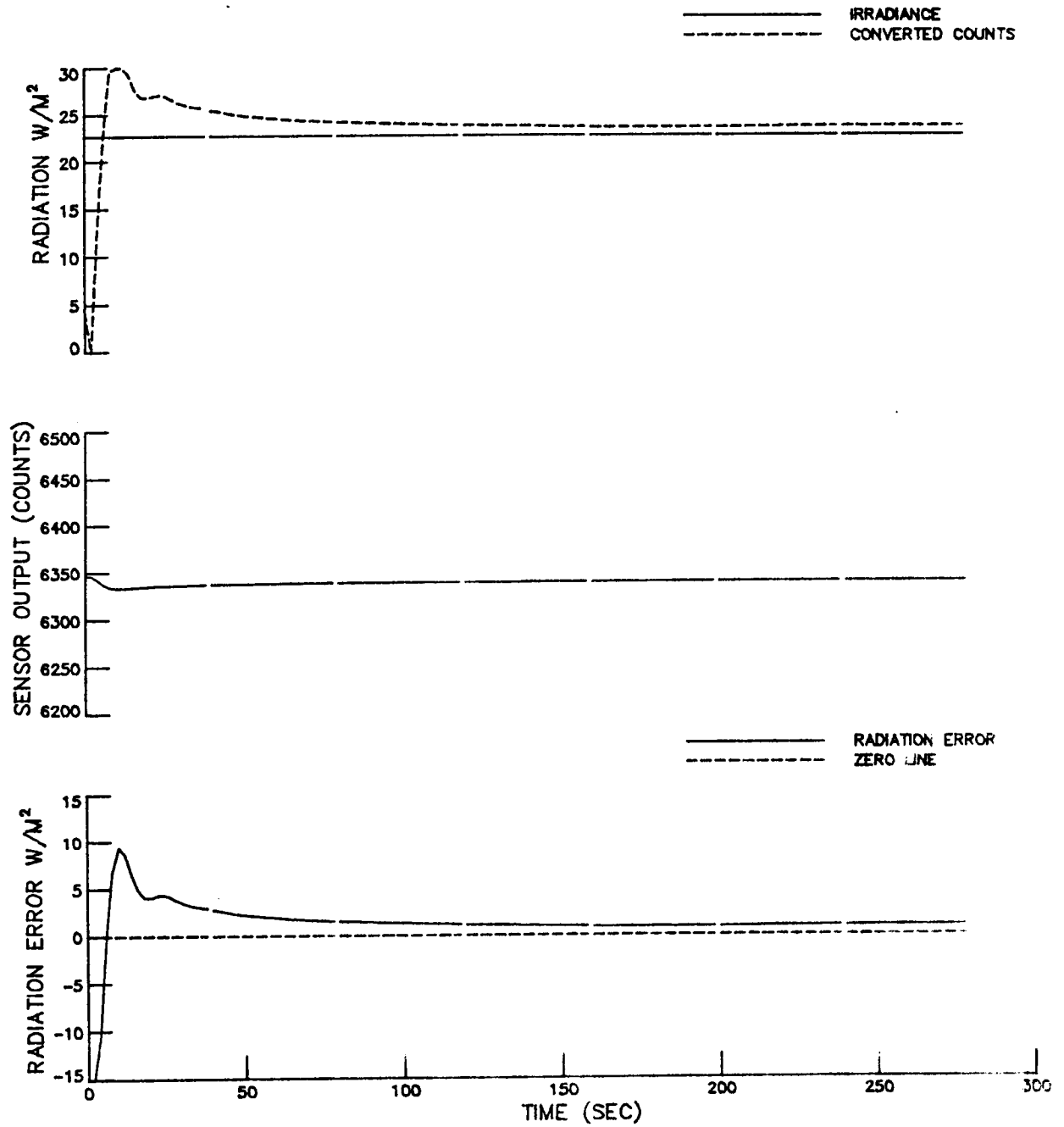
DOME FILTER SHORT WAVE TRANSMITTANCE CHANGED BY .005 (NOMINAL .946)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

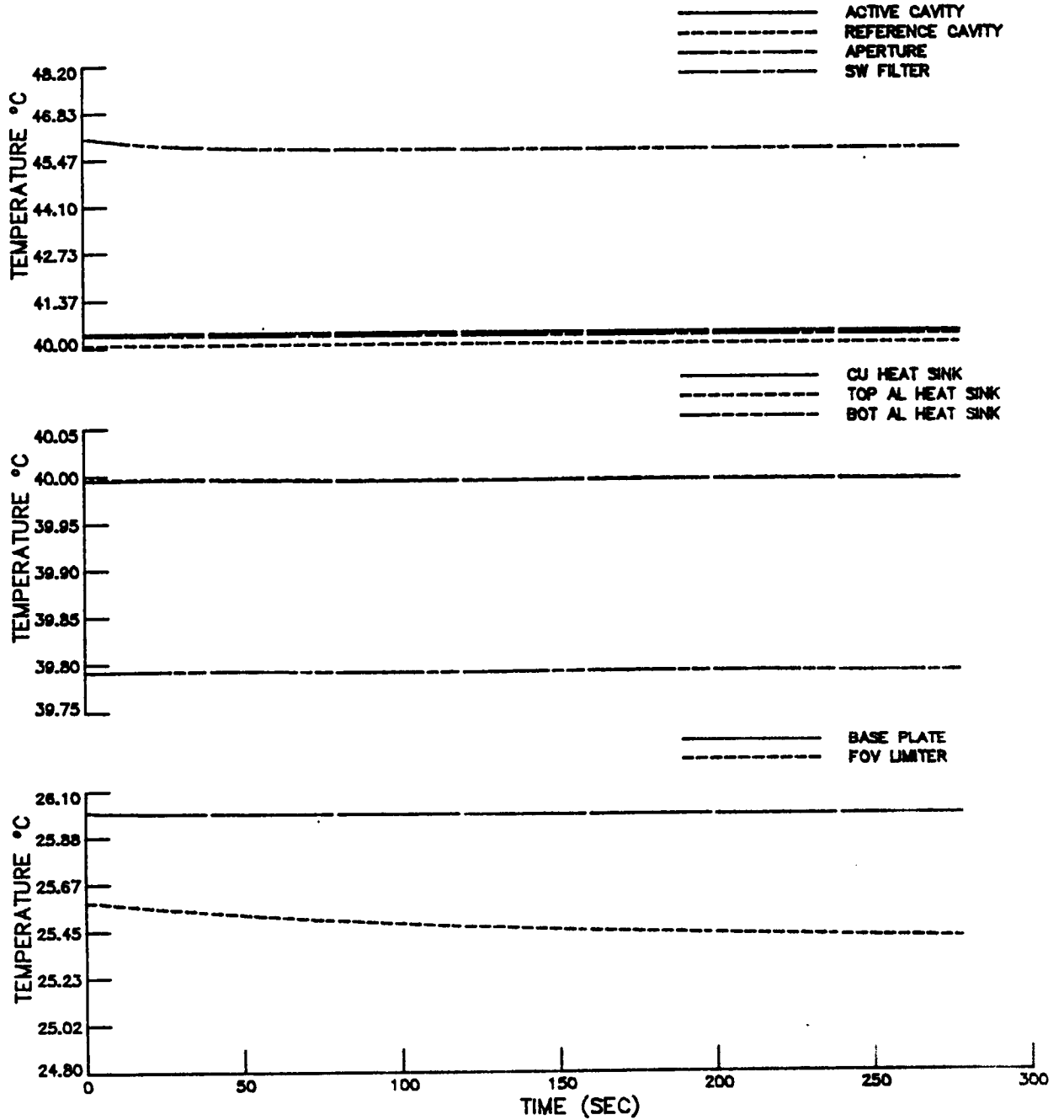
DOME FILTER LONG WAVE EMISSIVITY CHANGED BY .005 (NOMINAL .69)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

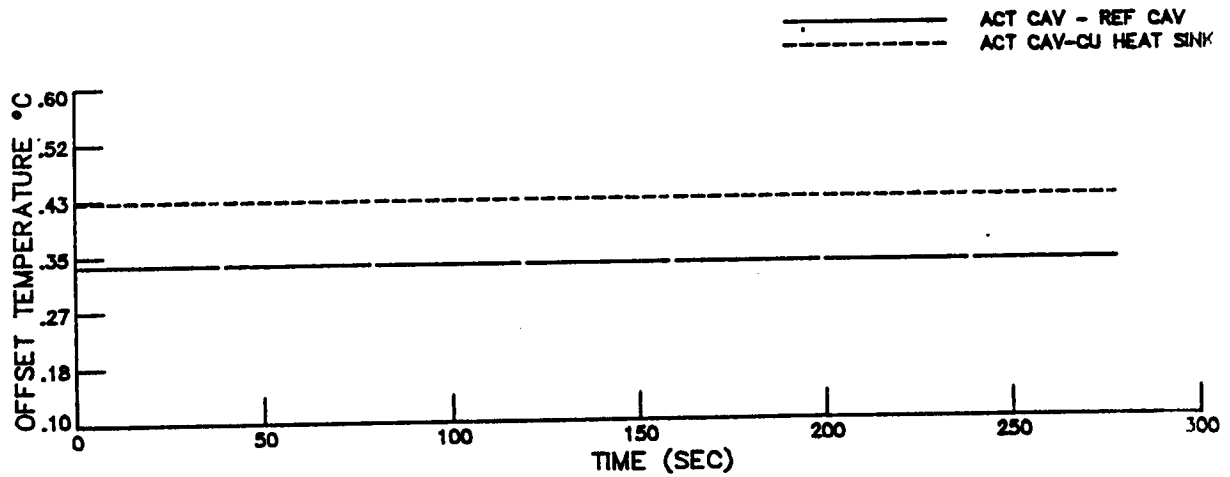
DOME FILTER LONG WAVE EMISSIVITY CHANGED BY .005 (NOMINAL .69)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV FILTERED SIMULATION

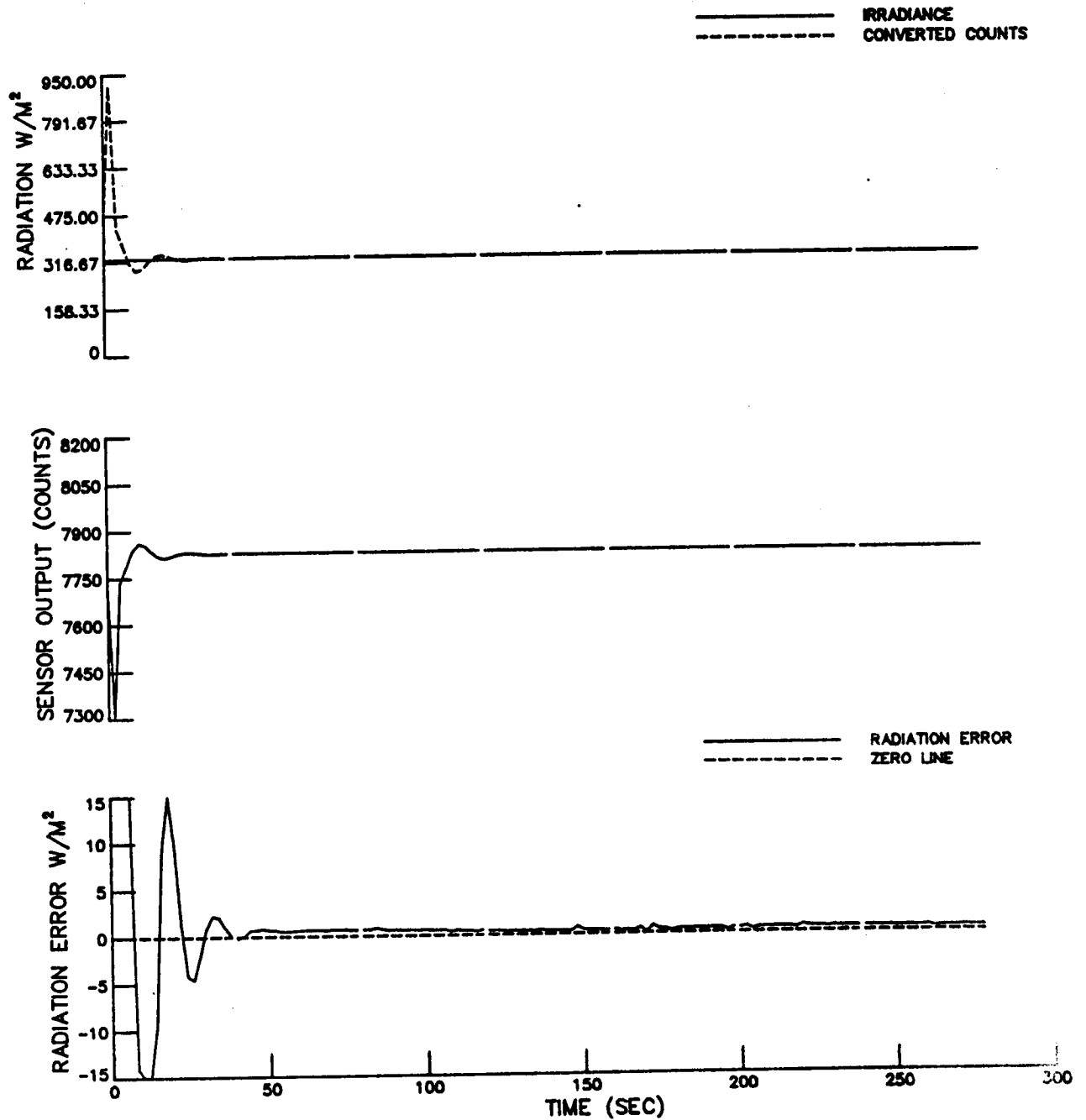
DOME FILTER LONG WAVE EMISSIVITY CHANGED BY .005 (NOMINAL .69)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

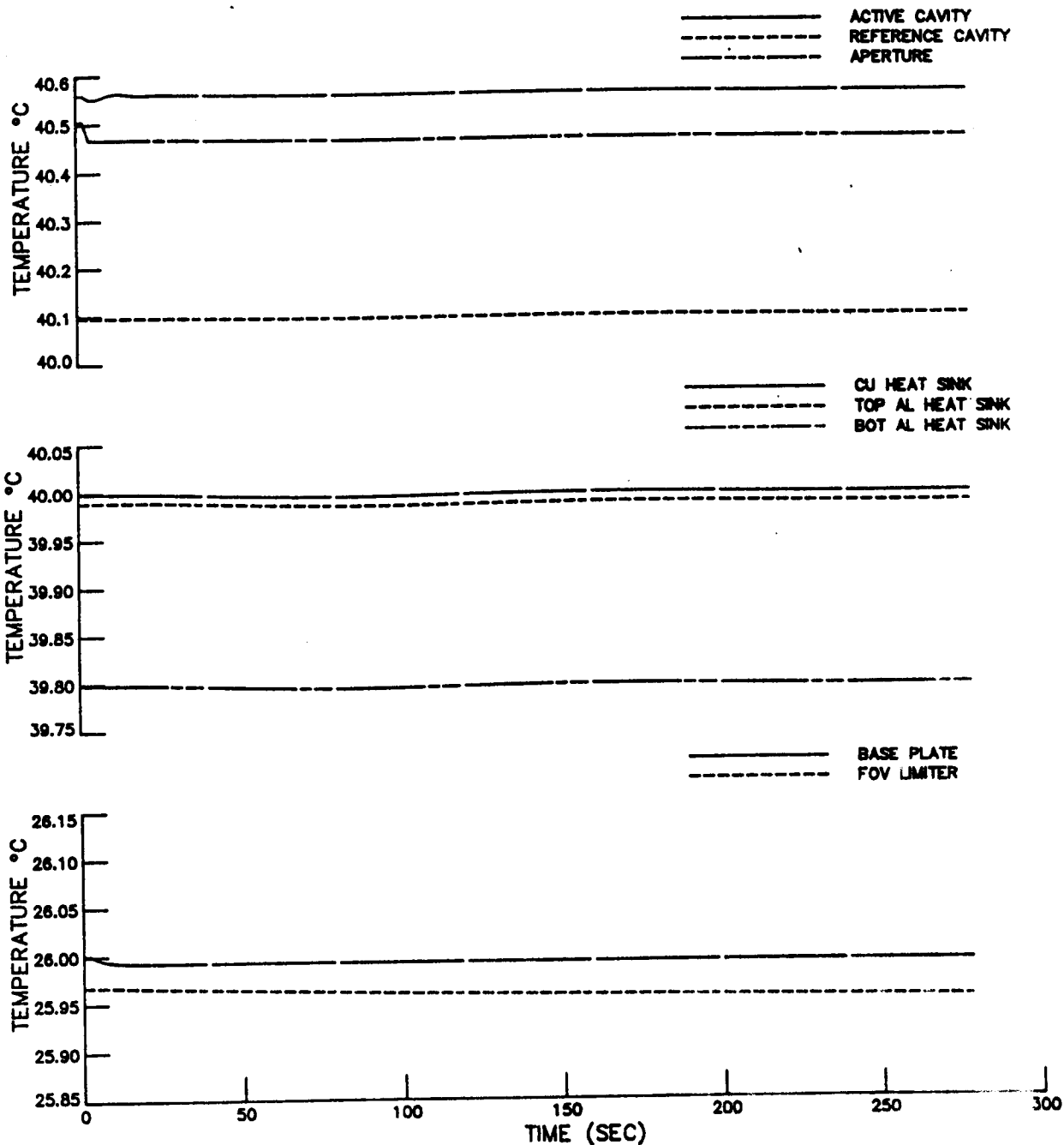
SENSOR BRIDGE VOLTAGE CHANGED BY .01 v (NOMINAL 10 v)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

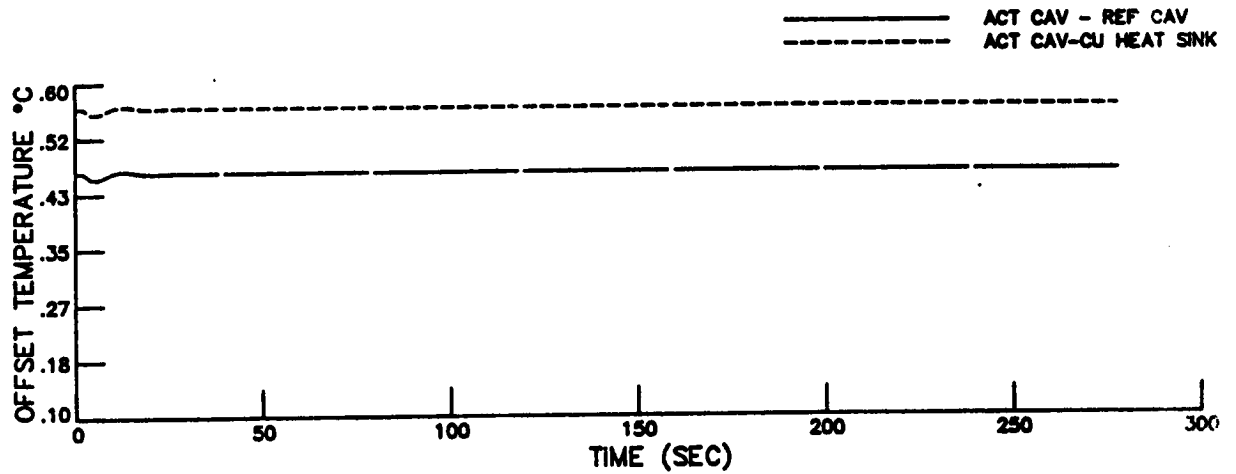
SENSOR BRIDGE VOLTAGE CHANGED BY .01 v (NOMINAL 10 v)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

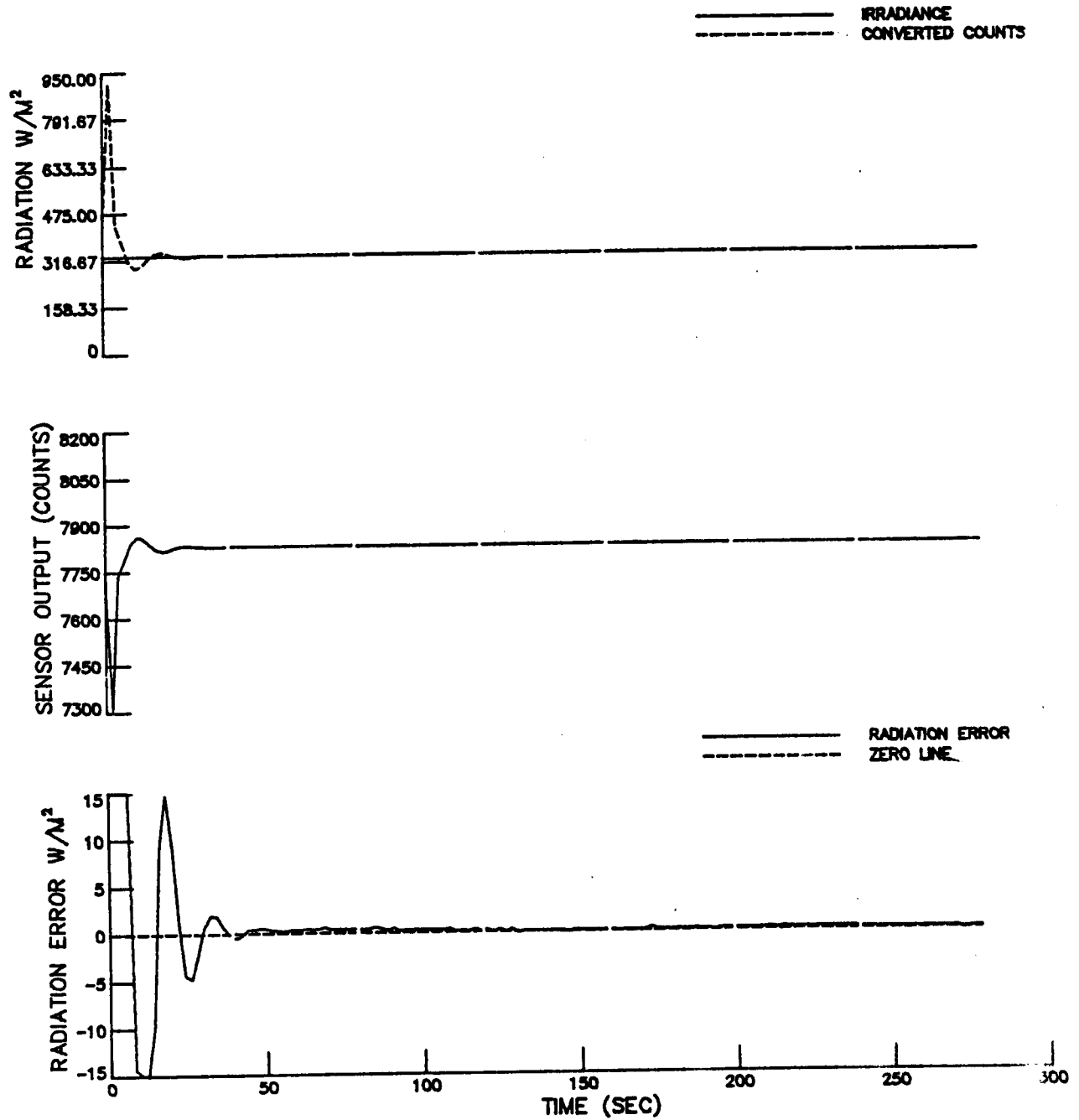
SENSOR BRIDGE VOLTAGE CHANGED BY .01 v (NOMINAL 10 v)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

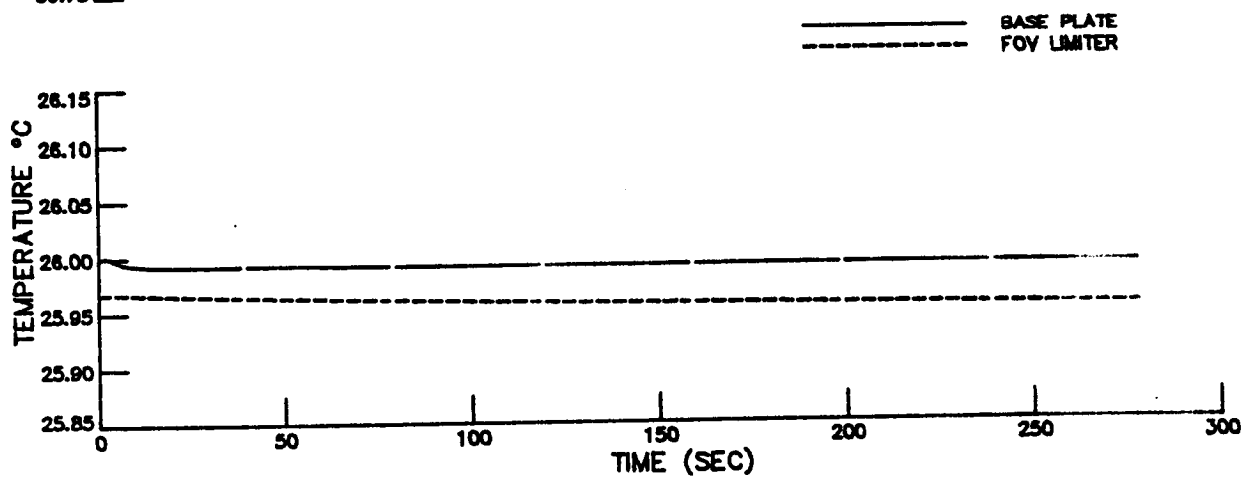
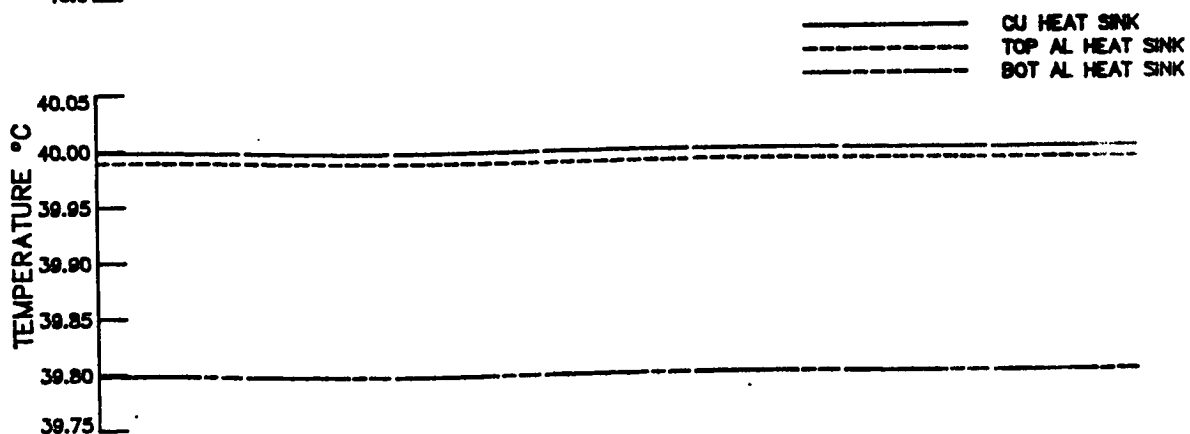
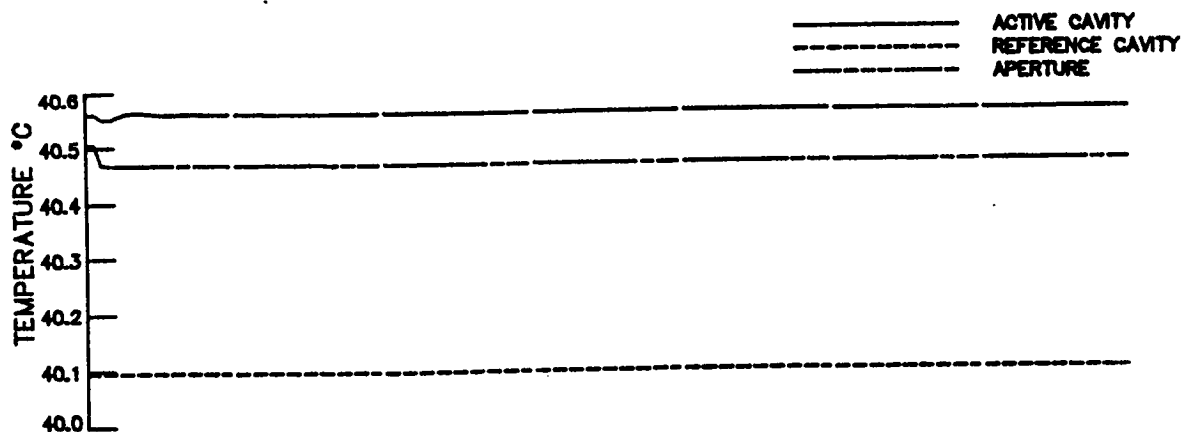
ACTIVE CAVITY HEATER RESISTANCE CHANGED BY $.1 \Omega$ (NOMINAL 1492.3Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

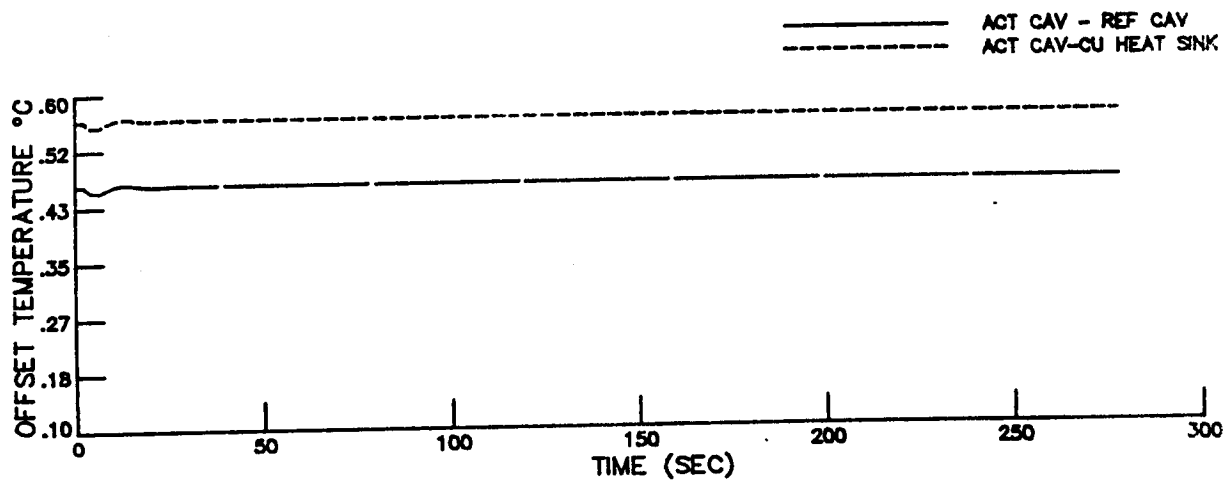
ACTIVE CAVITY HEATER RESISTANCE CHANGED BY $.1 \Omega$ (NOMINAL 1492.3Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

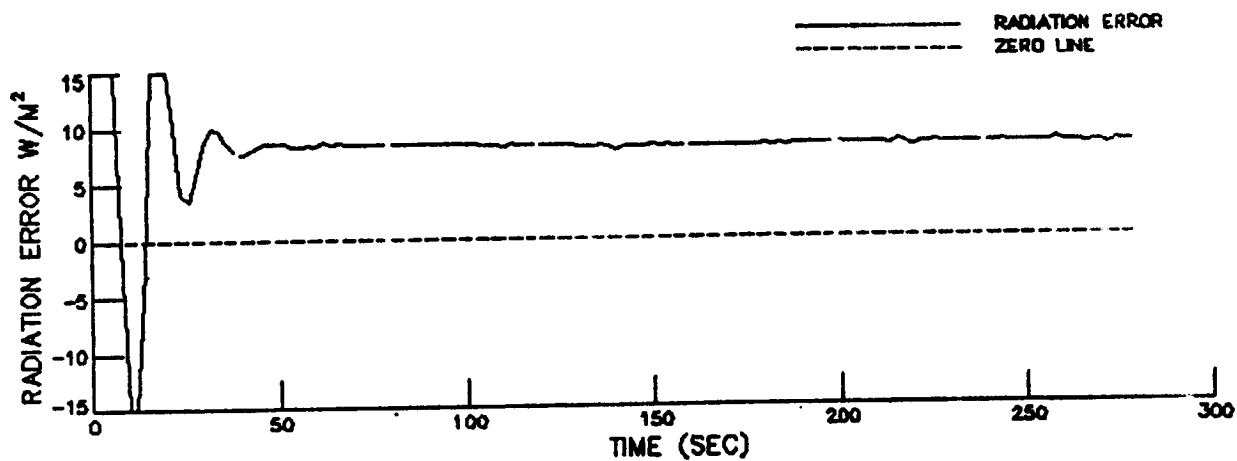
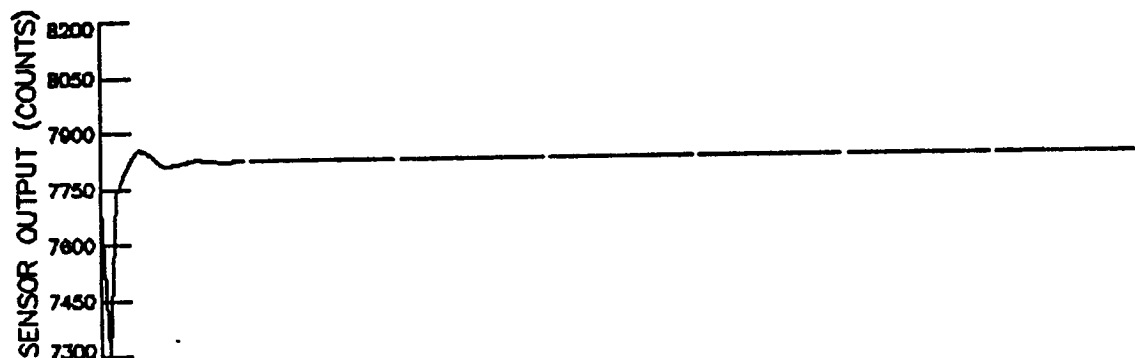
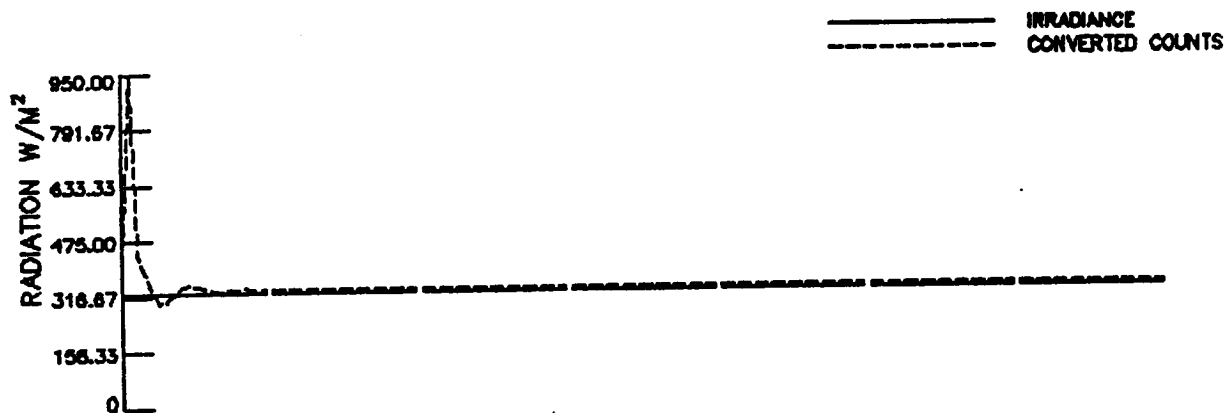
ACTIVE CAVITY HEATER RESISTANCE CHANGED BY $.1 \Omega$ (NOMINAL 1492.3Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

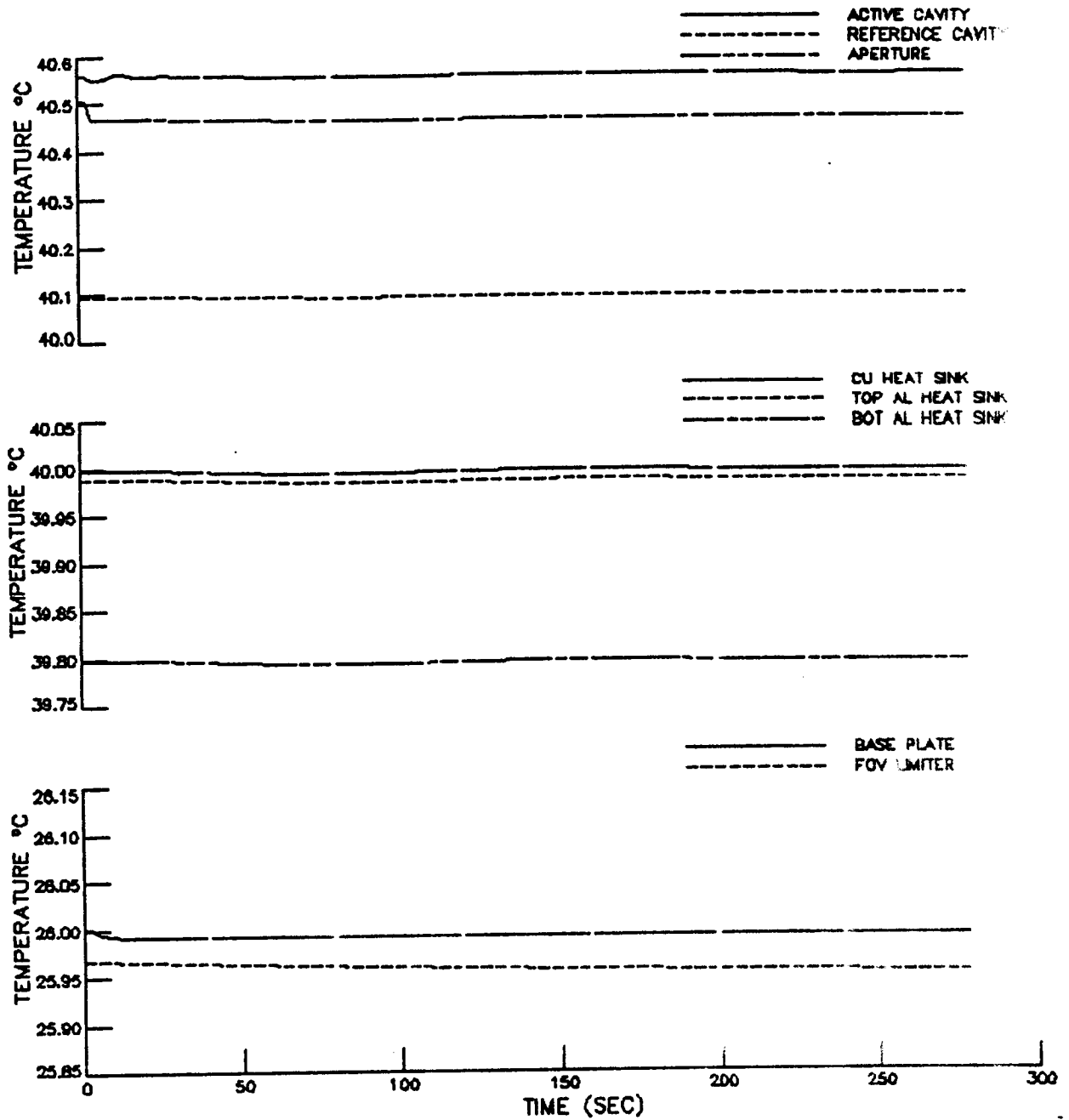
ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2319.1Ω @ 40°C)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

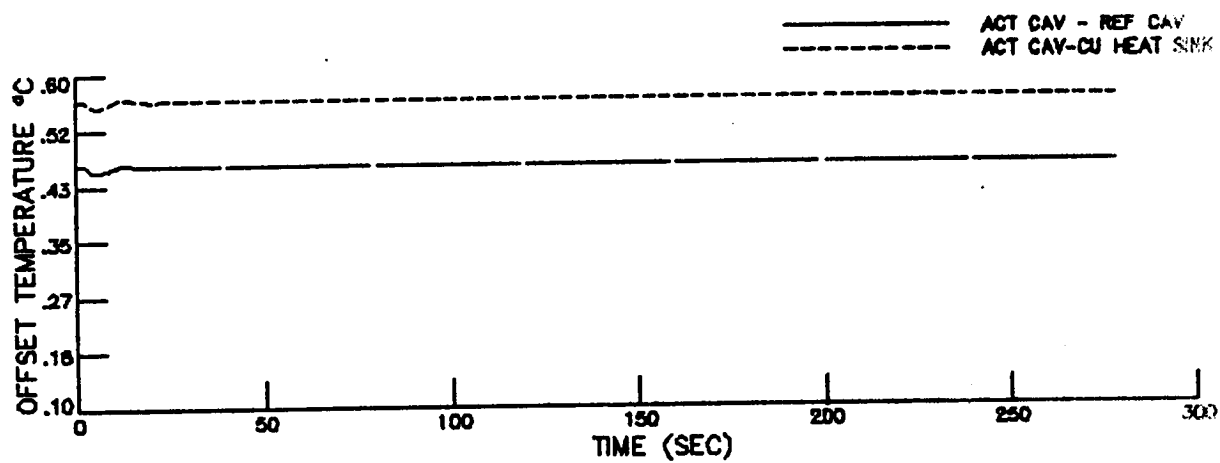
ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2319.1Ω @ 40°C)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MPOV TOTAL SIMULATION

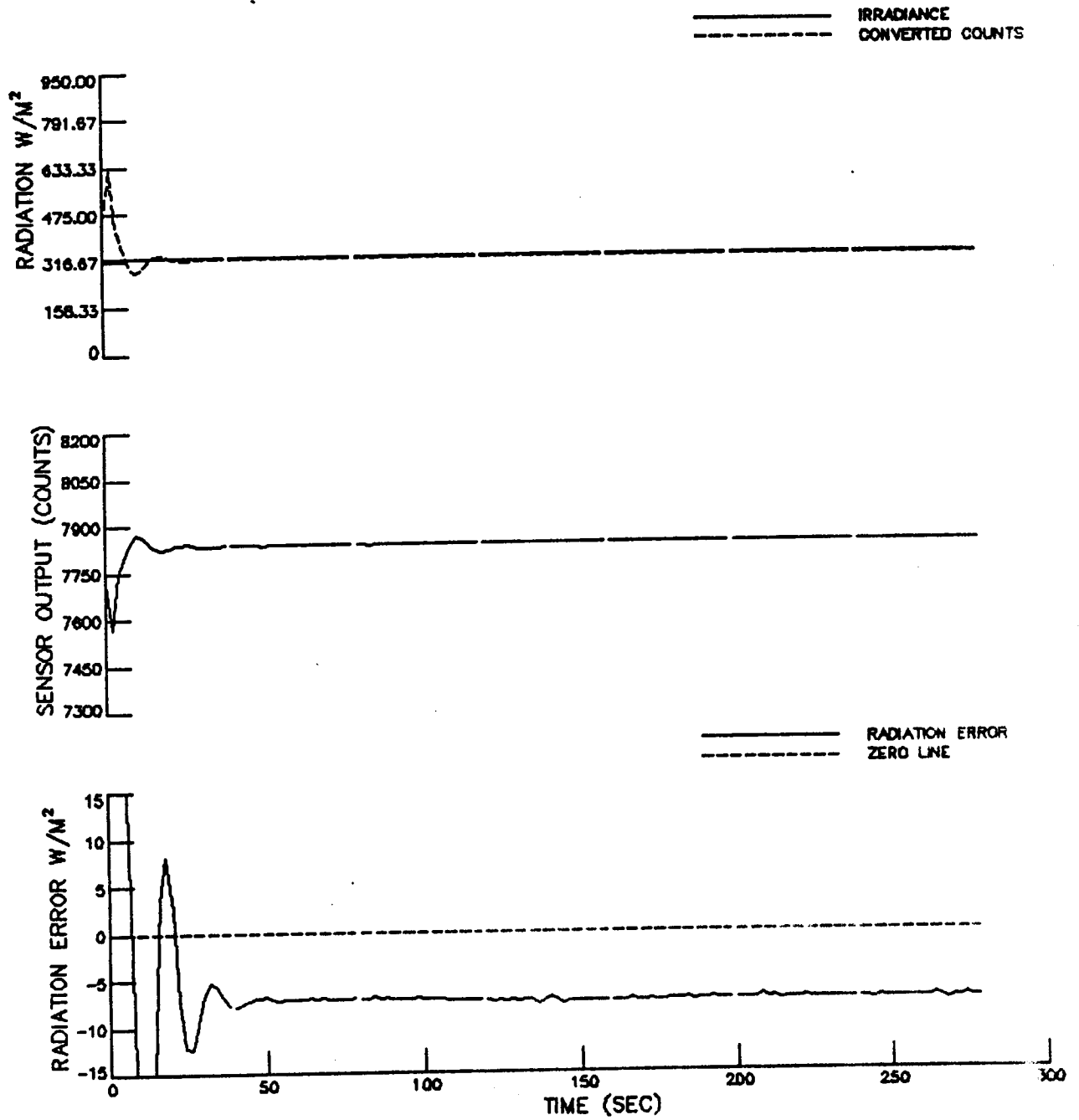
ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2319.1Ω @ 40°C)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

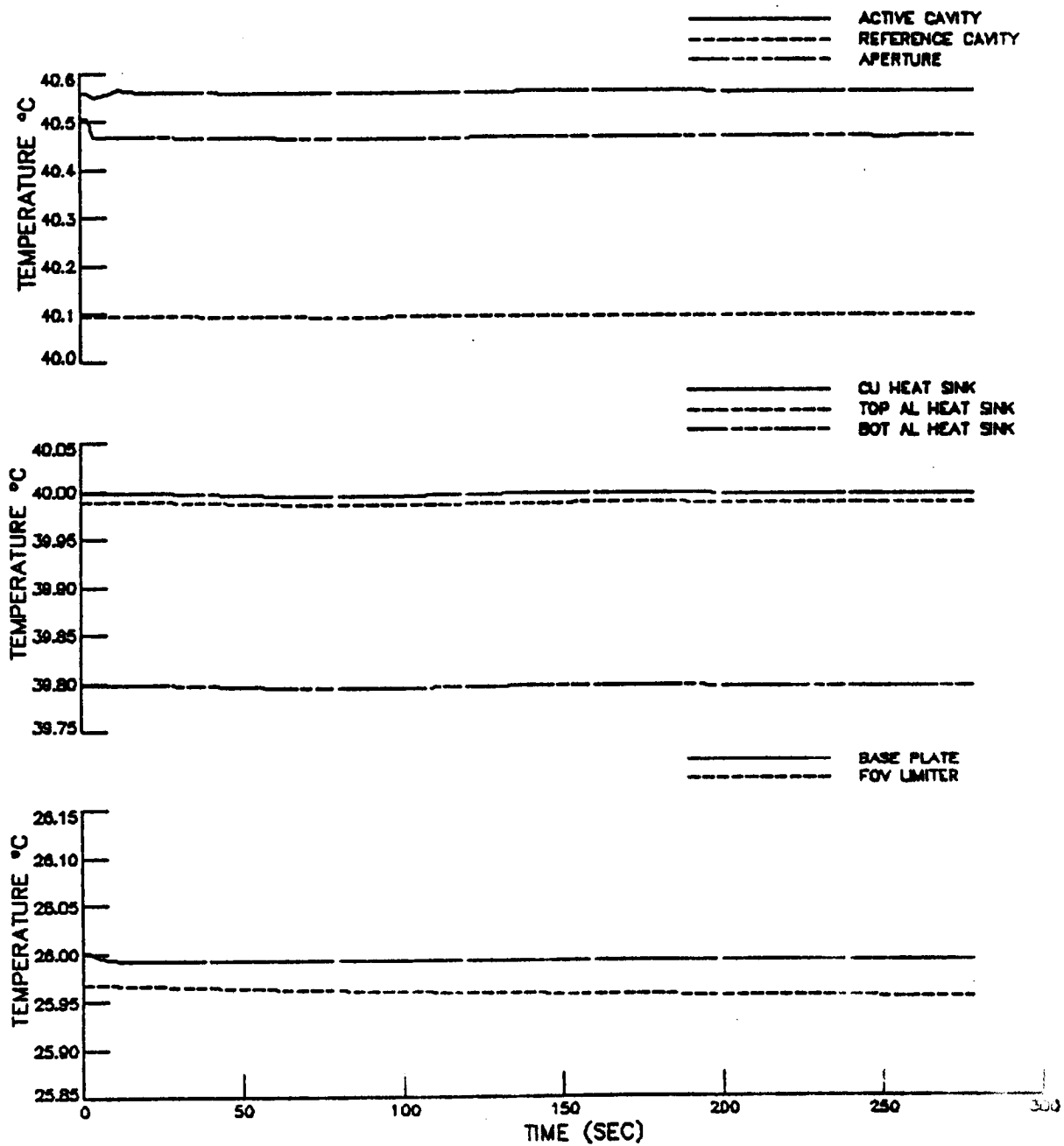
REFERENCE CAVITY SENSOR WIRE RESISTANCE CHANGED BY .01 Ω (NOMINAL 2327.1 Ω @ 40°C)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

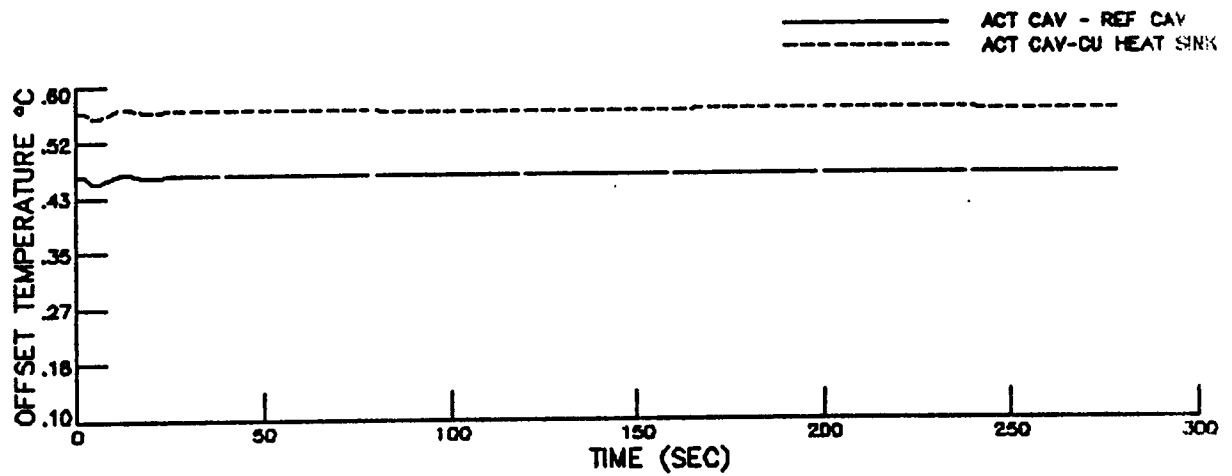
REFERENCE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2327.1Ω @ 40°C)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

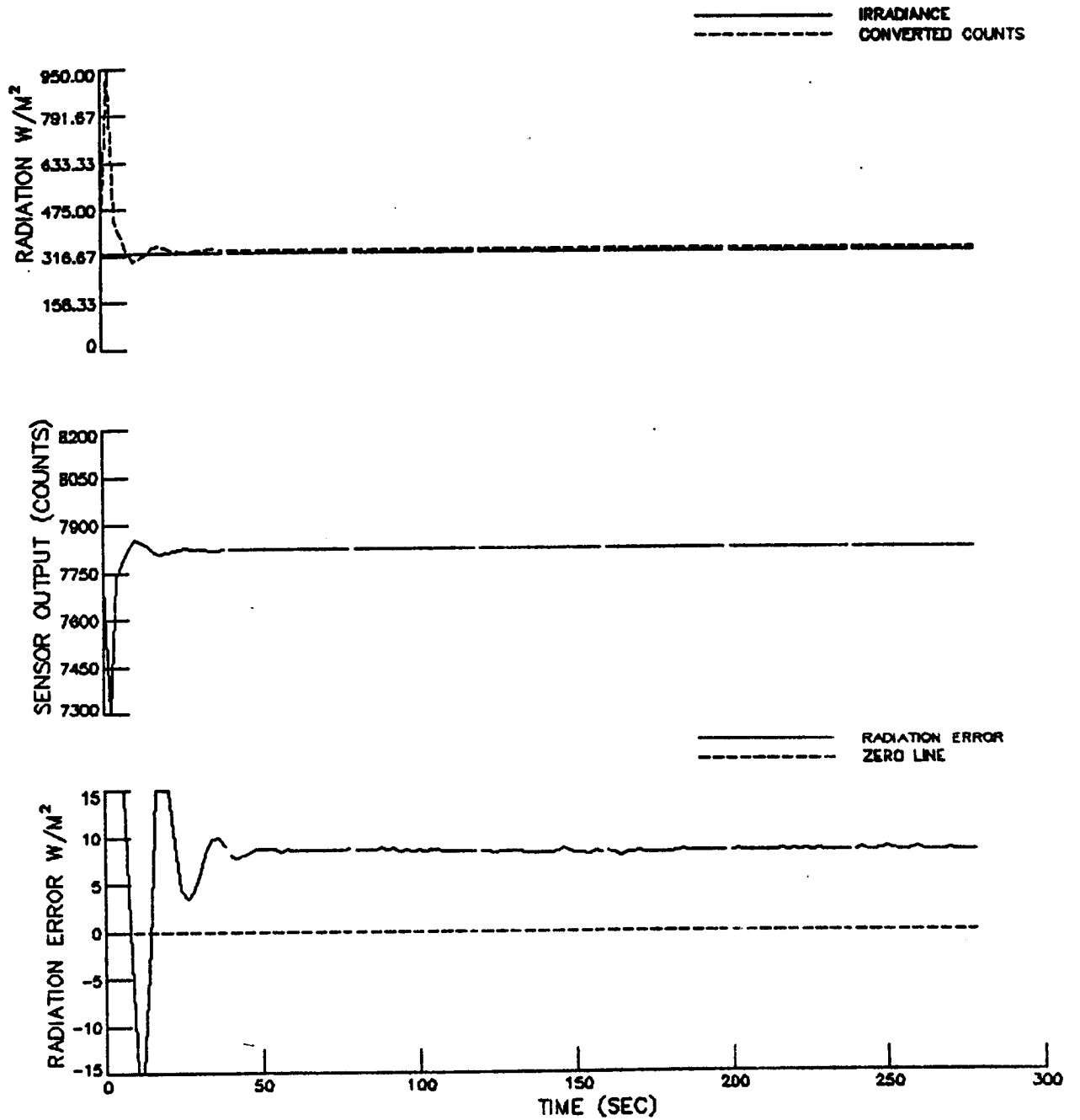
REFERENCE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2327.1Ω @ 40°C)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

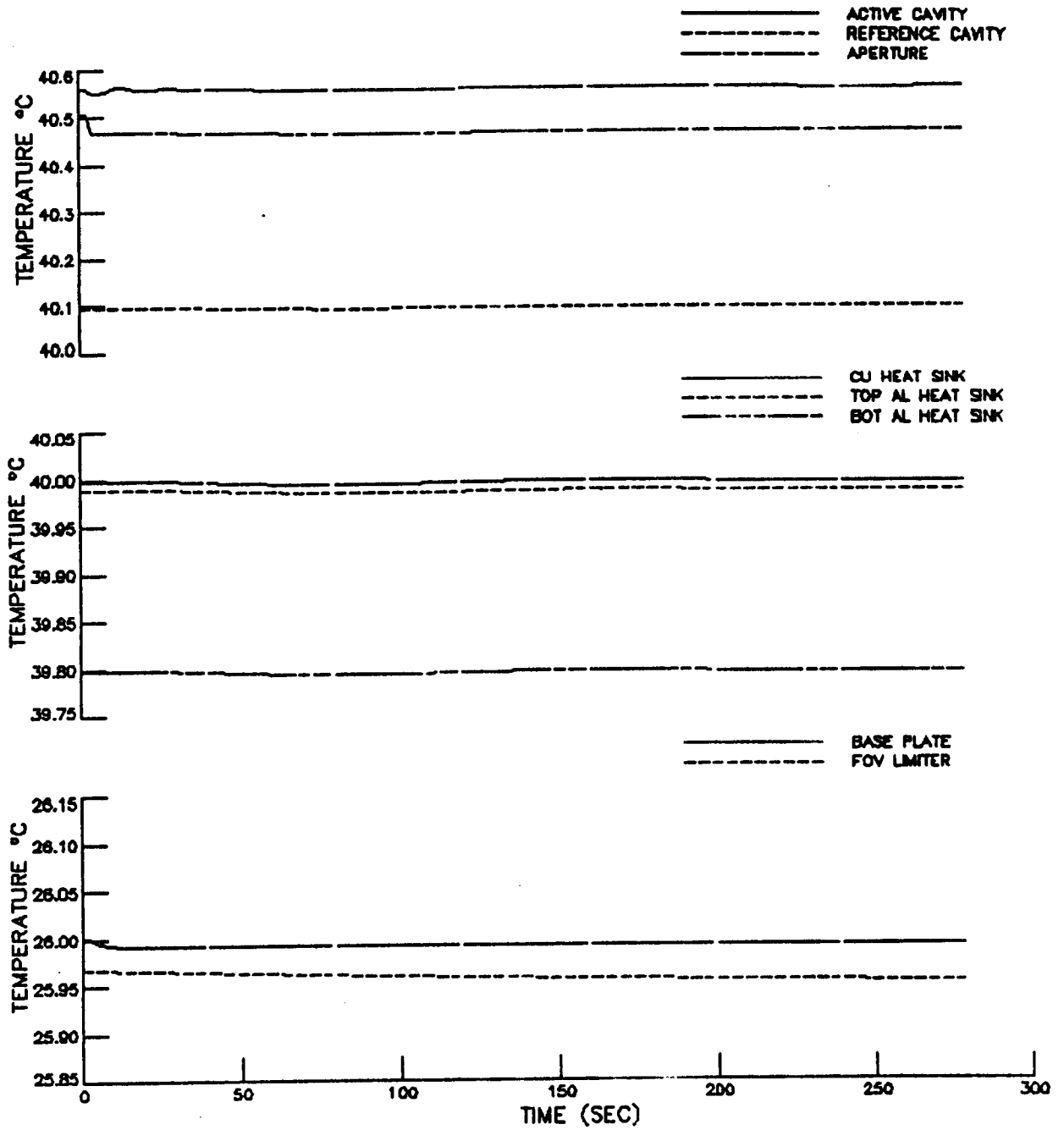
SENSOR BRIDGE RESISTANCE CHANGED BY .01 Ω (NOMINAL 2210 Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

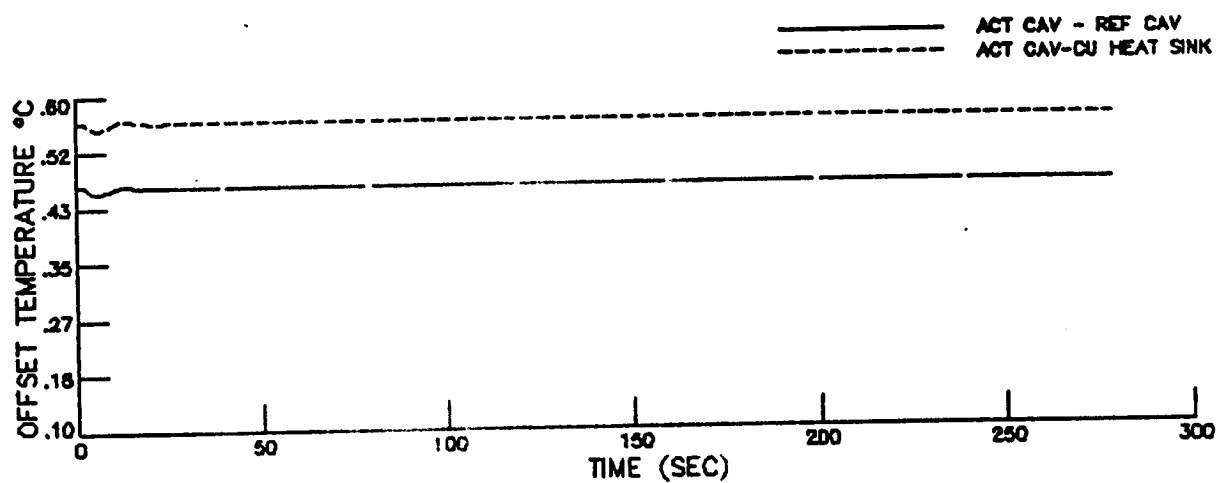
SENSOR BRIDGE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2210Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

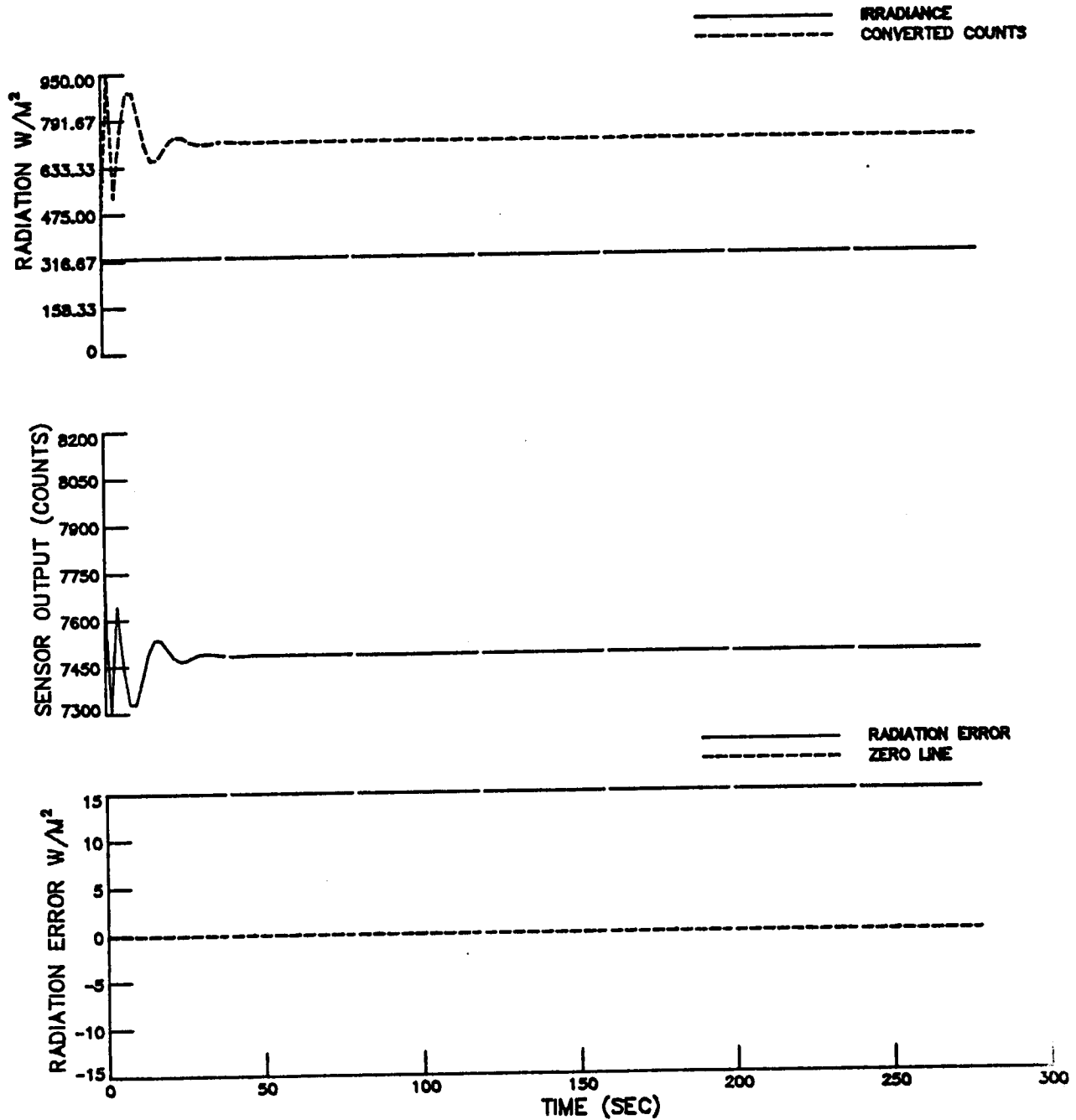
SENSOR BRIDGE RESISTANCE CHANGED BY $.01 \Omega$ (NOMINAL 2210Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

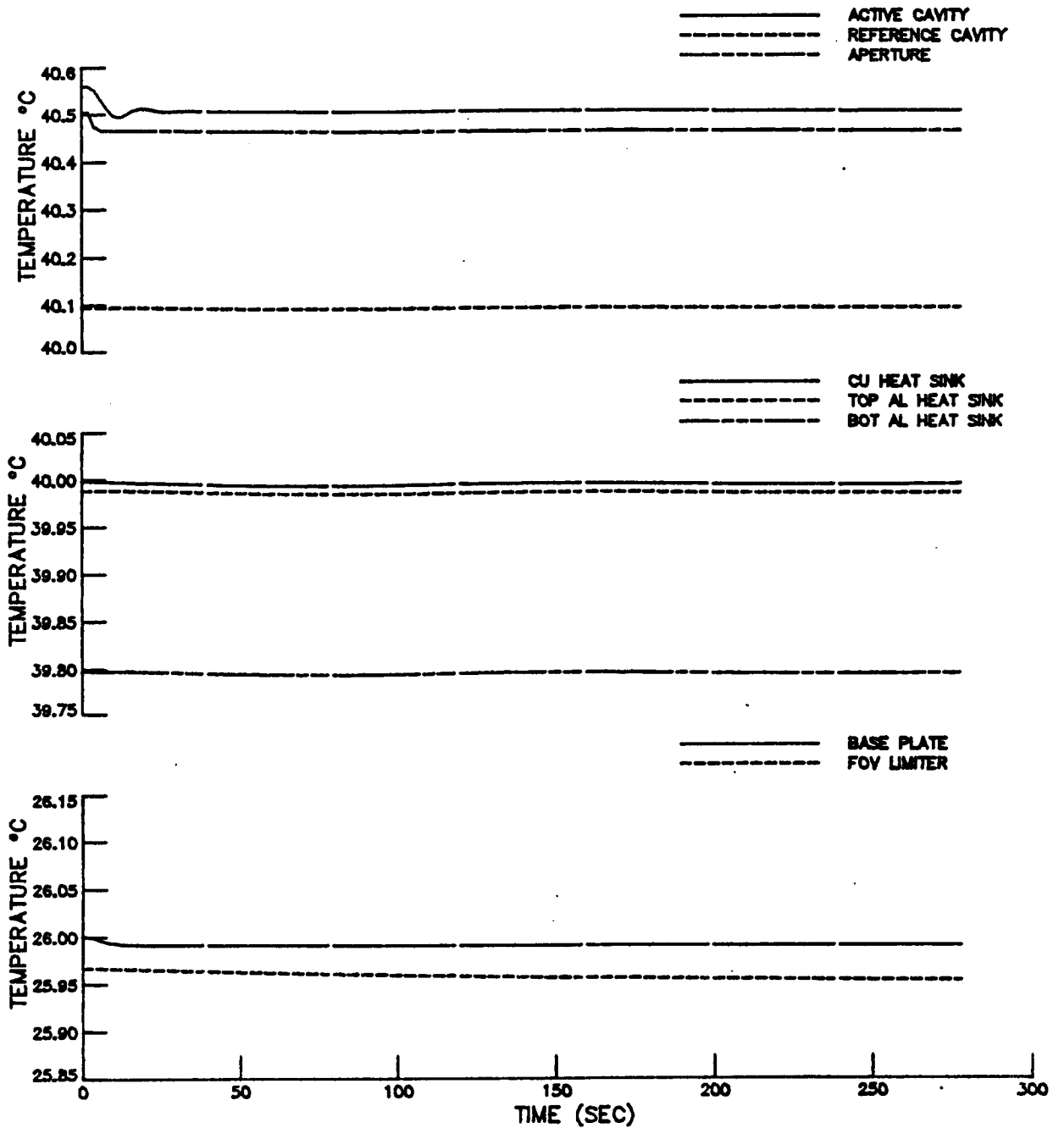
ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.5 \Omega$ (NOMINAL 2319.1Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

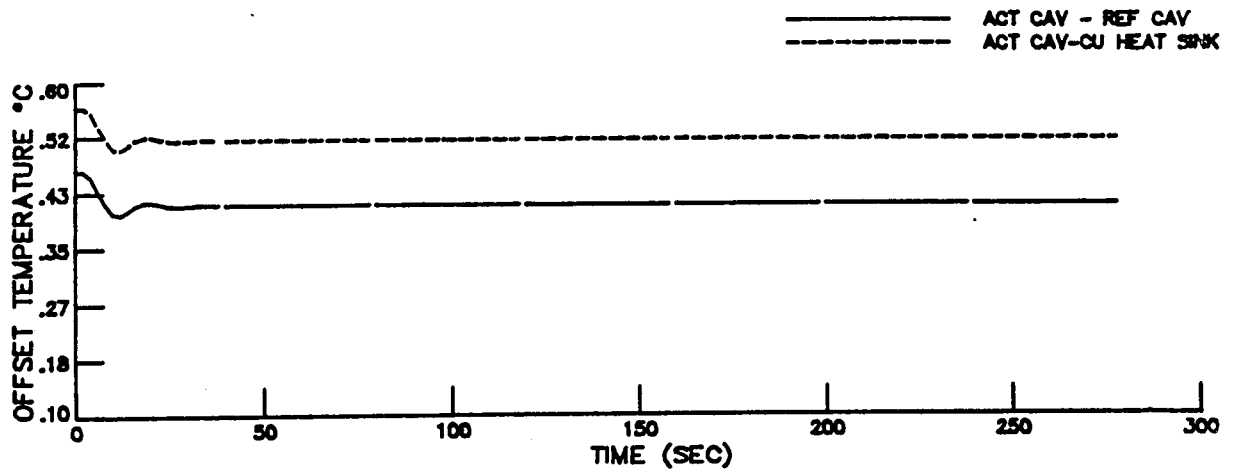
ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY .5 Ω (NOMINAL 2319.1 Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

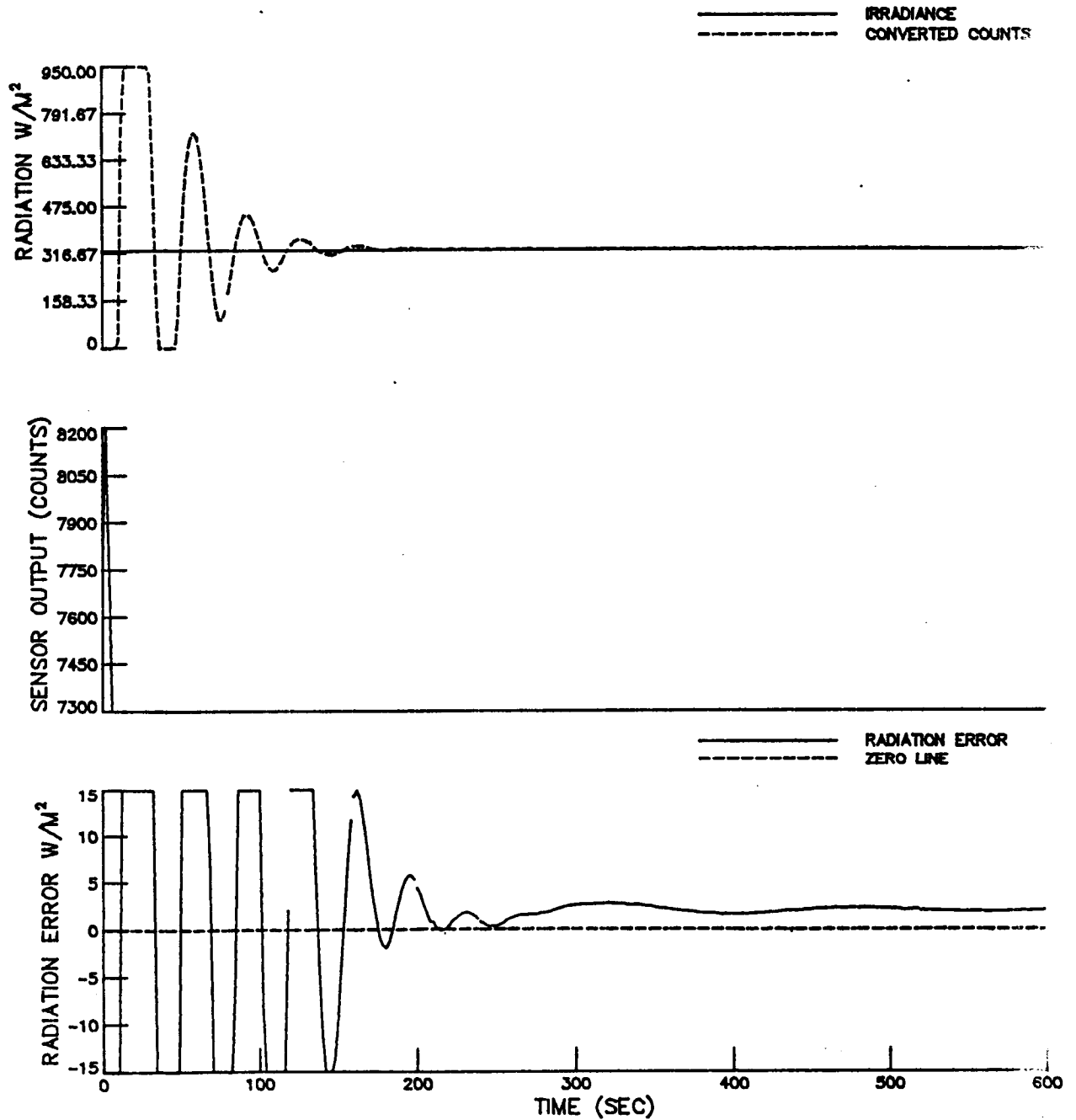
ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.5 \Omega$ (NOMINAL 2319.1Ω)



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

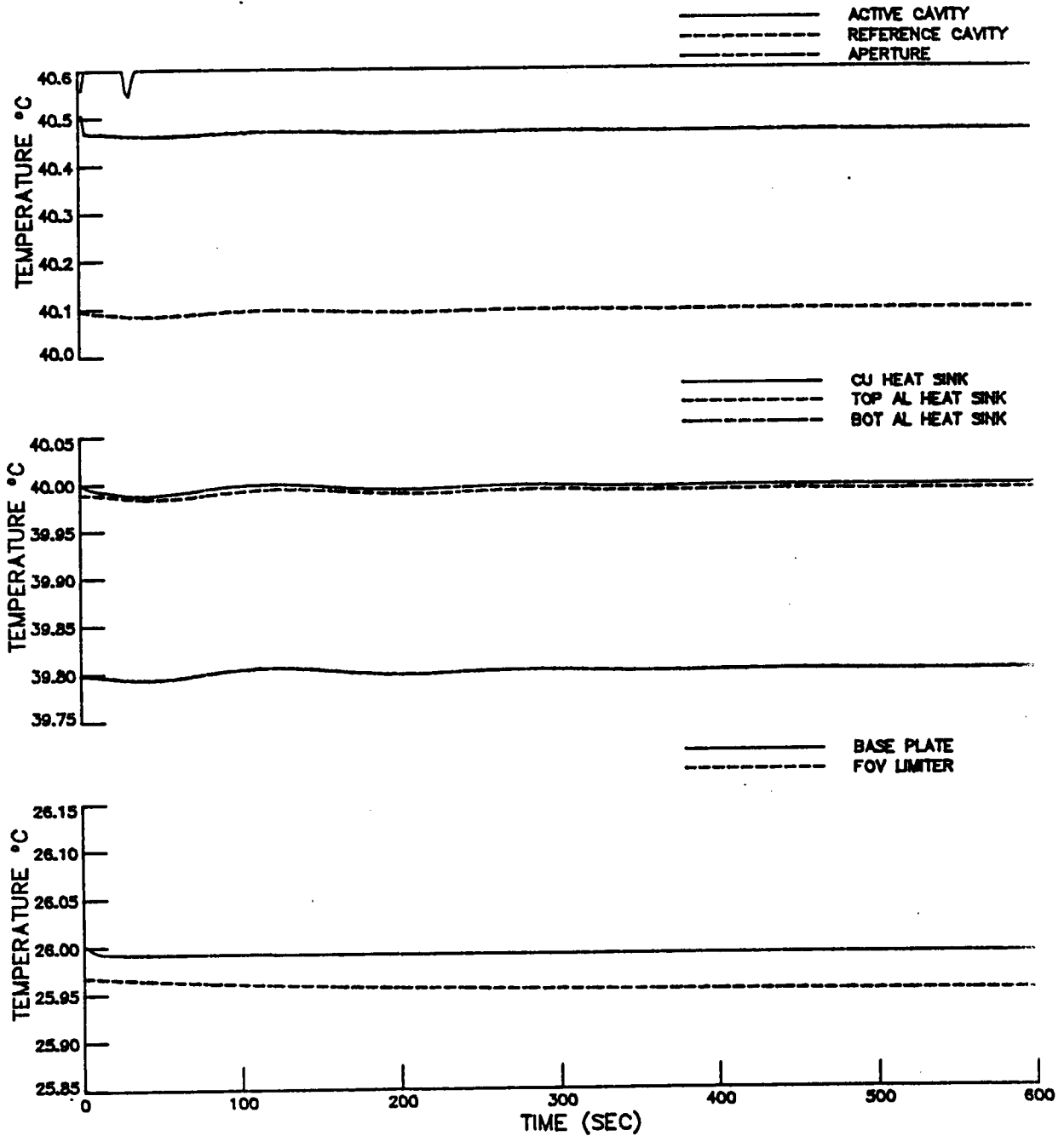
REDUCED CONDUCTANCE BETWEEN NODES 1 AND 2



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

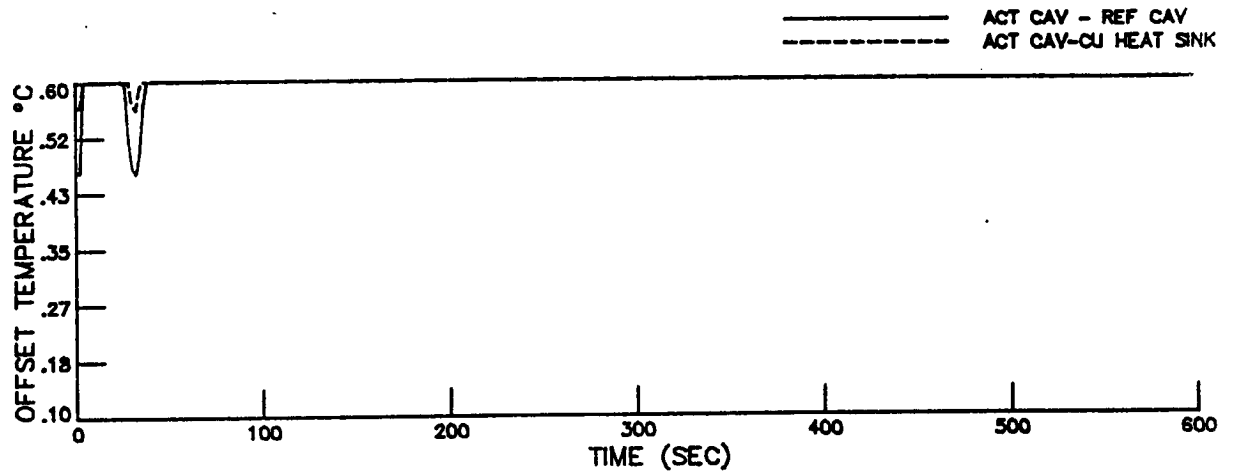
REDUCED CONDUCTANCE BETWEEN NODES 1 AND 2



SENSITIVITY ANALYSIS USING DYNAMIC MODEL

MFOV TOTAL SIMULATION

REDUCED CONDUCTANCE BETWEEN NODES 1 AND 2

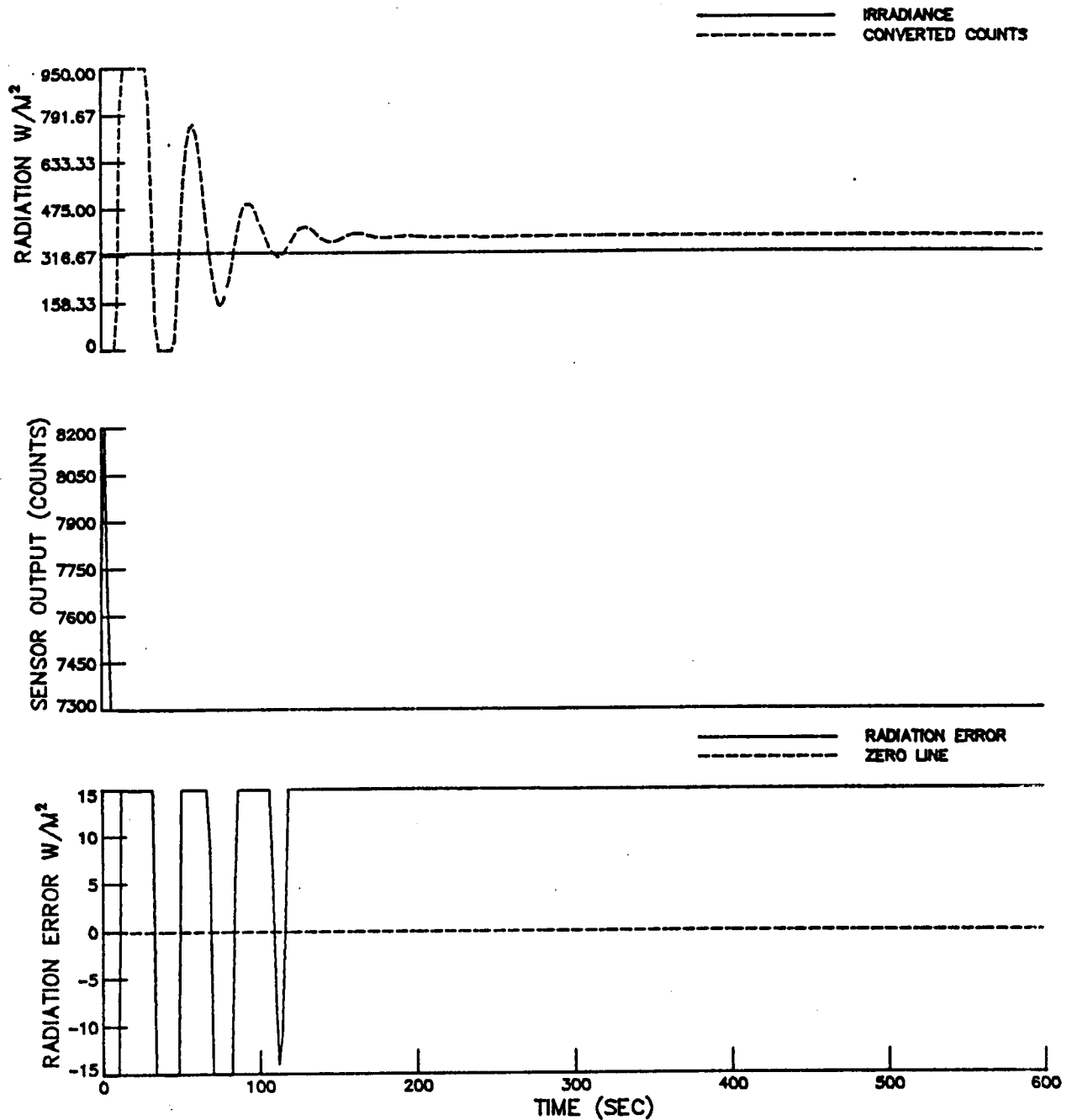


SENSITIVITY ANALYSIS USING DYNAMIC MODEL

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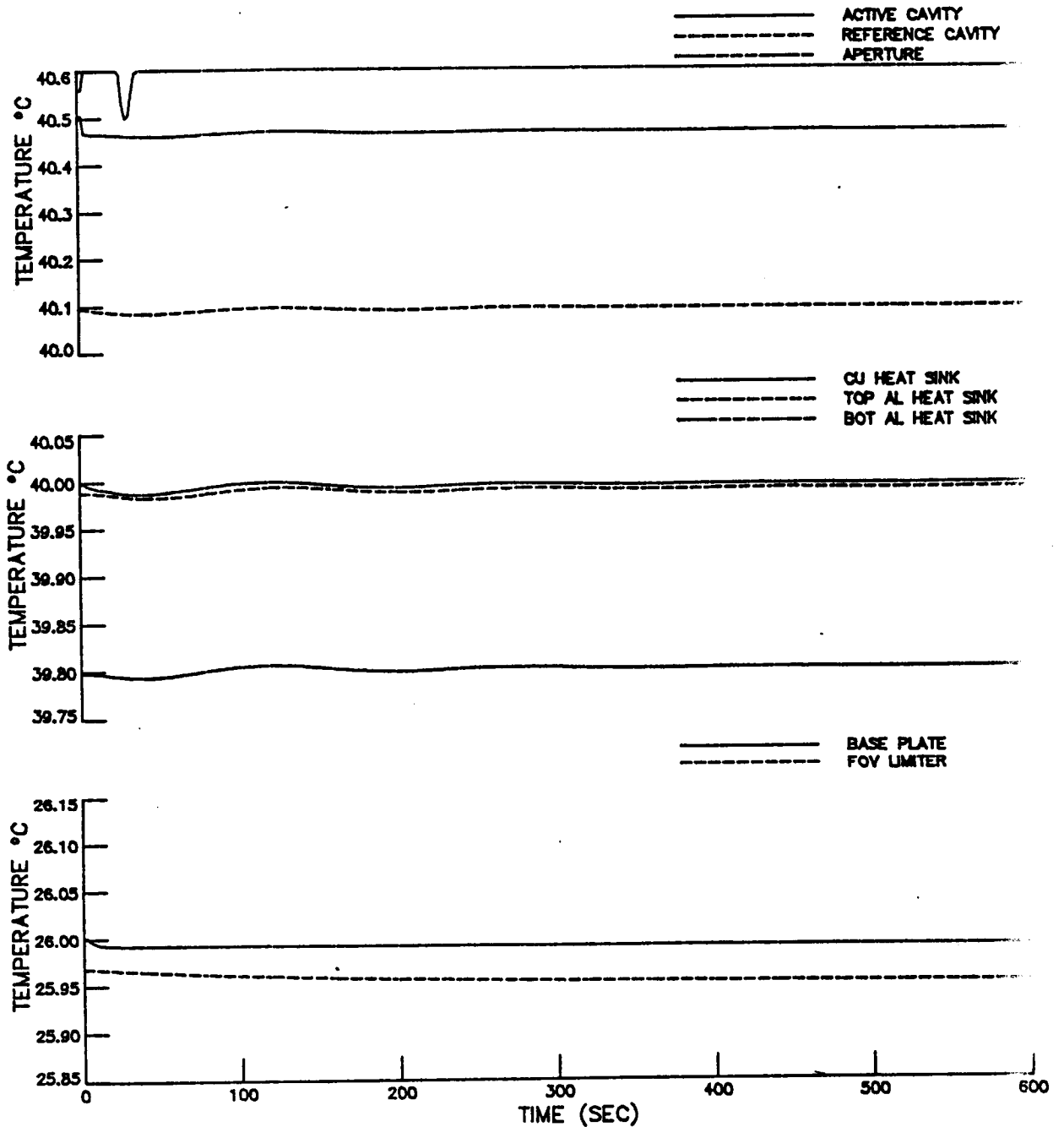


SENSITIVITY ANALYSIS USING DYNAMIC MODEL

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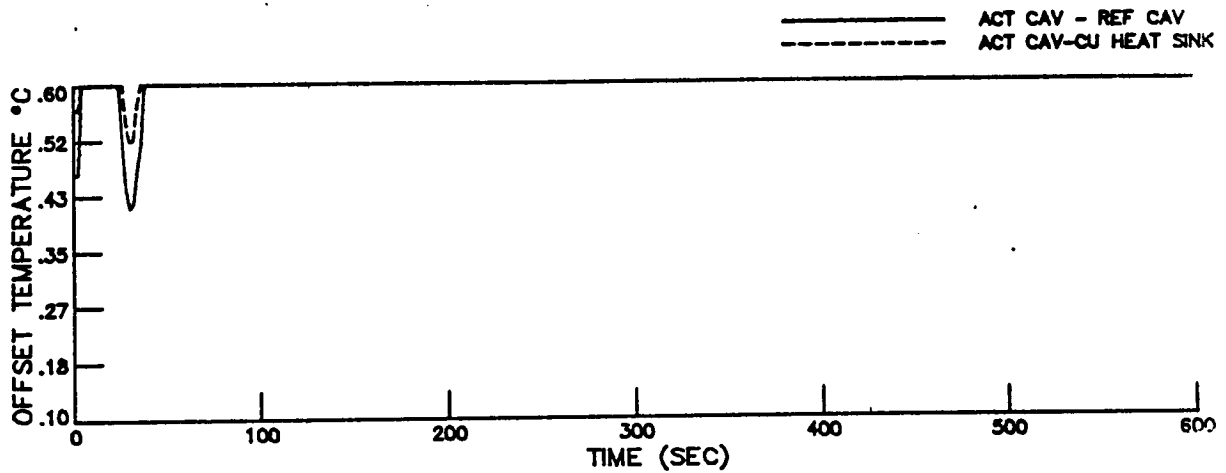


SENSITIVITY ANALYSIS USING DYNAMIC MODEL

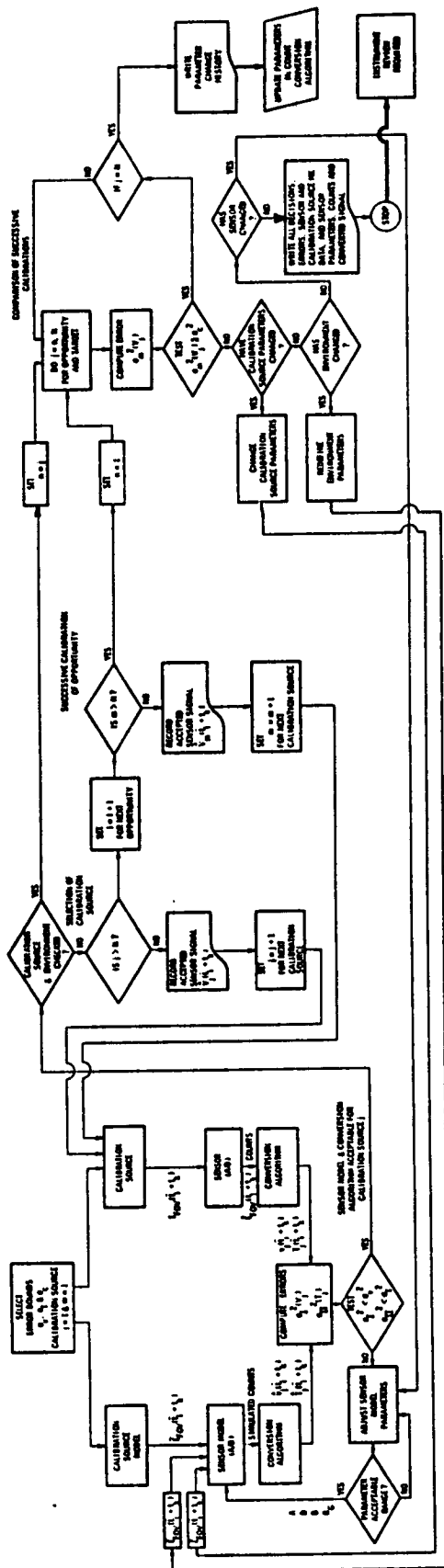
MFOV TOTAL SIMULATION

REDUCED CONDUCTANCE BETWEEN NODES 1 AND 2

ACTIVE CAVITY SENSOR WIRE RESISTANCE CHANGED BY $.5 \Omega$ (NOMINAL 2319.1Ω)



**ORIGINAL PAGE IS
OF POOR QUALITY**



CONCLUSIONS

- DYNAMIC MODELS HELP THE UNDERSTANDING OF THE ERBE INSTRUMENTS
- SIMPLE COUNT CONVERSION ALGORITHMS HAVE BEEN DEVELOPED
- STEADY STATE SIMULATION AND GROUND CALIBRATION RESULTS ARE IN GENERAL AGREEMENT
- THE ERRORS DUE TO COUNT CONVERSION ARE BEING ANALYZED BY SIMULATION
- INITIAL SENSITIVITY AND ERROR ANALYSES PROVIDE CONSTRUCTIVE RESULTS
- OVERALL APPROACH SEEMS WELL-SUITED TO ASSESS INSTRUMENT PERFORMANCE, ACCURACY AND VALIDATION

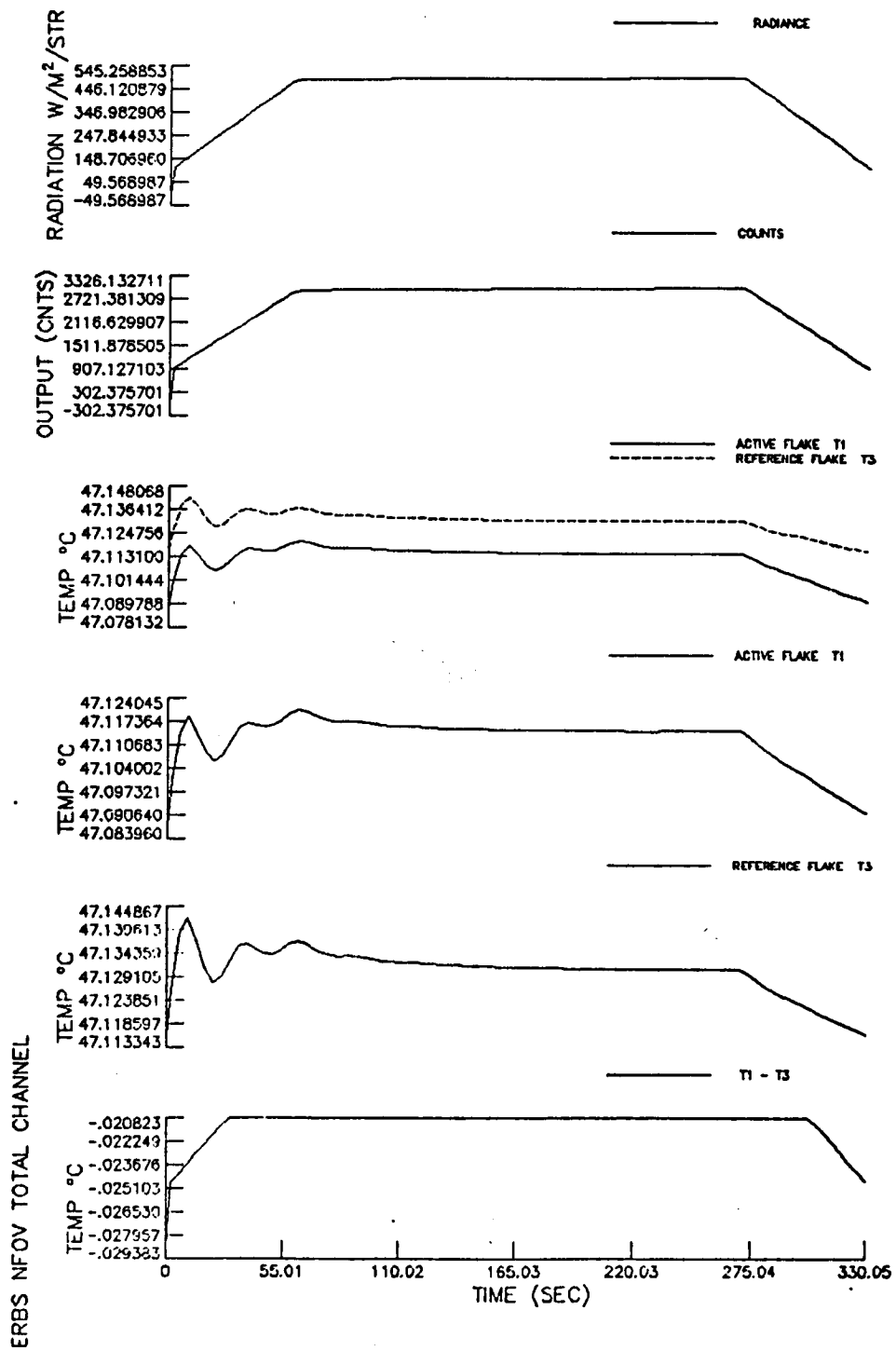
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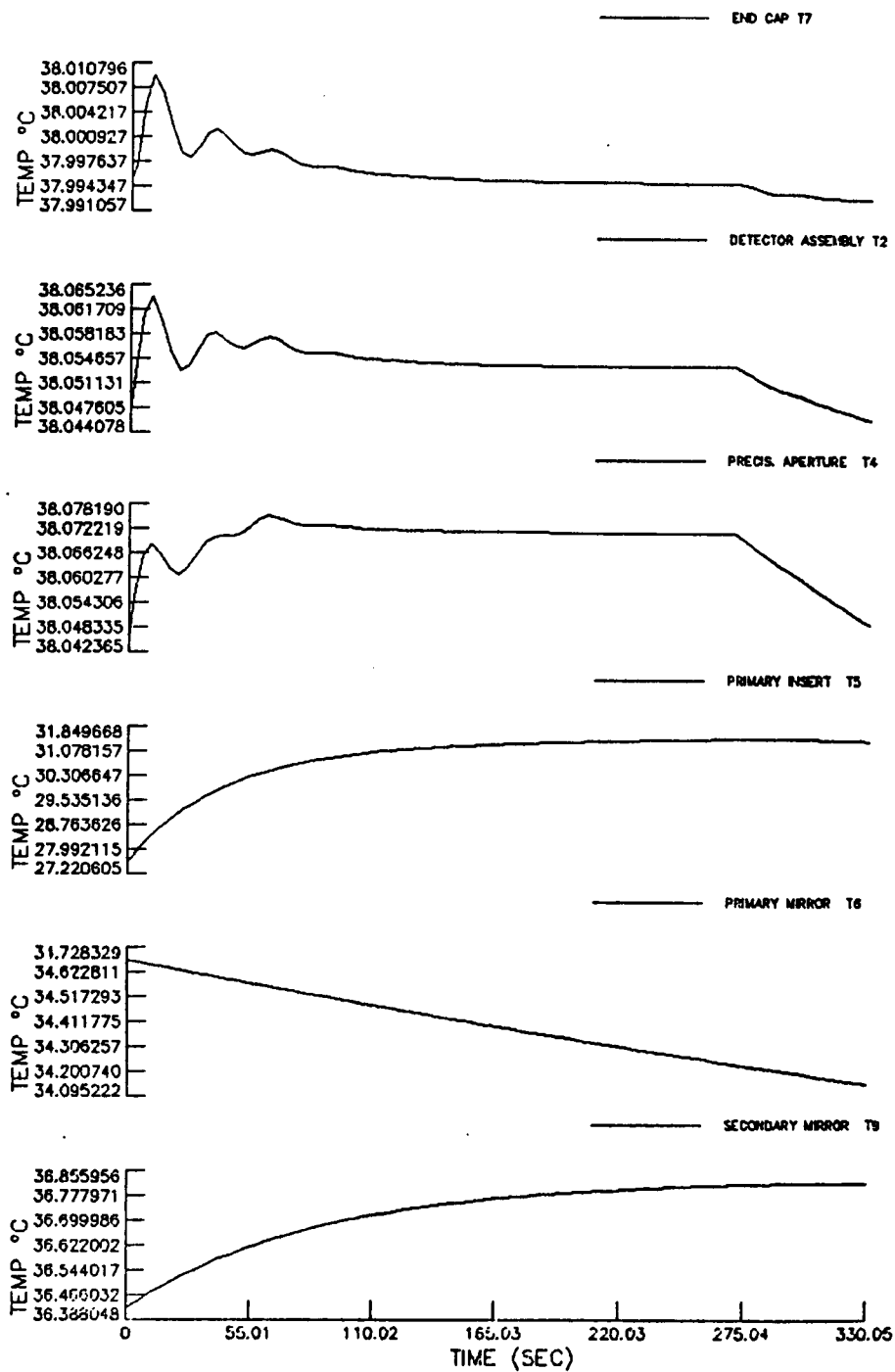
APPENDIX B

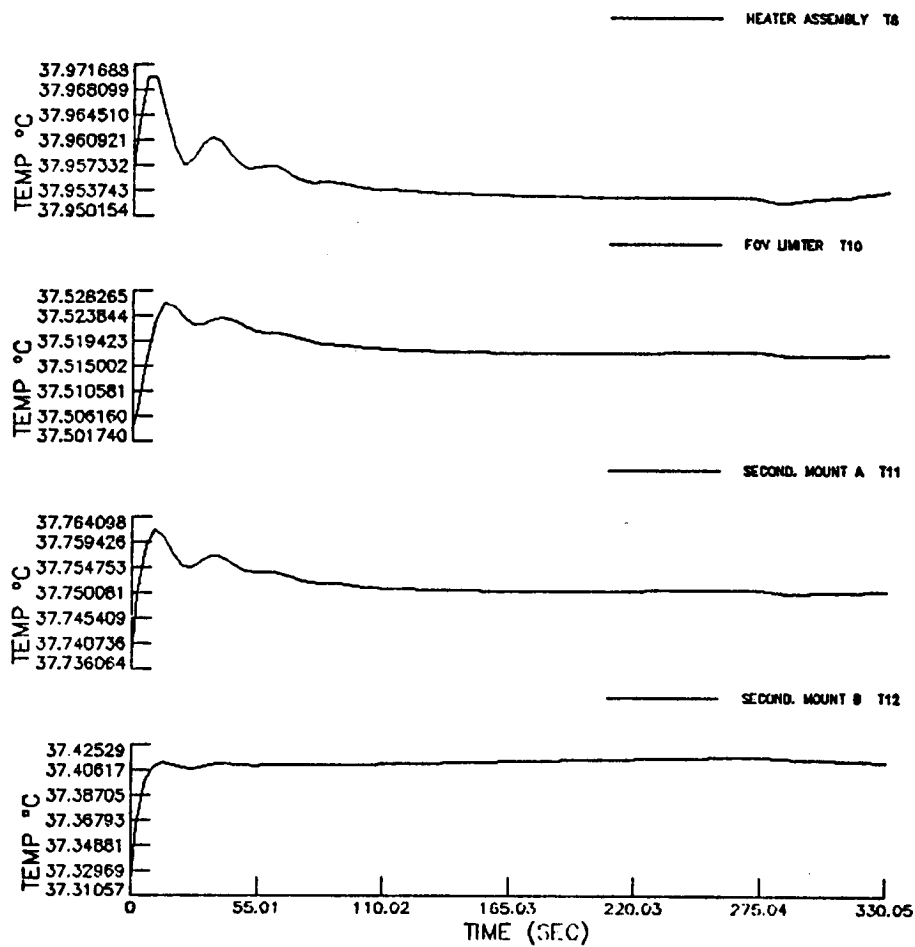
ERBE SCANNER SIMULATIONS

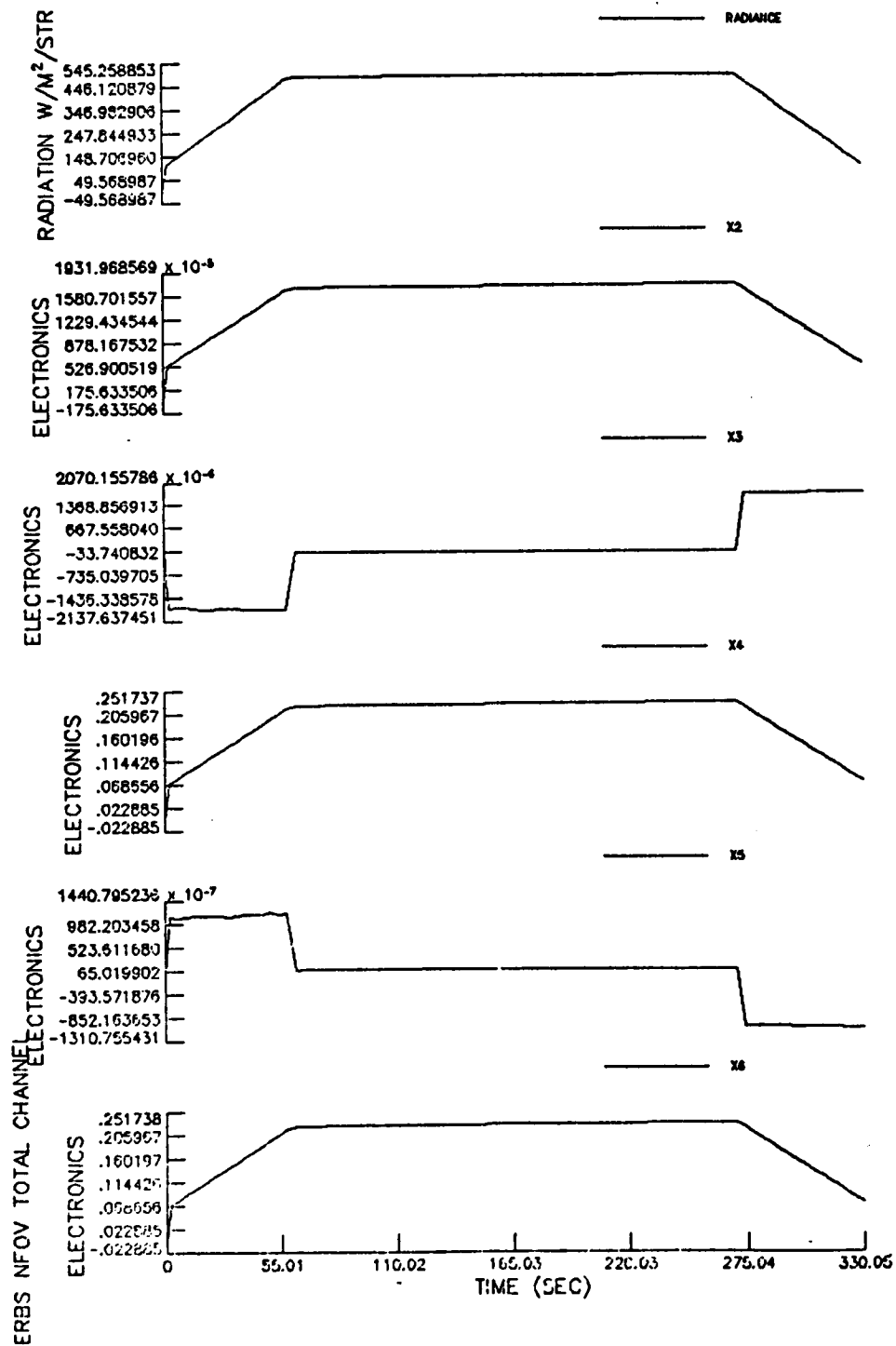
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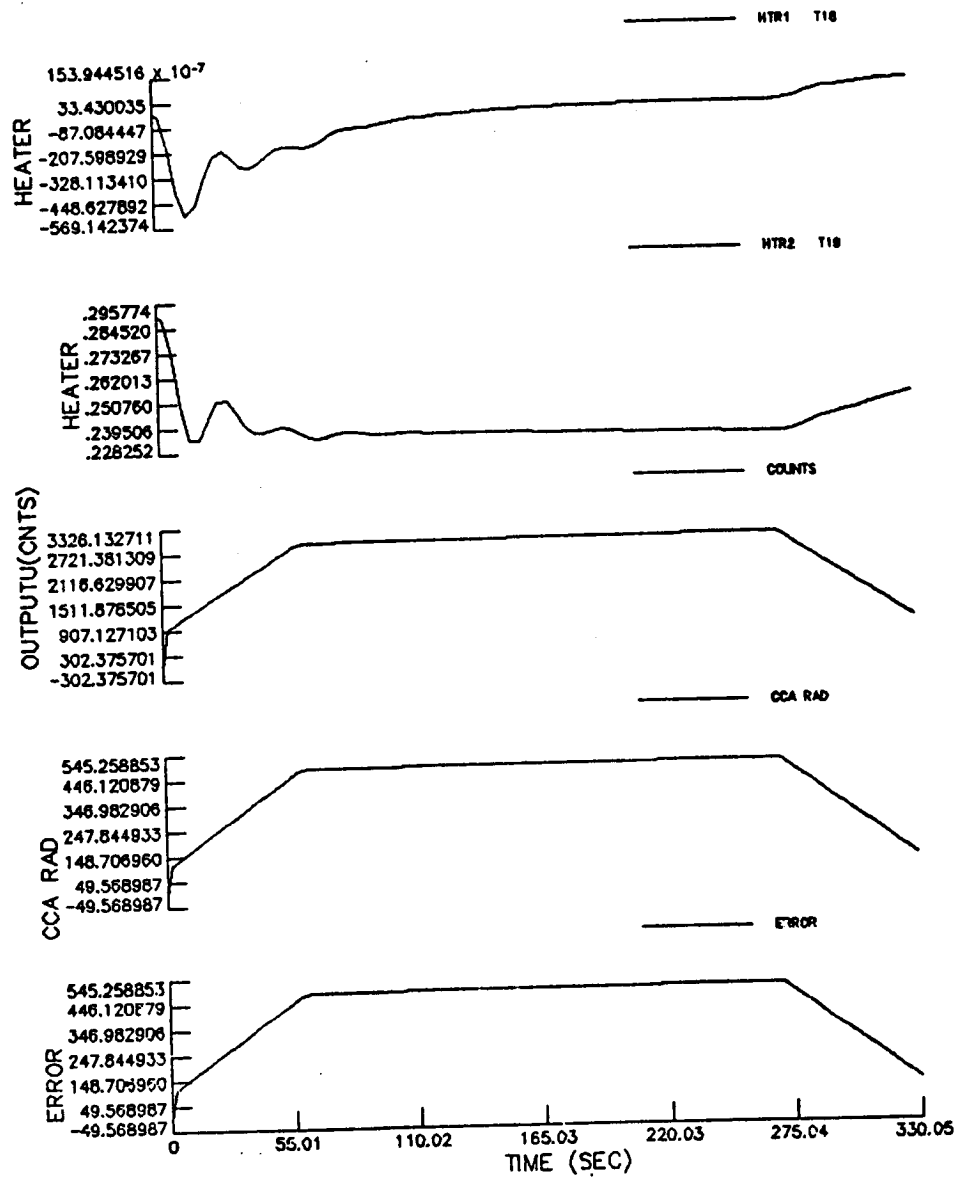
These graphical displays were produced during development of the simulation model. Please note that the plot of "OUTPUT (CNTS)" versus "TIME (SEC)" in the graph on pages 169, 173, 174, 177 and 180 does not display quantitatively absolute values for counts generated from the simulation model due to uncertain gain/offset parameters in the electronics.

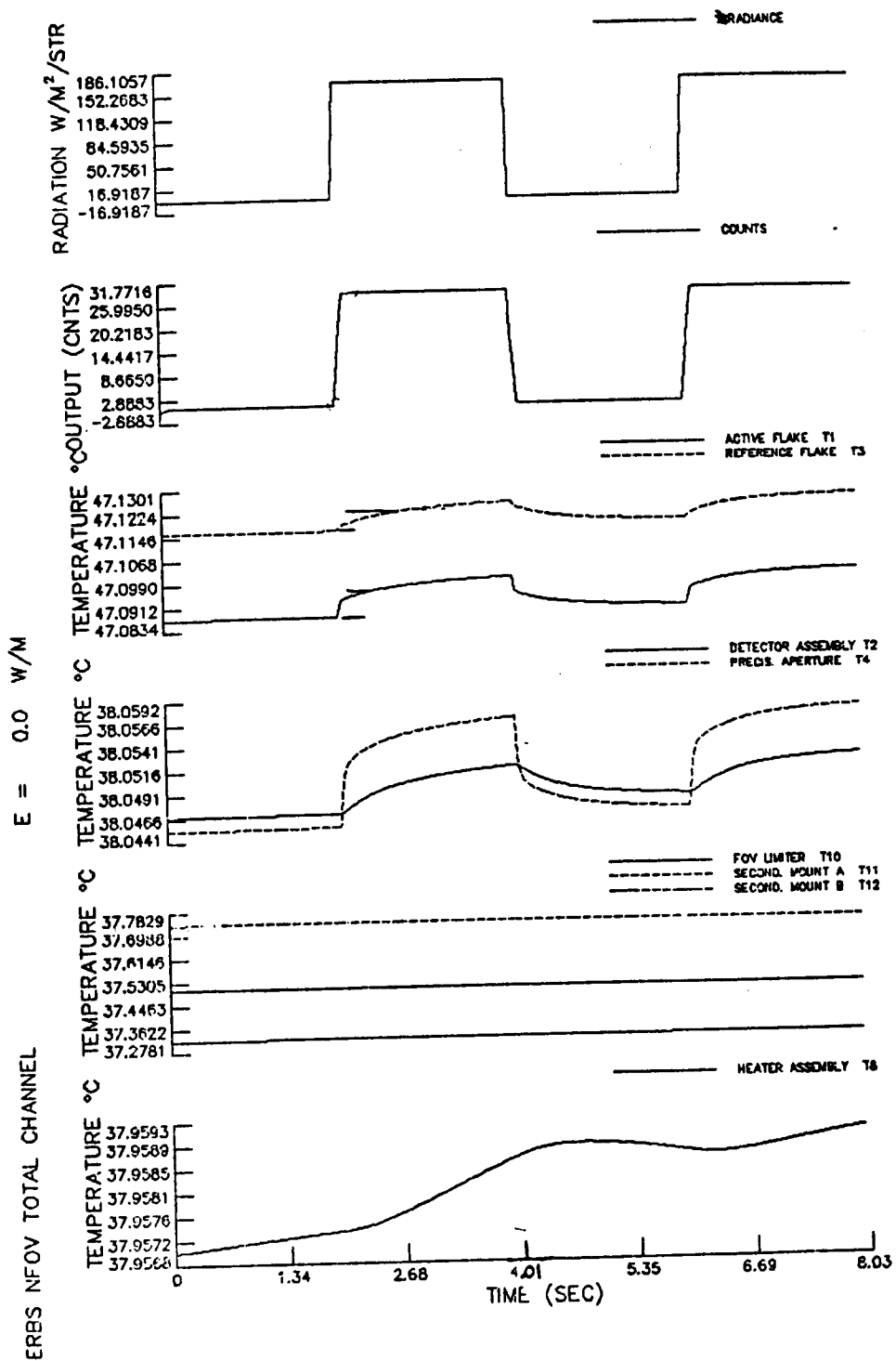


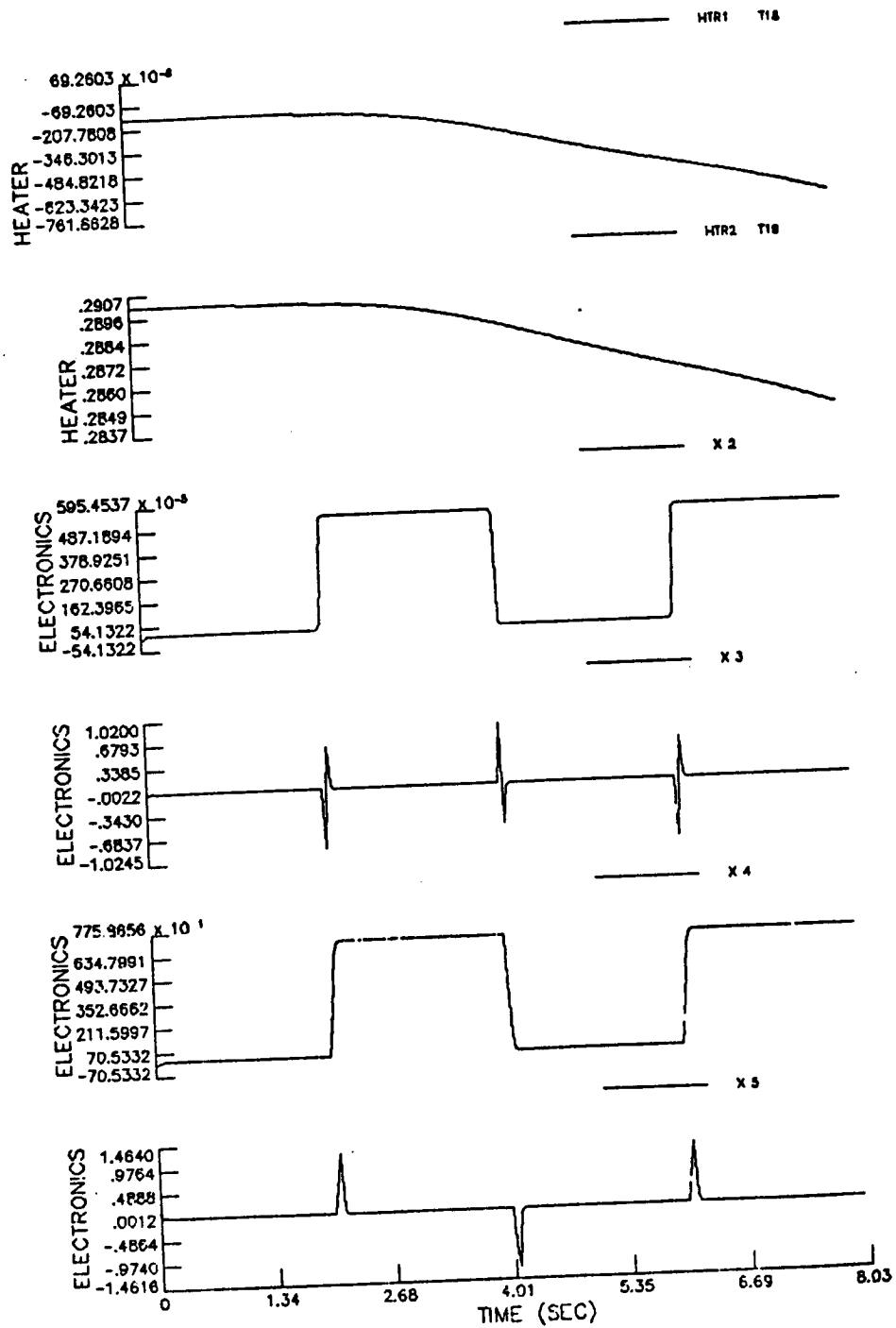


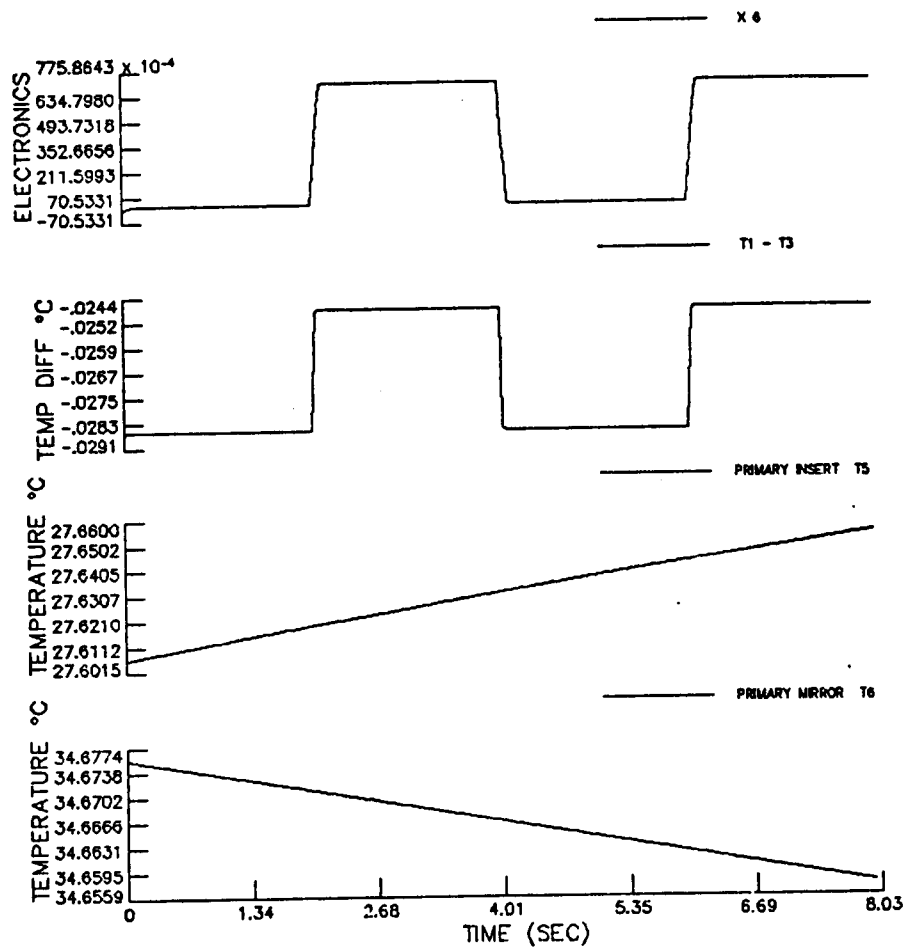


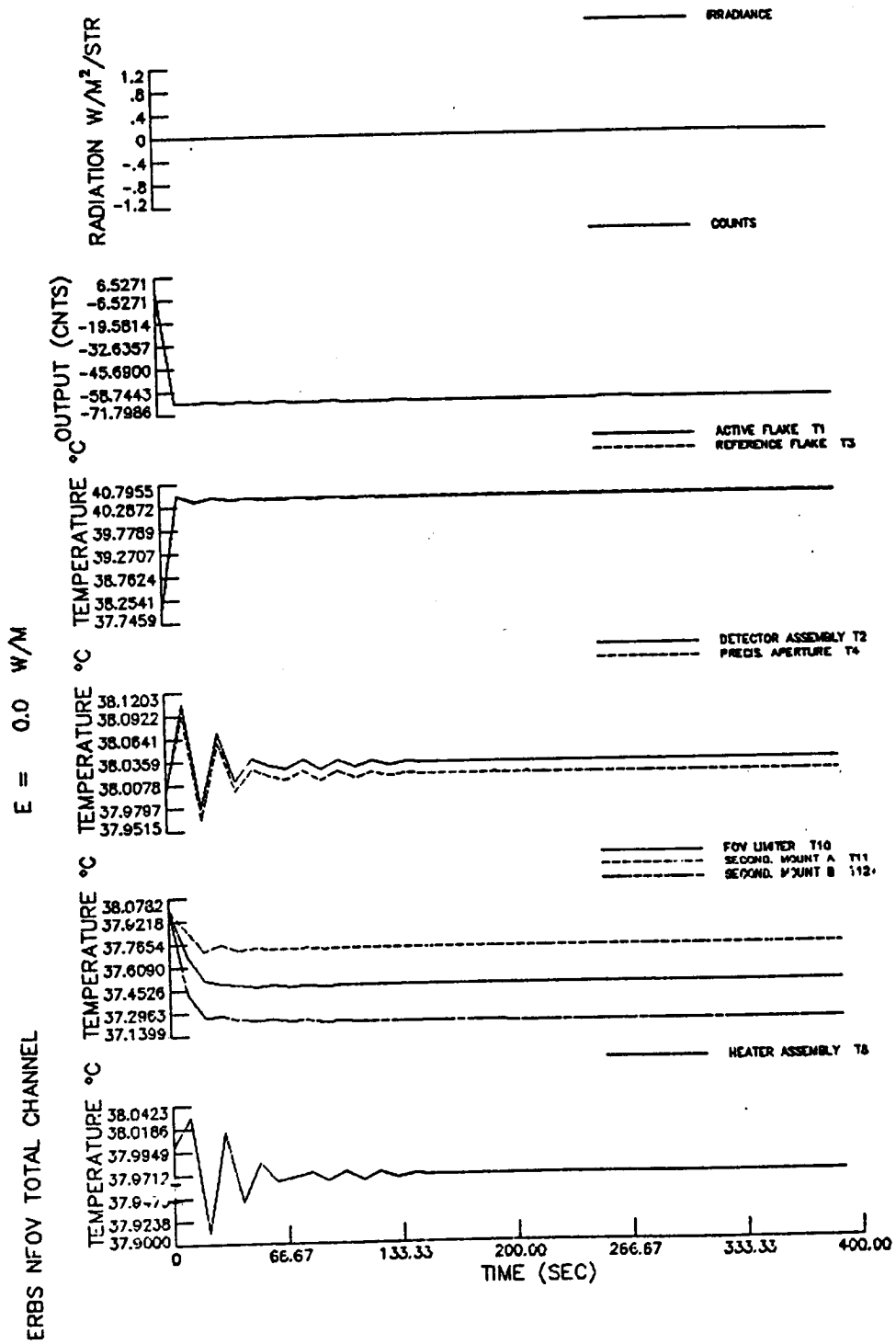


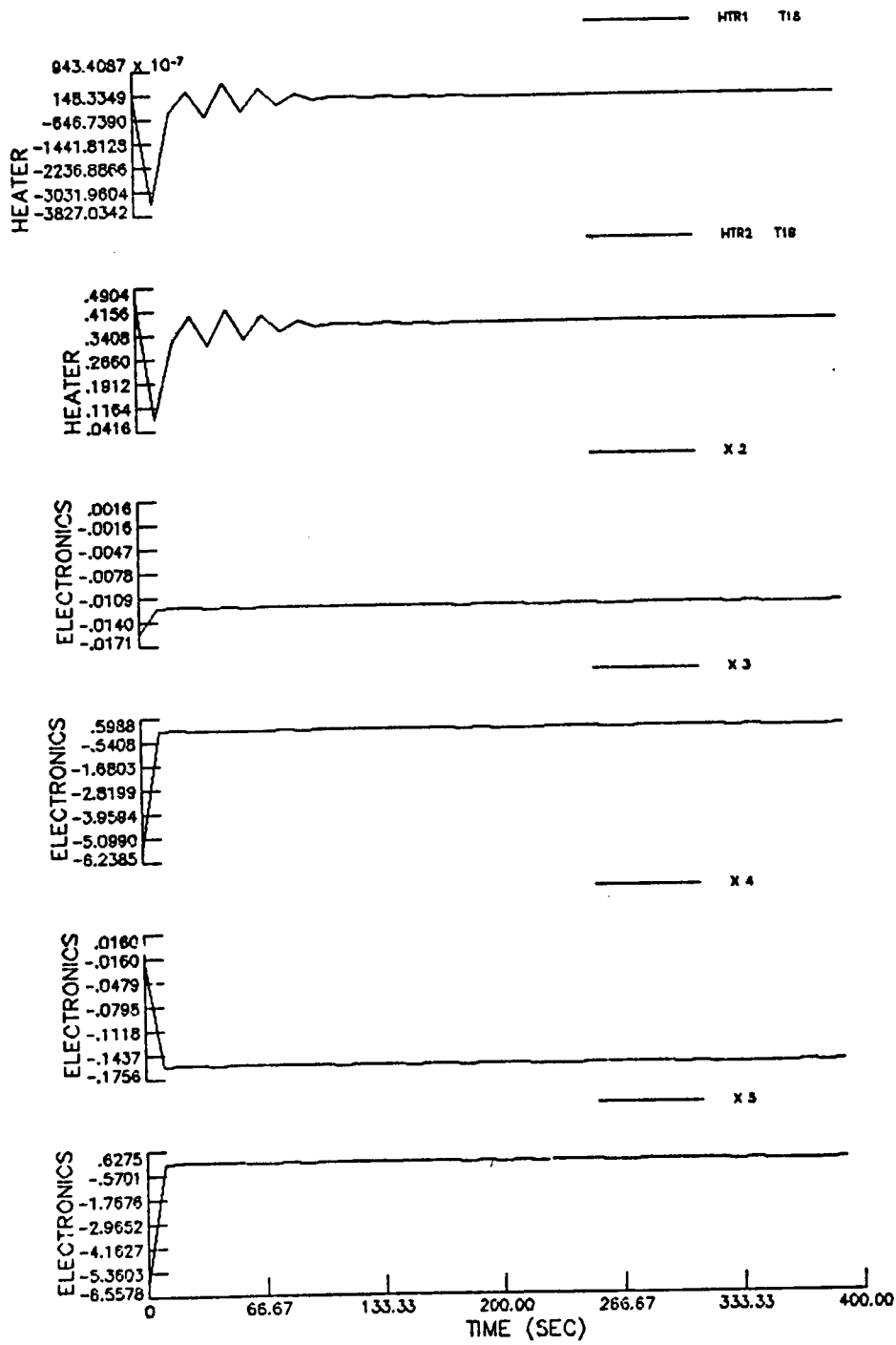


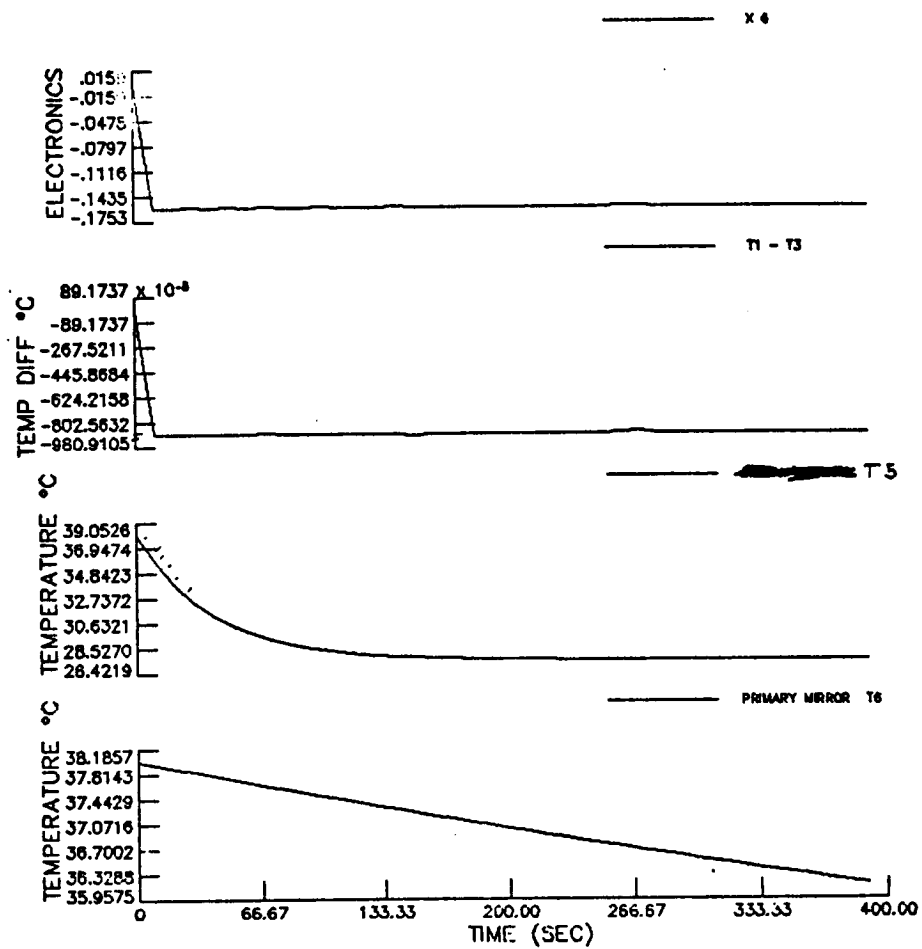


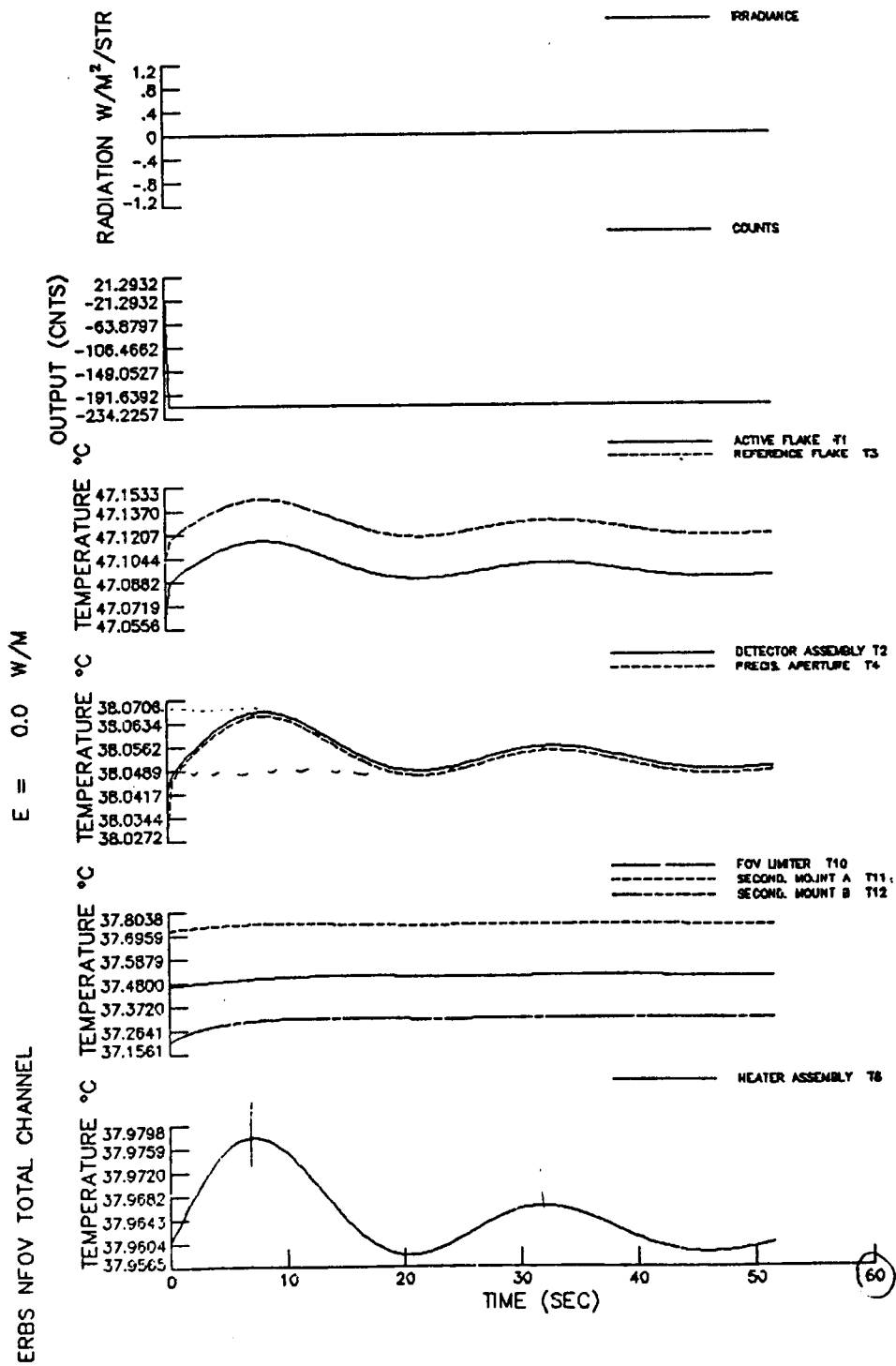


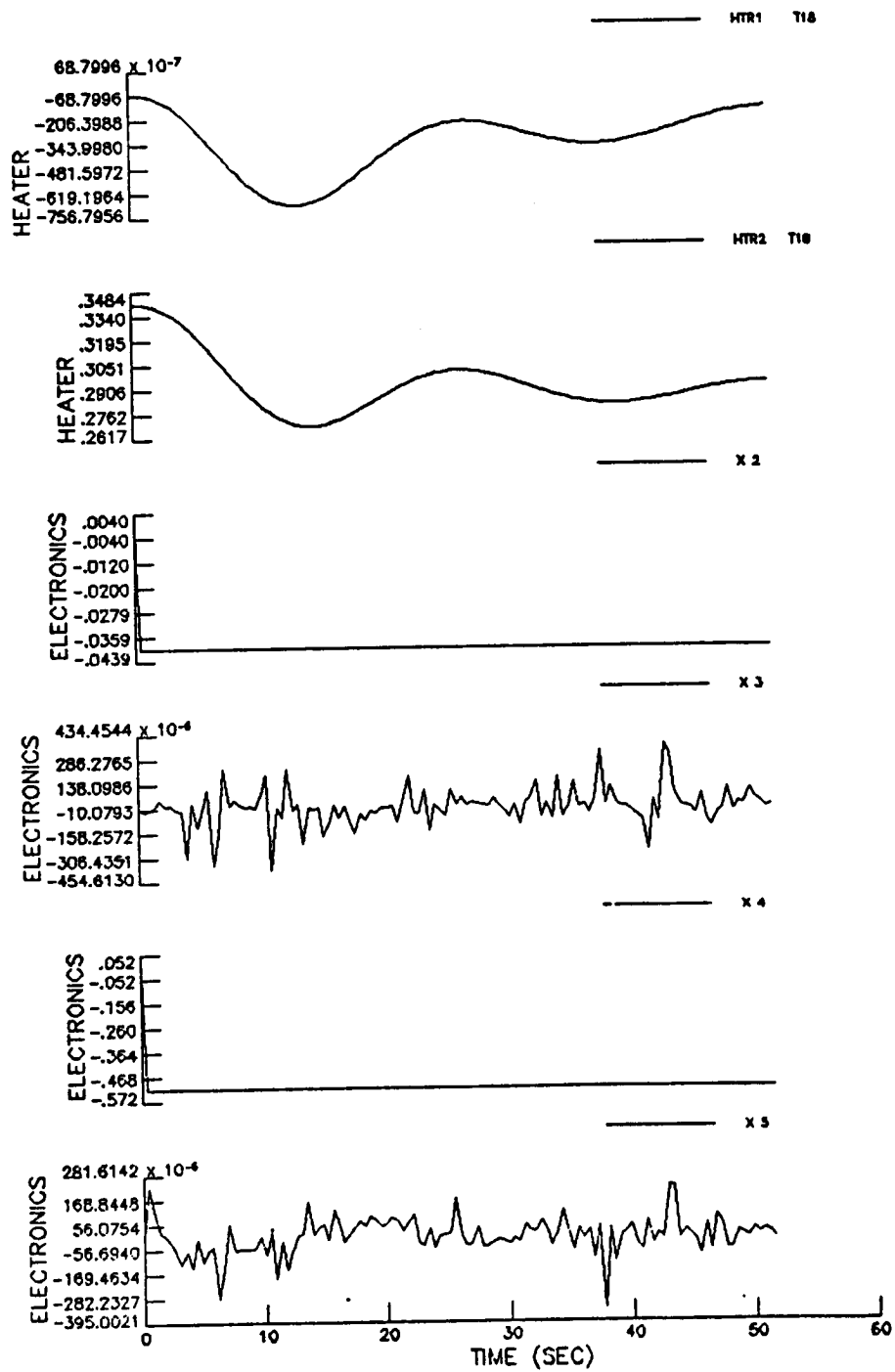


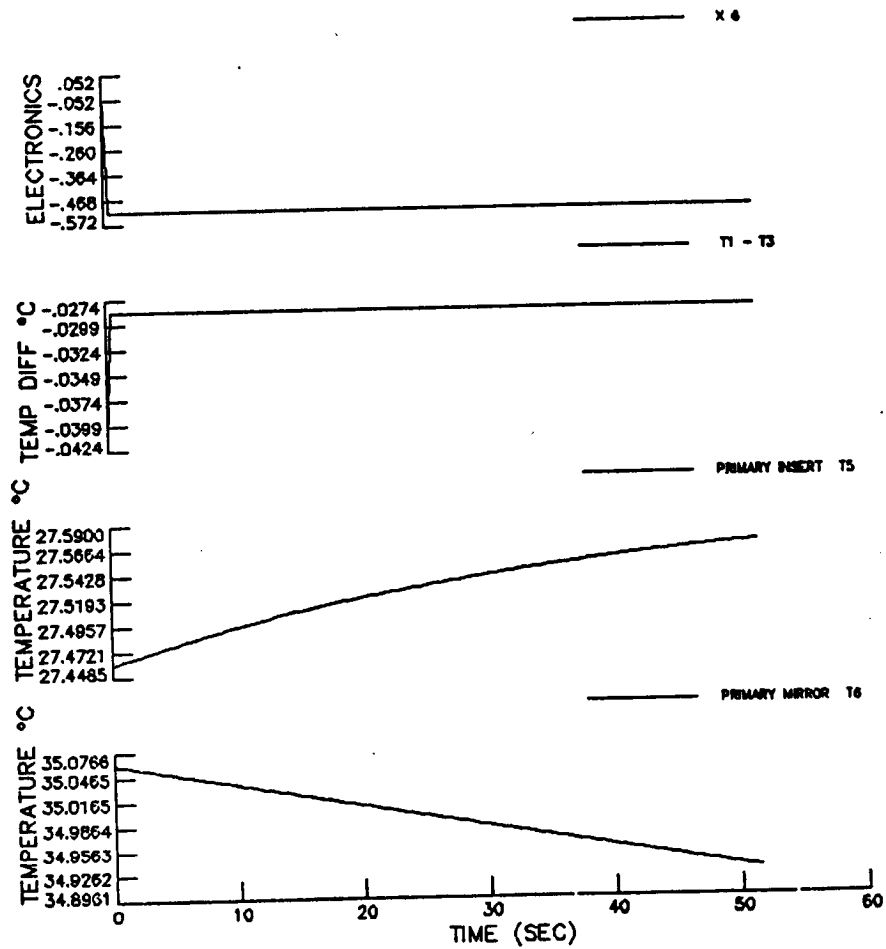












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APPENDIX C

ERBE NON-SCANNER THERMAL MODEL

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NON-SCANNER MODEL

The energy balance equation

$$M_k \dot{T}_k = \sum_{j=1}^n \frac{T_j - T_k}{R_{kj}} + A_k \phi_k + \dot{Q}_{\text{source}} \quad (1)$$

The second term in the right-hand side of the above equation is the radiation interaction of the k_{th} element to the other elements in an enclosure. This variables are defined by the following approach.

For the radiant interchange in an enclosure having both specular and diffuse surfaces, the surface properties are defined as the followings:

The reflectivity is

$$\rho_s = \rho_s^d + \rho_s^s$$

$$\rho_s = \rho_\ell^d + \rho_\ell^s$$

where subscripts, s and ℓ , signify short and long waves and superscripts, d and s, denote diffuse and specular.

For diffuse and gray approximation, the following realtionships are obtained

$$\rho_s + \epsilon_s + \tau_s = 1$$

$$\rho_\ell + \epsilon_\ell + \tau_\ell = 1.$$

From now, the subscripts for short and longwaves are omitted for brevity.

The radiosoty at the k_{th} surface is

$$M_k^d = \epsilon_k \sigma T_k^4 + \rho_k^d E_k + \tau_k \tilde{E}_k. \quad (2)$$

The incident radiant flux at the k_{th} surface is

$$E_k = \sum_j \bar{F}_{k-j} M_j^d \quad (3)$$

Combining the above two equations yields

$$M_j^d = \sum_i (\delta_{ji} - \rho_j^d \bar{F}_{j-i})^{-1} (\epsilon_k \Omega_i + \tau_i \tilde{E}_i) \quad (4)$$

From (3) and (4),

$$E_k = \sum_i (\epsilon_i \Omega_i + \tau_i \bar{E}_i) \sum_j \bar{F}_{k-j} P_{j-i} \quad (5)$$

where

$$P_{j-i} = (\delta_{ji} - \rho_j^d \bar{F}_{j-i})^{-1}$$

$$\Omega = \sigma T^4.$$

The radiant energy balance equation is the difference between the incident energy and the radiosoty. That is,

$$\begin{aligned} \phi_k &= E_k - M_k \\ &= E_k - \epsilon_k \Omega_k - \rho_k^d E_k - \rho_k^s E_k - \tau_k \tilde{E}_k \end{aligned} \quad (6)$$

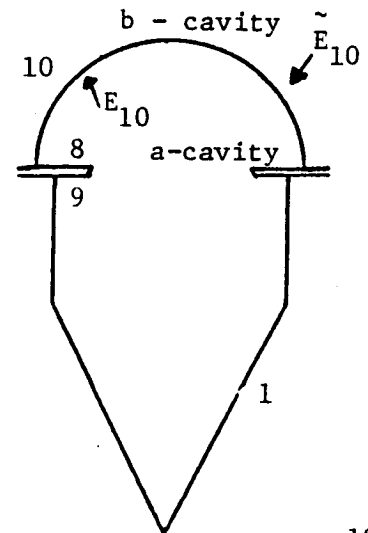
Substituting (5) into (6) we have

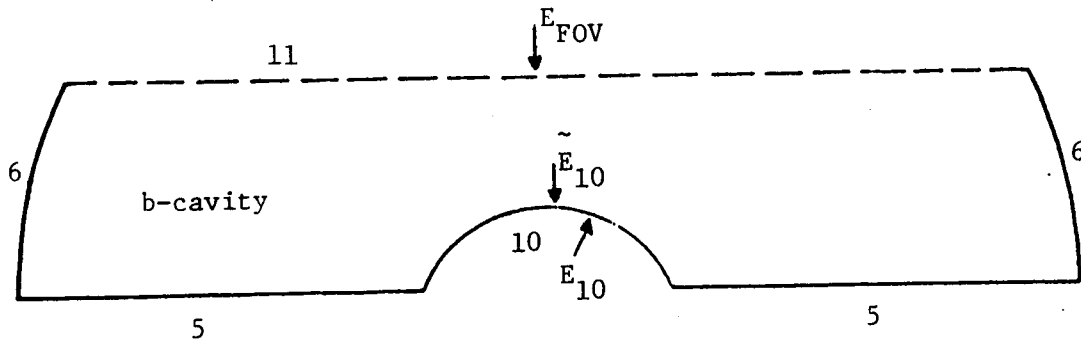
$$\begin{aligned} \phi_k &= \sum_i [(1 - \rho_k) \epsilon_i \sum_j \bar{F}_{k-j} P_{j-i} - \epsilon_k \delta_{ki}] \Omega_i \\ &+ \sum_i [(1 - \rho_k) \tau_i \sum_j \bar{F}_{k-j} P_{j-i} - \tau_k \delta_{ki}] \tilde{E}_i \end{aligned} \quad (7)$$

Let $H_{k-i} = \sum_j \bar{F}_{k-j} P_{j-i}$

Then, for the cavity a, the equation (5) can be expanded as

$$\begin{aligned} E_{10} &= H_{10-1} \epsilon_1 \Omega_1 \\ &+ [H_{10-8} \epsilon_8 + H_{10-9} \epsilon_9] \Omega_8 \\ &+ H_{10-10}^a \epsilon_{10} \Omega_{10} \\ &+ H_{10-10}^a \tau_{10} \tilde{E}_{10} \end{aligned} \quad (A)$$





For the cavity, b

$$\begin{aligned} \tilde{E}_{10} = & H_{10-5} \epsilon_5 \Omega_5 + H_{10-6} \epsilon_6 \Omega_6 + H_{10-10}^b \epsilon_{10} \Omega_{10} \\ & + H_{10-10}^b \tau_{10} E_{10} + H_{10-11}^b \tau_{11} E_{FOV} \end{aligned} \quad (B)$$

From (A) and (B),

$$\begin{aligned} E_{10} = & G \epsilon_1 H_{10-1} \Omega_1 \\ & + G \epsilon_5 \tau_{10} H_{10-5} H_{10-10}^a \Omega_5 \\ & + G \epsilon_6 \tau_{10} H_{10-6} H_{10-10}^a \Omega_6 \\ & + G [H_{10-8} \epsilon_8 + H_{10-9} \epsilon_9] \Omega_8 \\ & + G \epsilon_{10} [H_{10-10}^a + \tau_{10} H_{10-10}^a H_{10-10}^b] \Omega_{10} \\ & + G \tau_{10} \tau_{11} H_{10-10}^a H_{10-11} E_{FOV} \end{aligned}$$

where $G = (1 - \tau_{10}^2 H_{10-10}^a H_{10-10}^b)^{-1}$.

$$\begin{aligned} \tilde{E}_{10} = & G \epsilon_1 \tau_{10} H_{10-10}^b H_{10-1} \Omega_1 \\ & + G \epsilon_5 H_{10-5} \Omega_1 \\ & + G \epsilon_6 H_{10-6} \Omega_6 \\ & + G \tau_{10} H_{10-10}^b [H_{10-8} \epsilon_8 + H_{10-9} \epsilon_9] \Omega_8 \\ & + G \epsilon_{10} H_{10-10}^b [1 + \tau_{10} H_{10-10}^a] \Omega_{10} \\ & + G \tau_{11} H_{10-11}^b E_{FOV} \end{aligned}$$

$$\phi_i = \sum_j [(1 - \rho_i) H_{ij} \epsilon_j - \delta_{ij} \epsilon_j] \Omega_j + \sum_j [(1 - \rho_i) H_{ij} \tau_j - \delta_{ij} \tau_j] \tilde{E}_j$$

In cavity a:

$$\begin{aligned} \phi_1^a = & \epsilon_1 [(1 - \rho_1) H_{1-1} + (1 - \rho_1) \tau_{10}^2 G H_{1-10} H_{10-10}^b H_{10-1} - 1] \Omega_1 \\ & + (1 - \rho_1) \epsilon_5 \tau_{10} G H_{1-10} H_{10-5} \Omega_5 \\ & + (1 - \rho_1) \epsilon_6 \tau_{10} G H_{1-10} H_{10-6} \Omega_6 \\ & + (1 - \rho_1) \{ (\epsilon_8 H_{1-8} + \epsilon_9 H_{1-9}) + \tau_{10}^2 G H_{1-10} H_{10-10}^b (\epsilon_8 H_{10-8} \\ & \quad + H_{10-9} \epsilon_9) \} \Omega_8 \\ & + (1 - \rho_1) \epsilon_{10} H_{1-10} [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)] \Omega_{10} \\ & + (1 - \rho_1) \tau_{10} \tau_{11} G H_{1-10} H_{10-11} E_{FOV} \end{aligned}$$

$$\phi_8^a = \frac{A_8}{A_{(8+9)}} \phi_8 + \frac{A_9}{A_{(8+9)}} \phi_9$$

$$\begin{aligned} \phi_8 = & (1 - \rho_8) [\epsilon_1 H_{8-1} + \epsilon_1 \tau_{10}^2 G H_{8-10} H_{10-10}^b H_{10-1}] \Omega_1 \\ & + (1 - \rho_8) \epsilon_5 \tau_{10} G H_{8-10} H_{10-5} \Omega_5 \\ & + (1 - \rho_8) \epsilon_6 \tau_{10} G H_{8-10} H_{10-6} \Omega_6 \\ & + \{ \epsilon_8 [(1 - \rho_8) H_{8-8} - 1] + (1 - \rho_8) \tau_{10}^2 G H_{8-10} H_{10-10}^b (H_{10-8} \epsilon_8 \\ & \quad + H_{10-9} \epsilon_9) + (1 - \rho_8) \epsilon_9 H_{8-9} \} \Omega_8 \\ & + (1 - \rho_8) \epsilon_{10} H_{8-10} [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)] \Omega_{10} \\ & + (1 - \rho_8) \tau_{10} \tau_{11} G H_{8-10} H_{10-11} E_{FOV} \end{aligned}$$

$$\begin{aligned}
\phi_9 = & (1 - \rho_9) \epsilon_1 [H_{9-1} + \tau_{10}^2 G H_{9-10} H_{10-10}^b H_{10-1}] \Omega_1 \\
& + (1 - \rho_9) \epsilon_5 \tau_{10} G H_{9-10} H_{10-5} \Omega_5 \\
& + (1 - \rho_9) \epsilon_6 \tau_{10} G H_{9-10} H_{10-6} \Omega_6 \\
& + \{ (1 - \rho_9) \epsilon_9 H_{9-8} + \epsilon_9 [(1 - \rho_9) H_{9-9} - 1] + (1 - \rho_9) \tau_{10}^2 G H_{9-10} H_{10-10}^b \\
& \quad (H_{10-8} \epsilon_8 + H_{10-9} \epsilon_9) \} \Omega_8 \\
& + (1 - \rho_9) \epsilon_{10} H_{9-10} [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)] \Omega_{10} \\
& + (1 - \rho_9) \tau_{10} \tau_{11} G H_{9-10} H_{10-11} E_{FOV}
\end{aligned}$$

In Cavity b:

$$\begin{aligned}
\phi_5^b = & (1 - \rho_5) \epsilon_1 \tau_{10} G H_{5-10} H_{10-1} \Omega_1 \\
& + \epsilon_5 \{ [(1 - \rho_5) H_{5-5} - 1] + (1 - \rho_5) \tau_{10}^2 G H_{5-10} H_{10-5} H_{10-10}^a \} \Omega_5 \\
& + (1 - \rho_5) \epsilon_6 [H_{5-6} + \tau_{10}^2 G H_{5-10} H_{10-6} H_{10-10}^a] \Omega_6 \\
& + (1 - \rho_5) \tau_{10} G H_{5-10} (\epsilon_8 H_{10-8} + \epsilon_9 H_{10-9}) \Omega_8 \\
& + (1 - \rho_5) \epsilon_{10} H_{5-10} [1 + \tau_{10} G H_{10-10}^a (1 + \tau_{10} H_{10-10}^b)] \Omega_{10} \\
& + (1 - \rho_5) \tau_{10}^2 \tau_{11} G H_{5-10} H_{10-10}^a H_{10-11} E_{FOV}
\end{aligned}$$

$$\begin{aligned}
\phi_6^b = & \epsilon_1 (1 - \rho_6) \tau_{10} G H_{5-10} H_{10-1} \Omega_1 \\
& + (1 - \rho_6) \epsilon_6 [H_{6-5} + \tau_{10}^2 G H_{6-10} H_{10-5} H_{10-10}^a] \Omega_5 \\
& + \epsilon_6 \{ [(1 - \rho_6) H_{6-6} - 1] + (1 - \rho_6) \tau_{10}^2 G H_{6-10} H_{10-5} H_{10-10}^a \} \Omega_6 \\
& + (1 - \rho_6) \tau_{10} G H_{6-10} (\epsilon_8 H_{10-8} + \epsilon_9 H_{10-9}) \Omega_8 \\
& + (1 - \rho_6) \epsilon_{10} H_{5-10} [1 + \tau_{10} G H_{10-10}^a (1 + \tau_{10} H_{10-10}^b)] \Omega_{10} \\
& + (1 - \rho_6) \tau_{10}^2 \tau_{11} G H_{5-10} H_{10-10}^a H_{10-11} E_{FOV}
\end{aligned}$$

$$\begin{aligned}
\phi_{10}^b = & \epsilon_1 \tau_{10} G[(1 - \rho_{10}) H_{10-10}^b - 1] H_{10-1} \Omega_1 \\
& + \epsilon_5 H_{10-5} \{(1 - \rho_{10}) + \tau_{10}^2 G[(1 - \rho_{10}) H_{10-10}^b - 1] H_{10-10}^a\} \Omega_5 \\
& + \epsilon_6 H_{10-6} \{(1 - \rho_{10}) + \tau_{10}^2 G[(1 - \rho_{10}) H_{10-10}^b - 1] H_{10-10}^a\} \Omega_6 \\
& + \tau_{10} G[(1 - \rho_{10}) H_{10-10}^b - 1] (\epsilon_8 H_{10-8} + \epsilon_9 H_{10-9}) \Omega_8 \\
& + \epsilon_{10} [(1 - \rho_{10}) H_{10-10}^b - 1] \{1 + \tau_{10} G[H_{10-10}^a + \tau_{10} H_{10-10}^a H_{10-10}^b]\} \Omega_{10} \\
& + \tau_{10}^2 \tau_{11} G[(1 - \rho_{10}) H_{10-10}^b - 1] H_{10-10}^a H_{10-11} E_{FOV}
\end{aligned}$$

$$\phi_{10} \text{ filter} = \frac{A_{10a}}{A_{10(a+b)}} \phi_{10}^a + \frac{A_{10b}}{A_{10(a+b)}} \phi_{10}^b$$

Cavity C:

$$\begin{aligned}
\phi_1^c = & \epsilon_1 [(1 - \rho_1) H_{1-1} - 1] \Omega_1 + (1 - \rho_1) \epsilon_2 H_{1-2} \Omega_2 \\
& + (1 - \rho_1) \epsilon_4 H_{1-4} \Omega_4 \\
\phi_2^c = & (1 - \rho_2) \epsilon_1 H_{2-1} \Omega_1 + \epsilon_2 [(1 - \rho_2) H_{2-2} - 1] \Omega_2 \\
& + (1 - \rho_2) \epsilon_4 H_{2-4} \Omega_4 \\
\phi_4^c = & (1 - \rho_4) \epsilon_1 H_{4-1} \Omega_1 + (1 - \rho_4) \epsilon_2 H_{4-2} \Omega_2 \\
& + \epsilon_4 [(1 - \rho_4) H_{4-4} - 1] \Omega_4
\end{aligned}$$

where ϵ and ρ are ϵ^c and ρ^c .

Cavity d:

$$\begin{aligned}
\phi_2^d = & \epsilon_2 [(1 - \rho_2) H_{2-2} - 1] \Omega_2 + (1 - \rho_2) \epsilon_3 H_{2-3} \Omega_3 \\
& + (1 - \rho_2) \epsilon_7 H_{2-7} \Omega_7
\end{aligned}$$

$$\phi_3^d = (1 - \rho_3) \epsilon_2 H_{3-2} \Omega_2 + \epsilon_3 [(1 - \rho_3) H_{3-3} - 1] \Omega_3 \\ + (1 - \rho_3) \epsilon_7 H_{3-7} \Omega_7$$

$$\phi_7^d = (1 - \rho_7) \epsilon_2 H_{7-2} \Omega_2 + (1 - \rho_7) \epsilon_3 H_{7-3} \Omega_3 \\ + \epsilon_7 [(1 - \rho_7) H_{7-7} - 1] \Omega_7$$

Cavity e:

$$\phi_1^e = \epsilon_1 [(1 - \rho_1) H_{1-1} - 1] \Omega_1 + (1 - \rho_1) \epsilon_2 H_{1-2} \Omega_2 \\ + (1 - \rho_1) \epsilon_3 H_{1-3} \Omega_3$$

$$\phi_2^e = (1 - \rho_2) \epsilon_1 H_{2-1} \Omega_1 + \epsilon_2 [(1 - \rho_2) H_{2-2} - 1] \Omega_2 \\ + (1 - \rho_2) \epsilon_3 H_{2-3} \Omega_3$$

$$\phi_3^e = (1 - \rho_3) \epsilon_1 H_{3-1} \Omega_1 + (1 - \rho_3) \epsilon_2 H_{3-2} \Omega_2 \\ + \epsilon_3 [(1 - \rho_3) H_{3-3} - 1] \Omega_3$$

Cavity f:

$$\phi_3^f = \epsilon_3 [(1 - \rho_3) H_{3-3} - 1] \Omega_3 + (1 - \rho_3) \epsilon_7 H_{3-7} \Omega_7$$

$$\phi_7^f = (1 - \rho_7) \epsilon_3 H_{7-3} \Omega_3 + \epsilon_7 [(1 - \rho_7) H_{7-7} - 1] \Omega_7$$

The energy balance equation (1) is written for each node as the followings:

$$\dot{T}_1 = \frac{1}{M_1} \left\{ \frac{T_2 - T_1}{R_{1-2}} + A_1^c \phi_1^c + A_1^e \phi_1^e + A_1^a \phi_1^a + \dot{Q}_{H_1} \right\}$$

$$\dot{T}_2 = \frac{1}{M_2} \left\{ \frac{T_1 - T_2}{R_{2-1}} + \frac{T_4 - T_2}{R_{2-4}} + \frac{T_3 - T_2}{R_{2-3}} + \frac{T_7 - T_2}{R_{2-7}} + A_2^e \phi_2^e + \dot{Q}_2 \right\}$$

$$\dot{T}_3 = \frac{1}{M_3} \left\{ \frac{T_2 - T_3}{R_{3-2}} + A_3^e \phi_3^e + A_3^d \phi_3^d + A_3^f \phi_3^f + \dot{Q}_{H_3} \right\}$$

$$\dot{T}_4 = \frac{1}{M_4} \left\{ \frac{T_2 - T_4}{R_{4-2}} + \frac{T_5 - T_4}{R_{4-5}} + \frac{T_8 - T_4}{R_{4-8}} + A_4^c \phi_4^c + \dot{Q}_4 \right\}$$

$$\dot{T}_5 = \frac{1}{M_5} \left\{ \frac{T_4 - T_5}{R_{5-4}} + \frac{T_6 - T_5}{R_{5-6}} + \frac{T_{12} - T_5}{R_{5-12}} + A_5^b \phi_5^b \right\}$$

$$\dot{T}_6 = \frac{1}{M_6} \left\{ \frac{T_5 - T_6}{R_{6-5}} + A_6^b \phi_6^b \right\}$$

$$\dot{T}_7 = \frac{1}{M_7} \left\{ \frac{T_2 - T_7}{R_{2-7}} + A_7^d \phi_7^d + A_7^f \phi_7^f + \dot{Q}_7 \right\}$$

$$\dot{T}_8 = \frac{1}{M_8} \left\{ \frac{T_4 - T_8}{R_{8-4}} + \frac{T_5 - T_8}{R_{8-5}} + \frac{T_{10} - T_8}{R_{8-10}} + A_8^a \phi_8^a + A_9^a \phi_9^a \right\}$$

$$\dot{T}_{10} = \frac{1}{M_{10}} \left\{ \frac{T_4 - T_{10}}{R_{10-4}} + \frac{T_8 - T_{10}}{R_{10-8}} + A_{10}^a \phi_{10}^a + A_{10}^b \phi_{10}^b \right\}$$

Where $M = V \gamma C$. In addition there are three more equations which were derived from the electronics model. These equations are already discussed in Section II.

In the above equations, the ϕ s can be splitted as

$$\phi_k = (\phi_k)_{\text{shortwave}} + (\phi_k)_{\text{longwave}}$$

Substituting ϕ_k into the above equations and rearranging the constants yeild a following type of equation

$$\dot{T}_k = \sum_j A_{kj} T_j + \sum_j D_{kj} T_j^4 + B_{Qk} \dot{Q}_k + B_k E_{\text{FOV}}$$

where

$$A_{kj} = \frac{1}{M_k R_{k-j}}$$

D_{kj} = constants including all radiative parameters

$$B_{Qk} = \frac{A_k}{M_k}$$

B_k = constants related with the irradiance at FOV limiter of radiometers.

The above constants are described in the following pages.

$$\{A_{1-1} = -\frac{1}{M_1 R_{1-2}}, \quad A_{1-2} = \frac{1}{M_1 R_{1-2}}$$

$$\left\{ \begin{array}{l} A_{2-1} = \frac{1}{M_2 R_{2-1}}, \quad A_{2-2} = -\frac{1}{M_2} \left(\frac{1}{R_{2-1}} + \frac{1}{R_{2-4}} + \frac{1}{R_{2-3}} + \frac{1}{R_{2-7}} \right) \\ A_{2-3} = \frac{1}{M_2 R_{2-3}}, \quad A_{2-4} = \frac{1}{M_2 R_{2-4}}, \quad A_{2-7} = \frac{1}{M_2 R_{2-7}} \end{array} \right.$$

$$\{A_{3-2} = \frac{1}{M_3 R_{3-2}}, \quad A_{3-3} = \frac{1}{M_3 R_{3-2}}$$

$$\left\{ \begin{array}{l} A_{4-2} = \frac{1}{M_4 R_{4-2}}, \quad A_{4-4} = -\frac{1}{M_4} \left(\frac{1}{R_{4-2}} + \frac{1}{R_{4-5}} + \frac{1}{R_{4-8}} \right) \\ A_{4-5} = \frac{1}{M_4 R_{4-5}}, \quad A_{4-8} = \frac{1}{M_4 R_{4-8}} \end{array} \right.$$

$$\left\{ \begin{array}{l} A_{5-4} = \frac{1}{M_5 R_{5-4}}, \quad A_{5-5} = -\frac{1}{M_5} \left(\frac{1}{R_{5-4}} + \frac{1}{R_{5-6}} + \frac{1}{R_{5-12}} \right) \\ A_{5-6} = \frac{1}{M_5 R_{5-6}}, \quad A_{5-12} = \frac{1}{M_5 R_{5-12}} \end{array} \right.$$

$$\{A_{6-5} = \frac{1}{M_6 R_{6-5}} \quad , \quad A_{6-6} = - \frac{1}{M_6 R_{6-5}}$$

$$\{A_{7-2} = \frac{1}{M_7 R_{7-2}} \quad , \quad A_{7-7} = - \frac{1}{M_7 R_{7-2}}$$

$$\begin{cases} A_{8-4} = \frac{1}{M_8 R_{8-4}} \quad , \quad A_{8-5} = \frac{1}{M_8 R_{8-5}} \quad , \quad A_{8-5} = - \frac{1}{M_8} \left(\frac{1}{R_{8-4}} + \frac{1}{R_{8-5}} + \frac{1}{R_{8-10}} \right) \\ A_{8-10} = \frac{1}{M_8 R_{8-10}} \end{cases}$$

$$\{A_{10-4} = \frac{1}{M_{10} R_{10-4}} \quad , \quad A_{10-8} = \frac{1}{M_{10} R_{10-8}} \quad , \quad A_{10-10} = - \frac{1}{M_{10}} \left(\frac{1}{R_{10-4}} + \frac{1}{R_{10-8}} \right)$$

$$D_{1-1} = \frac{\sigma}{M_1} \{A_1^a \epsilon_1^a [(1 - \rho_1^a) H_{1-1}^a + (1 - \rho_1^a) \tau_{10}^2 G H_{1-10}^a H_{10-10}^b H_{10-1}^a - 1] \\ + A_1^c \epsilon_1^c [(1 - \rho_1^c) H_{1-1}^c - 1] + A_1^e \epsilon_1^e [(1 - \rho_1^e) H_{1-1}^e - 1]\}$$

$$D_{1-2} = \frac{\sigma}{M_1} \{A_1^c (1 - \rho_1^c) \epsilon_2^c H_{1-2}^c + A_1^e (1 - \rho_1^e) \epsilon_2^e H_{1-2}^e\}$$

$$D_{1-3} = \frac{\sigma}{M_1} A_1^e (1 - \rho_1^e) \epsilon_3^e H_{1-3}^e$$

$$D_{1-5} = \frac{\sigma}{M_1} \{A_1^a (1 - \rho_1^a) \epsilon_5^a \tau_{10} G H_{1-10}^a H_{10-5}^b\}$$

$$D_{1-4} = \frac{\sigma}{M_1} A_1^c (1 - \rho_1^c) \epsilon_4^c H_{1-4}^c$$

$$D_{1-6} = \frac{\sigma}{M_1} A_1^a (1 - \rho_1^a) \epsilon_6^a \tau_{10} G H_{1-10}^a H_{10-6}^b$$

$$D_{1-8} = \frac{\sigma}{M_1} A_1^a (1 - \rho_1^a) \{ (\epsilon_8^a H_{1-8}^a + \epsilon_9^a H_{1-9}^a) + \tau_{10}^2 G H_{1-10}^a H_{10-10}^b \\ (\epsilon_8^a H_{10-8}^a + \epsilon_9^a H_{10-9}^a) \}$$

$$D_{1-10} = \frac{\sigma}{M_1} A_1^a \epsilon_{10}^a H_{1-10}^a [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)]$$

$$B_1 = \frac{1}{M_1} A_1^a (1 - \rho_1^a) \tau_{10} \tau_{11} G H_{1-10}^a H_{10-11}^b$$

$$D_{2-1} = \frac{\sigma}{M_2} [A_2^c (1 - \rho_2^c) \epsilon_1^c H_{2-1}^c + A_2^e (1 - \rho_2^e) \epsilon_1^e H_{2-1}^e]$$

$$D_{2-2} = \frac{\sigma}{M_2} \{ A_2^c [(1 - \rho_2^c) H_{2-2}^c - 1] \epsilon_2^c + A_2^d \epsilon_2^d [(1 - \rho_2^d) H_{2-2}^d - 1] \\ + A_2^e \epsilon_2^e [(1 - \rho_2^e) H_{2-2}^e - 1] \}$$

$$D_{2-3} = \frac{\sigma}{M_2} \{ A_2^d (1 - \rho_2^d) \epsilon_3^d H_{2-3}^d + A_2^e (1 - \rho_2^e) \epsilon_3^e H_{2-3}^e \}$$

$$D_{2-4} = \frac{\sigma}{M_2} \{ A_2^c (1 - \rho_2^c) \epsilon_4^c H_{2-4}^c \}$$

$$D_{2-7} = \frac{\sigma}{M_2} A_2^d (1 - \rho_2^d) \epsilon_7^d H_{2-7}^d$$

$$D_{3-1} = \frac{\sigma}{M_3} A_3^e (1 - \rho_3^e) \epsilon_1^e H_{3-1}^e$$

$$D_{3-2} = \frac{\sigma}{M_3} \{ A_3^d (1 - \rho_3^d) \epsilon_2^d H_{3-2}^d + A_3^e (1 - \rho_3^e) \epsilon_2^e H_{3-2}^e \}$$

$$D_{3-3} = \frac{\sigma}{M_3} \{ A_3^d \epsilon_3^d [(1 - \rho_3^d) H_{3-3}^d - 1] + A_3^e \epsilon_3^e [(1 - \rho_3^e) H_{3-3}^e - 1] \\ + A_3^f \epsilon_3^f [(1 - \rho_3^f) H_{3-3}^f - 1] \}$$

$$D_{3-7} = \frac{\sigma}{M_3} \{A_3^d \epsilon_7^d (1 - \rho_3^d) H_{3-7}^d + A_3^f (1 - \rho_3^f) \epsilon_7^f H_{3-7}^f\}$$

$$D_{4-1} = \frac{\sigma}{M_4} A_4^c (1 - \rho_4^c) \epsilon_1^c H_{4-1}^c$$

$$D_{4-2} = \frac{\sigma}{M_4} A_4^c (1 - \rho_4^c) \epsilon_2^c H_{4-2}^c$$

$$D_{4-4} = \frac{\sigma}{M_4} A_4^c \epsilon_4^c [(1 - \rho_4^c) H_{4-4}^c - 1]$$

$$D_{5-1} = \frac{\sigma}{M_5} \{A_5^b (1 - \rho_5^b) \epsilon_1 \tau_{10} G H_{5-10}^b H_{10-1}^a\}$$

$$D_{5-5} = \frac{\sigma}{M_5} \{A_5^b [\epsilon_5^b [(1 - \rho_5^b) H_{5-5}^b - 1] + (1 - \rho_5^b) \tau_{10}^2 G H_{5-10}^b H_{10-5}^b H_{10-10}^a]\}$$

$$D_{5-6} = \frac{\sigma}{M_5} A_5^b (1 - \rho_5^b) \epsilon_5^b [H_{5-6}^b + \tau_{10}^2 G H_{5-10}^b H_{10-6}^b H_{10-10}^a]$$

$$D_{5-8} = \frac{\sigma}{M_5} A_5^b (1 - \rho_5^b) \tau_{10} G H_{5-10}^b (\epsilon_8^a H_{10-8}^a + \epsilon_9 H_{10-9})$$

$$D_{5-10} = \frac{\sigma}{M_5} A_5^b (1 - \rho_5^b) \epsilon_{10}^b H_{5-10}^b [1 + \tau_{10} G H_{10-10}^a (1 + \tau_{10} H_{10-10}^b)]$$

$$B_5 = \frac{1}{M_5} A_5^b (1 - \rho_5^b) \tau_{11} [\tau_{10}^2 G H_{5-10}^b H_{10-10}^a H_{10-11} + H_{5-11}]$$

$$D_{6-1} = \frac{\sigma}{M_6} A_6^b \epsilon_1^a (1 - \rho_6^b) \tau_{10} G H_{5-10}^b H_{10-1}^a$$

$$D_{6-5} = \frac{\sigma}{M_6} A_6^b (1 - \rho_6^b) \epsilon_6^b [H_{6-5}^b + \tau_{10}^2 G H_{6-10}^b H_{10-5}^b H_{10-10}^a]$$

$$D_{6-6} = \frac{\sigma}{M_6} A_6^b \epsilon_6^b \{[(1 - \rho_6^b) H_{6-6}^b - 1] + (1 - \rho_6^b) \tau_{10}^2 G H_{10-5}^b H_{10-10}^a\}$$

$$D_{6-8} = \frac{\sigma}{M_6} A_6^b (1 - \rho_6^b) \tau_{10} G H_{6-10}^b (\epsilon_8^a H_{10-8}^a + \epsilon_9^a H_{10-9}^a)$$

$$D_{6-10} = \frac{\sigma}{M_6} A_6^b (1 - \rho_6^b) \epsilon_{10}^b H_{6-10}^b [1 + \tau_{10} G H_{10-10}^a (1 - \tau_{10} H_{10-10}^b)]$$

$$B_6 = \frac{1}{M_6} A_6^b (1 - \rho_6^b) \tau_{11} [H_{6-11} + \tau_{10}^2 G H_{6-10}^b H_{10-10}^a H_{10-10}^b]$$

$$D_{7-2} = \frac{\sigma}{M_7} A_7^d (1 - \rho_7^d) \epsilon_2^d H_{7-2}^d$$

$$D_{7-3} = \frac{\sigma}{M_7} \{A_7^d (1 - \rho_7^d) \epsilon_3^d H_{7-3}^d + A_7^f (1 - \rho_7^f) \epsilon_3^f H_{7-3}^f\}$$

$$D_{7-7} = \frac{\sigma}{M_7} \{A_7^d \epsilon_7^d [(1 - \rho_7^d) H_{7-7}^d - 1] + A_7^f \epsilon_7^f [(1 - \rho_7^f) H_{7-7}^f - 1]\}$$

$$D_{8-1} = \frac{\sigma}{M_8} \left\{ \frac{A_8^a}{A_{(8+9)}^a} (1 - \rho_8^a) [\epsilon_1^a H_{8-1}^a + \epsilon_1^a \tau_{10}^2 G H_{8-10}^a H_{10-10}^b H_{10-1}^a] \right. \\ \left. + \frac{A_9^a}{A_{(8+9)}^a} (1 - \rho_9^a) \epsilon_1^a [H_{9-1}^a + \tau_{10}^2 G H_{9-10}^a H_{10-10}^b H_{10-1}^a] \right\} A_{(8+9)}$$

$$D_{8-5} = \frac{\sigma}{M_8} \{A_8^a (1 - \rho_8^a) \epsilon_5^b \tau_{10} G H_{8-10}^a H_{10-5}^b + A_9^a (1 - \rho_9^a) \epsilon_5^b \tau_{10} G H_{9-10}^a H_{10-5}^b\}$$

$$D_{8-6} = \frac{\sigma}{M_8} \{A_8^a (1 - \rho_8^a) \epsilon_6^b \tau_{10} G H_{8-10}^a H_{10-6}^b + A_9^a (1 - \rho_9^a) \epsilon_6^b \tau_{10} G H_{9-10}^a H_{10-6}^b\}$$

$$D_{8-8} = \frac{\sigma}{M_8} \{A_8^a [\epsilon_8^a ((1 - \rho_8^a) H_{8-8}^a - 1) + (1 - \rho_8^a) \tau_{10}^2 G H_{8-10}^a H_{10-10}^b (H_{10-8}^a \\ \epsilon_8^a + H_{10-9}^a \epsilon_9^a) + (1 - \rho_8^a) \epsilon_9^a H_{8-9}^a] + A_9^a [(1 - \rho_9^a) \epsilon_9^a H_{9-8}^a + \epsilon_9^a \\ ((1 - \rho_9^a) H_{9-9}^a - 1) + (1 - \rho_9^a) \tau_{10}^2 G H_{9-10}^a H_{10-10}^b (H_{10-8}^a \epsilon_8^a + H_{10-9}^a \epsilon_9^a)]\}$$

$$D_{8-10} = \frac{\sigma}{M_8} \{A_8 (1 - \rho_8^a) \epsilon_{10}^a H_{8-10}^a [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)] \\ + A_9 (1 - \rho_9^a) \epsilon_{10}^a H_{9-10}^a [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)]\}$$

$$B_8 = \frac{1}{M_8} \{A_8 (1 - \rho_8^a) \tau_{10} \tau_{11} G H_{8-10}^a H_{10-11}^b + A_9 (1 - \rho_9^a) \tau_{10} \tau_{11} G H_{9-10}^a H_{10-11}^b\}$$

$$D_{10-1} = \frac{\sigma}{M_{10}} \{A_{10}^a \epsilon_1^a H_{10-1}^a [(1 - \rho_{10}^a) + \tau_{10}^2 G ((1 - \rho_{10}^a) H_{10-10}^a - 1) H_{10-10}^b] \\ + A_{10}^b \epsilon_1^a \tau_{10} G [(1 - \rho_{10}^b) H_{10-10}^b - 1] H_{10-1}^a\}$$

$$D_{10-5} = \frac{\sigma}{M_{10}} \{A_{10}^a \epsilon_5^b \tau_{10} G [(1 - \rho_{10}^a) H_{10-10}^a - 1] H_{10-5}^b \\ + A_{10}^b \epsilon_5^b H_{10-5}^b [(1 - \rho_{10}^b) + \tau_{10}^2 G (1 - \rho_{10}^b) H_{10-10}^b - 1] H_{10-10}^a\}$$

$$D_{10-6} = \frac{\sigma}{M_{10}} \{A_{10}^a \epsilon_6^b \tau_{10} G [(1 - \rho_{10}^a) H_{10-10}^a - 1] H_{10-6}^b \\ + A_{10}^b \epsilon_6^b H_{10-6}^b [(1 - \rho_{10}^b) + \tau_{10}^2 G ((1 - \rho_{10}^b) H_{10-10}^b - 1) H_{10-10}^a]\}$$

$$D_{10-8} = \frac{\sigma}{M_{10}} \{A_{10}^a (\epsilon_8^a H_{10-8}^a + \epsilon_9^a H_{10-9}^a) [(1 - \rho_{10}^a) + \tau_{10}^2 G ((1 - \rho_{10}^a) H_{10-10}^a - 1) \\ H_{10-10}^b] + A_{10}^b \tau_{10} G [(1 - \rho_{10}^b) H_{10-10}^b - 1] (\epsilon_8^a H_{10-8}^a + \epsilon_9^a H_{10-9}^a)\}$$

$$D_{10-10} = \frac{\sigma}{M_{10}} \{A_{10}^a \epsilon_{10}^a [(1 - \rho_{10}^a) H_{10-10}^a - 1] [1 + \tau_{10} G H_{10-10}^b (1 + \tau_{10} H_{10-10}^a)] \\ + A_{10}^b \epsilon_{10}^b [(1 - \rho_{10}^b) H_{10-10}^b - 1] [1 + \tau_{10} G H_{10-10}^a (1 + \tau_{10} H_{10-10}^b)]\}$$

$$\begin{aligned}
B_{10} = & \frac{1}{M_{10}} \{ A_{10}^a \tau_{10} \tau_{11} G[(1 - \rho_{10}^a) H_{10-10}^a - 1] H_{10-11}^b \\
& + A_{10}^b \tau_{10} \tau_{11} G[(1 - \rho_{10}^b) H_{10-10}^b - 1] H_{10-10}^a H_{10-11}^b \\
& + A_{10}^b (1 - \rho_{10}^b) \tau_{11} H_{10-11}^b \}
\end{aligned}$$

$$\begin{aligned}
B_{10} = & \frac{1}{M_{10}} \tau_1 H_{10-11}^b \{ A_{10}^a \tau_{10} G[(1 - \rho_{10}^a) H_{10-10}^a - 1] \\
& + A_{10}^b (\tau_{10}^2 G H_{10-10}^a [(1 - \rho_{10}^b) H_{10-10}^b - 1] + 1 - \rho_{10}^b) \}
\end{aligned}$$

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APPENDIX D

ERBE SCANNER THERMAL MODEL EQUATIONS

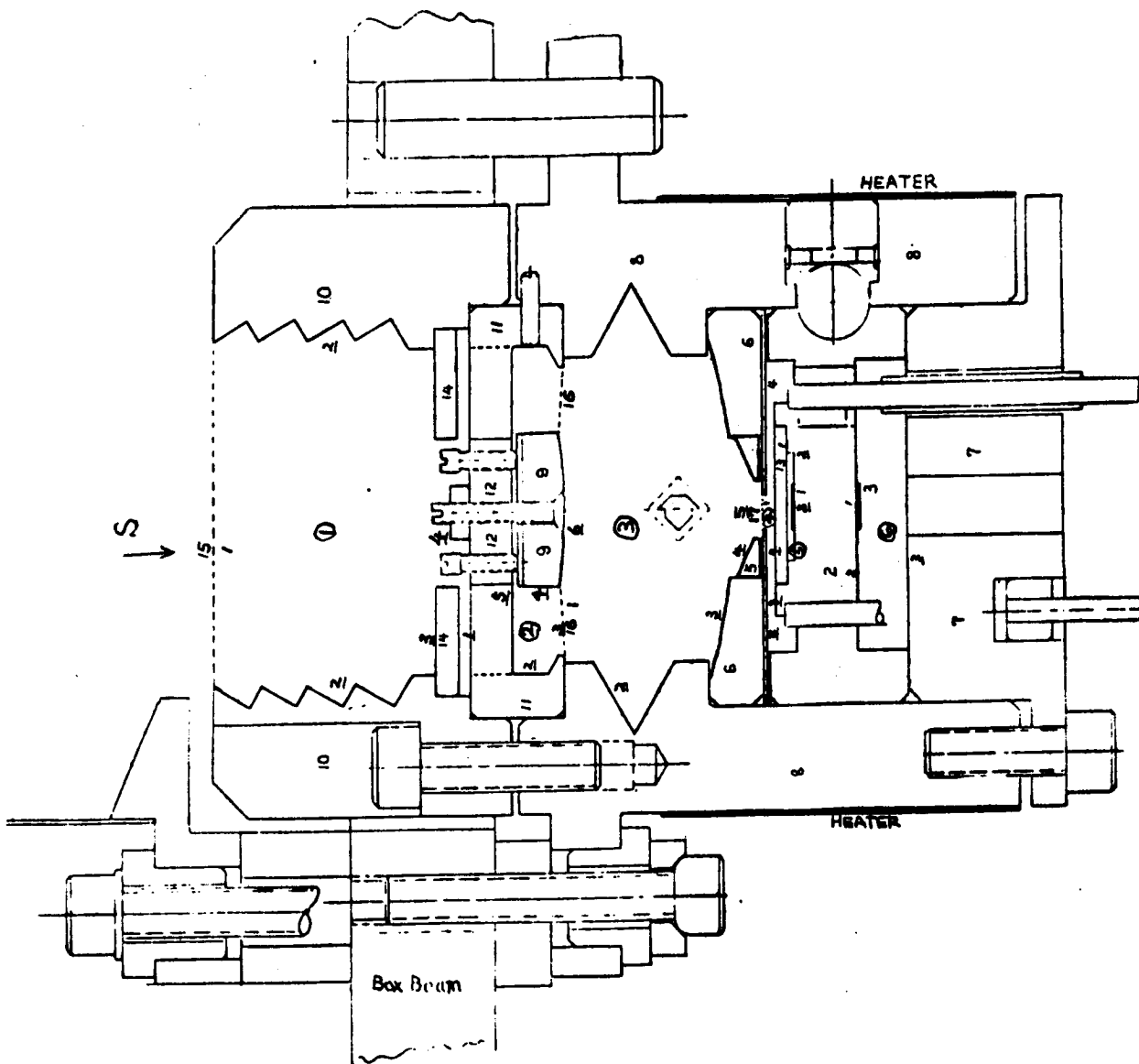
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SCANNER MODEL

The scanner model was developed by the same methods which were used for the non-scanner model development. The Equations (1) through (7) in the Appendix C are used to generate the constants of the thermal model Equation (24) in the Section 2.1.3.

To define the coefficients in the thermal model, the enclosure coupling technique was used as another alternative approach. The following pages contain those coefficients defined by using the enclosure coupling method.

The attached figure shows the scanner model which has node description and enclosure descriptions. The enclosure has the number which is circled around. The numbers describing enclosure surfaces are underlined. In the illustration of the coefficients, the word that explains the procedure of derivation was omitted because they were already shown in Appendix C.



1. Active Flake
2. Heat Sink & Detector Assy.
3. Compensating Flake
4. Precision Aperture
5. Insert, Primary
6. Primary Mirror
7. Cap Assembly
8. Housing & Heater
9. Secondary Mirror
10. FOV Limiter
11. Mount, Secondary (A)
12. Mount, Secondary (B)
13. LW or SW Filter
14. SW Filter
15. Fictitious Surface at FOV
16. Fictitious Surface at Secondary
17. Fictitious Surface at Aperture

SCANNER MODEL

$$\begin{cases} A_{1,1} = -\frac{1}{M_1 R_{1-2}} & , & A_{1,2} = \frac{1}{M_1 R_{1-2}} \\ A_{1,3} = A_{1,4} = A_{1,5} = A_{1,6} = A_{1,7} = A_{1,8} = A_{1,9} = A_{1,10} = A_{1,11} = A_{1,12} = \\ A_{1,13} = A_{1,14} = 0 \end{cases}$$

$$\begin{cases} A_{2,1} = \frac{1}{M_1 R_{2-1}} & , & A_{2,2} = -\left(\frac{1}{M_2}\right)\left(\frac{1}{R_{2-1}} + \frac{1}{R_{2-3}} + \frac{1}{R_{2-4}} + \frac{1}{R_{2-7}} + \frac{1}{R_{2-8}} + \frac{1}{R_{2-13}}\right) \\ A_{2,3} = \frac{1}{M_2 R_{2-3}} & , & A_{2,4} = \frac{1}{M_2 R_{2-4}} \\ A_{2,5} = A_{2,6} = 0 \\ A_{2,7} = \frac{1}{M_2 R_{2-7}} & , & A_{2,8} = \frac{1}{M_2 R_{2-8}} & , & A_{2,13} = \frac{1}{M_2 R_{2-13}} \\ A_{2,9} = A_{2,10} = A_{2,11} = A_{2,12} = A_{2,14} = 0 \end{cases}$$

$$\begin{cases} A_{3,1} = 0 & , & A_{3,2} = +\frac{1}{M_3 R_{3-2}} & , & A_{3,3} = -\frac{1}{M_3 R_{3-2}} \\ A_{3,4} = A_{3,5} = A_{3,6} = A_{3,7} = A_{3,8} = A_{3,9} = A_{3,10} = A_{3,11} = A_{3,12} = A_{3,13} = A_{3,14} = 0 \end{cases}$$

$$\begin{cases} A_{4,1} = 0 & , & A_{4,2} = \frac{1}{M_4 R_{4-2}} & , & A_{4,3} = 0 \\ A_{4,4} = -\left(\frac{1}{M_4}\right)\left(\frac{1}{R_{4-2}} + \frac{1}{R_{4-6}} + \frac{1}{R_{4-8}}\right) & , & A_{4,5} = 0 \\ A_{4,6} = \frac{1}{M_4 R_{4-6}} & , & A_{4,7} = 0 & , & A_{4,8} = \frac{1}{M_4 R_{4-8}} \\ A_{4,9} = A_{4,10} = A_{4,11} = A_{4,12} = A_{4,13} = A_{4,14} = 0 \end{cases}$$

$$\left\{ \begin{array}{l} A_{5,1} = A_{5,2} = A_{5,3} = A_{5,4} = 0 \\ A_{5,5} = -\frac{1}{M_5 R_{5-6}} \quad , \quad A_{5,6} = \frac{1}{M_5 R_{5-6}} \\ A_{5,7} = A_{5,8} = A_{5,9} = A_{5,10} = A_{5,11} = A_{5,12} = A_{5,13} = A_{5,14} = 0 \end{array} \right.$$

$$\left\{ \begin{array}{l} A_{6,1} = A_{6,2} = A_{6,3} = 0 \\ A_{6,4} = \frac{1}{M_6 R_{6-4}} \quad , \quad A_{6,5} = \frac{1}{M_6 R_{6-5}} \\ A_{6,6} = -\left(\frac{1}{M_6}\right)\left(\frac{1}{R_{6-4}} + \frac{1}{R_{6-5}} + \frac{1}{R_{6-8}}\right) \\ A_{6,7} = 0 \quad , \quad A_{6,8} = \frac{1}{M_6 R_{6-8}} \\ A_{6,9} = A_{6,10} = A_{6,11} = A_{6,12} = A_{6,13} = A_{6,14} = 0 \end{array} \right.$$

$$\left\{ \begin{array}{l} A_{7,1} = 0 \quad , \quad A_{7,2} = \frac{1}{M_7 R_{7-2}} \\ A_{7,3} = A_{7,4} = A_{7,5} = A_{7,6} = 0 \\ A_{7,7} = -\left(\frac{1}{M_7}\right)\left(\frac{1}{R_{7-2}} + \frac{1}{R_{7-8}}\right) \quad , \quad A_{7,8} = \frac{1}{M_7 R_{7-8}} \\ A_{7,9} = A_{7,10} = A_{7,11} = A_{7,12} = A_{7,13} = A_{7,14} = 0 \end{array} \right.$$

$$\left\{ \begin{array}{l} A_{8,1} = 0, \quad A_{8,2} = \frac{1}{M_8 R_{8-2}}, \quad A_{8,3} = 0, \quad A_{8,4} = \frac{1}{M_8 R_{8-4}} \\ A_{8,5} = 0, \quad A_{8,6} = \frac{1}{M_8 R_{8-6}}, \quad A_{8,7} = \frac{1}{M_8 R_{8-7}} \\ A_{8,8} = - \left(\frac{1}{M_8} \right) \left(\frac{1}{R_{8-2}} + \frac{1}{R_{8-4}} + \frac{1}{R_{8-6}} + \frac{1}{R_{8-7}} + \frac{1}{R_{8-10}} + \frac{1}{R_{8-11}} \right) \\ A_{8,9} = 0, \quad A_{8,10} = \frac{1}{M_8 R_{8-10}}, \quad A_{8,11} = \frac{1}{M_8 R_{8-11}} \\ A_{8,12} = A_{8,13} = A_{8,14} = 0 \end{array} \right.$$

$$\left\{ \begin{array}{l} A_{9,1} = A_{9,2} = A_{9,3} = A_{9,4} = A_{9,5} = A_{9,6} = A_{9,7} = A_{9,8} = 0 \\ A_{9,9} = - \frac{1}{M_9 R_{9-12}}, \quad A_{9,10} = 0, \quad A_{9,11} = 0, \quad A_{9,12} = \frac{1}{M_9 R_{9-12}} \\ A_{9,13} = A_{9,14} = 0 \end{array} \right.$$

$$\left\{ \begin{array}{l} A_{10,1} = A_{10,2} = A_{10,3} = A_{10,4} = A_{10,5} = A_{10,6} = A_{10,7} = 0 \\ A_{10,8} = \frac{1}{M_{10} R_{10-8}}, \quad A_{10-9} = 0, \\ A_{10-10} = - \left(\frac{1}{M_{10}} \right) \left(\frac{1}{R_{10-8}} + \frac{1}{R_{10-11}} + \frac{1}{R_{10-14}} \right) \\ A_{10-11} = \frac{1}{M_{10} R_{10-11}}, \quad A_{10-12} = 0, \quad A_{10,13} = 0, \quad A_{10,14} = \frac{1}{M_{10} R_{10-14}} \end{array} \right.$$

$$\begin{cases} A_{11,1} = A_{11,2} = A_{11,3} = A_{11,4} = A_{11,5} = A_{11,6} = A_{11,7} = 0 \\ A_{11,8} = \frac{1}{M_{11}R_{11-8}} , \quad A_{11,9} = 0 , \quad A_{11,10} = \frac{1}{M_{11}R_{11-10}} , \quad A_{11,12} = 0 \\ A_{11,13} = A_{11,14} = 0 , \quad A_{11,11} = - \left(\frac{1}{M_{11}} \right) \left(\frac{1}{R_{11-8}} + \frac{1}{R_{11-10}} \right) \end{cases}$$

$$\begin{cases} A_{12,1} = A_{12,2} = A_{12,3} = A_{12,4} = A_{12,5} = A_{12,6} = A_{12,7} = A_{12,8} = 0 \\ A_{12,9} = \frac{1}{M_{12}R_{12-9}} , \quad A_{12,10} = A_{12,11} = 0 , \quad A_{12,12} = - \frac{1}{M_{12}R_{12-9}} \\ A_{12,13} = 0 , \quad A_{12,14} = 0 \end{cases}$$

$$\begin{cases} A_{13,1} = 0 , \quad A_{13,2} = \frac{1}{M_{13}R_{13-2}} , \quad A_{13,3} = A_{13,4} = A_{13,5} = A_{13,6} = 0 \\ A_{13,7} = A_{13,8} = A_{13,9} = A_{13,10} = A_{13,11} = A_{13,12} = 0 \\ A_{13,13} = - \frac{1}{M_{13}R_{13-2}} , \quad A_{13,14} = 0 \end{cases}$$

$$\begin{cases} A_{14-1} = A_{14,2} = A_{14,3} = A_{14,4} = A_{14,5} = A_{14,6} = A_{14,7} = A_{14,8} = A_{14,9} = 0 \\ A_{14,10} = \frac{1}{M_{14}R_{14-10}} , \quad A_{14,11} = 0 , \quad A_{14,12} = 0 , \quad A_{14,13} = 0 \\ A_{14m14} = - \frac{1}{M_{14}R_{14-10}} \end{cases}$$

For the total channel, the filter glasses are removed. Accordingly, the $(\frac{1}{R_{i-j}})$ related with node number of filters have to set zero.

$$B_Q = [0, 0, 0, 0, 0, 0, 0, 0, \frac{1}{M_8}, 0, 0, 0, 0, 0, 0]$$

Scanner:

Let

$$(H_{kji})_A = (\sum_j L_{k-j} P_{j-i})_A,$$

where subscript A denotes the enclosure A.

Then let

$$G_1 = [1 - \tau_{13}^2 (H_{13j13})_D (H_{13m13})_F]^{-1}$$

$$S_1 = (H_{17m17})_F + \tau_{13}^2 G_1 (H_{17m13})_F (H_{13j13})_D (H_{13m17})_F$$

$$G_2 = [1 - \tau_{17}^2 S_1 (H_{17j17})_C]^{-1}$$

$$S_2 = (H_{16j16})_C + \tau_{17}^2 S_1 G_2 (H_{17j16})_C (H_{16j17})_C$$

$$G_3 = [1 - \tau_{16}^2 S_2 (H_{16m16})_B]^{-1}$$

$$S_3 = (H_{14m14})_B + \tau_{16}^2 S_2 G_3 (H_{14m16})_B (H_{16m14})_B$$

$$G_4 = [1 - \tau_{14}^2 S_3 (H_{14j14})_A]^{-1}$$

$$X_1 = F_{1,1}\Omega_1 + F_{1,2}\Omega_2 + F_{1,4}\Omega_4 + F_{1,5}\Omega_5 + F_{1,6}\Omega_6 + F_{1,8}\Omega_8 + F_{1,9}\Omega_9 \\ + F_{1,13}\Omega_{13}$$

where

$$F_{1,1} = \tau_{13} \epsilon_1 G_1 G_2 (H_{17m13})_F (H_{13j1})_D$$

$$F_{1,2} = \epsilon_2 G_2 \{ (H_{17m2})_F + \tau_{13} G_1 (H_{17m13})_F (H_{13j2})_D + \tau_{13}^2 G_1 (H_{17m13})_F \\ (H_{13j13})_D (H_{13m2})_F \}$$

$$F_{1,4} = \epsilon_4 G_2 \{ (H_{17m4})_F + \tau_{13}^2 G_1 (H_{17m13})_F (H_{13m13})_D (H_{13m4})_F \}$$

$$F_{1,5} = \tau_{17} \epsilon_5 S_1 G_2 (H_{17j5})_C$$

$$F_{1,6} = \tau_{17} \epsilon_6 S_1 G_2 (H_{17j6})_C$$

$$F_{1,8} = \tau_{17} \epsilon_8 S_1 G_2 (H_{17j8})_C$$

$$F_{1,9} = \tau_{17} \epsilon_9 S_1 G_2 (H_{17j9})_C$$

$$F_{1,13} = \epsilon_{13} G_2 \{ (H_{17m13})_F + \tau_{13} G_1 (H_{17m13})_F (H_{13j13})_D + \tau_{13}^2 G_1 (H_{17m13})_F \\ (H_{13j13})_D (H_{13m13})_F \}$$

$$X_2 = F_{2,1}\Omega_1 + F_{2,2}\Omega_2 + F_{2,4}\Omega_4 + F_{2,5}\Omega_5 + F_{2,6}\Omega_6 + F_{2,8}\Omega_8 + F_{2,9}\Omega_9 + F_{2,11}\Omega_{11} \\ + F_{2,12}\Omega_{12} + F_{2,13}\Omega_{13} + F_{2,14}\Omega_{14}$$

where

$$F_{2,1} = \tau_{13} \tau_{17} G_1 G_2 G_3 (H_{16j17})_C (H_{17m13})_F (H_{13j1})_D$$

$$F_{2,2} = \tau_{17} \epsilon_2 G_2 G_3 (H_{16j17})_C \{ (H_{17m2})_F + \tau_{13} G_1 (H_{17m13})_F (H_{13j2})_D + \tau_{13}^2 G_1 (H_{17m13})_F (H_{13j13})_D (H_{13m2})_F \}$$

$$F_{2,4} = \tau_{17} \epsilon_4 G_2 G_3 (H_{16j17})_C \{ (H_{17m4})_F + \tau_{13}^2 G_1 (H_{17m13})_F (H_{13m13})_D (H_{13m4})_F \}$$

$$F_{2,5} = \epsilon_5 G_3 \{ (H_{16j5})_C + \tau_{17}^2 S_1 G_2 (H_{16j17})_C (H_{17j5})_C \}$$

$$F_{2,6} = \epsilon_6 G_3 \{ (H_{16j6})_C + \tau_{17}^2 S_1 G_2 (H_{16j17})_C (H_{17j6})_C \}$$

$$F_{2,8} = \epsilon_8 G_3 \{ (H_{16j8})_C + \tau_{17}^2 S_1 G_2 (H_{16j17})_C (H_{17j8})_C \}$$

$$F_{2,9} = \epsilon_9 G_3 \{ (H_{16j9})_C + \tau_{16} S_2 (H_{16m9})_B + \tau_{17}^2 S_1 G_2 (H_{16j17})_C (H_{17j9})_C \}$$

$$F_{2,11} = \tau_{16} \epsilon_{11} S_2 G_3 (H_{16m11})_B$$

$$F_{2,12} = \tau_{16} \epsilon_{12} S_2 G_3 (H_{16m12})_B$$

$$F_{2,13} = \tau_{17} \epsilon_{13} G_2 G_3 (H_{16j17})_C (H_{17m13})_F \{ 1 + \tau_{13} G_1 (H_{13j13})_D + \tau_{13}^2 G_1 (H_{13j13})_D (H_{13m13})_F \}$$

$$F_{2,14} = \tau_{16} \epsilon_{14} S_2 G_3 (H_{16m14})_B$$

$$\begin{aligned}\check{E}_{14} = & U_{1,1}\Omega_1 + U_{1,2}\Omega_2 + U_{1,4}\Omega_3 + U_{1,5}\Omega_5 + U_{1,6}\Omega_6 + U_{1,8}\Omega_8 + U_{1,9}\Omega_9 + U_{1,10}\Omega_{10} \\ & + U_{1,11}\Omega_{11} + U_{1,12}\Omega_{12} + U_{1,13}\Omega_{13} + U_{1,14}\Omega_{14} + U_{1,15}E\end{aligned}$$

$$U_{1,1} = \tau_{16} G_4 (H_{14m16})_B F_{2,1}$$

$$U_{1,2} = \tau_{16} G_4 (H_{14m16})_B F_{2,2}$$

$$U_{1,4} = \tau_{16} G_4 (H_{14m16})_B F_{2,4}$$

$$U_{1,5} = \tau_{16} G_4 (H_{14m16})_B F_{2,5}$$

$$U_{1,6} = \tau_{16} G_4 (H_{14m16})_B F_{2,6}$$

$$U_{1,8} = \tau_{16} G_4 (H_{14m16})_B F_{2,8}$$

$$U_{1,9} = G_4 \{ \tau_{16} (H_{14m16})_B F_{2,9} + \epsilon_9 (H_{14m9})_B \}$$

$$U_{1,10} = \tau_{10} S_3 G_4 \epsilon_{10} (H_{14j10})_A$$

$$U_{1,11} = G_4 \{ \epsilon_{11} (H_{14m11})_B + \tau_{16} (H_{14m16})_B F_{2,11} \}$$

$$U_{1,12} = G_4 \{ \epsilon_{12} (H_{14m12})_B + \tau_{14} \epsilon_{12} S_3 (H_{14j12})_A + \tau_{16} (H_{14m16})_B F_{2,12} \}$$

$$U_{1,13} = \tau_{16} G_4 (H_{14m16})_B F_{2,13}$$

$$U_{1,14} = G_4 \{ \epsilon_{14} (H_{14m14})_B + \tau_{14} \epsilon_{14} S_3 (H_{14j14})_A + \tau_{16} (H_{14m16})_B F_{2,14} \}$$

$$U_{1,15} = \tau_{15} G_4 (H_{14j15})_A$$

$$\begin{aligned}\tilde{E}_{14} = & U_{2,1}\Omega_1 + U_{2,2}\Omega_2 + U_{2,4}\Omega_4 + U_{2,5}\Omega_5 + U_{2,6}\Omega_6 + U_{2,8}\Omega_8 + U_{2,9}\Omega_9 + U_{2,10}\Omega_{10} \\ & + U_{2,11}\Omega_{11} + U_{2,12}\Omega_{12} + U_{2,13}\Omega_{13} + U_{2,14}\Omega_{14} + U_{2,15}E\end{aligned}$$

$$U_{2,1} = \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,1}$$

$$U_{2,2} = \tau_{14} \tau_{16} G_4 (H_{14h14})_A (H_{14m16})_B F_{2,2}$$

$$U_{2,4} = \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,4}$$

$$U_{2,5} = \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,5}$$

$$U_{2,6} = \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,6}$$

$$U_{2,8} = \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,8}$$

$$U_{2,9} = G_4 (H_{14j14})_A \{ \tau_{14} \tau_{16} (H_{14m16})_B F_{2,9} + \tau_{14} \epsilon_9 (H_{14m9})_B \}$$

$$U_{2,10} = \epsilon_{10} (H_{14j10})_A \{ 1 + \tau_{14}^2 S_3 G_4 (H_{14j14})_A \}$$

$$U_{2,11} = G_4 (H_{14j14})_A \{ \tau_{14} \tau_{16} (H_{14m16})_B F_{2,11} + \tau_{14} \epsilon_{11} (H_{14m11})_B \}$$

$$\begin{aligned}U_{2,12} = & \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,12} + \epsilon_{12} (H_{14j12})_A [1 + \tau_{14}^2 S_3 G_4 \\ & (H_{14j14})_A] + \tau_{14} \epsilon_{12} G_4 (H_{14j14})_A (H_{14m12})_B\end{aligned}$$

$$U_{2,13} = \tau_{14} \tau_{16} G_4 (H_{14j14})_A (H_{14m16})_B F_{2,13}$$

$$U_{2,14} = (H_{14j14})_A \{ \tau_{14} \epsilon_{14} G_4 (H_{14m14})_B + \tau_{14} \tau_{16} G_4 (H_{14m16})_B F_{2,14} + \epsilon_{14} [1 + \tau_{14}^2 S_3 G_4 (H_{14j14})_A] \}$$

$$U_{2,15} = \tau_{15} (H_{14j15})_A \{ 1 + \tau_{14} G_4 (H_{14j14})_A \}$$

$$\hat{E}_{16} = U_{3,1}\Omega_1 + U_{3,2}\Omega_2 + U_{3,4}\Omega_4 + U_{3,5}\Omega_5 + U_{3,6}\Omega_6 + U_{3,8}\Omega_8 + U_{3,9}\Omega_9 + U_{3,10}\Omega_{10} \\ + U_{3,11}\Omega_{11} + U_{3,12}\Omega_{12} + U_{3,13}\Omega_{13} + U_{3,14}\Omega_{14} + U_{3,15}^E$$

$$U_{3,1} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,1} + \tau_{13} \tau_{17} G_1 G_2 G_3 \epsilon_1 (H_{16j17})_C (H_{17m13})_F \\ (H_{13j1})_D$$

$$U_{3,2} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,2} + \tau_{13} \tau_{17} \epsilon_2 G_1 G_2 G_3 (H_{16j17})_C (H_{17m13})_F \\ (H_{13j13})_D + \tau_{17} \epsilon_2 G_2 G_3 (H_{16j17})_C (H_{17m2})_F + \tau_{13}^2 \tau_{17} \epsilon_2 G_1 G_2 G_3 \\ (H_{16j17})_C (H_{17m13})_F (H_{13j13})_D * (H_{13m2})_F$$

$$U_{3,4} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,4} + \tau_{17} \epsilon_4 G_2 G_3 (H_{16j17})_C (H_{17m4})_F \\ + \tau_{13}^2 \tau_{17} \epsilon_4 G_1 G_2 G_3 (H_{16j17})_C (H_{17n13})_F (H_{13j13})_D (H_{13m4})_F$$

$$U_{3,5} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,5} + G_3 \epsilon_5 (H_{16j5})_C + \tau_{17}^2 \epsilon_5 S_1 G_2 (H_{16j17})_C \\ (H_{17j5})_C$$

$$U_{3,6} = \tau_{14} \tau_{16} G_3 (H_{16m14})_B U_{2,6} + \epsilon_6 G_3 (H_{16j6})_C + \tau_{17}^2 \epsilon_6 S_1 G_2 G_3 (H_{16j17})_C \\ (H_{17j6})_C$$

$$U_{3,8} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,8} + \epsilon_8 G_3 (H_{16j8})_C + \tau_{17}^2 \epsilon_8 S_1 G_2 G_3 (H_{16j17})_C \\ (H_{17j8})_C$$

$$U_{3,9} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,9} + \epsilon_9 G_3 (H_{16j9})_C + \tau_{17}^2 \epsilon_8 S_1 G_2 G_3 (H_{16j17})_C \\ (H_{17j9})_C + \tau_{16} \epsilon_9 S_2 G_3 (H_{17j9})_C$$

$$U_{3,10} = \tau_{14} \tau_{10} S_2 G_3 (H_{16m14})_B U_{2,10}$$

$$U_{3,11} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,11} + \tau_{16} \epsilon_{11} S_2 G_3 (H_{17j11})_C$$

$$U_{3,12} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,12} + \tau_{16} \epsilon_{12} S_2 G_3 (H_{17j12})_C$$

$$U_{3,13} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,13} + \tau_{17} \epsilon_{13} G_3 (H_{16j17})_C (H_{17m13})_F \\ + \tau_{13} \tau_{17} \epsilon_{13} G_1 G_2 G_3 (H_{16j17})_C (H_{13j13})_D (H_{17m13})_F [1 + \tau_{13} (H_{13m13})_F]$$

$$U_{3,14} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,14} + \tau_{16} \epsilon_{14} S_2 G_3 (H_{17j14})_C$$

$$U_{3,15} = \tau_{14} \tau_{16} S_2 G_3 (H_{16m14})_B U_{2,15}$$

$$\check{E}_{16} = U_{4,1}\Omega_1 + U_{4,2}\Omega_2 + U_{4,4}\Omega_4 + U_{4,5}\Omega_5 + U_{4,6}\Omega_6 + U_{4,8}\Omega_8 + U_{4,9}\Omega_9 + U_{4,10}\Omega_{10} \\ + U_{4,11}\Omega_{11} + U_{4,12}\Omega_{12} + U_{4,13}\Omega_{13} + U_{4,14}\Omega_{14} + U_{4,15}E$$

$$U_{4,1} = \tau_{14} (H_{16m14})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16m16})_B] U_{2,1} + \tau_{16} \tau_{17} G_3 (H_{16m16})_B \\ (H_{16j17})_C F_{1,1}$$

$$U_{4,2} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,2} + \tau_{16} \tau_{17} G_3 (H_{16ml6})_B \\ (H_{16j17})_C F_{1,2}$$

$$U_{4,4} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,4} + \tau_{16} \tau_{17} G_3 (H_{16ml6})_B \\ (H_{16j17})_C F_{1,4}$$

$$U_{4,5} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,5} + \tau_{16} \tau_{17} G_3 (H_{16ml6})_B \\ (H_{16j17})_C F_{1,5} + \tau_{16} \epsilon_5 G_3 (H_{16ml6})_B (H_{16j5})_C$$

$$U_{4,8} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,8} + \tau_{16} \tau_{17} G_3 (H_{16ml6})_B \\ (H_{16j17})_C F_{1,8} + \tau_{16} \epsilon_8 G_3 (H_{16ml6})_B (H_{16j8})_C$$

$$U_{4,9} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,9} + \tau_{16} \tau_{17} G_3 (H_{16ml6})_B \\ (H_{16j17})_C F_{1,9} + \epsilon_9 (H_{16m9})_B + \tau_{16} \epsilon_9 G_3 (H_{16ml6})_B (H_{16j9})_C \\ + \tau_{16}^2 \epsilon_9 S_2 G_3 (H_{16ml6})_B (H_{16m9})_B$$

$$U_{4,10} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,10}$$

$$U_{4,11} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,11} + \epsilon_{11} (H_{16ml1})_B \\ + \tau_{16}^2 \epsilon_{11} S_2 G_3 (H_{16ml6})_B (H_{16ml1})_B$$

$$U_{4,12} = \tau_{14} (H_{16ml4})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16ml6})_B] U_{2,12} + \epsilon_{12} (H_{16ml2})_B \\ + \tau_{16}^2 \epsilon_{12} S_2 G_3 (H_{16ml6})_B (H_{16ml2})_B$$

$$U_{4,13} = \tau_{14} (H_{16m14})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16m16})_B] U_{2,13} + \tau_{16} \tau_{17} G_3 (H_{16m16})_B$$

$$(H_{16j17})_C F_{1,13}$$

$$U_{4,14} = \tau_{14} (H_{16m14})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16m16})_B] U_{2,14} + \tau_{16}^2 \epsilon_{14} S_2 G_3 (H_{16m16})_B$$

$$(H_{16m14})_B + \epsilon_{14} (H_{16m14})_B$$

$$U_{4,15} = \tau_{14} (H_{16m14})_B [1 + \tau_{16}^2 S_2 G_3 (H_{16m16})_B] U_{2,15}$$

$$\bar{E}_{17} = U_{5,1}\Omega_1 + U_{5,2}\Omega_2 + U_{5,4}\Omega_4 + U_{5,5}\Omega_5 + U_{5,6}\Omega_6 + U_{5,8}\Omega_8 + U_{5,9}\Omega_9 + U_{5,10}\Omega_{10}$$

$$+ U_{5,11}\Omega_{11} + U_{5,12}\Omega_{12} + U_{5,13}\Omega_{13} + U_{5,14}\Omega_{14} + U_{5,15}E$$

$$U_{5,1} = \epsilon_1 \tau_{13} G_1 G_2 (H_{17m13})_F (H_{13j1})_D + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,1}$$

$$U_{5,2} = \epsilon_2 G_2 (H_{17m2})_F + \epsilon_2 \tau_{13} G_1 G_2 (H_{17m13})_F [(H_{13j2})_D + (H_{13j13})_D (H_{13m2})_F$$

$$\tau_{13}] + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,2}$$

$$U_{5,4} = \epsilon_4 G_2 (H_{17m4})_F + \epsilon_4 \tau_{13}^2 G_1 G_2 (H_{17m13})_F (H_{13j13})_D (H_{13m4})_F + \tau_{16} \tau_{17}$$

$$S_1 G_2 (H_{17j16})_C U_{4,4}$$

$$U_{5,5} = \epsilon_5 \tau_{17} S_1 G_2 (H_{17j5})_C + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,5}$$

$$U_{5,6} = \epsilon_6 \tau_{17} S_1 G_2 (H_{17j6})_C + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,6}$$

$$U_{5,8} = \epsilon_8 \tau_{17} S_1 G_2 (H_{17j8})_C + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,8}$$

$$U_{5,9} = \epsilon_9 \tau_{17} S_1 G_2 (H_{17j9})_C + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,9}$$

$$U_{5,10} = \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,10}$$

$$U_{5,11} = \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,11}$$

$$U_{5,12} = \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,12}$$

$$U_{5,13} = \epsilon_{13} G_2 (H_{17m13})_F + \epsilon_{13} \tau_{13} G_1 G_2 (H_{17m13})_F (H_{13j13})_D [1 + \tau_{13} (H_{13m13})_F] \\ + \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,13}$$

$$U_{5,14} = \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,14}$$

$$U_{5,15} = \tau_{16} \tau_{17} S_1 G_2 (H_{17j16})_C U_{4,15}$$

$$\hat{E}_{17} = U_{6,1}\Omega_1 + U_{6,2}\Omega_2 + U_{6,4}\Omega_4 + U_{6,5}\Omega_5 + U_{6,6}\Omega_6 + U_{6,8}\Omega_8 + U_{6,9}\Omega_9 + U_{6,10}\Omega_{10} \\ + U_{6,11}\Omega_{11} + U_{6,12}\Omega_{12} + U_{6,13}\Omega_{13} + U_{6,14}\Omega_{14} + U_{6,15}E$$

$$U_{6,1} = \epsilon_1 \tau_{13} \tau_{17} G_1 G_2 (H_{17j17})_C (H_{17m13})_F (H_{13j1})_D + \tau_{16} G_2 (H_{17j16})_C U_{4,1}$$

$$U_{6,2} = \epsilon_2 \tau_{17} G_2 (H_{17j17})_C (H_{17m2})_F + \epsilon_2 \tau_{13} \tau_{17} G_1 G_2 (H_{17j17})_C (H_{17m13})_F \\ [(H_{13j2})_D + \tau_{13} (H_{13j13})_D (H_{13m2})_F] + \tau_{16} G_2 (H_{17j16})_C U_{4,2}$$

$$U_{6,4} = \epsilon_4 \tau_{17} G_2 (H_{17j17})_C (H_{17m4})_F + \epsilon_4 \tau_{13}^2 \tau_{17} G_1 G_2 (H_{17j17})_C (H_{17m13})_F \\ (H_{13j13})_D (H_{13m4})_F + \tau_{16} G_2 (H_{17j16})_C U_{4,4}$$

$$U_{6,5} = \epsilon_5 G_2(H_{17j5})_C + \tau_{16} G_2(H_{17j16})_C U_{4,5}$$

$$U_{6,6} = \epsilon_6 G_2(H_{17j6})_C + \tau_{16} G_2(H_{17j16})_C U_{4,6}$$

$$U_{6,8} = \epsilon_8 G_2(H_{17j8})_C + \tau_{16} G_2(H_{17j16})_C U_{4,8}$$

$$U_{6,9} = \epsilon_9 G_2(H_{17j9})_C + \tau_{16} G_2(H_{17j16})_C U_{4,9}$$

$$U_{6,10} = \tau_{16} G_2(H_{17j16})_C U_{4,10}$$

$$U_{6,11} = \tau_{16} G_2(H_{17j16})_C U_{4,11}$$

$$U_{6,12} = \tau_{16} G_2(H_{17j16})_C U_{4,12}$$

$$U_{6,13} = \epsilon_{13} \tau_{13} \tau_{17} G_1 G_2(H_{17j17})_C (H_{17m13})_F (H_{13j13})_D [1 + \tau_{13} (H_{13m13})_F]$$

$$+ \epsilon_{13} \tau_{17} G_2(H_{17j17})_C (H_{17m13})_F + \tau_{16} G_2(H_{17j16})_C U_{4,13}$$

$$U_{6,14} = \tau_{16} G_2(H_{17j16})_C U_{4,14}$$

$$U_{6,15} = \tau_{16} G_2(H_{17j16})_C U_{4,15}$$

$$E_{13} = U_{7,1}\Omega_1 + U_{7,2}\Omega_2 + U_{7,4}\Omega_4 + U_{7,5}\Omega_5 + U_{7,6}\Omega_6 + U_{7,8}\Omega_8 + U_{7,9}\Omega_9 + U_{7,10}\Omega_{10}$$

$$+ U_{7,11}\Omega_{11} + U_{7,12}\Omega_{12} + U_{7,13}\Omega_{13} + U_{7,14}\Omega_{14} + U_{7,15}E$$

$$U_{7,1} = \epsilon_1 G_1(H_{13j1})_D + \tau_{13} \tau_{17} G_1(H_{13j13})_D (H_{13m17})_F U_{6,1}$$

$$U_{7,2} = \epsilon_2 G_1 (H_{13j2})_D + \epsilon_2 \tau_{13} G_1 (H_{13j13})_D (H_{13m2})_F + \tau_{13} \tau_{17} G_1 (H_{13j13})_D \\ (H_{13m17})_F U_{6,2}$$

$$U_{7,4} = \epsilon_4 \tau_{13} G_1 (H_{13j13})_D (H_{13m4})_F + \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,4}$$

$$U_{7,5} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,5}$$

$$U_{7,6} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,6}$$

$$U_{7,8} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,8}$$

$$U_{7,9} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,9}$$

$$U_{7,10} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,10}$$

$$U_{7,11} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{k3m17})_F U_{6,11}$$

$$U_{7,12} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,12}$$

$$U_{7,13} = \epsilon_{13} G_1 (H_{13j13})_D + \epsilon_{13} \tau_{13} G_1 (H_{13j13})_D (H_{13m13})_F + \tau_{13} \tau_{17} G_1 \\ (H_{13j13})_D (H_{13m17})_F U_{6,13}$$

$$U_{7,14} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,14}$$

$$U_{7,15} = \tau_{13} \tau_{17} G_1 (H_{13j13})_D (H_{13m17})_F U_{6,15}$$

$$\bar{E}_{13} = U_{8,1}\Omega_1 + U_{8,2}\Omega_2 + U_{8,4}\Omega_4 + U_{8,5}\Omega_5 + U_{8,6}\Omega_6 + U_{8,8}\Omega_8 + U_{8,9}\Omega_9 + U_{8,10}\Omega_{10} \\ + U_{8,11}\Omega_{11} + U_{8,12}\Omega_{12} + U_{8,13}\Omega_{13} + U_{8,14}\Omega_{14} + U_{8,15}E$$

$$U_{8,1} = \epsilon_1 \tau_{13} G_1 (H_{13m13})_F (H_{13j1})_D + \tau_{17} G_1 (H_{13m17})_F U_{6,1}$$

$$U_{8,2} = \epsilon_2 \tau_{13} G_1 (H_{13m13})_F (H_{13j2})_D + \epsilon_2 G_2 (H_{13m2})_F + \tau_{17} G_1 (H_{13m17})_F U_{6,2}$$

$$U_{8,4} = \epsilon_4 G_1 (H_{13m4})_F + \tau_{17} G_1 (H_{13m17})_F U_{6,4}$$

$$U_{8,5} = \tau_{17} G_1 (H_{13m17})_F U_{6,5}$$

$$U_{8,6} = \tau_{17} G_1 (H_{13m17})_F U_{6,6}$$

$$U_{8,8} = \tau_{17} G_1 (H_{13m17})_F U_{6,8}$$

$$U_{8,9} = \tau_{17} G_1 (H_{13m17})_F U_{6,9}$$

$$U_{8,10} = \tau_{17} G_1 (H_{13m17})_F U_{6,10}$$

$$U_{8,11} = \tau_{17} G_1 (H_{13m17})_F U_{6,11}$$

$$U_{8,12} = \tau_{17} G_1 (H_{13m17})_F U_{6,12}$$

$$U_{8,13} = \epsilon_{13} \tau_{13} G_1 (H_{13m13})_F (H_{13j13})_D + \epsilon_{13} G_1 (H_{13m13})_F + \tau_{17} G_1 (H_{13m17})_F U_{6,13}$$

$$U_{8,14} = \tau_{17} G_1 (H_{13m17})_F U_{6,14}$$

$$U_{8,15} = \tau_{17} G_1 (H_{13m17})_F U_{6,15}$$

$$D_{1,1} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \epsilon_1 (H_{1j1})_D - \epsilon_1 + (1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,1}]_{S,L}$$

$$D_{1,2} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \epsilon_2 (H_{1j2})_D + (1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,2}]_{S,L}$$

$$D_{1,3} = 0$$

$$D_{1,4} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,4}]_{S,L}$$

$$D_{1,5} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,5}]_{S,L}$$

$$D_{1,6} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,6}]_{S,L}$$

$$D_{1,7} = 0$$

$$D_{1,8} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,8}]_{S,L}$$

$$D_{1,9} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,9}]_{S,L}$$

$$D_{1,10} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,10}]_{S,L}$$

$$D_{1,11} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,11}]_{S,L}$$

$$D_{1,12} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,12}]_{S,L}$$

$$D_{1,13} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{1,13} + (1-\rho_1) \epsilon_{13} (H_{1j13})_D]_{S,L}$$

$$D_{1,14} = \left(\frac{\sigma A_1}{M_1}\right) [(1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,14}]_{S,L}$$

$$D_{2,1} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2)[\epsilon_1(H_{2j1})_D + \tau_{13}(H_{2j13})_D U_{8,1}] + A_{2F}(1-\rho_2)[\tau_{13}(H_{2m13})_F U_{7,1} + \tau_{17}(H_{2m17})_F U_{6,1}]\}_{S,L}$$

$$D_{2,2} = \left(\frac{\sigma}{M_2}\right) \{A_{2E}[(1-\rho_2) \epsilon_2(H_{2j2})_E - \epsilon_2] + A_{2D}[(1-\rho_2) \epsilon_2(H_{2j2}) - \epsilon_2 + (1-\rho_2) \tau_{13}(H_{2j13})_D U_{8,2}] + A_{2F}[(1-\rho_2) \epsilon_2(H_{2m2})_F - \epsilon_2 + (1-\rho_2) \tau_{13}(H_{2m13})_F U_{7,2} + (1-\rho_2) \tau_{17}(H_{2m17})_F U_{6,2}]\}_{S,L}$$

$$D_{2,3} = \left(\frac{\sigma}{M_2}\right) [A_{2E}(1-\rho_2) \epsilon_3(H_{2j3})_E]_{S,L}$$

$$D_{2,4} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}(H_{2j13})_D U_{8,4} + A_{2F}[(1-\rho_2) \epsilon_4(H_{2m4})_F + (1-\rho_2) \tau_{13}(H_{2m13})_F U_{7,4} + (1-\rho_2) \tau_{17}(H_{2m17})_F U_{6,4}]\}_{S,L}$$

$$D_{2,5} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}(H_{2j13})_D U_{8,5} + A_{2F}(1-\rho_2)[\tau_{13}(H_{2m13})_F U_{7,5} + \tau_{17}(H_{2m17})_F U_{6,5}]\}_{S,L}$$

$$D_{2,6} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}(H_{2j13})_D U_{8,6} + A_{2F}(1-\rho_2)[\tau_{13}(H_{2m13})_F U_{7,6} + \tau_{17}(H_{2m17})_F U_{6,6}]\}_{S,L}$$

$$D_{2,7} = \left(\frac{\sigma}{M_2}\right) [A_{2E}(1-\rho_2) \epsilon_7(H_{2j7})_E]_{S,L}$$

$$D_{2,8} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}(H_{2j13})_D U_{8,8} + A_{2F}(1-\rho_2)[\tau_{13}(H_{2m13})_F U_{7,8} + \tau_{17}(H_{2m17})_F U_{6,8}]\}_{S,L}$$

$$D_{2,9} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}^{(H_{2j13})_D} U_{8,9} + A_{2F}(1-\rho_2) [\tau_{13}^{(H_{2m13})_F} U_{7,9} \\ + \tau_{17}^{(H_{2m17})_F} U_{6,9}]\} S, L$$

$$D_{2,10} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}^{(H_{2j13})_D} U_{8,10} + A_{2F}(1-\rho_2) [\tau_{13}^{(H_{2m13})_F} U_{7,10} \\ + \tau_{17}^{(H_{2m17})_F} U_{6,10}]\} S, L$$

$$D_{2,11} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}^{(H_{2j13})_D} U_{8,11} + A_{2F}(1-\rho_2) [\tau_{13}^{(H_{2m13})_F} U_{7,11} \\ + \tau_{17}^{(H_{2m17})_F} U_{6,11}]\} S, L$$

$$D_{2,12} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}^{(H_{2j13})_D} U_{8,12} + A_{2F}(1-\rho_2) [\tau_{13}^{(H_{2m13})_F} U_{7,12} \\ + \tau_{17}^{(H_{2m17})_F} U_{6,12}]\} S, L$$

$$D_{2,13} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) [\epsilon_{13}^{(H_{2j13})_D} + \tau_{13}^{(H_{2j13})_D} U_{8,13}] + A_{2F}(1-\rho_2) [\epsilon_{13}^{(H_{2m13})_F} \\ + \tau_{13}^{(H_{2m13})_F} U_{7,13} + \tau_{17}^{(H_{2m17})_F} U_{6,13}]\} S, L$$

$$D_{2,14} = \left(\frac{\sigma}{M_2}\right) \{A_{2D}(1-\rho_2) \tau_{13}^{(H_{2j13})_D} U_{8,14} + A_{2F}(1-\rho_2) [\tau_{13}^{(H_{2m13})_F} U_{7,14} \\ + \tau_{17}^{(H_{2m17})_F} U_{6,14}]\} S, L$$

$$D_{3,1} = 0$$

$$D_{3,2} = \left(\frac{\sigma}{M_3}\right) A_3 [(1-\rho_3) \epsilon_2^{(H_{3j2})_E}] S, L$$

$$D_{3,3} = \left(\frac{\sigma}{M_3}\right) A_3 [(1-\rho_3) \epsilon_3 (H_{3j3})_E - \epsilon_3]_{S,L}$$

$$D_{3,4} = D_{3,5} = D_{3,6} = 0$$

$$D_{3,7} = \left(\frac{\sigma}{M_3}\right) A_3 [(1-\rho_3) \epsilon_7 (H_{3j7})_E]_{S,L}$$

$$D_{3,8} = D_{3,9} = D_{3,10} = D_{3,11} = D_{3,12} = D_{3,13} = D_{3,14} = 0$$

$$D_{4,1} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,1} + \tau_{17} (H_{4m17})_F U_{6,1}] \}_{S,L}$$

$$D_{4,2} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\epsilon_2 (H_{4m2})_F + \tau_{13} (H_{4m13})_F U_{7,2} + \tau_{17} (H_{4m17})_F U_{6,2}] \}_{S,L}$$

$$D_{4,3} = 0$$

$$D_{4,4} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\epsilon_4 (H_{4m4})_F - \epsilon_4 + (1-\rho_4) \tau_{13} (H_{4m13})_F U_{7,4} + (1-\rho_4) \tau_{17} (H_{4m17})_F U_{6,4}] \}_{S,L}$$

$$D_{4,5} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,5} + \tau_{17} (H_{4m17})_F U_{6,5}] \}_{S,L}$$

$$D_{4,6} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,6} + \tau_{17} (H_{4m17})_F U_{6,6}] \}_{S,L}$$

$$D_{4,7} = 0$$

$$D_{4,8} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,8} + \tau_{17} (H_{4,17})_F U_{6,8}] \}_{S,L}$$

$$D_{4,9} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,9} + \tau_{17} (H_{4m17})_F U_{6,9}] \}_{S,L}$$

$$D_{4,10} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,10} + \tau_{17} (H_{4m17})_F U_{6,10}] \}_{S,L}$$

$$D_{4,11} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,11} + \tau_{17} (H_{4m17})_F U_{6,11}] \}_{S,L}$$

$$D_{4,12} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,12} + \tau_{17} (H_{4m17})_F U_{6,12}] \}_{S,L}$$

$$D_{4,13} = \left(\frac{\sigma A_4}{M_4}\right) \{ [\epsilon_{13} (H_{4m13})_F + \tau_{13} (H_{4m13})_F U_{7,13} + \tau_{17} (H_{4m17})_F U_{6,13}] (1-\rho_4) \}_{S,L}$$

$$D_{4,14} = \left(\frac{\sigma A_4}{M_4}\right) \{ (1-\rho_4) [\tau_{13} (H_{4m13})_F U_{7,14} + \tau_{17} (H_{4m17})_F U_{6,14}] \}_{S,L}$$

$$D_{5,1} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16} (H_{5j16})_c U_{4,1} + \tau_{17} (H_{5j17})_c U_{5,1}] \}_{S,L}$$

$$D_{5,2} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16} (H_{5j16})_c U_{4,2} + \tau_{17} (H_{5j17})_c U_{5,2}] \}_{S,L}$$

$$D_{5,3} = 0$$

$$D_{5,4} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16} (H_{5j16})_c U_{4,4} + \tau_{17} (H_{5j17})_c U_{5,4}] \}_{S,L}$$

$$D_{5,5} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16} (H_{5j16})_c U_{4,5} + \tau_{17} (H_{5j17})_c U_{5,5}] + [(1-\rho_5) \epsilon_5 (H_{5j5})_c - \epsilon_5] \}_{S,L}$$

$$D_{5,6} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\epsilon_6 (H_{5j6})_c + \tau_{16} (H_{5j16})_c U_{4,6} + \tau_{17} (H_{5j17})_c U_{5,6}] \}_{S,L}$$

$$D_{5,7} = 0$$

$$D_{5,8} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\epsilon_8^{(H_{5j8})} c + \tau_{16}^{(H_{5j16})} c U_{4,8} + \tau_{17}^{(H_{5j17})} c U_{5,8}] \}_{S,L}$$

$$D_{5,9} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\epsilon_9^{(H_{5j9})} c + \tau_{16}^{(H_{5j16})} c U_{4,9} + \tau_{17}^{(H_{5j17})} c U_{5,9}] \}_{S,L}$$

$$D_{5,10} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16}^{(H_{5j16})} c U_{4,10} + \tau_{17}^{(H_{5j17})} c U_{5,10}] \}_{S,L}$$

$$D_{5,11} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16}^{(H_{5j16})} c U_{4,11} + \tau_{17}^{(H_{5j17})} c U_{5,11}] \}_{S,L}$$

$$D_{5,12} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16}^{(H_{5j16})} c U_{4,12} + \tau_{17}^{(H_{5j17})} c U_{5,12}] \}_{S,L}$$

$$D_{5,13} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16}^{(H_{5j16})} c U_{4,13} + \tau_{17}^{(H_{5j17})} c U_{5,13}] \}_{S,L}$$

$$D_{5,14} = \left(\frac{\sigma A_5}{M_5}\right) \{ (1-\rho_5) [\tau_{16}^{(H_{5j16})} c U_{4,14} + \tau_{17}^{(H_{5j17})} c U_{5,14}] \}_{S,L}$$

$$D_{6,1} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16}^{(H_{6j16})} c U_{4,1} + \tau_{17}^{(H_{6j17})} c U_{5,1}] \}_{S,L}$$

$$S_{6,2} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16}^{(H_{6j16})} c U_{4,2} + \tau_{17}^{(H_{6j17})} c U_{5,2}] \}_{S,L}$$

$$D_{6,3} = 0$$

$$D_{6,4} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16}^{(H_{6j16})} c U_{4,4} + \tau_{17}^{(H_{6j17})} c U_{5,4}] \}_{S,L}$$

$$D_{6,5} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\epsilon_5 (H_{6j5})_c + \tau_{16} (H_{6j16})_c U_{4,5} + \tau_{17} (H_{6j17})_c U_{5,5}] \}_{S,L}$$

$$D_{6,6} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\epsilon_6 (H_{6j6})_c + \tau_{16} (H_{6j16})_c U_{4,6} + \tau_{17} (H_{6j17})_c U_{5,6} - \epsilon_6] \}_{S,L}$$

$$D_{6,7} = 0$$

$$D_{6,8} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\epsilon_8 (H_{6j8})_c + \tau_{16} (H_{6j16})_c U_{4,8} + \tau_{17} (H_{6j17})_c U_{5,8}] \}_{S,L}$$

$$D_{6,9} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\epsilon_9 (H_{6j9})_c + \tau_{16} (H_{6j16})_c U_{4,9} + \tau_{17} (H_{6j17})_c U_{5,9}] \}_{S,L}$$

$$D_{6,10} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16} (H_{6j16})_c U_{4,10} + \tau_{17} (H_{6j17})_c U_{5,10}] \}_{S,L}$$

$$D_{6,11} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16} (H_{6j16})_c U_{4,11} + \tau_{17} (H_{6j17})_c U_{5,11}] \}_{S,L}$$

$$D_{6,12} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16} (H_{6j16})_c U_{4,12} + \tau_{17} (H_{6j17})_c U_{5,12}] \}_{S,L}$$

$$D_{6,13} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16} (H_{6j16})_c U_{4,13} + \tau_{17} (H_{6j17})_c U_{5,13}] \}_{S,L}$$

$$D_{6,14} = \left(\frac{\sigma A_6}{M_6}\right) \{ (1-\rho_6) [\tau_{16} (H_{6j16})_c U_{4,14} + \tau_{17} (H_{6j17})_c U_{5,15}] \}_{S,L}$$

$$D_{7,1} = 0$$

$$D_{7,2} = \left(\frac{\sigma A_7}{M_7}\right) \{ (1-\rho_7) \epsilon_2 (H_{7j2})_E \}_{S,L}$$

$$D_{7,3} = \left(\frac{\sigma A_7}{M_7}\right) \{ (1-\rho_7) \epsilon_3^{(H_{7j3})} \}_E S,L$$

$$D_{7,4} = D_{7,5} = D_{7,6} = 0$$

$$D_{7,7} = \left(\frac{\sigma A_7}{M_7}\right) \{ (1-\rho_7) \epsilon_7^{(H_{7j7})} \}_E - \epsilon_7 \}_S,L$$

$$D_{7,8} = D_{7,9} = D_{7,10} = D_{7,11} = D_{7,12} = D_{7,13} = D_{7,14} = 0$$

$$D_{8,1} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16}^{(H_{8j16})} \epsilon_{4,1} + \tau_{17}^{(H_{8j17})} \epsilon_{5,1}] \}_S,L$$

$$D_{8,2} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16}^{(H_{8j16})} \epsilon_{4,2} + \tau_{17}^{(H_{8j17})} \epsilon_{5,2}] \}_S,L$$

$$D_{8,3} = 0$$

$$D_{8,4} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16}^{(H_{8j16})} \epsilon_{4,4} + \tau_{17}^{(H_{8j17})} \epsilon_{5,4}] \}_S,L$$

$$D_{8,5} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\epsilon_5^{(H_{8j5})} \epsilon_{4,5} + \tau_{16}^{(H_{8j16})} \epsilon_{4,5} + \tau_{17}^{(H_{8j17})} \epsilon_{5,5}] \}_S,L$$

$$D_{8,6} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\epsilon_6^{(H_{8j6})} \epsilon_{4,6} + \tau_{16}^{(H_{8j16})} \epsilon_{4,6} + \tau_{17}^{(H_{8j17})} \epsilon_{5,6}] \}_S,L$$

$$D_{8,7} = 0$$

$$D_{8,8} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\epsilon_8^{(H_{8j8})} \epsilon_{4,8} + \tau_{16}^{(H_{8j16})} \epsilon_{4,8} + \tau_{17}^{(H_{8j17})} \epsilon_{5,8}] - \epsilon_8 \}_S,L$$

$$D_{8,9} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [e_9 (H_{8j9})_c + \tau_{16} (H_{8j16})_c U_{4,9} + \tau_{17} (H_{8j17})_c U_{5,9}] \}_{S,L}$$

$$D_{8,10} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16} (H_{8j16})_c U_{4,10} + \tau_{17} (H_{8j17})_c U_{5,10}] \}_{S,L}$$

$$D_{8,11} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16} (H_{8j16})_c U_{4,11} + \tau_{17} (H_{8j17})_c U_{5,11}] \}_{S,L}$$

$$D_{8,12} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16} (H_{8j16})_c U_{4,12} + \tau_{17} (H_{8j17})_c U_{5,12}] \}_{S,L}$$

$$D_{8,13} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16} (H_{8j16})_c U_{4,13} + \tau_{17} (H_{8j17})_c U_{5,13}] \}_{S,L}$$

$$D_{8,14} = \left(\frac{\sigma A_8}{M_8}\right) \{ (1-\rho_8) [\tau_{16} (H_{8j16})_c U_{4,14} + \tau_{17} (H_{8j17})_c U_{5,14}] \}_{S,L}$$

$$D_{9,1} = \left(\frac{\sigma}{M_9}\right) \{ A_{9B} (1-\rho_9)_B [\tau_{14} (H_{9m14})_B U_{2,1} + \tau_{16} (H_{9m16})_B U_{3,1}] + A_{9c} (1-\rho_9)_c \\ [\tau_{16} (H_{9j16})_c U_{4,1} + \tau_{17} (H_{9j17})_c U_{5,1}] \}_{S,L}$$

$$D_{9,2} = \left(\frac{\sigma}{M_9}\right) \{ A_{9B} (1-\rho_9)_B [\tau_{14} (H_{9m14})_B U_{2,2} + \tau_{16} (H_{9m16})_B U_{3,2}] + A_{9c} (1-\rho_9)_c \\ [\tau_{16} (H_{9j16})_c U_{4,2} + \tau_{17} (H_{9j17})_c U_{5,2}] \}_{S,L}$$

$$D_{9,3} = 0$$

$$D_{9,4} = \left(\frac{\sigma}{M_9}\right) \{ A_{9B} (1-\rho_9)_B [\tau_{14} (H_{9m14})_B U_{2,4} + \tau_{16} (H_{9m16})_B U_{3,4}] + A_{9c} (1-\rho_9)_c \\ [\tau_{16} (H_{9j16})_c U_{4,4} + \tau_{17} (H_{9j17})_c U_{5,4}] \}_{S,L}$$

$$D_{9,5} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\tau_{14}(H_{9m14})_B U_{2,5} + \tau_{16}(H_{9m16})_B U_{3,5}] + A_{9c}(1-\rho_9)_c$$

$$[\epsilon_5(H_{9j5})_c + \tau_{16}(H_{9j16})_c U_{4,5} + \tau_{17}(H_{9j17})_c U_{5,5}]\}_{S,L}$$

$$D_{9,6} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\tau_{14}(H_{9m14})_B U_{2,6} + \tau_{16}(H_{9m16})_B U_{3,6}] + A_{9c}(1-\rho_9)_c$$

$$[\epsilon_6(H_{9j6})_c + \tau_{16}(H_{9j16})_c U_{4,6} + \tau_{17}(H_{9j17})_c U_{5,6}]\}_{S,L}$$

$$D_{9,7} = 0$$

$$D_{9,8} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\tau_{14}(H_{9m14})_B U_{2,8} + \tau_{16}(H_{9m16})_B U_{3,8}] + A_{9c}(1-\rho_9)_c$$

$$[\epsilon_8(H_{9j6})_c + \tau_{16}(H_{9j16})_c U_{4,8} + \tau_{17}(H_{9j17})_c U_{5,8}]\}_{S,L}$$

$$D_{9,9} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\epsilon_9(H_{9m9})_B + \tau_{14}(H_{9m14})_B U_{4,9} + \tau_{16}(H_{9j16})_c U_{5,9}] - A_{9B} \epsilon_9$$

$$+ A_{9c}(1-\rho_9)_c [\epsilon_9(H_{9j9})_c + \tau_{16}(H_{9j16})_c U_{4,9} + \tau_{17}(H_{9j17})_c U_{5,9}] - A_{9c} \epsilon_9\}_{S,L}$$

$$D_{9,10} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\tau_{14}(H_{9m14})_B U_{2,10} + \tau_{16}(H_{9m16})_B U_{3,10}] + A_{9c} [\tau_{16}$$

$$(H_{9j16})_c U_{4,10} + \tau_{17}(H_{9j17})_c U_{5,10}]\}_{S,L}$$

$$D_{9,11} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\epsilon_{11}(H_{9m11})_B + \tau_{14}(H_{9m14})_B U_{2,11} + \tau_{16}(H_{9m16})_B U_{3,11}]$$

$$+ A_{9c}(1-\rho_9)_c [\tau_{16}(H_{9j16})_c U_{4,11} + \tau_{17}(H_{9j17})_c U_{5,11}]\}_{S,L}$$

$$D_{9,12} = \left(\frac{\sigma}{M_9}\right) \{A_{9B}(1-\rho_9)_B [\epsilon_{12}(H_{9m12})_B + \tau_{14}(H_{9m14})_B U_{2,12} + \tau_{16}(H_{9m16})_B U_{3,12}]$$

$$+ A_{9c}(1-\rho_9)_c [\tau_{16}(H_{9j16})_c U_{4,12} + \tau_{17}(H_{9j17})_c U_{5,12}]\}_{S,L}$$

$$D_{9,13} = \left(\frac{\sigma}{M_9}\right)\{A_{9B}(1-\rho_9)_B[\tau_{14}(H_{9m14})_B U_{2,3} + \tau_{16}(H_{9m16})_m U_{3,13}] + A_{9c}(1-\rho_9)_c$$

$$[\tau_{16}(H_{9j16})_c U_{4,13} + \tau_{17}(H_{9j17})_c U_{5,13}]\}_{S,L}$$

$$D_{9,14} = \left(\frac{\sigma}{M_9}\right)\{A_{9B}(1-\rho_9)_B[\epsilon_{14}(H_{9m14})_B + \tau_{14}(H_{9m14})_B U_{2,14} + \tau_{16}(H_{9m16})_B U_{3,14}]$$

$$+ A_{9c}(1-\rho_9)_c[\tau_{16}(H_{9j16})_c U_{4,14} + \tau_{17}(H_{9j17})_c U_{5,14}]\}_{S,L}$$

$$D_{10,1} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,1}\}_{S,L}$$

$$D_{10,2} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,2}\}_{S,L}$$

$$D_{10,3} = 0$$

$$D_{10,4} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,4}\}_{S,L}$$

$$D_{10,5} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,5}\}_{S,L}$$

$$D_{10,6} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,6}\}_{S,L}$$

$$D_{10,7} = 0$$

$$D_{10,8} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,8}\}_{S,L}$$

$$D_{10,9} = \left(\frac{\sigma A_{10}}{M_{10}}\right)\{(1-\rho_{10}) \tau_{14}(H_{10j14})_A U_{1,9}\}_{S,L}$$

$$D_{10,10} = \left(\frac{\sigma A_{10}}{M_{10}}\right) \{ (1-\rho_{10}) \epsilon_{10} (H_{10j10})_A - \epsilon_{10} + (1-\rho_{10}) \tau_{14} (H_{10j14})_A U_{1,10} \}_{S,L}$$

$$D_{10,11} = \left(\frac{\sigma A_{10}}{M_{10}}\right) \{ (1-\rho_{10}) \tau_{14} (H_{10j14})_A U_{1,11} \}$$

$$D_{10,12} = \left(\frac{\sigma A_{10}}{M_{10}}\right) \{ (1-\rho_{10}) [\epsilon_{12} (H_{10j12})_A + \tau_{14} (H_{10j14})_A U_{1,12}] \}_{S,L}$$

$$D_{10,13} = \left(\frac{\sigma A_{10}}{M_{10}}\right) \{ (1-\rho_{10}) \tau_{14} (H_{10j14})_A U_{1,13} \}_{S,L}$$

$$D_{10,14} = \left(\frac{\sigma A_{10}}{M_{10}}\right) \{ \epsilon_{14} (H_{10j14})_A + \tau_{14} (H_{10j14})_A U_{1,14} \}$$

$$D_{11,1} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,1} + \tau_{16} (H_{11ml6})_B U_{3,1}] \}_{S,L}$$

$$D_{11,2} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,2} + \tau_{16} (H_{11ml6})_B U_{3,2}] \}_{S,L}$$

$$D_{11,3} = 0$$

$$D_{11,4} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,4} + \tau_{16} (H_{11ml6})_B U_{3,4}] \}_{S,L}$$

$$D_{11,5} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,5} + \tau_{16} (H_{11ml6})_B U_{3,5}] \}_{S,L}$$

$$D_{11,6} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,6} + \tau_{16} (H_{11ml6})_B U_{3,6}] \}_{S,L}$$

$$D_{11,7} = 0$$

$$D_{11,8} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,8} + \tau_{16} (H_{11ml6})_B U_{3,8}] \}_{S,L}$$

$$D_{11,9} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ \epsilon_9 (H_{11m9})_B + \tau_{14} (H_{11ml4})_B U_{2,9} + \tau_{16} (H_{11ml6})_B U_{3,9} \} S, L$$

$$D_{11,10} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ \tau_{14} (H_{11ml4})_B U_{2,10} + \tau_{16} (H_{11ml6})_B U_{3,10} \} S, L$$

$$D_{11,11} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) \epsilon_{11} (H_{11ml1})_B - \epsilon_{11} + \tau_{14} (H_{11ml4})_B U_{2,11} + \tau_{16} (H_{11ml6})_B U_{3,11} \} S, L$$

$$D_{11,12} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\epsilon_{12} (H_{11ml2})_B + \tau_{14} (H_{11ml4})_B U_{2,12} + \tau_{16} (H_{11ml6})_B U_{3,12}] \} S, L$$

$$D_{11,13} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\tau_{14} (H_{11ml4})_B U_{2,13} + \tau_{16} (H_{11ml6})_B U_{3,13}] \} S, L$$

$$D_{11,14} = \left(\frac{\sigma A_{11}}{M_{11}}\right) \{ (1-\rho_{11}) [\epsilon_{14} (H_{11ml4})_B + \tau_{14} (H_{11ml4})_B U_{2,14} + \tau_{16} (H_{11ml6})_B U_{3,14}] \} S, L$$

$$D_{12,1} = \left(\frac{\sigma}{M_{12}}\right) \{ A_{12A} (1-\rho_{12})_A \tau_{14} (H_{12j14})_A U_{1,1} + A_{12B} (1-\rho_{12})_B [\tau_{14} (H_{12ml4})_B U_{2,1} + \tau_{16} (H_{12ml6})_B U_{3,1}] \} S, L$$

$$D_{12,2} = \left(\frac{\sigma}{M_{12}}\right) \{ A_{12A} (1-\rho_{12})_A \tau_{14} (H_{12j14})_A U_{1,2} + A_{12B} (1-\rho_{12})_B [\tau_{14} (H_{12ml4})_B U_{2,2} + \tau_{16} (H_{12ml6})_B U_{3,2}] \} S, L$$

$$D_{12,3} = 0$$

$$D_{12,4} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A}^{(1-\rho_{12})} \tau_{14}^{(H_{12j14})} U_{1,4} + A_{12B}^{(1-\rho_{12})} [\tau_{14}^{(H_{12m14})} U_{2,4} + \tau_{16}^{(H_{12m16})} U_{3,4}]\}_{S,L}$$

$$D_{12,5} = \left(\frac{\sigma}{M_{12}}\right) A_{12A}^{(1-\rho_{12})} \tau_{14}^{(H_{12j14})} U_{1,5} + A_{12B}^{(1-\rho_{12})} [\tau_{14}^{(H_{12m14})} U_{2,5} + \tau_{16}^{(H_{12m16})} U_{3,5}]\}_{S,L}$$

$$D_{12,6} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A}^{(1-\rho_{12})} \tau_{14}^{(H_{12j14})} U_{1,6} + A_{12B}^{(1-\rho_{12})} [\tau_{14}^{(H_{12m14})} U_{2,6} + \tau_{16}^{(H_{12m16})} U_{3,6}]\}_{S,L}$$

$$D_{12,7} = 0$$

$$D_{12,8} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A}^{(1-\rho_{12})} \tau_{14}^{(H_{12j14})} U_{1,8} + A_{12B}^{(1-\rho_{12})} [\tau_{14}^{(H_{12m14})} U_{2,8} + \tau_{16}^{(H_{12m16})} U_{3,8}]\}_{S,L}$$

$$D_{12,9} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A}^{(1-\rho_{12})} \tau_{14}^{(H_{12j14})} U_{1,9} + A_{12B}^{(1-\rho_{12})} [\epsilon_9^{(H_{12m9})} U_{2,9} + \tau_{14}^{(H_{12m14})} U_{2,9} + \tau_{16}^{(H_{12m16})} U_{3,9}]\}_{S,L}$$

$$D_{12,10} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A}^{(1-\rho_{12})} [\epsilon_{10}^{(H_{12j10})} + \tau_{12}^{(H_{12j14})} U_{1,10}] + A_{12B}^{(1-\rho_{12})} [\tau_{14}^{(H_{12m14})} U_{2,10} + \tau_{16}^{(H_{12m16})} U_{3,10}]\}_{S,L}$$

$$D_{12,11} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A}^{(1-\rho_{12})} \tau_{14}^{(H_{12j14})} U_{1,11} + A_{12B}^{(1-\rho_{12})} [\epsilon_{11}^{(H_{12m11})} U_{2,11} + \tau_{14}^{(H_{12m14})} U_{2,11} + \tau_{16}^{(H_{12m16})} U_{3,11}]\}_{S,L}$$

$$D_{12,12} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A} (1-\rho_{12})_A [\epsilon_{12} (H_{12j12})_A + \tau_{14} (H_{12j14})_A U_{1,12}] - A_{12A} \epsilon_{12} \\ + A_{12B} (1-\rho_{12})_B [\epsilon_{12} (H_{12m12})_B + \tau_{14} (H_{12j14})_B U_{2,12} + \tau_{16} (H_{12m16})_B \\ U_{3,12}] - A_{12B} \epsilon_{12}\}_{S,L}$$

$$D_{12,13} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A} (1-\rho_{12})_A \tau_{14} (H_{12j14})_A U_{1,13} + A_{12B} (1-\rho_{12})_B [\tau_{14} (H_{12m14})_B U_{2,13} \\ + \tau_{16} (H_{12m16})_B U_{3,13}]\}_{S,L}$$

$$D_{12,14} = \left(\frac{\sigma}{M_{12}}\right) \{A_{12A} (1-\rho_{12})_A [\epsilon_{12} (H_{12j14})_A + \tau_{14} (H_{12j14})_A U_{1,14} + A_{12B} (1-\rho_{12})_B \\ [\epsilon_{14} (H_{12m14})_B + \tau_{14} (H_{12m14})_B U_{2,4} + \tau_{16} (H_{12m16})_B U_{3,14}]]\}_{S,L}$$

$$D_{13,1} = \left(\frac{\sigma}{M_{13}}\right) \{A_{13F} (1-\rho_{13})_F [\tau_{17} (H_{13m17})_F U_{6,1} + \tau_{13} (H_{13m13})_F U_{7,1}] - A_{13F} U_{7,1} \\ + A_{13D} (1-\rho_{13})_D [\epsilon_1 (H_{13j1})_D + \tau_{13} (H_{13j13})_D U_{8,1}] - A_{13D} \tau_{13} U_{8,1}\}_{S,L}$$

$$D_{13,2} = \left(\frac{\sigma}{M_{13}}\right) \{A_{13F} (1-\rho_{13})_F [\tau_{17} (H_{13m17})_F U_{6,2} + \tau_{13} (H_{13m13})_F U_{7,2} \\ + \epsilon_2 (H_{13m2})_F] - A_{13F} \tau_{13} U_{7,2} + A_{13D} (1-\rho_{13})_D [\epsilon_2 (H_{13j2})_D \\ + \tau_{13} (H_{13j13})_D U_{8,2}] - A_{13D} \tau_{13} U_{8,2}\}_{S,L}$$

$$D_{13,3} = 0$$

$$D_{14,4} = \left(\frac{\sigma}{M_{13}}\right) \{A_{13F} (1-\rho_{13})_F [\tau_{17} (H_{13m17})_F U_{6,4} + \tau_{13} (H_{13m13})_F + \epsilon_4 (H_{13m4})_F] - A_{13F} \\ \tau_{13} U_{7,4} + A_{13D} [(1-\rho_{13})_D \tau_{13} (H_{13j13})_D - \tau_{13}] U_{8,4}\}_{S,L}$$

$$D_{13,5} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,5} + \tau_{13}(H_{13m13})_F U_{7,5}] - A_{13F}$$

$$\tau_{13} U_{7,5} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,5}\}_{S,L}$$

$$D_{13,6} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,6} + \tau_{13}(H_{13m13})_F U_{7,6}] - A_{13F}$$

$$\tau_{13} U_{7,6} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,6}\}_{S,L}$$

$$D_{13,7} = 0$$

$$D_{13,8} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,8} + \tau_{13}(H_{13m13})_F U_{7,8}] - A_{13F}$$

$$\tau_{13} U_{7,8} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,8}\}_{S,L}$$

$$D_{13,9} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,9} + \tau_{13}(H_{13m13})_F U_{7,9}] - A_{13F}$$

$$\tau_{13} U_{7,9} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,9}\}_{S,L}$$

$$D_{13,10} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,10} + \tau_{13}(H_{13m13})_F U_{7,10}] - A_{13F}$$

$$\tau_{13} U_{7,10} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,10}\}_{S,L}$$

$$D_{13,11} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,11} + \tau_{13}(H_{13m13})_F U_{7,11}] - A_{13F}$$

$$\tau_{13} U_{7,11} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,11}\}_{S,L}$$

$$D_{13,12} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,12} + \tau_{13}(H_{13m13})_F U_{7,12}] - A_{13F}$$

$$\tau_{13} U_{7,12} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,12}\}_{S,L}$$

$$D_{13,13} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,13} + \tau_{13}(H_{13m13})_F U_{7,13} + \epsilon_{13} \\ (H_{13m13})_F] - A_{13F}(\epsilon_{13} + \tau_{13} U_{7,13}) + A_{13D}(1-\rho_{13})_D[\epsilon_{13}(H_{13j13})_D \\ + \tau_{13}(H_{13j13})_D U_{8,13}] - A_{13D}(\epsilon_{13} + \tau_{13} U_{8,13})\}_{S,L}$$

$$D_{13,14} = \left(\frac{\sigma}{M_{13}}\right)\{A_{13F}(1-\rho_{13})_F[\tau_{17}(H_{13m17})_F U_{6,14} + \tau_{13}(H_{13m13})_F U_{7,14}] - A_{13F} \\ \tau_{13} U_{7,14} + A_{13D}[(1-\rho_{13})_D \tau_{13}(H_{13j13})_D - \tau_{13}] U_{8,14}\}_{S,L}$$

$$D_{14,1} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14}[(1-\rho_{14})(H_{14j14})_A - 1] U_{1,1} + A_{14B}[(1-\rho_{14}) \tau_{14} \\ (H_{14m14})_B - \tau_{14}] U_{2,1} + A_{14B} \tau_{16}(1-\rho_{14})(H_{14m16})_B U_{3,1}\}_{S,L}$$

$$D_{14,2} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14}[(1-\rho_{14})(H_{14j14})_A - 1] U_{1,2} + A_{14B}[(1-\rho_{14}) \tau_{14} \\ (H_{14m14})_B - \tau_{14}] U_{2,2} + A_{14B} \tau_{16}(1-\rho_{14})(H_{14m16})_B U_{3,2}\}_{S,L}$$

$$D_{14,3} = 0$$

$$D_{14,4} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14}[(1-\rho_{14})(H_{14j14})_A - 1] U_{1,4} + A_{14B} \tau_{14}[(1-\rho_{14}) \\ (H_{14m14})_B - 1] U_{2,4} + A_{14B} \tau_{16}(1-\rho_{14})(H_{14m16})_B U_{3,4}\}_{S,L}$$

$$D_{14,5} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14}[(1-\rho_{14})(H_{14j14})_A - 1] U_{1,5} + A_{14B} \tau_{14}[(1-\rho_{14}) \\ (H_{14m14})_B - 1] U_{2,5} + A_{14B} \tau_{16}(1-\rho_{14})(H_{14m16})_B U_{3,5}\}_{S,L}$$

$$D_{14,6} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14} [(1-\rho_{14})(H_{14j14})_A - 1] U_{1,6} + A_{14B} \tau_{14} [(1-\rho_{14})(H_{14m14})_B - 1] U_{2,6} + A_{14B} \tau_{16} (1-\rho_{14})(H_{14m16})_B U_{3,6}\}_{S,L}$$

$$D_{14,7} = 0$$

$$D_{14,8} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14} [(1-\rho_{14})(H_{14h14})_A - 1] U_{1,8} + A_{14B} \tau_{14} [(1-\rho_{14})(H_{14m14})_B - 1] U_{2,8} + A_{14B} \tau_{16} (1-\rho_{14})(H_{14m16})_B U_{3,8}\}_{S,L}$$

$$D_{14,9} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14} [(1-\rho_{14})(H_{14j14})_A - 1] U_{1,9} + A_{14B} (1-\rho_{14})_B [\epsilon_9 (H_{14m9})_B + \tau_{14} (H_{14m14})_B U_{2,9} + \tau_{16} (H_{14m16})_B U_{3,9}] - A_{14B} \tau_{14} U_{2,9}\}_{S,L}$$

$$D_{14,10} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} (1-\rho_{14}) [\epsilon_{10} (H_{14j10})_A + \tau_{14} (H_{14j14})_A U_{1,10}] - A_{14A} \tau_{14} U_{1,10} + A_{14B} (1-\rho_{14}) [\tau_{14} (H_{14m14})_B U_{2,10} + \tau_{16} (H_{14m16})_B U_{3,10}] - A_{14B} \tau_{14} U_{2,10}\}_{S,L}$$

$$D_{14,11} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14} [(1-\rho_{14})(H_{14j14})_A - 1] U_{1,11} + A_{14B} (1-\rho_{14}) [\epsilon_{11} (H_{14m11})_B + \tau_{14} (H_{14m14})_B U_{2,11} + \tau_{16} (H_{14m16})_B U_{3,11}] - A_{14B} \tau_{14} U_{2,11}\}_{S,L}$$

$$D_{14,12} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} (1-\rho_{14}) [\epsilon_{12} (H_{14j12})_A + \tau_{14} (H_{14j14})_A U_{1,12}] - A_{14A} \tau_{14} U_{1,2} + A_{14B} (1-\rho_{14}) [\epsilon_{12} (H_{14m12})_B + \tau_{14} (H_{14m14})_B U_{2,12} + \tau_{16} (H_{14m16})_B U_{3,12}] - A_{14B} \tau_{14} U_{2,12}\}_{S,L}$$

$$D_{14,13} = \left(\frac{\sigma}{M_{14}}\right)\{A_{14A} \tau_{14} [(1-\rho_{14})(H_{14j14})_A - 1] U_{1,13} + A_{14B} (1-\rho_{14}) [\tau_{14} (H_{14m14})_B U_{2,13} + \tau_{16} (H_{14m16})_B U_{3,13}] - A_{14B} \tau_{14} U_{2,13}\}_{S,L}$$

$$D_{14,14} = \left(\frac{\sigma}{M_{14}}\right) \{A_{14A} [(1-\rho_{14}) (H_{14j14})_A - 1] (\epsilon_{14} + \tau_{14} U_{1,14}) + A_{14B} (1-\rho_{14})$$

$$[\epsilon_{14} (H_{14m14})_B + \tau_{14} (H_{14m14})_B U_{2,14} + \tau_{16} (H_{14m16})_B U_{3,14}] - A_{14B}$$

$$(\epsilon_{14} + \tau_{14} U_{2,14})\}_{S,L}$$

$$(B_1)_{S,L} = \left(\frac{1}{M_1}\right) [A_1 (1-\rho_1) \tau_{13} (H_{1j13})_D U_{8,15}]_{S,L}$$

$$(B_2)_{S,L} = \left(\frac{1}{M_2}\right) [A_{2D} (1-\rho_2) \tau_{13} (H_{2j13})_D U_{8,15} + A_{2F} (1-\rho_2) \{\tau_{13} (H_{2m13})_F U_{7,15}$$

$$+ \tau_{17} (H_{2m17})_F U_{6,15}\}]_{S,L}$$

$$(B_4)_{S,L} = \left(\frac{1}{M_4}\right) [A_4 (1-\rho_4) \{\tau_{13} (H_{4m13})_F U_{7,15} + \tau_{17} (H_{4m17})_F U_{6,15}\}]_{S,L}$$

$$(B_5)_{S,L} = \left(\frac{1}{M_5}\right) [A_5 (1-\rho_5) \{\tau_{16} (H_{5j16})_c U_{4,15} + \tau_{17} (H_{5j17})_c U_{5,15}\}]_{S,L}$$

$$(B_6)_{S,L} = \left(\frac{1}{M_6}\right) [A_6 (1-\rho_6) \{\tau_{16} (H_{6j16})_c U_{4,15} + \tau_{17} (H_{6j17})_c U_{5,15}\}]_{S,L}$$

$$(B_8)_{S,L} = \left(\frac{1}{M_8}\right) [A_8 (1-\rho_8) \{\tau_{16} (H_{8j16})_c U_{4,15} + \tau_{17} (H_{8j17})_c U_{5,15}\}]_{S,L}$$

$$(B_9)_{S,L} = \left(\frac{1}{M_9}\right) [A_{9B} (1-\rho_9) \{\tau_{14} (H_{9m14})_B U_{2,15} + \tau_{16} (H_{9m16})_B U_{3,15}\} + A_{9c} (1-\rho_9)$$

$$\{\tau_{16} (H_{9j16})_c U_{4,15} + \tau_{17} (H_{9j17})_c U_{5,15}\}]_{S,L}$$

$$(B_{10})_{S,L} = \left(\frac{1}{M_{10}}\right) [A_{10} (1-\rho_{10}) \{\tau_{14} (H_{10j14})_A U_{1,15} + \tau_{15} (H_{10j15})_A\}]_{S,L}$$

$$(B_{11})_{S,L} = \left(\frac{1}{M_{11}}\right) [A_{11} (1-\rho_{11}) \{\tau_{14} (H_{11m14})_B U_{2,15} + \tau_{16} (H_{11m16})_B U_{3,15}\}]_{S,L}$$

$$(B_{12})_{S,L} = \left(\frac{1}{M_{12}}\right) [A_{12A} (1-\rho_{12}) \{ \tau_{14} (H_{12j14})_A U_{1,15} + \tau_{15} (H_{12j15})_A \} + A_{12B} (1-\rho_{12}) \\ \{ \tau_{14} (H_{12ml4})_B U_{2,15} + \tau_{16} (H_{12ml6})_B U_{3,15} \}]_{S,L}$$

$$(B_{13})_{S,L} = \left(\frac{1}{M_{13}}\right) [A_{13D} \tau_{13} \{ (1-\rho_{13}) (H_{13j13})_D - 1 \} U_{8,15} + A_{13F} \tau_{13} \{ (1-\rho_{13}) \\ (H_{13ml3})_F - 1 \} U_{7,15} + A_{13F} (1-\rho_{13}) \tau_{17} (H_{13ml7})_F U_{6,15}]_{S,L}$$

$$(B_{14})_{S,L} = \left(\frac{1}{M_{14}}\right) [A_{14A} \tau_{14} \{ (1-\rho_{14}) (H_{14j14})_A - 1 \} U_{1,5} + A_{14A} (1-\rho_{14}) \tau_{15} \\ (H_{14j15})_A + A_{14B} \tau_{14} \{ (1-\rho_{14}) (H_{14ml4})_B - 1 \} U_{2,15} + A_{14B} \tau_{16} \\ (1-\rho_{14}) (H_{14ml6})_B U_{3,15}]_{S,L}$$

For the total channel,

$$(B_i)_{S,L} \quad B_i$$

$$\tau_{13} = \tau_{14} = 1 ,$$

$$\epsilon_{13} = \epsilon_{14} = 0 .$$

Scanner Model Equations

Total Channel

$$\dot{T}_i = A_{ij} T_j + D_{ij} T_j^4 + B_{Qi} \dot{Q} + B_i E + b + e$$

SW/LW Channels

$$\dot{T}_i = A_{ij} T_j + D_{ij} T_j^4 + B_{Qi} \dot{Q} + (B_i)_S E_S + (B_i)_L E_L + b + e$$

Scanner (using Enclosure Coupling Method)

$$E = \begin{bmatrix} E^1 \\ E^2 \\ E^3 \\ E^4 \\ E^5 \\ E^6 \end{bmatrix}$$

where $m = 6$

For each cavity (or enclosure in the scanner model)

$$E^1 = \begin{bmatrix} E^1_1 \\ E^1_2 \\ E^1_3 \\ E^1_4 \end{bmatrix} \quad E^2 = \begin{bmatrix} E^2_1 \\ E^2_2 \\ E^2_3 \\ E^2_4 \\ E^2_5 \end{bmatrix} \quad E^3 = \begin{bmatrix} E^3_1 \\ E^3_2 \\ E^3_3 \\ E^3_4 \\ E^3_5 \\ E^3_6 \end{bmatrix} \quad E^4 = \begin{bmatrix} E^4_1 \\ E^4_2 \\ E^4_3 \\ E^4_4 \end{bmatrix} \quad E^5 = \begin{bmatrix} E^5_1 \\ E^5_2 \\ E^5_3 \end{bmatrix} \quad E^6 = \begin{bmatrix} E^6_1 \\ E^6_2 \\ E^6_3 \end{bmatrix}$$

where $E^1_1 = E^1_{15}$, $E^1_2 = E^1_{10}$, $E^1_3 = E^1_{14}$, $E^1_4 = E^1_{12}$

$$E^2_1 = E^2_{14}$$
 , $E^2_2 = E^2_{11}$, $E^2_3 = E^2_{k6}$, $E^2_4 = E^2_9$, $E^2_5 = E^2_{12}$

$$E^3_1 = E^3_{10}$$
 , $E^3_2 = E^3_8$, $E^3_3 = E^3_6$, $E^3_4 = E^3_5$, $E^3_5 = E^3_{17}$, $E^3_6 = E^3_9$

$$E^4_1 = E^4_{17}$$
 , $E^4_2 = E^4_4$, $E^4_3 = E^4_2$, $E^4_4 = E^4_{13}$

$$E^5_1 = E^5_{13}$$
 , $E^5_2 = E^5_2$, $E^5_3 = E^5_1$

$$E^6_1 = E^6_3$$
 , $E^6_2 = E^6_2$, $E^6_3 = E^6_7$

$$\tilde{E}^1 = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \end{bmatrix} E^2$$

$$\tilde{E}^2 = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} E^1 + \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} E^3$$

$$\tilde{E}^3 = \begin{bmatrix} 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \end{bmatrix} E^2 + \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} E^4$$

$$\tilde{E}^4 = \begin{bmatrix} 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} E^3 + \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \\ 1 & 0 & 0 \end{bmatrix} E^5$$

$$\tilde{E}^5 = \begin{bmatrix} 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} E^4, \quad \tilde{E}^6 = 0$$

Thus D matrix is formed by the above \tilde{E} s.

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1	2	3	4	5	6
$A_{11} A_{12} A_{13} A_{14}$ $A_{21} A_{22} A_{23} A_{24}$ $A_{31} A_{32} A_{33} A_{34}$ $A_{41} A_{42} A_{43} A_{44}$					
	$A_{11} A_{12} A_{13} A_{14} A_{15}$ $A_{21} A_{22} A_{23} A_{24} A_{25}$ $A_{31} A_{32} A_{33} A_{34} A_{35}$ $A_{41} A_{42} A_{43} A_{44} A_{45}$ $A_{51} A_{52} A_{53} A_{54} A_{55}$				
		$A_{11} A_{12} A_{13} A_{14} A_{15} A_{16}$ $A_{21} A_{22} A_{23} A_{24} A_{25} A_{26}$ $A_{31} A_{32} A_{33} A_{34} A_{35} A_{36}$ $A_{41} A_{42} A_{43} A_{44} A_{45} A_{46}$ $A_{51} A_{52} A_{53} A_{54} A_{55} A_{56}$ $A_{61} A_{62} A_{63} A_{64} A_{65} A_{66}$			
			$A_{11} A_{12} A_{13} A_{14}$ $A_{21} A_{22} A_{23} A_{24}$ $A_{31} A_{32} A_{33} A_{34}$ $A_{41} A_{42} A_{43} A_{44}$		
				$A_{11} A_{12} A_{13}$ $A_{21} A_{22} A_{23}$ $A_{31} A_{32} A_{33}$	
					$A_{11} A_{12} A_{13}$ $A_{21} A_{22} A_{23}$ $A_{31} A_{32} A_{33}$

$$A = (I - d_E)^{-1} E \epsilon$$

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APPENDIX E

ERBE SOLAR MONITOR THERMAL MODEL

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SOLAR MONITOR MODEL

As described in the non-scanner and scanner modeling efforts, the same notations are used for constants and variables in developing the energy balance equation of the solar monitor.

To avoid the burden to write the same equations as used in developing the other models, the same variable name in the equation such as R_{ij} is here simply used.

A general form of the equation for the solar monitor which is divided into 19 nodes is presented here as

$$\dot{T} = AT + DT^4 + B_Q \dot{Q} + BE + b.$$

Now the computational algorithm for the constants of the above equation has been developed by using the enclosure coupling technique. It is available and can be used for all 8 non-scanner, scanner and solar monitor models.

The following constants, A_{ij} , represent the coefficients of thermal conduction terms in the energy balance equations.

$$A_{11} = -1/(R_{13}M_1)$$

$$A_{12} = 0$$

$$A_{13} = 1/(R_{13}M_1)$$

$$A_{21} = 0$$

$$A_{22} = -1/(R_{23}M_2)$$

$$A_{23} = 1/(R_{23}M_2)$$

	1	2	3	4	5	6
$C =$	$\begin{bmatrix} c_{11} & c_{12} & c_{13} & c_{14} \\ c_{21} & c_{22} & c_{23} & c_{24} \\ c_{31} & c_{32} & c_{33} & c_{34} \\ c_{41} & c_{42} & c_{43} & c_{44} \end{bmatrix}$					
		$\begin{bmatrix} c_{11} & c_{12} & c_{13} & c_{14} & c_{15} \\ c_{21} & c_{22} & c_{23} & c_{24} & c_{25} \\ c_{31} & c_{32} & c_{33} & c_{34} & c_{35} \\ c_{41} & c_{42} & c_{43} & c_{44} & c_{45} \\ c_{51} & c_{52} & c_{53} & c_{54} & c_{55} \end{bmatrix}$				
			$\begin{bmatrix} c_{11} & c_{12} & c_{13} & c_{14} & c_{15} & c_{16} \\ c_{21} & c_{22} & c_{23} & c_{24} & c_{25} & c_{26} \\ c_{31} & c_{32} & c_{33} & c_{34} & c_{35} & c_{36} \\ c_{41} & c_{42} & c_{43} & c_{44} & c_{45} & c_{46} \\ c_{51} & c_{52} & c_{53} & c_{54} & c_{55} & c_{56} \\ c_{61} & c_{62} & c_{63} & c_{64} & c_{65} & c_{66} \end{bmatrix}$			
				$\begin{bmatrix} c_{11} & c_{12} & c_{13} & c_{14} \\ c_{21} & c_{22} & c_{23} & c_{24} \\ c_{31} & c_{32} & c_{33} & c_{34} \\ c_{41} & c_{42} & c_{43} & c_{44} \end{bmatrix}$		
				$\begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix}$		
					$\begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{21} & c_{22} & c_{23} \\ c_{31} & c_{32} & c_{33} \end{bmatrix}$	

$$C = (I - \rho^d E)^{-1} E \tau.$$

$$\Phi = [(I - \rho - \tau D)(I - CD)^{1-} A - \epsilon] \Omega + [(I - \rho - \tau D)(I - CD)^{1-} C - \tau] F S$$

$$A_{31} = 1/(R_{31}M_3)$$

$$A_{32} = 1/(R_{32}M_3)$$

$$A_{33} = -(\frac{1}{R_{31}} + \frac{1}{R_{32}} + \frac{1}{R_{34}})/M_3$$

$$A_{34} = 1/(R_{34}M_3)$$

$$A_{43} = 1/(R_{43}M_4)$$

$$A_{44} = -(\frac{1}{R_{43}} + \frac{1}{R_{45}} + \frac{1}{R_{47}} + \frac{1}{R_{4-19}})/M_4$$

$$A_{45} = 1/(R_{45}M_4)$$

$$A_{47} = 1/(R_{47}M_4)$$

$$A_{4-19} = 1/(R_{4-19}M_4)$$

$$A_{54} = 1/(R_{54}M_5)$$

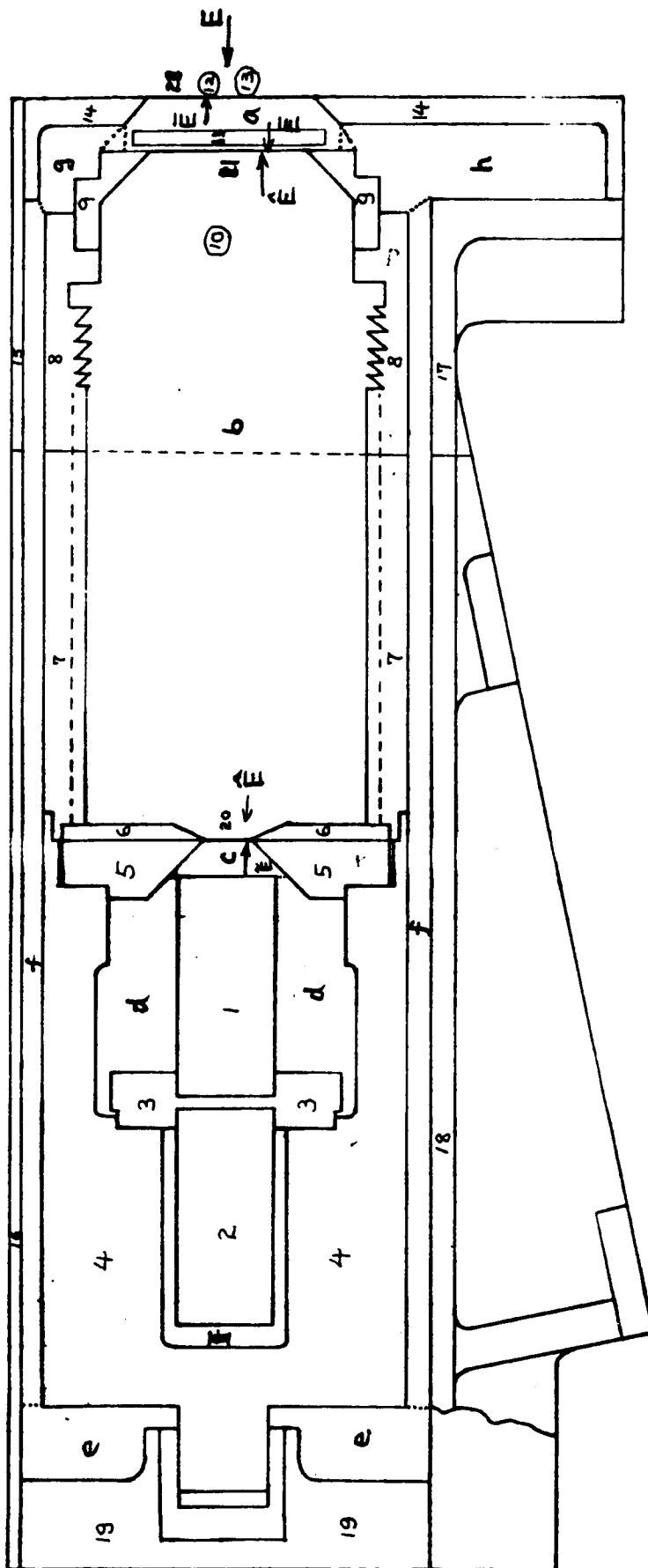
$$A_{55} = -(\frac{1}{R_{54}} + \frac{1}{R_{56}})/M_5$$

$$A_{56} = 1/(R_{56}M_5)$$

$$A_{65} = 1/(R_{65}M_6)$$

$$A_{66} = -(\frac{1}{R_{65}} + \frac{1}{R_{67}})/M_6$$

$$A_{67} = 1/(R_{67}M_6)$$



SOLAR MONITOR MODEL

1. Active Cavity
2. Reference Cavity
3. Button, Duplex Cone Cavities
4. Heat Sink
5. Aperture, Primary
6. Shield, Aperture
7. Body (Rear Barrel)
8. Body (Front Barrel)
9. Aperture, Secondary (or FOV Limiter)
10. Motor Housing
11. Chopper Blade
12. Chopper Pendulum
13. Motor, Chopper
14. FOV Limiter
15. Front Top Housing Cover
16. Rear Top Housing Cover
17. Front Housing
18. Rear Housing
19. Holder, Heat Sink
20. Fictitious Surface
- p. Temperature Probe
- E. Total Incident Radiation
- E_a . Irradiance in enclosure a
- E_b . Irradiance in enclosure b
- E_c . Rate of Incident Radiation From a Neighboring Surface
- a-h. Cavities Within The Sensor, i.e., Enclosures or Spaces Between Sensor Nodes

$$A_{74} = 1/(R_{74}M_7)$$

$$A_{76} = 1/(R_{76}M_7)$$

$$A_{77} = -(\frac{1}{R_{74}} + \frac{1}{R_{76}} + \frac{1}{R_{78}})/M_7$$

$$A_{78} = 1/(R_{78}M_7)$$

$$A_{87} = 1/(R_{87}M_8)$$

$$A_{88} = 1(\frac{1}{R_{87}} + \frac{1}{R_{89}} + \frac{1}{R_{8-10}})/M_8$$

$$A_{89} = 1/(R_{89}M_8)$$

$$A_{8-10} = 1/(R_{8-10}M_8)$$

$$A_{9-8} = 1/(R_{9-8}M_9)$$

$$A_{9-9} = -\frac{1}{R_{9-8}M_9}$$

$$A_{10-8} = 1/(R_{10-8}M_{10})$$

$$A_{10-10} = -1/(R_{10-8}M_{10}) - 1/(R_{10-12}M_{10}) - 1/(R_{10-13}M_{10})$$

$$A_{10-12} = 1/(R_{10-12}M_{10})$$

$$A_{10-13} = 1/(R_{10-13}M_{10})$$

$$A_{11-12} = 1/(R_{11-12}M_{11})$$

$$A_{11-11} = -1/(R_{11-12}M_{11})$$

$$A_{12-11} = 1/(R_{12-11}M_{12})$$

$$A_{12-12} = -(1/R_{12-11} + 1/R_{12-13})/M_{12}$$

$$A_{12-13} = 1/(R_{12-11}M_{12})$$

$$A_{13-12} = 1/(R_{13-12}M_{12})$$

$$A_{13-13} = -(1/R_{13-12} + 1/R_{13-10})/M_{12}$$

$$A_{13-10} = 1/(R_{13-10}M_{13})$$

$$A_{14-14} = -(1/R_{14-15} + 1/R_{14-17})/M_{14}$$

$$A_{14-15} = 1/(R_{14-15}M_{14})$$

$$A_{14-17} = 1/(R_{14-17}M_{14})$$

$$A_{15-15} = -1/(R_{15-16}M_{15}) - 1/(R_{15-14}R_{15})$$

$$A_{15-16} = 1/(R_{15-16}M_{15})$$

$$A_{15-14} = 1/(R_{15-14}M_{15})$$

$$A_{16-15} = 1/(R_{16-15}M_{16})$$

$$A_{16-16} = -(1/R_{16-15} + 1/R_{16-19})/M_{16}$$

$$A_{16-19} = 1/(R_{16-19}M_{16})$$

$$A_{17-14} = 1/(R_{17-14} M_{17})$$

$$A_{17-17} = -(1/R_{17-14} + 1/R_{17-18})/M_{17}$$

$$A_{17-18} = 1/(R_{17-18} M_{17})$$

$$A_{18-17} = 1/(R_{18-17} M_{18})$$

$$A_{18-18} = -(1/R_{18-17} + 1/R_{18-19})/M_{18}$$

$$A_{18-19} = 1/(R_{18-19} M_{18})$$

$$A_{19-4} = 1/(R_{19-4} M_{19})$$

$$A_{19-16} = 1/(R_{19-16} M_{19})$$

$$A_{19-18} = 1/(R_{19-18} M_{19})$$

$$A_{19-19} = -(1/R_{19-4} + 1/R_{19-16} + 1/R_{19-18})/M_{19}$$

For the radiative interaction term, considering the specular reflection,

$$\rho = \rho^d + \rho^s \quad (1)$$

$$\epsilon + \rho + \tau = 1 \quad (2)$$

$$M_k^d = \epsilon_k \sigma T_k^4 + \rho_k^d E_k + \tau_k \bar{E}_k \quad (3)$$

$$E_k = \sum_j E_{k-j} M_j^d \quad (4)$$

The energy balance equation is now

$$\phi_k = E_k = M_k \quad (5)$$

where

$$M_k = S_k E_k + M_k^d .$$

Thus

$$\kappa_k = E_k = \epsilon_k \Omega_k - \rho_k^d E_k - \rho_k^s E_k - \tau_k \bar{E}_k \quad (6)$$

where

$$\Omega_k = \sigma T_k^4 .$$

From (3) and (4),

$$M_j^d = \sum_{i=1}^n (\delta_{ij} - \rho_j^d E_{j-1})^{-1} (\epsilon_i \Omega_i + \tau_i \bar{E}_i) \quad (7)$$

From (7) through (4)

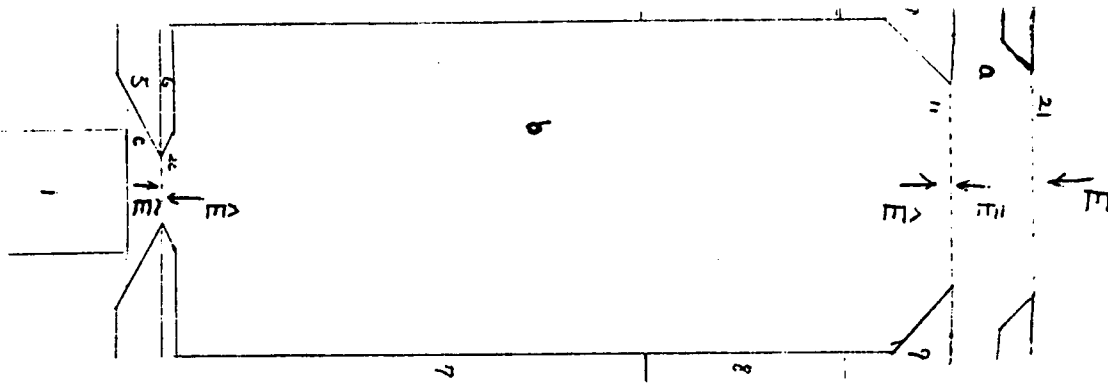
$$\begin{aligned} E_k &= \sum_j E_{k-j} \sum_i P_{j-i} (\epsilon_i \Omega_i + \tau_i \bar{E}_i) \\ &= \sum_i (\epsilon_i \Omega_i + \tau_i \bar{E}_i) \sum_{j=1}^n E_{k-j} P_{j-i} \end{aligned} \quad (8)$$

Equation (6) becomes

$$\begin{aligned} \phi_k &= (1 - \rho_k) E_k - \epsilon_k \Omega_k - \tau_k \bar{E}_k \\ &= \sum_i [(1 - \rho_k) \epsilon_i \sum_j E_{k-j} P_{j-i} - \epsilon_k \delta_{ki}] \Omega_i \\ &\quad + \sum_i [(1 - \rho_k) \tau_i \sum_j E_{k-j} P_{j-i} - \tau_k \delta_{ki}] \bar{E}_i \end{aligned} \quad (9)$$

Let

$$\begin{aligned} H_{kji} &= \sum_j E_{k-j} P_{j-i} \\ H_{nm\ell} &= \sum_m E_{n-m} P_{m-\ell} \end{aligned}$$



From the above drawing, the enclosures, a, b and c, are found. For each enclosure, the incoming radiance to a certain surface can be described by Equation (8).

Choose the enclosure b and c and couple them together, then one can define the transmitted radiance through the filtering surface.

First,

			Index
enclosure b	-	elements 6,7,8,9,11,20	i,j,k
enclosure c	-	elements 1,5,20	l,m,n

From Equation (8)

$$E_k = \sum_i \epsilon_i \Omega_i \sum_j E_{k-j} P_{j-i} + \sum_i \tau_i \bar{E}_i \sum_j E_{k-j} P_{j-i}$$

where in this equation i, j and k are dummy index

or

$$E_k = \sum_i \epsilon_i \Omega_i H_{kji} + \sum_i \tau_i \bar{E}_i H_{kji}$$

For the enclosure c, Equation (8) becomes

$$\tilde{E}_{20} = \left(\sum_l \epsilon_l \Omega_l H_{20ml} \right)_c + \tau_{20} \hat{E}_{20} (H_{20m20})_c \quad (10)$$

where l, m and n = 1, 5 and 20.

Likewise, for the enclosure b,

$$\hat{E}_{20} = \left(\sum_i \epsilon_i \Omega_i H_{20ji} \right)_b + \tau_{20} \tilde{E}_{20} (H_{20j20})_b + \tau_{11} \bar{E}_{11} (H_{20j11})_b \quad (11)$$

where i and $j = 6, 7, 8, 9, 11$ and 20 .

Combining (10) and (11) to obtain \tilde{E}_{20} and \hat{E}_{20} as a function of \bar{E}_{11} , we have

$$\begin{aligned} \tilde{E}_{20} = G_1 [& \left(\sum_{\ell} \epsilon_{\ell} \Omega_{\ell} H_{20m\ell} \right)_c + \tau_{20} (H_{20m20})_c \left(\sum_i \epsilon_i \Omega_i H_{20ji} \right)_b \\ & + \tau_{11} \tau_{20} (H_{20m20})_c (H_{20j11})_b \bar{E}_{11}] \end{aligned} \quad (12)$$

and

$$\begin{aligned} \hat{E}_{20} = G_1 [& \left(\sum_i \epsilon_i \Omega_i H_{20ji} \right)_n + \tau_{20} (H_{20j20})_b \left(\sum_{\ell} \epsilon_{\ell} \Omega_{\ell} H_{20m\ell} \right)_c \\ & + \tau_{11} (H_{20j11})_b \bar{E}_{11}] \end{aligned} \quad (13)$$

where

$$G_1 = [1 - \tau_{20}^2 (H_{20j20})_b (H_{20m20})_c]^{-1}.$$

The radiation energy balances in the enclosures b and c are

$$\begin{aligned} \phi_k \Big|_b = & \sum_i [(1 - \rho_k) \epsilon_i H_{kji} - \epsilon_k \delta_{ki}] \Omega_i \\ & + \sum_i [(1 - \rho_k) \tau_i H_{kji} - \tau_k \delta_{ki}] (\tilde{E}_{i=20} \text{ or } \bar{E}_{i=11}) \end{aligned} \quad (14)$$

and

$$\begin{aligned} \phi_n \Big|_b &= \sum_{\ell} [(1 - \rho_n) e_{\ell n m \ell}^H - e_{n n \ell}^{\delta}] \Omega_{\ell} \\ &+ \sum_{\ell} [(1 - \rho_n) \tau_{\ell n m \ell}^H - \tau_{n n \ell}^{\delta}] \hat{E}_{\ell} \end{aligned} \quad (15)$$

Second,

		Index
enclosure d	- elements 11,14,21	ℓ, m, n
enclosure b	- elements 6,7,8,9,11,20	i, j, k

From Equation (8)

$$\hat{E}_{11} = \left(\sum_i \epsilon_{i i}^{\Omega} H_{11 j i} \right)_b + \tau_{11} (H_{11 j 11})_b \bar{E}_{11} + \tau_{20} (H_{11 j 20})_b \tilde{E}_{20} \quad (16)$$

and

$$\bar{E}_{11} = \left(\sum_{\ell} \epsilon_{\ell \ell}^{\Omega} H_{11 m \ell} \right)_a + \tau_{11} (H_{11 m 11})_a \hat{E}_{11} + \tau_{21} (H_{11 m 21})_a E \quad (17)$$

Substituting (12) into (16), Equation (16) becomes

$$\begin{aligned} \hat{E}_{11} &= \left(\sum_i \epsilon_{i i}^{\Omega} H_{11 j i} \right)_b + \tau_{20} G_1 (H_{11 j 20})_b \left(\sum_{\ell} \epsilon_{\ell \ell}^{\Omega} H_{20 m \ell} \right)_c \\ &+ \tau_{20}^2 G_1 (H_{11 j 20})_b (H_{20 m 20})_c \left(\sum_i \epsilon_{i i}^{\Omega} H_{20 j i} \right)_b \\ &+ [\tau_{11} (H_{11 j 11})_b + \tau_{11} \tau_{20}^2 G_1 (H_{11 j 20})_b (H_{20 m 20})_c (H_{2-j 11})_b] \bar{E}_{11} \end{aligned} \quad (18)$$

Substituting (18) into (17)

$$\begin{aligned} \bar{E}_{11} &= G_2 \left[\left(\sum_{\ell} \epsilon_{\ell \ell}^{\Omega} H_{n m \ell} \right)_a + \tau_{21} (H_{11 m 21})_a E + \tau_{11} (H_{11 m 11})_a \left(\sum_i \epsilon_{i i}^{\Omega} H_{11 j i} \right)_b \right. \\ &+ \tau_{11} \tau_{20} G_1 (H_{11 m 11})_a (H_{11 j 20})_b \left(\sum_{\ell} \epsilon_{\ell \ell}^{\Omega} H_{20 m \ell} \right)_c \\ &\left. + \tau_{11} \tau_{20}^2 G_1 (H_{11 m 11})_a (H_{11 j 20})_b (H_{20 m 20})_c \left(\sum_i \epsilon_{i i}^{\Omega} H_{20 j i} \right)_b \right] \end{aligned} \quad (19)$$

where

$$G_2 = [1 - \tau_{11}^2 S_1 (H_{11m11})_a]^{-1}$$

$$S_1 = (H_{11j11})_b + \tau_{20}^2 G_1 (H_{11j20})_b (H_{20m20})_c (H_{20j11})_b$$

On the other hand, substitution (17) into (18)

$$\begin{aligned} \hat{E}_{11} = & G_2 [(\sum_i \epsilon_i \Omega_i H_{11ji})_b + \tau_{20} G_1 (H_{11j20})_b (\sum_\ell \epsilon_\ell \Omega_\ell H_{20m\ell})_c \\ & + \tau_{20}^2 G_1 (H_{11j20})_b (H_{20m20})_c (\sum_i \epsilon_i \Omega_i H_{20ji})_b \\ & + \tau_{11} S_1 (\sum_\ell \epsilon_\ell \Omega_\ell H_{11m\ell})_a + \tau_{11} \tau_{21} S_1 (H_{11m21})_a E] \end{aligned} \quad (20)$$

For the enclosure a: nodes: 11,14,21

$$\begin{aligned} \phi_{11} \Big|_a = & [(1 - \rho_{11}) \epsilon_{11} \overset{0 \text{ for open position}}{\cancel{(H_{11m11})_a}} - \epsilon_{11}] \Omega_{11} + (1 - \rho_{14}) \epsilon_{14} (H_{11m14})_a \Omega_{14} \\ & + [1 - \rho_{11}) \tau_{21} (H_{11m21})_a E + [(1 - \rho_{11}) \tau_{11} \cancel{(H_{11m11})_a} - \tau_{11}] \hat{E}_{11} \\ & \qquad \qquad \qquad \searrow \downarrow \\ & \qquad \qquad \qquad 0 \text{ for closed position} \end{aligned}$$

$$\begin{aligned} \phi_{14} \Big|_a = & (1 - \rho_{14}) \overset{0 \text{ for open position}}{\cancel{\epsilon_{11}}} (H_{14m11})_a \Omega_{11} + [(1 - \rho_{14}) G_{14} (H_{14m14})_a - \epsilon_{14}] \Omega_{14} \\ & + (1 - \rho_{14}) \tau_{11} (H_{14m11})_a \hat{E}_{11} + (1 - \rho_{14}) \tau_{21} (H_{14m21})_a E \end{aligned}$$

For the enclosure b: nodes: 6,7,8,9,11,20

$$\begin{aligned} \phi_6|_b &= [(1 - \rho_6) \epsilon_6 (H_{6j6})_b - \epsilon_6] \Omega_6 + (1 - \rho_6) \epsilon_7 (H_{6j7})_b \Omega_7 \\ &+ (1 - \rho_6) \epsilon_8 (H_{6j8})_b \Omega_8 + (1 - \rho_6) \epsilon_9 (H_{6j9})_b \Omega_9 + (1 - \rho_6) \overset{0 \text{ for open position}}{\epsilon_{11} (H_{6j11})_b} \Omega_{11} \\ &+ (1 - \rho_6) \overset{0 \text{ for closed position}}{\epsilon_{11} (H_{6j11})_b} \bar{E}_{11} + (1 - \rho_6) \tau_{20} (H_{6j20})_b \tilde{E}_{20} \end{aligned}$$

$$\begin{aligned} \phi_7|_b &= (1 - \rho_7) \epsilon_6 (H_{7j6})_b \Omega_6 + [(1 - \rho_7) \epsilon_7 (H_{7j7})_b - \epsilon_7] \Omega_7 \\ &+ (1 - \rho_7) \epsilon_8 (H_{7j8})_b \Omega_8 + (1 - \rho_7) \epsilon_9 (H_{7j9})_b \Omega_9 + (1 - \rho_7) \overset{0 \text{ for open position}}{\epsilon_{11} (H_{7j11})_b} \Omega_{11} \\ &+ (1 - \rho_7) \underset{0 \text{ for closed position}}{\epsilon_{11} (H_{7j11})_b} \bar{E}_{11} + (1 - \rho_7) \tau_{20} (H_{7j20})_b \tilde{E}_{20} \end{aligned}$$

$$\phi_8|_b = (1 - \rho_8) \epsilon_6 (H_{8j6})_b \Omega_6 + (1 - \rho_8) \epsilon_7 (H_{8j7})_b \Omega_7 + [(1 - \rho_8) \epsilon_8$$

$$(H_{8j8})_b - \epsilon_8] \Omega_8$$

$$\begin{aligned} &+ (1 - \rho_8) \epsilon_9 (H_{8j9})_b \Omega_9 + (1 - \rho_8) \overset{0 \text{ for open position}}{\epsilon_{11} (H_{8j11})_b} \Omega_{11} \\ &+ (1 - \rho_8) \underset{0 \text{ for closed position}}{\epsilon_{11} (H_{8j11})_b} \bar{E}_{11} + (1 - \rho_8) \tau_{20} (H_{8j20})_b \tilde{E}_{20} \end{aligned}$$

$$\begin{aligned}
\phi_9 \Big|_b &= (1 - \rho_9) \epsilon_6^{(H_{9j6})_b} \Omega_6 + (1 - \rho_9) \epsilon_7^{(H_{9j7})_b} \Omega_7 + (1 - \rho_9) \epsilon_8^{(H_{9j8})_b} \Omega_8 \\
&+ [(1 - \rho_9) \epsilon_9^{(H_{9j9})_b} - \epsilon_9] \Omega_9 + (1 - \rho_9) \overset{0 \text{ for open position}}{\epsilon_{11}^{(H_{9j11})_b}} \Omega_{11} \\
&+ (1 - \rho_9) \underset{0 \text{ for closed position}}{\tau_{11}^{(H_{9j11})_b}} \bar{E}_{11} + (1 - \rho_9) \tau_{20}^{(H_{9j20})_b} \tilde{E}_{20}
\end{aligned}$$

$$\begin{aligned}
\phi_{11} \Big|_b &= (1 - \rho_{11}) \epsilon_6^{(H_{11j6})_b} \Omega_6 + (1 - \rho_{11}) \epsilon_7^{(H_{11j7})_b} \Omega_7 + (1 - \rho_{11}) \epsilon_8^{(H_{11j8})_b} \Omega_8 \\
&+ (1 - \rho_{11}) \epsilon_9^{(H_{11j9})_b} \Omega_9 + [(1 - \rho_{11}) \overset{0 \text{ for open position}}{\epsilon_{11}^{(H_{11j11})_b}} - \epsilon_{11}] \Omega_{11} \\
&+ (1 - \rho_{11}) \tau_{20}^{(H_{11j20})_b} \tilde{E}_{20} + [(1 - \rho_{11}) \underset{0 \text{ for closed position}}{\tau_{11}^{(H_{11j11})_b}} - \tau_{11}] \bar{E}_{11}
\end{aligned}$$

$$\begin{aligned}
\phi_{20} \Big|_b &= (1 - \rho_{20}) \epsilon_6^{(H_{20j6})_b} \Omega_6 + (1 - \rho_{20}) \epsilon_7^{(H_{20j7})_b} \Omega_7 \\
&+ (1 - \rho_{20}) \epsilon_8^{(H_{20j8})_b} \Omega_8 + (1 - \rho_{20}) \epsilon_9^{(H_{20j9})_b} \Omega_9 + (1 - \rho_{20}) \overset{0 \text{ for open position}}{\epsilon_{11}^{(H_{20j11})_b}} \Omega_{11} \\
&+ (1 - \rho_{20}) \tau_{11}^{(H_{20j11})_b} \bar{E}_{11} + [(1 - \rho_{20}) \tau_{20}^{(H_{20j20})_b} - \tau_{20}] \tilde{E}_{20}
\end{aligned}$$

Enclosure c: Nodes: 1,5,20

$$\phi_1 \Big|_c = [(1 - \rho_1) \epsilon_1 (H_{1j1})_c - \epsilon_1] \Omega_1 + (1 - \rho_1) \epsilon_5 (H_{1j5})_c \Omega_5 \\ + (1 - \rho_1) \tau_{20} (H_{1j20})_c \hat{E}_{20}$$

$$\phi_5 \Big|_c = (1 - \rho_5) \epsilon_1 (H_{5j1})_c \Omega_1 + [(1 - \rho_5) \epsilon_5 (H_{5h5})_c - \epsilon_5] \Omega_5 \\ + (1 - \rho_5) \tau_{20} (H_{5j20})_c \hat{E}_{20}$$

$$\phi_{20} \Big|_c = (1 - \rho_{20}) \epsilon_1 (H_{20j1})_c \Omega_1 + (1 - \rho_{20}) \epsilon_5 (H_{20h5})_c \Omega_5 \\ + [(1 - \rho_{20}) \tau_{20} (H_{20j20})_c - \tau_{20}] \hat{E}_{20}$$

For the enclosure d: Nodes: 1,j,k = 1,3,4,5

$$\phi_1 \Big|_d = [(1 - \rho_1) \epsilon_1 (H_{1j1})_d - \epsilon_1] \Omega_1 + (1 - \rho_1) \epsilon_3 (H_{1j3})_d \Omega_3 \\ + (1 - \rho_1) \epsilon_4 (H_{1j4})_d \Omega_4 + (1 - \rho_1) \epsilon_5 (H_{1j5})_d \Omega_5$$

$$\phi_3 \Big|_d = (1 - \rho_3) \epsilon_1 (H_{3j1})_d \Omega_1 + [(1 - \rho_3) \epsilon_3 (H_{3j3})_d - \epsilon_3] \Omega_3 \\ + (1 - \rho_3) \epsilon_4 (H_{3j4})_d \Omega_4 + (1 - \rho_3) \epsilon_5 (H_{3j5})_d \Omega_5$$

$$\phi_4 \Big|_d = (1 - \rho_4) \epsilon_1 (H_{4j1})_d \Omega_1 + (1 - \rho_4) \epsilon_3 (H_{4j3})_d \Omega_3 \\ + [(1 - \rho_4) \epsilon_4 (H_{4j4})_d - \epsilon_4] \Omega_4 + (1 - \rho_4) \epsilon_5 (H_{4j5})_d \Omega_5$$

$$\begin{aligned}\phi_5 \Big|_d &= (1-\rho_5) \epsilon_1 (H_{5j1})_d \Omega_1 + (1-\rho_5) \epsilon_3 (H_{5j3})_d \Omega_3 \\ &+ (1-\rho_5) \epsilon_4 (H_{5j4})_d \Omega_4 + [(1-\rho_5) \epsilon_5 (H_{5j5})_d - \epsilon_5] \Omega_5\end{aligned}$$

For the enclosure e: Nodes: 4,16,18,19

$$\begin{aligned}\phi_4 \Big|_e &= [(1-\rho_4) \epsilon_4 (H_{4j4})_e - \epsilon_4] \Omega_4 + (1-\rho_4) \epsilon_{16} (H_{4j16})_e \Omega_{16} \\ &+ (1-\rho_4) \epsilon_{18} (H_{4j18})_e \Omega_{18} + (1-\rho_4) \epsilon_{19} (H_{4j19})_e \Omega_{19} \\ \phi_{16} \Big|_e &= (1-\rho_{16}) \epsilon_4 (H_{16j4})_e \Omega_4 + [(1-\rho_{16}) \epsilon_{16} (H_{16j16})_e - \epsilon_{16}] \Omega_{16} \\ &+ (1-\rho_{16}) \epsilon_{18} (H_{16j18})_e \Omega_{18} + (1-\rho_{16}) \epsilon_{19} (H_{16j19})_e \Omega_{19} \\ \phi_{18} \Big|_e &= (1-\rho_{18}) \epsilon_4 (H_{18j4})_e \Omega_4 + (1-\rho_{18}) \epsilon_{16} (H_{18j16})_e \Omega_{16} \\ &+ [(1-\rho_{18}) \epsilon_{18} (H_{18j18})_e - \epsilon_{18}] \Omega_{18} + (1-\rho_{18}) \epsilon_{19} (H_{18j19})_e \Omega_{19} \\ \phi_{19} \Big|_e &= (1-\rho_{19}) \epsilon_4 (H_{19j4})_e \Omega_4 + (1-\rho_{19}) \epsilon_{16} (H_{19j16})_e \Omega_{16} \\ &+ (1-\rho_{19}) \epsilon_{18} (H_{19j18})_e \Omega_{18} + [(1-\rho_{19}) \epsilon_{19} (H_{19j19})_e - \epsilon_{19}] \Omega_{19}\end{aligned}$$

For the enclosure f: Nodes: 4,7,8,14,15,16,17,18

$$\begin{aligned}\phi_4 \Big|_f &= [(1-\rho_4) \epsilon_4 (H_{4j4})_f - \epsilon_4] \Omega_4 + (1-\rho_4) \epsilon_7 (H_{4j7})_f \Omega_7 \\ &+ (1-\rho_4) \epsilon_8 (H_{4j8})_f \Omega_8 + (1-\rho_4) \epsilon_{14} (H_{4j14})_f \Omega_{14} + (1-\rho_4) \epsilon_{15} (H_{4j15})_f \Omega_{15} \\ &+ (1-\rho_4) \epsilon_{16} (H_{4j16})_f \Omega_{16} + (1-\rho_4) \epsilon_{17} (H_{4j17})_f \Omega_{17} + (1-\rho_4) \epsilon_{18} (H_{4j18})_f \Omega_{18}\end{aligned}$$

$$\begin{aligned}
\phi_7 \Big|_f &= (1-\rho_7) \epsilon_4 (H_{7j4})_f \Omega_4 + [(1-\rho_7) \epsilon_7 (H_{7j7})_f - \epsilon_7] \Omega_7 + (1-\rho_7) \epsilon_8 \\
&\quad (H_{7j8})_j \Omega_8 \\
&+ (1-\rho_7) \epsilon_{14} (H_{7j14})_f \Omega_{14} + (1-\rho_7) \epsilon_{15} (H_{7j15})_f \Omega_{15} + (1-\rho_7) \epsilon_{16} \\
&\quad (H_{7j16})_f \Omega_{16} \\
&+ (1-\rho_7) \epsilon_{17} (H_{7j17})_f \Omega_{17} + (1-\rho_7) \epsilon_{18} (H_{7j18})_f \Omega_{18}
\end{aligned}$$

$$\begin{aligned}
\phi_8 \Big|_f &= (1-\rho_8) \epsilon_4 (H_{8j4})_f \Omega_4 + (1-\rho_8) \epsilon_7 (H_{8j7})_f \Omega_7 + [(1-\rho_8) \epsilon_8 \\
&\quad (H_{8j8})_f - \epsilon_8] \Omega_8 \\
&+ (1-\rho_8) \epsilon_{14} (H_{8j14})_f \Omega_{14} + (1-\rho_8) \epsilon_{15} (H_{8j15})_f \Omega_{15} + (1-\rho_8) \epsilon_{16} \\
&\quad (H_{8j16})_f \Omega_{16} \\
&+ (1-\rho_8) \epsilon_{17} (H_{8j17})_f \Omega_{17} + (1-\rho_8) \epsilon_{18} (H_{8j18})_f \Omega_{18}
\end{aligned}$$

$$\begin{aligned}
\phi_{14} \Big|_f &= (1-\rho_{14}) \epsilon_4 (H_{14j14})_f \Omega_4 + (1-\rho_{14}) \epsilon_7 (H_{14j7})_f \Omega_7 + (1-\rho_{14}) \epsilon_8 \\
&\quad (H_{14j8})_f \Omega_8 \\
&+ [(1-\rho_{14}) \epsilon_{14} (H_{14j14})_f - \epsilon_{14}] \Omega_{14} + (1-\rho_{14}) \epsilon_{15} (H_{14j15})_f \Omega_{15} + \\
&\quad (1-\rho_{14}) \epsilon_{16} (H_{14j16})_f \Omega_{16} \\
&+ (1-\rho_{14}) \epsilon_{17} (H_{14j17})_f \Omega_{17} + (1-\rho_{14}) \epsilon_{18} (H_{14j18})_f \Omega_{18}
\end{aligned}$$

$$\begin{aligned}
\phi_{15} \Big|_f &= (1-\rho_{15}) \epsilon_4 (H_{15j4})_f \Omega_4 + (1-\rho_{15}) \epsilon_7 (H_{15j7})_f \Omega_7 + (1-\rho_{15}) \epsilon_8 \\
&\quad (H_{15j8})_f \Omega_8 \\
&+ (1-\rho_{15}) \epsilon_{14} (H_{15j14})_f \Omega_{14} + [(1-\rho_{15}) \epsilon_{15} (H_{15j15})_f - \epsilon_{15}] \Omega_{15} + \\
&\quad (1-\rho_{15}) \epsilon_{16} (H_{15j16})_f \Omega_{16} \\
&+ (1-\rho_{15}) \epsilon_{17} (H_{15j17})_f \Omega_{17} + (1-\rho_{15}) \epsilon_{18} (H_{15j18})_f \Omega_{18}
\end{aligned}$$

$$\begin{aligned}
\phi_{16} \Big|_f &= (1-\rho_{16}) \epsilon_4 (H_{16j4})_f \Omega_4 + (1-\rho_{16}) \epsilon_7 (H_{16j7})_f \Omega_7 + (1-\rho_{16}) \epsilon_8 \\
&\quad (H_{16j8})_f \Omega_8 \\
&+ (1-\rho_{16}) \epsilon_{14} (H_{16j14})_f \Omega_{14} + (1-\rho_{16}) \epsilon_{15} (H_{16j15})_f \Omega_{15} + [(1-\rho_{16}) \epsilon_{16} \\
&\quad (H_{16j16})_f - \epsilon_{16}] \Omega_{16} \\
&+ (1-\rho_{16}) \epsilon_{17} (H_{16j17})_f \Omega_{17} + (1-\rho_{16}) \epsilon_{18} (H_{16j18})_f \Omega_{18}
\end{aligned}$$

$$\begin{aligned}
\phi_{17} \Big|_f &= (1-\rho_{17}) \epsilon_4 (H_{17j4})_f \Omega_4 + (1-\rho_{17}) \epsilon_7 (H_{17j7})_f \Omega_7 + (1-\rho_{17}) \epsilon_8 \\
&\quad (H_{17j8})_f \Omega_8 \\
&+ (1-\rho_{17}) \epsilon_{14} (H_{17j14})_f \Omega_{14} + (1-\rho_{17}) \epsilon_{15} (H_{17j15})_f \Omega_{15} + (1-\rho_{17}) \epsilon_{16} \\
&\quad (H_{17j16})_f \Omega_{16} \\
&+ [(1-\rho_{17}) \epsilon_{17} (H_{17j17})_f - \epsilon_{17}] \Omega_{17} + (1-\rho_{17}) \epsilon_{18} (H_{17j18})_f \Omega_{18}
\end{aligned}$$

$$\begin{aligned}
\phi_{18} \Big|_f &= (1-\rho_{18}) \epsilon_4 (H_{18j4})_f \Omega_f + (1-\rho_{18}) \epsilon_7 (H_{18j7})_f \Omega_7 + (1-\rho_{18}) \epsilon_8 \\
&\quad (H_{18j18})_f \Omega_8 \\
&+ (1-\rho_{18}) \epsilon_{14} (H_{18j14})_f \Omega_{14} + (1-\rho_{18}) \epsilon_{15} (H_{18j15})_f \Omega_{15} + (1-\rho_{18}) \\
&\quad \epsilon_{16} (H_{18j16})_f \Omega_{16} \\
&+ (1-\rho_{18}) \epsilon_{17} (H_{18j17})_f \Omega_{17} + [(1-\rho_{18}) \epsilon_{18} (H_{18j18})_f - \epsilon_{18}] \Omega_{18}
\end{aligned}$$

For the enclosure g: Nodes: 8,9,14

$$\begin{aligned}
\phi_8 \Big|_g &= [(1-\rho_8) \epsilon_8 (H_{8j8})_g - \epsilon_8] \Omega_8 + (1-\rho_8) \epsilon_9 (H_{8j9})_f \Omega_9 \\
&+ (1-\rho_8) \epsilon_{14} (H_{8j14})_g \Omega_{14}
\end{aligned}$$

$$\begin{aligned}
\phi_9 \Big|_g &= (1-\rho_9) \epsilon_8 (H_{9j8})_g \Omega_8 + [(1-\rho_9) \epsilon_9 (H_{9j9})_g - \epsilon_9] \Omega_9 \\
&+ (1-\rho_9) \epsilon_{14} (H_{9j14})_g \Omega_{14}
\end{aligned}$$

$$\begin{aligned}
\phi_{14} \Big|_g &= (1-\rho_{14}) \epsilon_8 (H_{14j8})_g \Omega_8 + (1-\rho_{14}) \epsilon_9 (H_{14j9})_g \Omega_9 \\
&+ [(1-\rho_{14}) \epsilon_{14} (H_{14j14})_g - \epsilon_{14}] \Omega_{14}
\end{aligned}$$

For the enclosure h: Nodes: 8,9,14,17 - i,j,k

$$\begin{aligned}
\phi_8 \Big|_h &= [(1-\rho_8) \epsilon_8 (H_{8j8})_h - \epsilon_8] \Omega_8 + (1-\rho_8) \epsilon_9 (H_{8j9})_h \Omega_9 \\
&+ (1-\rho_8) \epsilon_{14} (H_{8j14})_h \Omega_{14} + (1-\rho_8) \epsilon_{17} (H_{8j17})_h \Omega_{17}
\end{aligned}$$

$$\begin{aligned}\phi_9 \Big|_h &= (1-\rho_9) \epsilon_8 (H_{9j8})_h \Omega_8 + [(1-\rho_9) \epsilon_9 (H_{9j9})_h - \epsilon_9] \Omega_9 \\ &+ (1-\rho_9) \epsilon_{14} (H_{9j14})_h \Omega_{14} + (1-\rho_9) \epsilon_{17} (H_{9j17})_h \Omega_{17}\end{aligned}$$

$$\begin{aligned}\phi_{14} \Big|_h &= (1-\rho_{14}) \epsilon_8 (H_{14j8})_h \Omega_8 + (1-\rho_{14}) \epsilon_9 (H_{14j9})_h \Omega_9 \\ &+ [(1-\rho_{14}) \epsilon_{14} (H_{14j14})_h - \epsilon_{14}] \Omega_{14} + (1-\rho_{14}) \epsilon_{17} (H_{14j17})_h \Omega_{17}\end{aligned}$$

$$\begin{aligned}\phi_{17} \Big|_h &= (1-\rho_{17}) \epsilon_8 (H_{17j8})_g \Omega_8 + (1-\rho_{17}) \epsilon_9 (H_{17j9})_h \Omega_9 \\ &+ (1-\rho_{17}) \epsilon_{14} (H_{17j14})_h \Omega_{14} + [(1-\rho_{17}) \epsilon_{17} (H_{17j17})_h - \epsilon_{17}] \Omega_{17}\end{aligned}$$

For the enclosure I: Nodes: 2,3,4 - i,j,k

$$\begin{aligned}\phi_2 \Big|_i &= [(1-\rho_2) \epsilon_2 (H_{2j2})_i - \epsilon_2] \Omega_2 + (1-\rho_2) \epsilon_3 (H_{2j3})_i \Omega_3 \\ &+ (1-\rho_2) \epsilon_4 (H_{2j4})_i \Omega_4\end{aligned}$$

$$\begin{aligned}\phi_3 \Big|_i &= (1-\rho_3) \epsilon_2 (H_{3j2})_i \Omega_2 + [(1-\rho_3) \epsilon_3 (H_{3j3})_i - \epsilon_3] \Omega_3 \\ &+ (1-\rho_3) \epsilon_4 (H_{3j4})_i \Omega_4\end{aligned}$$

$$\begin{aligned}\phi_4 \Big|_i &= (1-\rho_4) \epsilon_2 (H_{4j2})_i \Omega_2 + (1-\rho_4) \epsilon_3 (H_{4j3})_i \Omega_3 \\ &+ [(1-\rho_4) \epsilon_4 (H_{4j4})_i - \epsilon_4] \Omega_4\end{aligned}$$

The following are the constants for the radiation transfer terms in the energy balance equation.

$$G_1 = [1 - \tau_{20}^2 (H_{20j20})_b (H_{20m20})_c]^{-1}$$

$$S_1 = (H_{11j11})_b + \tau_{20}^2 G_1 (H_{11j20})_b (H_{20m20})_c (H_{20j11})_b$$

$$G_2 = [1 - \tau_{11}^2 S_1 (H_{11m11})_a]^{-1}$$

$$U_k = G_2 \epsilon_k [(H_{11hk})_b + \tau_{20}^2 G_1 (H_{11j20})_b (H_{20m20})_c (H_{20jk})_b]$$

$$V_k = G_1 \epsilon_k \tau_{20} [(H_{20j20})_b (H_{20mk})_c + \tau_{11}^2 G_1 G_2 (H_{20j11})_b (H_{11m11})_a (H_{11j20})_b (H_{20mk})_c]$$

$$Y_k = G_1 \epsilon_k [(H_{20mk})_c + \tau_{11}^2 \tau_{20}^2 G_1 G_2 (H_{20m20})_c (H_{20j11})_b (H_{11m11})_a (H_{11j20})_b (H_{20mk})_c]$$

$$W_k = G_1 [(H_{20jk})_b \epsilon_k + \tau_{11}^2 (H_{11m11})_a (H_{20j11})_b U_k]$$

$$D_{11} = \left(\frac{\sigma}{M_1}\right) \{A_{1c} [(1-\rho_1) \epsilon_1 (H_{1j1})_c - \epsilon_1] + A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c V_1 + A_{1d} [(1-\rho_1) \epsilon_1 (H_{1j1})_d - e_1]\}$$

$$D_{12} = 0$$

$$D_{13} = \left(\frac{\sigma}{M_1}\right) A_{1d} (1-\rho_1) \epsilon_3 (H_{1j3})_d$$

$$D_{14} = \left(\frac{\sigma}{M_1}\right) A_{1d} (1-\rho_1) \epsilon_4 (H_{1j4})_d$$

$$D_{15} = \left(\frac{\sigma}{M_1}\right) \{A_{1c} [(1-\rho_1) \epsilon_5 (H_{1j5})_c + (1-\rho_1) \tau_{20} (H_{1j20})_c v_5] + A_{1d} (1-\rho_1) \epsilon_5 (H_{1j5})_d\}$$

$$D_{16} = \left(\frac{\sigma}{M_1}\right) A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c w_6$$

$$D_{17} = \left(\frac{\sigma}{M_1}\right) A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c w_7$$

$$D_{18} = \left(\frac{\sigma}{M_1}\right) A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c w_8$$

$$D_{19} = \left(\frac{\sigma}{M_1}\right) A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c w_9$$

$$D_{1-10} = 0$$

$$D_{1-11} = \left(\frac{\sigma}{M_1}\right) A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c [w_{11} + G_1 G_2 \tau_{11} \epsilon_{11} (H_{20j11})_b (H_{11m11})_a]$$

$$D_{1-12} = D_{1-13} = 0$$

$$D_{1-14} = \left(\frac{\sigma}{M_1}\right) A_{1c} (1-\rho_1) \tau_{20} (H_{1j20})_c \tau_{11} \epsilon_{14} G_1 G_2 (H_{20j11})_b (H_{11m14})_c$$

$$D_{1-15} = D_{1-16} = D_{1-17} = D_{1-18} = D_{1-19} = 0$$

$$B_1 = \left(\frac{A_{1c}}{M_1}\right) \tau_{11} \tau_{20} \tau_{21} (1-\rho_1) G_1 G_2 (H_{1j20})_c (H_{20j11})_b (H_{11m21})_a$$

$$D_{2-1} = 0$$

$$D_{2-2} = \left(\frac{\sigma}{M_2}\right) A_{2i} [(1-\rho_2) \epsilon_2 (H_{2j2})_i - \epsilon_2]$$

$$D_{2-3} = \left(\frac{\sigma}{M_1}\right) A_{2i} (1-\rho_2) \epsilon_3 (H_{2j3})_i$$

$$D_{2-4} = \left(\frac{\sigma}{M_2}\right) A_{2i} (1-\rho_2) \epsilon_4 (H_{2j4})_i$$

$$\begin{aligned} D_{2-5} &= D_{2-6} = D_{2-7} = D_{2-8} = D_{2-9} = D_{2-10} = D_{2-11} = D_{2-12} = D_{2-13} = D_{2-14} \\ &= D_{2-15} = D_{2-16} = D_{2-17} = D_{2-18} = D_{2-19} = 0 \end{aligned}$$

$$D_{3-1} = \left(\frac{\sigma}{M_3}\right) A_{3d} (1-\rho_3) \epsilon_1 (H_{3j1})_d$$

$$D_{3-2} = \left(\frac{\sigma}{M_3}\right) A_{3d} (1-\rho_3) \epsilon_2 (H_{3j2})_i$$

$$D_{3-3} = \left(\frac{\sigma}{M_3}\right) \{A_{3d} [(1-\rho_3) \epsilon_3 (H_{3j3})_d - \epsilon_3] + A_{3i} [(1-\rho_3) \epsilon_3 (H_{3j3})_i - \epsilon_3]\}$$

$$D_{3-4} = \left(\frac{\sigma}{M_3}\right) \{A_{3d} (1-\rho_3) \epsilon_4 (H_{3j4})_d + A_{3i} (1-\rho_3) \epsilon_4 (H_{3j4})_i\}$$

$$D_{3-5} = \left(\frac{\sigma}{M_3}\right) A_{3d} (1-\rho_3) \epsilon_5 (H_{3j5})_d$$

$$\begin{aligned} D_{3-6} &= D_{3-7} = D_{3-8} = D_{3-9} = D_{3-10} = D_{3-11} = D_{3-12} = D_{3-13} = D_{3-14} = D_{3-15} \\ &= D_{3-16} = D_{3-17} = D_{3-18} = D_{3-19} = 0 \end{aligned}$$

$$D_{4-1} = \left(\frac{\sigma}{M_4}\right) A_{4d} (1-\rho_4) \epsilon_1 (H_{4j1})_d$$

$$D_{4-2} = \left(\frac{\sigma}{M_4}\right) A_{4i} (1-\rho_4) \epsilon_2 (H_{4j2})_i$$

$$D_{4-3} = \left(\frac{\sigma}{M_4}\right) \{A_{4d} (1-\rho_4) \epsilon_3 (H_{4j3})_d + A_{4i} (1-\rho_4) \epsilon_3 (H_{4h3})_i\}$$

$$D_{4-4} = \left(\frac{\sigma}{M_4}\right) \{A_{4d} [(1-\rho_4) \epsilon_4 (H_{4j4})_d - \epsilon_4] + A_{4e} [(1-\rho_4) \epsilon_4 (H_{4j4})_e - \epsilon_4] \\ + A_{4f} [(1-\rho_4) \epsilon_4 (H_{4j4})_f - \epsilon_4] + A_{4i} [(1-\rho_4) \epsilon_4 (H_{4j4})_i - \epsilon_4]\}$$

$$D_{4-5} = \left(\frac{\sigma}{M_4}\right) \{A_{4d} (1-\rho_4) \epsilon_5 (H_{4j5})_d\}$$

$$D_{4-6} = 0$$

$$D_{4-7} = \left(\frac{\sigma}{M_4}\right) A_{4f} (1-\rho_4) \epsilon_7 (H_{4j7})_f$$

$$D_{4-8} = \left(\frac{\sigma}{M_4}\right) A_{4f} (1-\rho_4) \epsilon_8 (H_{4j8})_f$$

$$D_{4-9} = D_{4-10} = D_{4-11} = D_{4-12} = D_{4-13} = 0$$

$$D_{4-14} = \left(\frac{\sigma}{M_4}\right) A_{4f} (1-\rho_4) \epsilon_{14} (H_{4j14})_f$$

$$D_{4-15} = \left(\frac{\sigma}{M_4}\right) A_{4f} (1-\rho_4) \epsilon_{15} (H_{4j15})_f$$

$$D_{4-16} = \left(\frac{\sigma}{M_4}\right) \{A_{4e} (1-\rho_4) \epsilon_{16} (H_{4j16}) + A_{4f} (1-\rho_4) \epsilon_{16} (H_{4j16})_f\}$$

$$D_{4-17} = \left(\frac{\sigma}{M_4}\right) A_{4f} (1-\rho_4) \epsilon_{17} (H_{4j17})_f$$

$$D_{4-18} = \left(\frac{\sigma}{M_4}\right) \{A_{4e} (1-\rho_4) \epsilon_{18} (H_{4j18})_e + A_{4f} (1-\rho_4) \epsilon_{18} (H_{4j18})_f\}$$

$$D_{4-19} = \left(\frac{\sigma}{M_4}\right) A_{4e} (1-\rho_4) \epsilon_{19} (H_{4j19})_e$$

$$D_{5-1} = \left(\frac{\sigma}{M_5}\right) \{A_{5c} (1-\rho_5) [\epsilon_1 (H_{5j1})_c + \tau_{20} (H_{5j20})_c v_1] + A_{5d} (1-\rho_5) \epsilon_1 \\ (H_{5j1})_d\}$$

$$D_{5-2} = 0$$

$$D_{5-3} = \left(\frac{\sigma}{M_5}\right) A_{5d} (1-\rho_5) \epsilon_3 (H_{5j3})_d$$

$$D_{5-4} = \left(\frac{\sigma}{M_5}\right) A_{5d} (1-\rho_5) \epsilon_4 (H_{5j4})_d$$

$$D_{5-5} = \left(\frac{\sigma}{M_5}\right) \{A_{5d} [(1-\rho_5) \epsilon_5 (H_{5h5})_c - \epsilon_5 + (1-\rho_5) \tau_{20} (H_{5j20})_c V_5] \\ + A_{5d} [(1-\rho_5) \epsilon_5 (H_{5j5})_d - \epsilon_5]_d\}$$

$$D_{5-6} = \left(\frac{\sigma}{M_5}\right) A_{5c} (1-\rho_5) \tau_{20} (H_{5j20})_c W_6$$

$$D_{5-7} = \left(\frac{\sigma}{M_5}\right) A_{5c} (1-\rho_5) \tau_{20} (H_{5j20})_c W_7$$

$$D_{5-8} = \left(\frac{\sigma}{M_5}\right) A_{5c} (1-\rho_5) \tau_{20} (H_{5j20})_c W_8$$

$$D_{5-9} = \left(\frac{\sigma}{M_5}\right) A_{5c} (1-\rho_5) \tau_{20} (H_{5j20})_c W_9$$

$$D_{5-10} = 0$$

$$D_{5-11} = \left(\frac{\sigma}{M_5}\right) A_{5c} (1-\rho_5) \tau_{20} (H_{5j20})_c [W_{11} + G_1 G_2 \tau_{11} \epsilon_{11} (H_{20j11})_b \\ (H_{11m11})_a \epsilon_{14}]$$

$$D_{5-12} = D_{5-13} = 0$$

$$D_{5-14} = \left(\frac{\sigma}{M_5}\right) A_{5c} \tau_{11} \tau_{20} G_1 G_2 (1-\rho_5) (H_{5j20})_c (H_{20j11})_b (H_{11m14})_a \epsilon_{14}$$

$$D_{5-15} = D_{5-16} = D_{5-17} = D_{5-18} = D_{5-19} = 0$$

$$B_5 = \left(\frac{A_{5c}}{M_5}\right) (1-\rho_5) \tau_{11} \tau_{20} \tau_{21} G_1 G_2 (H_{5j20})_c (H_{20j11})_b (H_{11m21})_a$$

$$D_{6-1} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) [\tau_{11}^2 \tau_{20} \epsilon_1 G_1 G_2 (H_{6j11})_b (H_{11m11})_a (H_{11j20})_b \\ (H_{20m1})_c + \tau_{20} (H_{8j20})_b Y_1]$$

$$D_{6-2} = 0, \quad D_{6-3} = 0, \quad D_{6-4} = 0$$

$$D_{6-5} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) [\tau_{11}^2 \tau_{20} \epsilon_5 G_1 G_2 (H_{6j11})_b (H_{11m11})_a (H_{11j20})_c \\ + \tau_{20} (H_{6j20})_b Y_5]$$

$$D_{6-6} = \left(\frac{\sigma}{M_6}\right) A_{6b} \{ (1-\rho_6) \epsilon_6 (H_{6j6})_b - \epsilon_6 + (1-\rho_6) [\tau_{11}^2 (H_{6j11})_b (H_{11m11})_a U_6 \\ + \tau_{20}^2 (H_{6j20})_b (H_{20m20})_c W_6] \}$$

$$D_{6-7} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) \{ \epsilon_7 (H_{6j7})_b + \tau_{11}^2 (H_{6j11})_b (H_{11m11})_a U_7 \\ + \tau_{20}^2 (H_{6j20})_b (H_{20m20})_c W_7 \}$$

$$D_{6-8} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) \{ \epsilon_8 (H_{6j8})_b + \tau_{11}^2 (H_{6j11})_b (H_{11m11})_a U_8 \\ + \tau_{20}^2 (H_{6j20})_b (H_{20m20})_c W_8 \}$$

$$D_{6-9} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) \{ \epsilon_9 (H_{6j9})_b + \tau_{11}^2 (H_{6j11})_b (H_{11m11})_a U_9 \\ + \tau_{20}^2 (H_{6j20})_b (H_{20m20})_c W_9 \}$$

$$D_{6-10} = 0$$

$$D_{6-11} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) \{ \epsilon_{11} (H_{6j11})_b + \tau_{11} (H_{6j11})_b (H_{11m11})_a \\ (\tau_{11} U_{11} + \epsilon_{11} G_2) + \tau_{20}^2 (H_{6j20})_b (H_{20m20})_c [W_{11} + G_1 G_2 \epsilon_{11} \tau_{11} \\ (H_{20j11})_b (H_{11m11})_a] \}$$

$$D_{6-12} = D_{6-13} = 0$$

$$D_{6-14} = \left(\frac{\sigma}{M_6}\right) A_{6b} (1-\rho_6) [\tau_{11} \epsilon_{14} G_2 (H_{6j11})_b (H_{11m14}) + \tau_{11} \tau_{20}^2 G_1 G_2 \epsilon_{14} \\ (H_{6j20})_b (H_{20m20})_c (H_{20j11})_b (H_{11m11})_a]$$

$$D_{5-15} = D_{6-16} = D_{6-17} = D_{6-18} = D_{6-19} = 0$$

$$B_6 = \left(\frac{A_{6b}}{M_6}\right) (1-\rho_6) \tau_{11} \tau_{21} G_2 [(H_{6j11})_b (H_{11m21})_a + G_1 \tau_{20}^2 (H_{6j20})_b (H_{20m20})_c \\ (H_{20j11})_b (H_{11m21})_a]$$

$$D_{7-1} = \left(\frac{\sigma}{M_7}\right) A_{7b} (1-\rho_7) \{ \tau_{11}^2 \tau_{20} G_1 G_2 (H_{7j11})_b (H_{11m11})_a (H_{11j20})_b (H_{20m1})_c \epsilon_1 \\ + \tau_{20} (H_{7j20})_b Y_1 \}$$

$$D_{7-2} = D_{7-3} = 0$$

$$D_{7-4} = \left(\frac{\sigma}{M_1}\right) A_{7f} (1-\rho_7) \epsilon_4 (H_{7j4})_f.$$

$$D_{7-5} = \left(\frac{\sigma}{M_7}\right) A_{7b} (1-\rho_7) \tau_{20} \{ \tau_{11}^2 \epsilon_5 G_1 G_2 (H_{7j11})_b (H_{11m11})_a (H_{11j20})_b \\ (H_{20m5})_c + (H_{7j20})_b Y_5 \}$$

$$D_{7-6} = \left(\frac{\sigma}{M_7}\right) A_{7b} (1-\rho_7) \{ \epsilon_6 (H_{7j6})_b + \tau_{11}^2 (H_{7j11})_b (H_{11m11})_a U_6 + \tau_{20}^2 \\ (H_{7j20})_b (H_{20m20})_c W_6 \}$$

$$D_{7-7} = \left(\frac{\sigma}{M_7}\right) \{A_{7b} [(1-\rho_7) \epsilon_7 (H_{7j7})_b - \epsilon_7 + (1-\rho_7) \{\tau_{11}^2 (H_{7j11})_b (H_{11m11})_a U_7 \\ + \tau_{20}^2 (H_{7j20})_b (H_{20m20})_c W_7\}] + A_{7f} [(1-\rho_7) \epsilon_7 (H_{7j7})_f - \epsilon_7]\}$$

$$D_{7-8} = \left(\frac{\sigma}{M_7}\right) \{A_{7b} (1-\rho_7) [\epsilon_8 (H_{7j8})_b + \tau_{11}^2 (H_{7j11})_b (H_{11m11})_a U_8 \\ + \tau_{20}^2 (H_{7j20})_b (H_{20m20})_c W_8] + A_{7f} (1-\rho_7) \epsilon_8 (H_{7j8})_f\}$$

$$D_{7-9} = \left(\frac{\sigma}{M_7}\right) \{A_{7b} (1-\rho_7) [\epsilon_9 (H_{7j9})_b + \tau_{11}^2 (H_{7j11})_b (H_{11m11})_a U_9 \\ + \tau_{20}^2 (H_{7j20})_b (H_{20m20})_c W_9]\}$$

$$D_{7-10} = 0$$

$$D_{7-11} = \left(\frac{\sigma}{M_7}\right) A_{7b} (1-\rho_7) \{\epsilon_{11} (H_{7j11})_b + \tau_{11} (H_{7j11})_b (\tau_{11} U_{11} + \epsilon_{11} G_2) \\ (H_{1-m11})_a + \tau_{20}^2 (H_{7j20})_b (H_{20m20})_c [W_{11} + G_1 G_2 \epsilon_{11} \tau_{11} (H_{20j11})_b \\ (H_{11m11})_a]\}$$

$$D_{7-12} = D_{7-13} = 0$$

$$D_{7-14} = \left(\frac{\sigma}{M_7}\right) \{A_{7b} (1-\rho_7) \tau_{11} G_2 \epsilon_{14} [(H_{7j11})_b (H_{11m14})_a + \tau_{20}^2 G_1 (H_{7j20})_b \\ (H_{20m20})_c (H_{20j11})_b (H_{11m14})_a] + A_{7f} (1-\rho_7) \epsilon_{14} (H_{7j14})_f\}$$

$$D_{7-15} = \left(\frac{\sigma}{M_7}\right) A_{7f} (1-\rho_7) \epsilon_{15} (H_{7j15})_f$$

$$D_{7-16} = \left(\frac{\sigma}{M_7}\right) A_{7f} (1-\rho_7) \epsilon_{16} (H_{7j16})_f$$

$$D_{7-17} = \left(\frac{\sigma}{M_7}\right) A_{7f} (1-\rho_7) \epsilon_{17} (H_{7j17})_f$$

$$D_{7-18} = \left(\frac{\sigma}{M_7}\right) A_{7f} (1-\rho_7) \epsilon_{18} (H_{7j18})_f$$

$$D_{7-19} = 0$$

$$B_7 = \left(\frac{A_{7b}}{M_7}\right) (1-\rho_7) \tau_{11} G_2 \{ \tau_{21} (H_{7j11})_b (H_{11m21})_a + \tau_{20}^2 \tau_{21} G_1 (H_{7j20})_b \\ (H_{20m20})_c (H_{20j11})_b (H_{11m21})_a \}$$

$$D_{8-1} = \left(\frac{\sigma}{M_8}\right) A_{8b} (1-\rho_8) \{ \tau_{11}^2 \tau_{20} \epsilon_1 G_1 G_2 (H_{8j11})_b (H_{11m11})_a (H_{11j20})_b (H_{20m1})_c \\ + \tau_{20} (H_{8j20})_b Y_1 \}$$

$$D_{8-2} = D_{8-3} = 0$$

$$D_{8-4} = \left(\frac{\sigma}{M_8}\right) A_{8f} (1-\rho_8) \epsilon_4 (H_{8j4})_f$$

$$D_{8-5} = \left(\frac{\sigma}{M_8}\right) A_{8b} (1-\rho_8) \{ \tau_{11}^2 \tau_{20} \epsilon_5 G_1 G_2 (H_{8j11})_b (H_{11m11})_a (H_{11j20})_b \\ (H_{20m5})_c + \tau_{20} (H_{8j20})_b Y_5 \}$$

$$D_{8-6} = \left(\frac{\sigma}{M_8}\right) A_{8b} (1-\rho_8) \{ \epsilon_6 (H_{8j6})_b + \tau_{11}^2 (H_{8j11})_b (H_{11m11})_a U_6 \\ + \tau_{20}^2 (H_{8j20})_b (H_{20m20})_c W_6 \}$$

$$D_{8-7} = \left(\frac{\sigma}{M_8}\right) \{ A_{8b} (1-\rho_8) [\epsilon_7 (H_{8j7})_b + \tau_{11}^2 (H_{8j11})_b (H_{11m11})_a U_7 + \tau_{20}^2 \\ (H_{8j20})_b (H_{20m20})_c W_7] + A_{8f} (1-\rho_8) \epsilon_7 (H_{8j7})_f \}$$

$$D_{8-8} = \left(\frac{\sigma}{M_8}\right) \{A_{8b} [(1-\rho_8) \epsilon_8 (H_{8j8})_b - \epsilon_8 + (1-\rho_8) [\tau_{11}^2 (H_{8j11})_b (H_{11ml1})_a U_8 \\ + \tau_{20}^2 (H_{8j20})_b (H_{20m20})_c W_8]] + A_{8f} [(1-\rho_8) \epsilon_8 (H_{8j8})_f - \epsilon_8] + A_{8g} \\ [(1-\rho_8) \epsilon_8 (H_{8j8})_g - \epsilon_8] + A_{8h} [(1-\rho_8) \epsilon_8 (H_{8j8})_h - \epsilon_8]\}$$

$$D_{8-9} = \left(\frac{\sigma}{M_8}\right) \{A_{8b} (1-\rho_8) [\epsilon_9 (H_{8j9})_b + \tau_{11}^2 (H_{8j11})_b (H_{11ml1})_a U_9 + \tau_{20}^2 \\ (H_{8j20})_b (H_{20m20})_c W_9] + A_{8g} (1-\rho_8) \epsilon_9 (H_{8j9})_g + A_{8h} (1-\rho_8) \epsilon_9 \\ (H_{8j9})_h\}$$

$$D_{8-10} = \left(\frac{\sigma}{M_8}\right) \times 0 = 0$$

$$D_{8-11} = \left(\frac{\sigma}{M_8}\right) A_{8b} (1-\rho_8) \{\epsilon_{11} (H_{8j11})_b + \tau_{11} (H_{8j11})_b (H_{11ml1})_a (\tau_{11} U_{11} \\ + \epsilon_{11} G_2) + \tau_{20}^2 (H_{8j20})_b (H_{20m20})_c [W_{11} + \tau_{11} \epsilon_{11} G_1 G_2 (H_{20j11})_b \\ (H_{11ml1})_a]\}$$

$$D_{8-12} = D_{8-13} = 0$$

$$D_{8-14} = \left(\frac{\sigma}{M_8}\right) \{A_{8b} (1-\rho_8) \tau_{11} \epsilon_{14} G_2 [(H_{8j11})_b (H_{11ml4})_a - \tau_{20}^2 G_1 (H_{8j20})_b \\ (H_{20m20})_c (H_{20j11})_b (H_{11ml4})_a] + A_{8f} (1-\rho_8) \epsilon_{14} (H_{8j14})_f \\ + A_{8g} (1-\rho_8) \epsilon_{14} (H_{8j14})_g + A_{8h} (1-\rho_8) \epsilon_{14} (H_{8j14})_h\}$$

$$D_{8-15} = \left(\frac{\sigma}{M_8}\right) A_{8f} (1-\rho_8) \epsilon_{15} (H_{8j15})_f$$

$$D_{8-16} = \left(\frac{\sigma}{M_8}\right) A_{8f} (1-\rho_8) \epsilon_{16} (H_{8j16})_f$$

$$D_{8-17} = \left(\frac{\sigma}{M_8}\right) [A_{8f} (1-\rho_8) \epsilon_{17} (H_{8j17})_f + A_{8h} (1-\rho_8) \epsilon_{17} (H_{9j17})_h]$$

$$D_{8-19} = 0$$

$$B_8 = \left(\frac{A_{8b}}{M_8}\right) (1-\rho_8) \tau_{11} \tau_{21} G_2 \{ (H_{11m21})_a + \tau_{20}^2 G_1 (H_{8j20})_b (H_{20m20})_b (H_{11m21})_a \}$$

$$D_{9-1} = \left(\frac{\sigma}{M_9}\right) A_{9b} (1-\rho_9) \tau_{20} \{ \tau_{11}^2 G_1 G_2 (H_{9j11})_b (H_{11m11})_a (H_{11j20})_b (H_{20m1})_b \epsilon_1 + (H_{9j20})_b Y_1 \}$$

$$D_{9-2} = 0, \quad D_{9-3} = 0, \quad D_{9-4} = 0$$

$$D_{9-5} = \left(\frac{\sigma}{M_9}\right) A_{9b} (1-\rho_9) \tau_{20} \{ \tau_{11}^2 \epsilon_5 G_1 G_2 (H_{9j11})_b (H_{11m11})_a (H_{11j20})_b (H_{20m5})_c + (H_{9j20})_b Y_5 \}$$

$$D_{9-6} = \left(\frac{\sigma}{M_9}\right) A_{9b} (1-\rho_9) \{ \epsilon_6 (H_{9j6})_b + \tau_{11}^2 (H_{9j11})_b (H_{11m11})_a U_6 + \tau_{20}^2 (H_{9j20})_b (H_{20m20})_c W_6 \}$$

$$D_{9-7} = \left(\frac{\sigma}{M_9}\right) A_{9b} (1-\rho_9) \{ \epsilon_7 (H_{9j7})_b + \tau_{11}^2 (H_{9j11})_b (H_{11m11})_a U_7 + \tau_{20}^2 (H_{9j20})_b (H_{20m20})_c W_7 \}$$

$$D_{9-8} = \left(\frac{\sigma}{M_9}\right) \{ A_{9b} (1-\rho_9) [\epsilon_8 (H_{9j8})_b + \tau_{11}^2 (H_{9j11})_b (H_{11m11})_a U_8 + \tau_{20}^2 (H_{9j20})_b (H_{20m20})_c W_8 + A_{9g} (1-\rho_9) \epsilon_8 (H_{9j8})_g + A_{9h} (1-\rho_9) \epsilon_8 (H_{9j8})_h] \}$$

$$D_{9-9} = \left(\frac{\sigma}{M_9}\right) \{A_{9b} [(1-\rho_9) \epsilon_9 (H_{9j9})_b - \epsilon_9 + (1-\rho_9) [\tau_{11}^2 (H_{9j11})_b (H_{11m11})_a U_9 \\ + \tau_{20}^2 (H_{9j20})_b (H_{20m20})_c W_9]] + A_{9g} [(1-\rho_9) \epsilon_9 (H_{9j9})_g - \epsilon_9] + A_{9h} \\ [(1-\rho_9) \epsilon_9 (H_{9j9})_h - \epsilon_9]\}$$

$$D_{9-10} = 0$$

$$D_{9-11} = \left(\frac{\sigma}{M_9}\right) A_{9b} (1-\rho_9) \{\epsilon_{11} (H_{9j11})_b + \tau_{11} (H_{9j11})_b (H_{11m11})_a (\tau_{11} U_{11} + \\ \epsilon_{11} G_2) + \tau_{20}^2 (H_{9j20})_b (H_{20m20})_c [W_{11} + \epsilon_{11} \tau_{11} G_1 G_2 (H_{20j11})_b \\ (H_{11m11})_a]\}$$

$$D_{9-12} = D_{9-13} = 0$$

$$D_{9-14} = \left(\frac{\sigma}{M_9}\right) \{A_{9b} \tau_{11} \epsilon_{14} G_2 [(H_{9j11})_b (H_{11m14})_a + \tau_{20}^2 G_1 (H_{9j20})_b (H_{20m20})_c \\ (H_{20j11})_b (H_{11m14})_a] + A_{9g} (1-\rho_9) \epsilon_{14} (H_{9j14})_g + A_{9h} (1-\rho_9) \epsilon_{14} \\ (H_{9j14})_h\}$$

$$D_{9-15} = D_{9-16} = 0$$

$$D_{9-17} = \left(\frac{\sigma}{M_9}\right) A_{9h} (1-\rho_9) \epsilon_{17} (H_{9j19})_h$$

$$D_{9-18} = D_{9-19} = 0$$

$$B_9 = \left(\frac{A_{8b}}{M_9}\right) (1-\rho_9) \tau_{11} \tau_{21} G_2 \{(H_{9j11})_b (H_{11m21})_a + \tau_{20}^2 G_1 (H_{9j20})_b (H_{20m20})_c \\ (H_{20j11})_b (H_{11m21})_a\}$$

$$D_{10-1} = D_{10-2} = D_{10-3} = D_{10-4} = D_{10-5} = D_{10-6} = D_{10-7} = D_{10-8} = D_{10-9} =$$

$$D_{10-10} = D_{10-11} = D_{10-12} = D_{10-13} = D_{10-14} = D_{10-15} = D_{10-16} =$$

$$D_{10-17} = D_{10-18} = D_{10-19} = 0$$

$$D_{11-1} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{\tau_{11}^2 \tau_{20} G_1 G_2 \epsilon_1 [(1-\rho_{11}) (H_{11j11})_b - 1] (H_{11m11})_a (H_{11j20})_c \\ + (1-\rho_{11}) \tau_{20} (H_{11j20})_b Y_1\} + A_{11a} \tau_{11} \tau_{20} \epsilon_1 G_1 G_2 [(1-\rho_{11}) \\ (H_{11m11})_a - 1] (H_{11j20})_a (H_{20m1})_c]$$

$$D_{11-2} = D_{11-3} = D_{11-4} = 0$$

$$D_{11-5} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{\tau_{11}^2 \tau_{20} G_1 G_2 \epsilon_5 [(1-\rho_{11}) (H_{11j11})_n - 1] (H_{11m11})_a (H_{11j20})_b \\ (H_{20m5})_c + (1-\rho_{11}) \tau_{20} (H_{11j20})_b Y_5\} + A_{11a} \tau_{11} \tau_{20} \epsilon_5 G_1 G_5 \\ [(1-\rho_{11}) (H_{11m11})_a - 1] (H_{11j20})_b (H_{20m5})_c]$$

$$D_{11-6} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{(1-\rho_{11}) \epsilon_6 (H_{11j6})_b + \tau_{11}^2 [(1-\rho_{11}) (H_{11j11})_b - 1] \\ (H_{11m11})_a U_6 + \tau_{20}^2 (1-\rho_{11}) (H_{11j20})_b (H_{20m20})_c W_6\} + A_{11a} \tau_{11} \\ [(1-\rho_{11}) (H_{11m11})_a - 1] U_6]$$

$$D_{11-7} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{(1-\rho_{11}) \epsilon_7 (H_{11j7})_b + \tau_{11}^2 [(1-\rho_{11}) (H_{11j11})_b - 1] U_7 \\ + \tau_{20}^2 (1-\rho_{11}) (H_{11j20})_b (H_{20m20})_c W_9\} + A_{11a} \tau_{11} [(1-\rho_{11}) \\ (H_{11m11})_a - 1] U_7]$$

$$D_{11-8} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{ (1-\rho_{11}) \epsilon_8 (H_{11j8})_b + \tau_{11}^2 [(1-\rho_{11}) (H_{11j11})_b - 1] \\ (H_{11m11})_a U_8 + \tau_{20}^2 (1-\rho_{11}) (H_{11j20})_b (H_{20m20})_c W_8 \} + A_{11a} \tau_{11} \\ [(1-\rho_{11}) (H_{11m11})_a - 1] U_8]$$

$$D_{11-9} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{ (1-\rho_{11}) \epsilon_9 (H_{11j9})_b + \tau_{11}^2 [(1-\rho_{11}) (H_{11j11})_b - 1] \\ (H_{11m11})_a U_9 + \tau_{20}^2 (1-\rho_{11}) (H_{11j20})_b (H_{20m20})_c W_9 \} + A_{11a} \tau_{11} \\ [(1-\rho_{11}) (H_{11m11})_a - 1] U_9]$$

$$D_{11-10} = 0$$

$$D_{11-11} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{ (1-\rho_{11}) \epsilon_{11} (H_{11j11})_b - \epsilon_{11} + \tau_{11} [(1-\rho_{11}) (H_{11j11})_b - 1] \\ (H_{11m11})_a (\tau_{11} U_{11} + e_{11} G_2) + \tau_{20}^2 (1-\rho_{11}) (H_{11j20})_b (H_{1120m20})_c \\ [W_{11} + \epsilon_{11} \tau_{11} G_1 G_2 (H_{20j11})_b (H_{11m11})_a] \} + A_{11a} \{ (1-\rho_{11}) \epsilon_{11} \\ (H_{11m11})_a - \epsilon_{11} + \tau_{11} [(1-\rho_{11}) (H_{11m11})_a - 1] [U_{11} + \tau_{11} \epsilon_{11} G_2 \\ S_1 (H_{11m11})_a] \}]]$$

$$D_{11-12} = D_{11-13} = 0$$

$$D_{11-14} = \left(\frac{\sigma}{M_{11}}\right) [A_{11b} \{ \tau_{11} \epsilon_{14} [(1-\rho_{11}) (H_{11j11})_b - 1] G_2 (H_{11m14})_a + (1-\rho_{11}) \tau_{11} \\ \tau_{20}^2 \epsilon_{14} G_1 G_2 (H_{11j20})_b (H_{20m20})_c (H_{20j11})_b (H_{11m14})_a \} + A_{11a} (1-\rho_{11}) \\ \epsilon_{14} S_1 G_2 [(1-\rho_{11}) (H_{11m11})_a]]]$$

$$D_{11-15} = D_{11-16} = D_{11-17} = D_{11-18} = D_{11-19} = 0$$

$$B_{11} = \left(\frac{1}{M_{11}}\right) [A_{11b} \{\tau_{11} \tau_{21} G_2 [(1-\rho_{11}) (H_{11j11})_b - 1] (H_{11m21})_a + (1-\rho_{11}) \tau_{11} \tau_{20}^2 \tau_{21} G_1 G_2 (H_{11j20})_b (H_{20m20})_c (H_{20j11})_b (H_{11m2k})\} + A_{11a} \{(1-\rho_{11}) \tau_{21} (H_{11m21})_a + \tau_{11}^2 \tau_{21} S_1 G_2 [(1-\rho_{11}) (H_{11m11})_a - 1] (H_{11m21})_a\}]$$

$$D_{12-1} = D_{12-2} = D_{12-3} = D_{12-4} = D_{12-5} = D_{12-6} = D_{12-7} = D_{12-8} = D_{12-9} = \\ D_{12-10} = D_{12-11} = D_{12-12} = D_{12-13} = D_{12-14} = D_{12-15} = D_{12-16} = \\ D_{12-17} = D_{12-18} = D_{12-19} = 0$$

$$D_{13-1} = D_{13-2} = D_{13-3} = D_{13-4} = D_{13-5} = D_{13-6} = D_{13-7} = D_{13-8} = D_{13-9} = \\ D_{13-10} = D_{13-11} = D_{13-12} = D_{13-13} = D_{13-14} = D_{13-15} = D_{13-16} = \\ D_{13-17} = D_{13-18} = D_{13-19} = 0$$

$$D_{14-1} = D_{14-2} = D_{14-3} = D_{14-4} = D_{14-5} = D_{14-6} = D_{14-7} = 0$$

$$D_{14-8} = \left(\frac{\sigma}{M_{14}}\right) [A_{14g} (1-\rho_{14}) \epsilon_8 (H_{14j8})_g + A_{14h} (1-\rho_{14}) \epsilon_8 (H_{14j8})_h]$$

$$D_{14-9} = \left(\frac{\sigma}{M_{14}}\right) [A_{14g} (1-\rho_{14}) \epsilon_9 (H_{14j9})_g + A_{14h} (1-\rho_{14}) \epsilon_9 (H_{14j9})_h]$$

$$D_{14-10} = D_{14-11} = D_{14-12} = D_{14-13} = 0$$

$$D_{14-14} = \left(\frac{\sigma}{M_{14}}\right) \{A_{14g} \epsilon_{14} [(1-\rho_{14}) (H_{14j14})_g - 1] + A_{14h} \epsilon_{14} [(1-\rho_{14}) (H_{14j14})_h - 1]\}$$

$$D_{14-15} = D_{14-16} = 0$$

$$D_{14-17} = \left(\frac{\sigma}{M_{14}}\right) A_{14h} (1-\rho_{14}) \epsilon_{17} (H_{14j17})_h$$

$$D_{14-18} = D_{14-19} = 0$$

$$D_{15-1} = D_{15-2} = D_{15-3} = 0$$

$$D_{14-4} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_4 (H_{15j4})_f$$

$$D_{15-5} = D_{15-6} = 0$$

$$D_{15-7} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_7 (H_{15j7})_f$$

$$D_{15-8} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_8 (H_{15j8})_f$$

$$D_{15-9} = D_{15-10} = D_{15-11} = D_{15-12} = D_{15-13} = 0$$

$$D_{15-14} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_{14} (H_{15j14})_f$$

$$D_{15-15} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} [(1-\rho_{15}) \epsilon_{15} (H_{15j15})_f - \epsilon_{15}]$$

$$D_{15-16} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_{16} (H_{15j16})_f$$

$$D_{15-17} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_{17} (H_{15j17})_f$$

$$D_{15-18} = \left(\frac{\sigma}{M_{15}}\right) A_{15f} (1-\rho_{15}) \epsilon_{18} (H_{15j18})_f$$

$$D_{15-19} = 0$$

$$D_{16-1} = D_{16-2} = D_{16-3} = 0$$

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$$D_{16-4} = \left(\frac{\sigma}{M_{16}}\right) \{A_{16e} (1-\rho_{16}) \epsilon_4 (H_{16j4})_e + A_{16f} (1-\rho_{16}) \epsilon_4 (H_{16j4})_f\}$$

$$D_{16-5} = D_{16-6} = 0$$

$$D_{16-7} = \left(\frac{\sigma}{M_{16}}\right) A_{16f} (1-\rho_{16}) \epsilon_7 (H_{16j7})_f$$

$$D_{16-8} = \left(\frac{\sigma}{M_{16}}\right) A_{16f} (1-\rho_{16}) \epsilon_8 (H_{16j8})_f$$

$$D_{16-9} = D_{16-10} = D_{16-11} = D_{16-12} = D_{16-13} = 0$$

$$D_{16-14} = \left(\frac{\sigma}{M_{16}}\right) A_{16f} (1-\rho_{16}) \epsilon_{14} (H_{16j14})_f$$

$$D_{16-15} = \left(\frac{\sigma}{M_{16}}\right) A_{16f} (1-\rho_{16}) \epsilon_{15} (H_{16j15})_f$$

$$D_{16-16} = \left(\frac{\sigma}{M_{16}}\right) \{A_{16e} [(1-\rho_{16}) (H_{16j16})_e - 1] + A_{16f} \epsilon_{16} [(1-\rho_{16}) (H_{16j16})_f - 1]\}$$

$$D_{16-17} = \left(\frac{\sigma}{M_{16}}\right) A_{16f} (1-\rho_{16}) \epsilon_{17} (H_{16j17})_f$$

$$D_{16-18} = \left(\frac{\sigma}{M_{16}}\right) \{A_{16e} (1-\rho_{16}) \epsilon_{18} (H_{16j18}) + A_{16f} (1-\rho_{16}) \epsilon_{18} (H_{16j18})_f\}$$

$$D_{16-19} = \left(\frac{\sigma}{M_{16}}\right) A_{16f} (1-\rho_{16}) \epsilon_{19} (H_{16j19})_e$$

$$D_{17-1} = D_{17-2} = D_{17-3} = 0$$

$$D_{17-4} = \left(\frac{\sigma}{M_{17}}\right) A_{17f} (1-\rho_{17}) \epsilon_4 (H_{17j4})_f$$

$$D_{17-5} = D_{17-6} = 0$$

$$D_{17-7} = \left(\frac{\sigma}{M_{17}}\right) A_{17f} (1-\rho_{17}) \epsilon_7 (H_{17j7})_f$$

$$D_{17-8} = \left(\frac{\sigma}{M_{17}}\right) \{A_{17f}^{(1-\rho_{17})} \epsilon_8 (H_{17j8})_f + A_{17h}^{(1-\rho_{17})} \epsilon_8 (H_{17j8})_h\}$$

$$D_{17-9} = \left(\frac{\sigma}{M_{17}}\right) A_{17h}^{(1-\rho_{17})} \epsilon_9 (H_{17j9})_h$$

$$D_{17-10} = D_{17-11} = D_{17-12} = D_{17-13} = 0$$

$$D_{17-14} = \left(\frac{\sigma}{M_{17}}\right) \{A_{17f}^{(1-\rho_{17})} \epsilon_{14} (H_{17j14})_f + A_{17h}^{(1-\rho_{17})} \epsilon_{14} (H_{17j14})_h\}$$

$$D_{17-15} = \left(\frac{\sigma}{M_{17}}\right) A_{17f}^{(1-\rho_{17})} \epsilon_{15} (H_{17j15})_f$$

$$D_{17-16} = \left(\frac{\sigma}{M_{17}}\right) A_{17f}^{(1-\rho_{17})} \epsilon_{16} (H_{17j16})_f$$

$$D_{17-17} = \left(\frac{\sigma}{M_{17}}\right) \{A_{17f} \epsilon_{17} [(1-\rho_{17}) (H_{17j17})_f - 1] + A_{17h} \epsilon_{17} [(1-\rho_{17}) (H_{17j17})_h - 1]\}$$

$$D_{17-18} = D_{17-19} = 0$$

$$D_{18-1} = D_{18-2} = D_{18-3} = 0$$

$$D_{18-4} = \left(\frac{\sigma}{M_{18}}\right) \{A_{18e}^{(1-\rho_{18})} \epsilon_4 (H_{18j4})_e + A_{18f}^{(1-\rho_{18})} \epsilon_4 (H_{18j4})_h\}$$

$$D_{18-5} = D_{18-6} = 0$$

$$D_{18-7} = \left(\frac{\sigma}{M_{18}}\right) A_{18f}^{(1-\rho_{18})} \epsilon_7 (H_{18j7})_f$$

$$D_{18-8} = \left(\frac{\sigma}{M_{18}}\right) A_{18f}^{(1-\rho_{18})} \epsilon_8 (H_{18j8})_f$$

$$D_{18-9} = D_{18-10} = D_{18-11} = D_{18-12} = D_{18-13} = 0$$

$$D_{18-14} = \left(\frac{\sigma}{M_{18}}\right) A_{18f}^{(1-\rho_{18})} \epsilon_{14} (H_{18j14})_f$$

$$D_{18-15} = \left(\frac{\sigma}{M_{18}}\right) A_{18f} (1-\rho_{18}) \epsilon_{15} (H_{18j15})_f$$

$$D_{18-16} = \left(\frac{\sigma}{M_{18}}\right) \{A_{18e} (1-\rho_{18}) \epsilon_{16} (H_{18j16}) + A_{18f} (1-\rho_{18}) \epsilon_{16} (H_{18j16})_f\}$$

$$D_{18-17} = \left(\frac{\sigma}{M_{18}}\right) A_{18f} (1-\rho_{18}) \epsilon_{17} (H_{18j17})_f$$

$$D_{18-18} = \left(\frac{\sigma}{M_{18}}\right) \{A_{18e} \epsilon_{18} [(1-\rho_{18})(H_{18j18})_e - 1] + A_{18f} \epsilon_{18} [(1-\rho_{18})(H_{18j18})_f - 1]\}$$

$$D_{18-19} = \left(\frac{\sigma}{M_{18}}\right) A_{18e} (1-\rho_{18}) \epsilon_{19} (H_{18j19})_e$$

$$D_{19-1} = D_{19-2} = D_{19-3} = 0$$

$$D_{19-4} = \left(\frac{\sigma}{M_{19}}\right) A_{19e} (1-\rho_{19}) \epsilon_4 (H_{19j4})_e$$

$$D_{19-5} = D_{19-6} = D_{19-7} = D_{19-8} = D_{19-9} = D_{19-10} = D_{19-11} = D_{19-12} = D_{19-13} =$$

$$D_{19-14} = D_{19-15} = 0$$

$$D_{19-16} = \left(\frac{\sigma}{M_{19}}\right) A_{19e} (1-\rho_{19}) \epsilon_{16} (H_{19j16})_e$$

$$D_{19-17} = 0$$

$$D_{19-18} = \left(\frac{\sigma}{M_{18}}\right) A_{19e} (1-\rho_{19}) \epsilon_{18} (H_{19j18})_e$$

$$D_{19-19} = \left(\frac{\sigma}{M_{18}}\right) A_{19e} \epsilon_{19} [(1-\rho_{19})(H_{19j19})_e - 1]$$

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16. Abstract Dynamic models and computer simulations were developed for the radiometric sensors utilized in the Earth Radiation Budget Experiment (ERBE). The models were developed to understand sensor performance, improve measurement accuracy by updating model parameters and provide the constants needed for the count conversion algorithms. Model simulations were compared with the sensor's actual responses demonstrated in the ground and inflight calibrations. The models consider thermal and radiative exchange effects, surface specularity, spectral dependence of a filter, radiative interactions among an enclosure's nodes, partial specular and diffuse enclosure surface characteristics and steady-state and transient sensor responses. Relatively few sensor nodes were chosen for the models since there is an accuracy tradeoff between increasing the number of nodes and approximating parameters such as the sensor's size, material properties, geometry, enclosure surface characteristics, etc. Given that the temperature gradients within a node and between nodes are small enough, approximating with only a few nodes does not jeopardize the accuracy required to perform the parameter estimates and error analyses.					
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